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MEMORANDUM

INVESTIGATION OF THE PRESSURE DISTRIBUTION OVER A MODEL
OF AN INTERCONTINENTAL BALLISTIC MISSILE FOR SEVERAL
NOSE SHAPES AT MACH NUMBERS OF 1.57, 2.29,
2.98, 3.96, AND 4.65

By John G. Presnell, Jr., and James D. Church

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Langley Field, Va.

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INVESTIGATION OF THE PRESSURE DISTRIBUTION OVER A MODEL
OF AN INTERCONTINENTAL BALLISTIC MISSILE FOR SEVERAL
NOSE SHAPES AT MACH NUMBERS OF 1.57, 2.29,
2.98, 3.96, AND 4.65*

By John G. Presnell, Jr., and James D. Church

SUMMARY

An investigation of the pressure distribution over a model of an intercontinental ballistic missile has been conducted in the Langley Unitary Plan wind tunnel. Data were obtained for five nose shapes at Mach numbers of 1.57, 2.29, 2.98, 3.96, and 4.65 at a Reynolds number per foot of approximately 4.5×10^6 at $M = 1.57$ and 3.8×10^6 at the other Mach numbers. Included in the results are boundary-layer characteristics of the noses and the base of the model which were measured with rakes. Data were obtained with both natural and fixed transition for three nose shapes. Tests were conducted over an angle-of-attack range from approximately -10° to 10° at about 0° angle of sideslip. The data are listed in pressure-coefficient form and presented with schlieren photographs without analysis.

INTRODUCTION

Pressure-distribution tests, including boundary-layer measurements, of a model of an intercontinental ballistic missile (ICBM) have been conducted at the Langley Unitary Plan wind tunnel.

During its period of flight in the atmosphere (ascent and descent), an ICBM is subjected to high airloads and temperatures because of surrounding atmosphere and extreme airspeeds. In designing such a missile for minimum weight with maximum performance, an accurate knowledge of the viscous heating and airloads encountered is essential.

The purpose of the present tests was to obtain pressure-distribution and boundary-layer data on an ICBM model in order to predict more accurately the airloads to be encountered and to aid in the evaluation of heat-transfer information. Tests were made at Mach numbers of 1.57, 2.29, 2.98, 3.96, and 4.65 through an angle-of-attack range from approximately -10° to 10° at about 0° angle of sideslip.

SYMBOLS

C_p	pressure coefficient, $\frac{p_l - p}{q}$
q	free-stream dynamic pressure, lb/sq ft
p	free-stream static pressure, lb/sq ft
p_l	measured pressure at each orifice, lb/sq ft
M	free-stream Mach number
α	angle of attack referred to missile center line, deg
β	angle of sideslip referred to missile center line, deg
l	reference model length, in.
x	distance from model station 0 to center of each particular orifice measured along model center line, in.
y	distance from model surface outboard to center of particular rake orifice, in.
θ	angular location of orifices about model center line ($\theta = 0^{\circ}$ at top of model, increasing clockwise facing downstream), deg

APPARATUS AND MODEL

The tests were conducted in both the high and low Mach number test sections of the Langley Unitary Plan wind tunnel. Tests were made in the low Mach number test section at $M = 1.57$ and in the high Mach number test section from $M = 2.29$ to 4.65 . This tunnel is of the variable-pressure, return-flow type with two test sections measuring 4 feet square

and approximately 7 feet long. Mach number may be varied continuously from approximately 1.5 to 2.9 in the low Mach number test section and from 2.3 to 4.7 in the high Mach number test section by means of asymmetric sliding-block nozzles.

Each test section has a central support system consisting of a variable-angle sting support which rides on a horizontal strut. This remotely operated sting support may be traversed across the strut as well as being varied in angle with respect to the relative wind. This allows the model to be positioned near the center of the test section at all angles of attack in order to reduce the likelihood of wall-reflected shock waves striking the model.

The model consisted of five interchangeable nose shapes attached to a cylindrical secondary rocket stage which in turn was attached to a larger first-stage rocket. (See fig. 1 for details and dimensions and fig. 2 for photographs.) The model was mounted to the test-section central support system by a sting from the base of the first-stage rocket. The instrumentation of the model consisted of 265 pressure orifices including 4 rakes for boundary-layer measurements having a total of 40 tubes. The base of the model was equipped with two manifolds: One measured base pressure ($x/l = 1.00$) and the other measured sting-cavity pressure ($x/l = 0.921$). The orifice locations of the model instrumentation are illustrated in figure 1.

Pressure tubes led from the model instrumentation through the sting and permanent test-section pressure instrumentation to liquid manometers outside the test section. Manometer readings were recorded photographically.

Two manometer fluids were utilized; rake pressures and pressures at the first five orifices at $\theta = 180^\circ$ were measured with mercury-filled manometers and pressures at all other orifices were measured with alkazene-filled manometers.

TESTS

Tests were made through an angle-of-attack range from approximately -10° to 10° at 0° angle of sideslip.

The test conditions of Mach number, stagnation and dynamic pressure, and Reynolds number per foot are listed in the following table:

M	Static pressure, lb/sq ft	Dynamic pressure, q , lb/sq ft	Reynolds number per ft
1.57	587.4	1,013	4.47×10^6
2.29	231.5	852	3.77
2.98	116.2	720	3.77
3.96	50.5	555	3.78
4.65	28.7	436	3.79

All configurations were tested at $M = 1.57$, and those considered to be of the most interest were tested at the other Mach numbers.

In order to determine the effect of fixing boundary-layer transition on the pressure distribution, transition strips $3/16$ inch wide of No. 180 carborundum grains imbedded in Krylon were placed around the nose as shown in figures 1(b) and 1(c) for noses I, II, and III.

In order to prevent adverse condensation effects, the dewpoint temperature for all tests was maintained below -30° F except at $M = 4.65$ where it was maintained below -20° F. The stagnation temperature was maintained at 150° F for $M = 1.57$, 2.29, and 2.98 and at 175° F for $M = 3.96$ and 4.65.

PRESENTATION OF RESULTS

The results have been plotted and orifice readings which were obviously in error have been deleted. Rake data were obtained simultaneously with the overall pressure distribution and slight interference effects of the rakes appear.

The data are presented for angles of attack of $\pm 10.1^\circ$, $\pm 6.3^\circ$, $\pm 3^\circ$, and 0° with the model oriented so that the orifices located along $\theta = 180^\circ$ measured positive pressures with positive angle of attack. In utilizing the data, a more complete pressure distribution may be obtained by combining data obtained with positive and negative angles of attack of equal magnitude.

Typical schlieren photographs are presented in figure 3; the tabulated pressure coefficients are presented in the following figures and tables:

Nose	M	Transition	Table number
I	1.57	Natural	1
	2.98	Fixed	2
	2.98	Fixed	3
II	1.57	Natural	4
		Fixed	5
III	1.57	Natural	6
	2.29		7
	2.98		8
	3.96		9
	4.65		10
	1.57	Fixed	11
	2.29		12
	2.98		13
	3.96		14
	4.65		15
IV	1.57	Natural	16
	2.29		17
	2.98		18
	3.96		19
	4.65		20
V	1.57	Natural	21
	2.29		22
	2.98		23
	3.96		24
	4.65		25

CORRECTIONS AND ACCURACY

Tunnel calibrations indicate that some flow angularity exists in both test sections. For the present investigation the model was oriented so that measurable angularity existed only in the angle-of-sideslip plane. All tests were conducted at 0° geometric sideslip angle; hence, the absolute value of this angle corresponds to the flow angularities listed as follows:

M	β , deg
1.57	-0.4
2.23	-.4
2.98	-.2
3.96	-.5
4.65	-.9

The angles of attack have been corrected for deflections of the model-sting combination under static load and are accurate to $\pm 0.1^\circ$.

Random pressure-coefficient errors encountered in data processing are about $\pm 0.005 C_p$. Inherent errors in pressure coefficients arise in testing because of the variation of local Mach number in that part of the test section occupied by the model. These errors affect the determination of static pressure p and dynamic pressure q from Mach number and a measured stagnation pressure. Local Mach numbers from this investigation vary about ± 0.015 from the values listed in the preceding table. An indication of the effect of this Mach number variation on pressure coefficients is given in the following table:

M	C_p	ΔC_p
1.57	1.564 1.000 .502 .050	0.018 .015 .014 .012
2.98	1.765 .994 .501 .050	.026 .016 .010 .005
4.65	1.801 .989 .501 .050	.024 .014 .007 .002

At a Mach number of 1.57 for all configurations at angles of attack of $\pm 6.3^\circ$ and $\pm 10.1^\circ$ wall-reflected shock waves impinged on the rear of the model invalidating the data presented for values of x/l greater than 0.90, including the base boundary-layer rakes.

Langley Research Center,
National Aeronautics and Space Administration,
Langley Field, Va., October 2, 1958.

TABLE 1. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH NATURAL TRANSITION AT $M = 1.57$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.554			-0.055
-0.037							1.145			-0.037
-0.018							1.115			-0.018
.002							1.079			.002
.021							1.038			.021
.041							1.015			.041
.060							1.008			.060
.075	.275	.042	.000	-0.023	-0.015	-0.009	-0.008			.075
.085	.313	.053	.014	.006	.008	.015	.024	.033	.011	.085
.096	.393	.094	.054	.054	.064	.071	.067	-.001	.090	.096
.126		.101	.065	.048	.053	.055	.063	.054	.090	.126
.143							1.210			.143
.154	-0.003	-0.169	-0.190	-0.200	-0.190	-0.179	-0.173	-0.209		.154
.171							1.142			.171
.189							1.112			.189
.206	.021	-0.150	-0.164	-0.165	-0.108	-0.094	-0.087	-0.128		.206
.223							1.062			.223
.241							1.047			.241
.258	.025	-0.152	-0.155	-0.126	-0.066	-0.044	-0.035	-0.078	-0.150	.258
.275							1.027			.275
.293							1.019			.293
.310	.025	-0.147	-0.135	-0.087	-0.049	-0.026	-0.007	-0.056	-0.144	.310
.327							1.000			.327
.344							1.003			.344
.362	.017	-0.134	-0.098	-0.056	-0.051	-0.025	.000	-0.045	-0.123	.362
.379							1.001			.379
.396	.016	-0.121	-0.090	-0.048	-0.051	-0.024	.010	-0.041	-0.109	.396
.408							1.002			.408
.426	.237	.074	.097	.084	.061	.158	.251	.087	.093	.426
.451	.219	.071	.060	.075	.059	.108	.161	.071	.451	
.476	.213	.048	.042	.063	.039	.082	.110	.076	.055	.476
.501	.204	.040	.035	.046	.021	.067	.091	.050	.047	.501
.530	-0.004	-0.132	-0.138	-0.113	-0.087	-0.133	-0.110	-0.076	-0.128	.530
.548							1.123			.548
.566							1.096			.566
.584	.030	-0.120	-0.106	-0.084	-0.092	-0.091	-0.055	-0.074	-0.115	.584
.602							1.053			.602
.620							1.044			.620
.638	.048	-0.107		-0.082	-0.053				-0.102	.638
.655							1.029			.655
.673							1.024			.673
.691	.033	-0.121	-0.109	-0.071	-0.045	-0.049	-0.023	-0.035	-0.099	.691
.709							1.019			.709
.727							1.022			.727
.745	.034	-0.120		-0.054	-0.037		-0.019		-0.089	.745
.763							1.018			.763
.781							1.028			.781
.799							1.021			.799
.817	.042	-0.102	-0.070	-0.041	-0.035	-0.031	-0.009	-0.024	-0.075	.817
.834							1.016			.834
.852							1.007			.852
.870							1.008			.870
.888							1.009			.888
.906							1.003			.906
.921	-0.224						1.002		-0.071	.921
.924	.041	-0.075			-0.031	-0.031				.924
.947							1.002			.947
.964							1.016			.964
.981	.087	-0.065	-0.028			-0.013	-0.004		-0.050	.981
1.000	-0.230						1.015			1.000

Orifice station, y_{in}	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y_{in}	Cp top	Cp side	
1.010	1.690	.784	.031	.880	1.423	
1.030	1.648	1.480	.094	1.593	1.057	
1.050	1.703	1.702	.156	1.670	1.531	
1.070	1.719	1.704	.219	1.594	1.620	
1.090	1.728	1.609	.281	1.597	1.493	
1.110	1.740	1.557	.344	1.589	1.469	
1.130	1.733	1.557	.406	1.590	1.492	
1.150	1.724	1.550	.469	1.595	1.485	
1.170	1.705	1.536	.750	1.581		
1.190	1.619	1.438	1.000	1.582		
			1.250	1.575		
			1.500	1.568		

TABLE 1. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(b) $\alpha = -6.3^\circ$.

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.555			-0.055
-0.037							.199			-0.037
-0.018							.162			-0.018
.002		.204					.123	.147		.002
.021							.069			.021
.041							.046			.041
.060							.019			.060
.075	.205	.072	.046	.025	.019	.019	.019			.075
.085	.240	.136	.110	.110	.105	.102	.094	.074		.085
.096	.323						.094	.073	.117	.096
.126		.139	.119	.102	.096	.090	.091	.106	.132	.126
.143							.194			.143
.154	-0.055	-0.137	-0.149	-0.160	-0.169	-0.166	.159	-0.165		.154
.171							.141			.171
.189							.112			.189
.206	-0.022	-0.103	-0.110	-0.125	-0.100	-0.091	.091	-0.108		.206
.223							.063			.223
.241							.047			.241
.258	-0.012	-0.091	-0.092	-0.079	-0.049	-0.037	.035	-0.054	-0.087	.258
.275							.030			.275
.293							.023			.293
.310	-0.007	-0.069	-0.065	-0.052	-0.031	-0.018	.015	-0.032	-0.069	.310
.327							.012			.327
.344							.008			.344
.362	-0.006	-0.057	-0.050	-0.034	-0.018	-0.006	.001	-0.021	-0.055	.362
.379							.003	-0.016	-0.050	.379
.396	-0.008	-0.054	-0.046	-0.031	-0.014	-0.006	.020			.396
.408							.170	.138	.145	.408
.426	.208	.142	.150	.143	.134	.152				.426
.451	.190	.129	.119	.115	.118	.125	.141			.451
.476	.182	.105	.096	.096	.098	.104	.114	.101	.104	.476
.501	.173	.093	.086	.079	.067	.077	.093	.078	.093	.501
.530	-0.027	-0.092	-0.101	-0.107	-0.082	-0.088	.098	-0.079	-0.094	.530
.548							.074			.548
.566							.060			.566
.584	-0.002	-0.075	-0.073	-0.065	-0.066	-0.065	.053	-0.059	-0.071	.584
.602							.044			.602
.620							.036			.620
.638		.000	-0.056		-0.051	-0.040				.638
.655							.020			.655
.673							.019			.673
.691	.003	-0.061	-0.058	-0.049	-0.031	-0.030	.008	-0.026	-0.053	.691
.709							.014			.709
.727							.006			.727
.745	.006	-0.054		-0.033	-0.021		.010		-0.043	.745
.763							.007			.763
.781							.014			.781
.799							.004			.799
.817	.017	-0.043	-0.036	-0.026	-0.015	-0.009	.004	-0.010	-0.025	.817
.834							.002			.834
.852							.007			.852
.870							.008			.870
.888							.007			.888
.906							.012			.906
.921	-0.173	-0.030	-0.028		-0.016	-0.005			-0.039	.921
.924							.011			.924
.947							.020			.947
.964							.012			.964
.981	.061	-0.017	-0.007			.017	.031	.048	-0.021	.981
1.000	-0.178						.004			1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.384	.949	.031	.752	.541
.030	1.605	1.531	.094	1.330	.971
.050	1.689	1.701	.156	1.623	1.331
.070	1.712	1.705	.219	1.596	1.555
.090	1.725	1.657	.281	1.574	1.563
.110	1.730	1.619	.344	1.575	1.530
.130	1.722	1.602	.406	1.574	1.536
.150	1.710	1.591	.469	1.581	1.540
.170	1.689	1.567	.750	1.573	
.190	1.595	1.462	1.000	1.572	
			1.250	1.562	
			1.500	1.565	

TABLE 1. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.556			-0.055
-0.037							.254			-0.037
-0.018							.217			-0.018
.002							.172			.002
.021							.109			.021
.041							.081			.041
.060							.048			.060
.075	.089	.075	.057	.052	.049	.048				.075
.085	.107	.099	.090	.075	.073	.078	.095			.085
.096	.158	.143	.147	.139	.132	.122	.128	.168		.096
.126							.123			.126
.143							.178			.143
.154	.162	.152	.138	.130	.124		.141			.154
.171							.147			.171
.189							.123			.189
.206	.051	.080	.082	.095	.084	.081	.083	.086		.206
.223							.064			.223
.241							.055			.241
.258	.034	.054	.057	.056	.050	.044	.048	.048	.050	.258
.275							.039			.275
.293							.032			.293
.310	.024	.035	.039	.033	.027	.017	.017	.025	.038	.310
.327							.011			.327
.344							.010			.344
.362	.012	.029	.021	.013	.014	.009	.008	.011	.026	.362
.379							.010			.379
.396	.013	.018	.019	.020	.012	.007	.000	.013	.019	.396
.408							.017			.408
.426	.190	.172	.172	.170	.164	.170	.168	.185	.169	.426
.451	.171	.154	.149	.147	.147	.144	.147			.451
.476	.158	.134	.128	.127	.118	.117	.119	.126	.132	.476
.501	.142	.120	.114	.103	.099	.096	.099	.109	.121	.501
.530	.053	.073	.079	.082	.079	.080	.079	.074	.076	.530
.548							.064			.548
.566							.052			.566
.584	.022	.048	.050	.052	.054	.054	.047	.045	.048	.584
.602							.035			.602
.620							.032			.620
.638	.009	.034		.028	.028				.034	.638
.655							.021			.655
.673							.021			.673
.691	.010	.028	.028	.029	.031	.027	.015	.022	.021	.691
.709							.015			.709
.727							.012			.727
.745	.004	.026		.022	.013		.004		.014	.745
.763							.007			.763
.781							.007			.781
.799							.004			.799
.817	.008	.015	.011	.010	.009	.004	.000	.003	.000	.817
.834							.002			.834
.852							.003			.852
.870							.005			.870
.888							.009			.888
.906							.006			.906
.921	.101									.921
.924	.002	.004		.010	.003		.010		.013	.924
.947							.011			.947
.964							.013			.964
.981	.050	.011	.008		.008	.007	.007	.004	.002	.981
1.000	.105									1.000

Nose rake			Base rake		
Orifice station, y , in.	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side
.010	.930	.598	.031	.648	.549
.030	1.386	1.235	.094	1.041	.823
.050	1.646	1.634	.156	1.241	1.057
.070	1.729	1.721	.219	1.530	1.279
.090	1.753	1.728	.281	1.566	1.443
.110	1.746	1.690	.344	1.583	1.525
.130	1.728	1.621	.406	1.574	1.560
.150	1.702	1.592	.469	1.576	1.564
.170	1.669	1.570	.750	1.569	
.190	1.558	1.472	1.000	1.569	
			1.250	1.571	
			1.500	1.565	

TABLE 1. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{L}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{L}$
	0	90	105	120	150	165	180	225	270	
-0.055						1.553				-0.055
-0.037						.305				-0.037
-0.018						.266				-0.018
.002						.217				.002
.021						.150				.021
.041						.116				.041
.060						.084				.060
.075	.118	.113	.117	.113	.109	.109	.114	.116		.075
.085	.142	.166	.160	.173	.176	.170	.158	.175	.184	.085
.096	.271									.096
.126		.169	.168	.164	.166	.162	.162	.164	.161	.126
.143							.157			.143
.154	-0.122	-0.116	-0.120	-0.122	-0.124	-0.125	-0.119	-0.129		.154
.171							.106			.171
.189							.088			.189
.206	-0.071	-0.072	-0.069	-0.080	-0.072	-0.071	-0.072	-0.073		.206
.223							.056			.223
.241							.048			.241
.258	-0.043	-0.043	-0.044	-0.043	-0.040	-0.037	-0.041	-0.041	-0.038	.258
.275							.039			.275
.293							.036			.293
.310	-0.028	-0.021	-0.024	-0.023	-0.032	-0.028	-0.026	-0.026	-0.025	.310
.327							.021			.327
.344							.013			.344
.362	-0.012	-0.019	-0.016	-0.011	-0.012	-0.008	-0.007	-0.010	-0.017	.362
.379							.010			.379
.396	-0.013	-0.009	-0.009	-0.009	-0.010	-0.009	-0.001	-0.011	-0.008	.396
.408							.010			.408
.426	.178	.182	.184	.182	.178	.182	.179	.180	.179	.426
.451	.158	.162	.160	.160	.160	.157	.160			.451
.476	.140	.142	.139	.141	.145	.140	.141	.143	.140	.476
.501	.118	.126	.130	.124	.120	.118	.120	.122	.127	.501
.530	-0.070	-0.064	-0.068	-0.072	-0.069	-0.071	-0.071	-0.067	-0.063	.530
.548							.055			.548
.566							.042			.566
.584	-0.040	-0.040	-0.039	-0.041	-0.043	-0.046	-0.039	-0.036	-0.041	.584
.602							.033			.602
.620							.023			.620
.638	-0.026	-0.026		-0.021	-0.013				-0.025	.638
.655							.020			.655
.673							.019			.673
.691	-0.016	-0.018	-0.019	-0.022	-0.029	-0.026	-0.015	-0.020	-0.012	.691
.709							.016			.709
.727							.014			.727
.745	-0.009	-0.016		-0.015	-0.013		-0.010		-0.006	.745
.763							.005			.763
.781							.012			.781
.799							.007			.799
.817	.004	-0.004	-0.005	-0.003	-0.008	-0.004	-0.001	-0.000	.009	.817
.834							.005			.834
.852							.002			.852
.870							.003			.870
.888							.008			.888
.906							.005			.906
.921	-0.082									.921
.924	.004	.004		-0.007	.006					.924
.947							.011			.947
.964							.012			.964
.981	.055	.029	.014		.011	.007				.981
1.000	-0.085						.009			1.000

Orifice station, $y, \text{in.}$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y, \text{in.}$	Cp top	Cp side
.010	.791	.631	.031	.506	.534
.030	1.216	1.282	.094	.739	.746
.050	1.550	1.649	.156	.911	.910
.070	1.706	1.723	.219	1.077	1.080
.090	1.744	1.727	.281	1.217	1.228
.110	1.740	1.684	.344	1.352	1.352
.130	1.718	1.623	.406	1.453	1.458
.150	1.685	1.595	.469	1.524	1.528
.170	1.649	1.570	.750	1.563	
.190	1.534	1.475	1.000	1.561	
			1.250	1.562	
			1.500	1.561	

TABLE 1. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.550			-0.055
-0.037							•363			-0.037
-0.018							•328			-0.018
•002							•279			•002
•021							•205			•021
•041							•164			•041
•060		•076	•085	•090	•121	•126	•127			•060
•075	•081	•104	•115	•125	•143	•149	•156	•143	•171	•075
•085	•106	•143	•155	•186	•212	•212	•201	•197		•085
•096	•231									•096
•126		•165	•176	•183	•201	•204	•209	•186	•151	•126
•143					•112	•101	•097	•094	•115	•143
•154	-•142	-•123	-•121	-•112						•154
•171										•171
•189										•189
•206	-•082	-•078	-•068	-•074	-•059	-•054	-•054	-•066		•206
•223										•223
•241										•241
•258	-•048	-•051	-•051	-•047	-•032	-•026	-•032	-•040	-•048	•258
•275										•275
•293										•293
•310	-•021	-•033	-•033	-•026	-•029	-•024	-•022	-•027	-•035	•310
•327										•327
•344										•344
•362	-•010	-•027	-•025	-•020	-•017	-•009	-•007	-•019	-•025	•362
•379										•379
•396	-•008	-•018	-•019	-•016	-•010	-•005	-•001	-•017	-•017	•396
•408										•408
•426	•168	•175	•180	•181	•184	•193	•189	•181	•169	•426
•451	•147	•155	•156	•160	•169	•168	•173		•146	•451
•476	•124	•134	•137	•144	•159	•162	•166	•148	•132	•476
•501	•102	•119	•130	•132	•138	•138	•143	•135	•118	•501
•530	-•076	-•071	-•071	-•070	-•058	-•058	-•055	-•061	-•072	•530
•548										•548
•566										•566
•584	-•050	-•046	-•042	-•040	-•034	-•034	-•025	-•031	-•049	•584
•602										•602
•620										•620
•638	-•029	-•033		-•021	-•006				-•032	•638
•655										•655
•673										•673
•691	-•016	-•026	-•027	-•025	-•027	-•021	-•008	-•023	-•021	•691
•709										•709
•727										•727
•745	-•007	-•022		-•019	-•012				-•014	•745
•763										•763
•781										•781
•799										•799
•817	•004	-•013	-•011	-•012	-•008	-•003	-•002	-•005	-•001	•817
•834										•834
•852										•852
•870										•870
•888										•888
•906										•906
•921	-•094									•921
•924	•011	•001		-•011	•001				-•008	•924
•947										•947
•964										•964
•981	•064	•019	•004							•981
1.000	-•102									1.000

Orifice station, y,in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side	
•010	•662	1.006	•031	•473	•546	
•030	1.007	1.454	•094	•662	•803	
•050	1.355	1.628	•156	•772	1.019	
•070	1.622	1.665	•219	•860	1.242	
•090	1.721	1.664	•281	•931	1.414	
•110	1.733	1.653	•344	1.008	1.509	
•130	1.712	1.638	•406	1.079	1.554	
•150	1.678	1.625	•469	1.150	1.567	
•170	1.636	1.601	•750	1.413		
•190	1.518	1.502	1.000	1.591		
			1.250	1.560		
			1.500	1.561		

TABLE 1. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.559			-0.055
-0.037							.438			-0.037
-0.018							.400			-0.018
.002		.203					.346			.002
.021							.271			.021
.041							.228			.041
.060							.180			.060
.075	.052	.095	.122	.148	.185	.200	.210	.164		.075
.085	.079	.144	.161	.206	.256	.264	.256	.215	.113	.085
.096	.198									.096
.126		.141	.164	.190	.241	.251	.264	.203	.130	.126
.143							.096			.143
.154	-0.161	-0.136	-0.124	-0.106	-0.075	-0.065	-0.059	-0.103		.154
.171							.052			.171
.189							.038			.189
.206	-0.091	-0.096	-0.081	-0.076	-0.040	-0.030	-0.028	-0.061		.206
.223							.016			.223
.241							.017			.241
.258	-0.041	-0.087	-0.075	-0.057	-0.023	-0.010	-0.014	-0.043	-0.083	.258
.275							.014			.275
.293							.011			.293
.310	-0.016	-0.068	-0.067	-0.050	-0.020	-0.009	-0.004	-0.037	-0.068	.310
.327							.005			.327
.344							.007			.344
.362	-0.005	-0.059	-0.052	-0.042	-0.025	-0.013	-0.009	-0.038	-0.052	.362
.379							.008			.379
.396	-0.004	-0.052	-0.055	-0.045	-0.016	-0.007	.003	-0.036	-0.048	.396
.408							.001			.408
.426	.175	.145	.151	.161	.190	.209	.208	.172	.146	.426
.451	.144	.131	.137	.147	.173	.179	.188			.451
.476	.117	.107	.115	.133	.166	.174	.178	.148	.105	.476
.501	.093	.093	.111	.123	.160	.169	.176	.140	.096	.501
.530	-0.097	-0.091	-0.082	-0.069	-0.043	-0.037	-0.034	-0.050	-0.092	.530
.548							.021			.548
.566							.009			.566
.584	-0.051	-0.071	-0.058	-0.046	-0.021	-0.014	-0.005	-0.031	-0.073	.584
.602							.001			.602
.620							.000			.620
.638	-0.029	-0.055		-0.035	-0.006		.017			.638
.655							.001			.655
.673							.004			.673
.691	-0.016	-0.058	-0.044	-0.037	-0.024	-0.010	-0.000	-0.029	-0.052	.691
.709							.000			.709
.727							.000			.727
.745	-0.003	-0.053		-0.038	-0.011		.002			.745
.763							.005			.763
.781							.003			.781
.799							.008			.799
.817	.008	-0.038	-0.039	-0.029	-0.007	.004	.011	-0.010	-0.024	.817
.834							.003			.834
.852							.015			.852
.870							.015			.870
.888							.013			.888
.906							.017			.906
.921	-0.160	.003	-0.023		-0.031	.006	.018			.921
.924							.019			.924
.947							.024			.947
.964							.025			.964
.981	.047	-0.009	-0.020			.011	.015	.017	.001	.981
1.000	-0.172									1.000

Orifice station, $y, \text{in.}$	Nose rake		Base rake		
	Cp , top	Cp , side	Orifice station, $y, \text{in.}$	Cp , top	Cp , side
.010	.549	.961	.031	.713	.545
.030	.810	.542	.094	1.248	.951
.050	1.079	1.709	.156	1.464	1.300
.070	1.373	1.702	.219	1.481	1.550
.090	1.614	1.662	.281	1.492	1.579
.110	1.712	1.619	.344	1.511	1.545
.130	1.705	1.598	.406	1.524	1.549
.150	1.677	1.587	.469	1.539	1.550
.170	1.635	1.563	.750	1.552	
.190	1.509	1.464	1.000	1.574	
			1.250	1.585	
			1.500	1.581	

TABLE 1. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH NATURAL TRANSITION AT $M = 1.57$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.557			-0.055
-0.037							.537			-0.037
-0.018							.496			-0.018
0.002							.434			0.002
0.021							.347			0.021
0.041							.301			0.041
0.060							.252			0.060
0.075							.283			0.075
0.085							.197			0.085
0.096							.242			0.096
0.126							.230			0.126
0.143							.086			0.143
0.154	-0.178	-0.164	-0.139	-0.103	-0.042	-0.020	-0.007	-0.088		0.154
0.171							.003			0.171
0.189							.005			0.189
0.206	-0.088	-0.145	-0.113	-0.084	-0.010	.008	.014	-0.052		0.206
0.223							.025			0.223
0.241							.021			0.241
0.258	-0.036	-0.141	-0.120	-0.080	-0.005	.021	.020	-0.045	-0.145	0.258
0.275							.016			0.275
0.293							.018			0.293
0.310	-0.011	-0.146	-0.123	-0.083	-0.009	.016	.026	-0.049	-0.142	0.310
0.327							.023			0.327
0.344							.018			0.344
0.362	-0.001	-0.131	-0.125	-0.090	-0.020	.005	.015	-0.058	-0.119	0.362
0.379							.014			0.379
0.396	-0.000	-0.114	-0.123	-0.096	-0.011	.013	.028	-0.061	-0.103	0.396
0.408							.020			0.408
0.426							.026			0.426
0.451							.023			0.451
0.476							.018			0.476
0.501							.015			0.501
0.530	-0.110	-0.127	-0.112	-0.090	-0.031	-0.015	-0.007	-0.055	-0.123	0.530
0.548							.007			0.548
0.566							.024			0.566
0.584	-0.055	-0.115	-0.093	-0.066	-0.004	.012	.025	-0.032	-0.115	0.584
0.602							.029			0.602
0.620							.031			0.620
0.638	-0.038	-0.104		-0.057	.014				-0.101	0.638
0.655							.034			0.655
0.673							.031			0.673
0.691	-0.019	-0.115	-0.096	-0.069	-0.012	.013	.032	-0.038	-0.098	0.691
0.709							.028			0.709
0.727							.027			0.727
0.745	-0.012	-0.114		-0.069	-0.000		.033		-0.086	0.745
0.763							.028			0.763
0.781							.017			0.781
0.799							.020			0.799
0.817	-0.004	-0.098	-0.098	-0.076	-0.002	.022	.035	-0.030	-0.069	0.817
0.834							.027			0.834
0.852							.036			0.852
0.870							.037			0.870
0.888							.041			0.888
0.906							.038			0.906
0.921	-0.215									0.921
0.924	-0.000	-0.069		-0.067	.000		.041		-0.065	0.924
0.947							.039			0.947
0.964							.045			0.964
0.981							.037	-0.013	-0.050	0.981
1.000	-0.222						.050			1.000

Orifice station, y/r	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/r	Cp top	Cp side
0.10	.482	.572	.031	.912	.449
0.30	.700	1.335	.094	1.660	1.070
0.50	.921	1.687	.156	1.592	1.527
0.70	1.158	1.710	.219	1.541	1.627
0.90	1.390	1.694	.281	1.540	1.500
1.10	1.574	1.599	.344	1.513	1.485
1.30	1.661	1.548	.406	1.493	1.504
1.50	1.650	1.541	.469	1.478	1.506
1.70	1.615	1.521	.750	1.418	
1.90	1.491	1.423	1.000	1.433	
			1.250	1.484	
			1.500	1.530	

TABLE 2. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 1.57$ (a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.564			-0.055
-0.037							1.55			-0.037
-0.018							1.125			-0.018
.002							1.077			.002
.021							1.011			.021
.041							1.041			.041
.060							1.020			.060
.075	.291	.037	-.001	-.025	-.017	-.011	1.010			.075
.085	.316	.053	.016	.005	.004	.011	1.019			.085
.096	.494	.097	.056	.056	.060	.065	1.061	-.003	.015	.096
.126							1.025			.126
.143							1.205			.143
.154	-.005	-.172	-.193	-.196	-.186	-.175	1.167	-.208		.154
.171							1.140			.171
.189							1.112			.189
.206	.022	-.152	-.163	-.166	-.113	-.097	1.088	-.128		.206
.223							1.064			.223
.241							1.048			.241
.258	.024	-.148	-.157	-.130	-.065	-.046	1.036	-.079	-.150	.258
.275							1.026			.275
.293							1.016			.293
.310	.023	-.150	-.139	-.084	-.047	-.028	1.011	-.053	-.147	.310
.327							1.008			.327
.344							1.005			.344
.362	.020	-.137	-.102	-.062	-.052	-.029	1.001	-.046	-.126	.362
.379							1.002			.379
.396	.016	-.118	-.093	-.048	-.048	-.028	1.008	-.040	-.107	.396
.408							1.000			.408
.426	.235	.072	.095	.084	.064	.146	1.249	.091	.094	.426
.451	.220	.066	.060	.072	.062	.099	1.158		.070	.451
.476	.211	.048	.039	.061	.046	.086	1.122	.078	.056	.476
.501	.200	.038	.033	.047	.024	.059	1.087	.051	.045	.501
.530	-.001	-.132	-.138	-.113	-.093	-.134	1.113	-.081	-.130	.530
.548							1.122			.548
.566							1.092			.566
.584	.026	-.120	-.104	-.086	-.094	-.089	1.056	-.076	-.118	.584
.602							1.052			.602
.620							1.045			.620
.638	.034	-.103		-.083	-.053		1.030		-.091	.638
.655							1.024			.655
.673							1.023			.673
.691	.032	-.125	-.111	-.072	-.046	-.050	1.024	-.034	-.096	.691
.709							1.020			.709
.727							1.018			.727
.745	.028	-.120		-.054	-.037		1.017			.745
.763							1.028			.763
.781							1.020			.781
.799							1.026			.799
.817	.046	-.098	-.066	-.039	-.034	-.030	1.007	-.026	-.072	.817
.834							1.014			.834
.852							1.010			.852
.870							1.003			.870
.888							1.004			.888
.906							1.002			.906
.921	-.222	.052	-.073		-.032	-.031		.001		.921
.924							1.003			.924
.947							1.002			.947
.964							1.017			.964
.981	.070	-.059	-.022			-.024	1.022		-.011	.981
1.000	-.228						1.017		-.045	1.000

Orifice station, y_{in}	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y_{in}	Cp top	Cp side
.010	1.209	.552	.031	1.871*	.430
.030	1.567	1.242	.094	1.591	1.070
.050	1.693	1.643	.156	1.650	1.535
.070	1.730	1.706	.219	1.581	1.633
.090	1.747	1.696	.281	1.593	1.512
.110	1.755	1.618	.344	1.578	1.483
.130	1.745	1.554	.406	1.584	1.506
.150	1.731	1.546	.469	1.586	1.494
.170	1.707	1.529	.750	1.574	
.190	1.606	1.436	1.000	1.577	
			1.250	1.577	
			1.500	1.583	

TABLE 2. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.553			-0.055
-0.037							2.01			-0.037
-0.018							1.70			-0.018
.002							1.17			.002
.021							0.68			.021
.041							0.43			.041
.060							0.18			.060
.075	.216	.071	.047	.025	.020	.019	.050	.071		.075
.085	.243	.089	.068	.055	.043	.043	.070		.118	.085
.096	.397	.138	.112	.110	.102	.101	.093			.096
.126		.141	.121	.106	.098	.092	.092	.105	.130	.126
.143							-1.192			.143
.154	-0.059	-0.137	-0.148	-0.159	-0.167	-0.164	-0.158	-0.165		.154
.171							-1.139			.171
.189							-0.111			.189
.206	-0.021	-0.107	-0.113	-0.127	-0.100	-0.091	-0.090	-0.108		.206
.223							-0.064			.223
.241							-0.048			.241
.258	-0.012	-0.089	-0.090	-0.079	-0.049	-0.037	-0.037	-0.055	-0.088	.258
.275							-0.031			.275
.293							-0.024			.293
.310	-0.012	-0.069	-0.064	-0.050	-0.030	-0.019	-0.016	-0.033	-0.070	.310
.327							-0.012			.327
.344							-0.009			.344
.362	-0.003	-0.057	-0.051	-0.036	-0.019	-0.008	-0.001	-0.022	-0.056	.362
.379							-0.001			.379
.396	-0.009	-0.053	-0.046	-0.032	-0.014	-0.008	0.001	-0.015	-0.049	.396
.408							0.21			.408
.426	.204	.143	.152	.145	.132	.148	.166	.138	.147	.426
.451	.191	.129	.119	.115	.118	.124	.140	.124	.124	.451
.476	.180	.104	.094	.095	.098	.103	.115	.103	.107	.476
.501	.167	.093	.084	.078	.066	.077	.094	.079	.091	.501
.530	-0.029	-0.092	-0.101	-0.108	-0.081	-0.085	-0.097	-0.081	-0.095	.530
.548							-0.072			.548
.566							-0.059			.566
.584	.001	-0.074	-0.074	-0.064	-0.065	-0.064	-0.053	-0.061	-0.075	.584
.602							-0.044			.602
.620							-0.037			.620
.638	.003	-0.053		-0.051	-0.041				-0.051	.638
.655							-0.021			.655
.673							-0.020			.673
.691	.003	-0.064	-0.060	-0.050	-0.031	-0.029	-0.008	-0.023	-0.050	.691
.709							-0.012			.709
.727							-0.003			.727
.745	.001	-0.057		-0.035	-0.021		-0.008		-0.045	.745
.763							-0.006			.763
.781							-0.014			.781
.799							-0.004			.799
.817	.021	-0.040	-0.033	-0.023	-0.013	-0.008	-0.004	-0.009	-0.025	.817
.834							-0.001			.834
.852							.005			.852
.870							.007			.870
.888							.010			.888
.906							.012			.906
.921	-0.159	.030	-0.030		-0.015	-0.003				.921
.924							.015			.924
.947							.012			.947
.964							.017			.964
.981	.054	-0.012	-0.010		-0.002		.001			.981
1.000	.054	-0.165					.007			1.000

Orifice station, y,in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side	
.010	1.059	.595	.031	.761	.532	
.030	1.483	1.268	.094	1.329	.965	
.050	1.672	1.652	.156	1.612	1.318	
.070	1.732	1.721	.219	1.581	1.557	
.090	1.753	1.716	.281	1.563	1.583	
.110	1.753	1.665	.344	1.565	1.542	
.130	1.739	1.597	.406	1.565	1.542	
.150	1.715	1.580	.469	1.568	1.542	
.170	1.690	1.561	.750	1.558		
.190	1.581	1.459	1.000	1.566		
			1.250	1.559		
			1.500	1.550		

TABLE 2. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	130	145	165	180	225	
-0.055							1.554			-0.055
-0.037							.251			-0.037
-0.018							.214			-0.018
.002							.154			.002
.021							.108			.021
.041							.078			.041
.060							.048			.060
.075	.164	.088	.075	.057	.053	.050	.079	.097		.075
.085	.189	.107	.100	.090	.076	.074	.123	.127	.168	.085
.096	.328	.161	.144	.148	.139	.133				.096
.126		.160	.149	.137	.131	.125	.123	.137	.151	.126
.143							.176			.143
.154	-0.094	-0.120	-0.130	-0.138	-0.145	-0.146	.140	.144		.154
.171							.123			.171
.189							.102			.189
.206	-0.049	-0.081	-0.083	-0.095	-0.086	-0.086	.086	.087		.206
.223							.068			.223
.241							.055			.241
.258	-0.031	-0.053	-0.058	-0.056	-0.048	-0.042	.043	.048	-0.052	.258
.275							.032			.275
.293							.025			.293
.310	-0.024	-0.036	-0.035	-0.027	-0.024	-0.018	.019	.023	-0.038	.310
.327							.016			.327
.344							.014			.344
.362	-0.012	-0.022	-0.022	-0.019	-0.017	-0.012	.010	.014	-0.024	.362
.379							.009			.379
.396	-0.015	-0.021	-0.021	-0.019	-0.008	-0.001	.006	.013	-0.019	.396
.408							.021			.408
.426	.190	.174	.175	.173	.162	.164	.163	.187	.170	.426
.451	.173	.155	.150	.143	.138	.137	.141		.149	.451
.476	.158	.131	.122	.121	.119	.118	.121	.126	.133	.476
.501	.141	.116	.112	.104	.100	.099	.102	.110	.114	.501
.530	-0.051	-0.073	-0.078	-0.083	-0.079	-0.081	.079	.074	-0.077	.530
.548							.061			.548
.566							.046			.566
.584	-0.024	-0.048	-0.051	-0.050	-0.048	-0.054	.050	.037	-0.051	.584
.602							.042			.602
.620							.034			.620
.638	-0.015	-0.029		-0.032	-0.028				-0.029	.638
.655							.022			.655
.673							.021			.673
.691	-0.010	-0.035	-0.034	-0.032	-0.030	-0.025	.012	-0.016	-0.025	.691
.709							.012			.709
.727							.008			.727
.745	-0.001	-0.027		-0.020	-0.012		.009		-0.017	.745
.763							.007			.763
.781							.012			.781
.799							.003			.799
.817	.009	-0.015	-0.014	-0.010	-0.005	-0.002	.001	-0.002	-0.000	.817
.834							.002			.834
.852							.003			.852
.870							.006			.870
.888							.008			.888
.906							.010			.906
.921										.921
.924	.018	.000		-0.008	.004		.012		-0.011	.924
.947							.009			.947
.964							.013			.964
.981	.049	.010	.004		.008	.009	.008	.003	.006	.981
1.000	-0.105									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp _{top}	Cp _{side}	Orifice station, y,in.	Cp _{top}	Cp _{side}
.010	.915	.606	.031	.654	.545
.030	1.343	1.171	.094	1.046	.824
.050	1.622	1.587	.156	1.337	1.051
.070	1.722	1.716	.219	1.522	1.270
.090	1.746	1.728	.281	1.552	1.435
.110	1.746	1.692	.344	1.571	1.517
.130	1.727	1.624	.406	1.563	1.551
.150	1.703	1.590	.469	1.566	1.561
.170	1.668	1.564	.750	1.556	
.190	1.555	1.463	1.000	1.561	
			1.250	1.558	
			1.500	1.558	

TABLE 2. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
-.055							1.555			-.055
-.037							.303			-.037
-.018							.265			-.018
.002							.205			.002
.021							.150			.021
.041							.117			.041
.060										.060
.075	.093	.090	.079	.086	.085	.084				.075
.085	.113	.117	.114	.108	.109	.114	.116			.085
.096	.143	.167	.161	.173	.175	.170	.158	.174	.183	.096
.126							.162	.164	.160	.126
.143							.162			.143
.154	-.122	-.116	-.120	-.122	-.124	-.125	-.119	-.129		.154
.171							.105			.171
.189							.088			.189
.206	-.071	-.072	-.069	-.080	-.072	-.072	-.073	-.074		.206
.223							.056			.223
.241							.049			.241
.258	-.043	-.043	-.044	-.044	-.041	-.037	-.042	-.041	-.039	.258
.275							.039			.275
.293							.038			.293
.310	-.029	-.022	-.025	-.024	-.032	-.026	-.026	-.026	-.025	.310
.327							.022			.327
.344							.013			.344
.362	-.012	-.018	-.016	-.011	-.013	-.008	-.008	-.010	-.016	.362
.379							.011			.379
.396	-.014	-.009	-.009	-.010	-.011	-.009	-.002	-.012	-.008	.396
.408							.011			.408
.426	.178	.182	.184	.181	.177	.182	.178	.179	.179	.426
.451	.158	.161	.160	.160	.159	.157	.160		.156	.451
.476	.140	.141	.139	.141	.144	.139	.140	.144	.140	.476
.501	.118	.126	.130	.123	.120	.117	.118	.121	.127	.501
.530	-.070	-.064	-.068	-.072	-.069	-.072	-.072	-.068	-.064	.530
.548							.056			.548
.566							.043			.566
.584	-.040	-.040	-.039	-.041	-.044	-.046	-.040	-.036	-.041	.584
.602							.034			.602
.620							.023			.620
.638	-.025	-.026		-.021	-.014					.638
.655							.020			.655
.673							.019			.673
.691	-.016	-.018	-.019	-.022	-.029	-.026	-.015	-.020	-.012	.691
.709							.016			.709
.727							.015			.727
.745	-.009	-.016		-.015	-.013		.010		-.006	.745
.763							.006			.763
.781							.011			.781
.799							.007			.799
.817	.004	-.005	-.005	-.004	-.009	-.005	-.001		.009	.817
.834							.005			.834
.852							.002			.852
.870							.002			.870
.888							.008			.888
.906							.006			.906
.921	-.081									.921
.924	.005	.004		-.007	.005		.008			.924
.947							.011			.947
.964							.012			.964
.981	.054	.028	.014		.011	.007	.006		.011	.981
1.000	-.085							.007		1.000

Orifice station, y , in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side	
.010	.776	.603	.031	.509	.535	
.030	1.167	1.180	.094	.738	.746	
.050	1.510	1.592	.156	.910	.911	
.070	1.688	1.713	.219	1.075	1.079	
.090	1.737	1.728	.281	1.214	1.230	
.110	1.739	1.696	.344	1.346	1.355	
.130	1.720	1.631	.406	1.449	1.457	
.150	1.687	1.594	.469	1.521	1.527	
.170	1.649	1.567	.750	1.564		
.190	1.530	1.471	1.000	1.563		
			.1250	1.563		
			1.500	1.558		

TABLE 2. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{y}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{y}{T}$
	0	90	105	120	150	165	180	225	270	
-.055							1.554			-.055
-.037							.362			-.037
-.018							.329			-.018
.002							.259			.002
.021							.206			.021
.041							.165			.041
.060							.127			.060
.075	.082	.090	.096	.096	.122	.127	.127			.075
.085	.107	.109	.123	.132	.143	.148	.156			.085
.096	.232	.159	.167	.192	.212	.211	.202			.096
.126		.163	.172	.180	.201	.204	.209	.185	.150	.126
.143							.130			.143
.154	-.142	-.122	-.119	-.112	-.102	-.098	-.094	-.115		.154
.171							.082			.171
.189							.066			.189
.206	-.083	-.078	-.068	-.074	-.059	-.055	-.054	-.065		.206
.223							.041			.223
.241							.038			.241
.258	-.048	-.052	-.052	-.047	-.033	-.027	-.031	-.040	-.048	.258
.275							.031			.275
.293							.028			.293
.310	-.022	-.034	-.033	-.027	-.029	-.025	-.024	-.026	-.035	.310
.327							.020			.327
.344							.018			.344
.362	-.011	-.027	-.025	-.020	-.017	-.009	-.007	-.019	-.026	.362
.379							.007			.379
.396	-.009	-.019	-.019	-.016	-.009	-.006	-.000	-.016	-.017	.396
.408							.009			.408
.426	.168	.175	.180	.181	.184	.192	.188	.182	.169	.426
.451	.146	.154	.156	.161	.169	.167	.172		.146	.451
.476	.123	.134	.136	.144	.160	.163	.166	.149	.132	.476
.501	.101	.119	.129	.132	.138	.138	.142	.136	.118	.501
.530	-.076	-.071	-.071	-.070	-.058	-.058	-.057	-.061	-.071	.530
.548							.042			.548
.566							.028			.566
.584	-.050	-.047	-.042	-.040	-.034	-.034	-.026	-.031	-.048	.584
.602							.021			.602
.620							.014			.620
.638	-.029	-.033		-.021	-.007				-.032	.638
.655							.011			.655
.673							.012			.673
.691	-.017	-.027	-.027	-.025	-.027	-.021	-.009	-.023	-.021	.691
.709							.011			.709
.727							.010			.727
.745	-.007	-.023		-.019	-.011				-.014	.745
.763							.004			.763
.781							.006			.781
.799							.010			.799
.817	.004	-.014	-.011	-.013	-.008	-.003	-.009	-.005	.001	.817
.834							.002			.834
.852							.005			.852
.870							.005			.870
.888							.011			.888
.906							.009			.906
.921	-.094						.012			.921
.924	.010	.000			-.011	.000		.012		.924
.947							.016			.947
.964							.009			.964
.981	.064	.018	.003			.010	.009	.009	.007	.981
1.000	-.101									1.000

Orifice station, $y, \text{in.}$	Nose rake			Base rake		
	Cp top	Cp side	Cp top	Orifice station, $y, \text{in.}$	Cp top	Cp side
.010	.657	.600	.469	.031	.546	
.030	.969	1.243	.657	.094	.807	
.050	1.292	1.643	.764	.156		1.020
.070	1.569	1.721	.853	.219	1.240	
.090	1.709	1.724	.925	.281	1.413	
.110	1.736	1.683	.997	.344	1.510	
.130	1.714	1.611	1.070	.406	1.556	
.150	1.683	1.587	1.140	.469	1.567	
.170	1.640	1.561	1.404	.750		
.190	1.514	1.466	1.548	1.000		
				1.250	1.563	
				1.500	1.564	

TABLE 2. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.556			-0.055
-0.037							•441			-0.037
-0.018							•401			-0.018
0.002		•222					•323	•298		0.002
0.021							•268			0.021
0.041							•225			0.041
0.060							•179			0.060
0.075	•052	•075	•096	•112	•165	•177	•209	•163		0.075
0.085	•078	•139	•159	•204	•256	•265	•258	•216	•114	0.085
0.096	•198									0.096
0.126		•142	•165	•190	•242	•252	•264	•204	•129	0.126
0.143							•096			0.143
0.154	-0.161	-0.136	-0.125	-0.107	-0.076	-0.066	-0.058	-0.101		0.154
0.171							•051			0.171
0.189							•037			0.189
0.206	-0.090	-0.096	-0.082	-0.076	-0.040	-0.031	-0.027	-0.061		0.206
0.223							•016			0.223
0.241							•016			0.241
0.258	-0.043	-0.084	-0.076	-0.058	-0.024	-0.009	•013	-0.042	-0.080	0.258
0.275							•014			0.275
0.293							•012			0.293
0.310	-0.016	-0.072	-0.066	-0.047	-0.022	-0.011	-0.006	-0.037	-0.071	0.310
0.327							•008			0.327
0.344							•010			0.344
0.362	-0.006	-0.059	-0.055	-0.046	-0.021	-0.009	-0.005	-0.036	-0.053	0.362
0.379							•001			0.379
0.396	-0.003	-0.051	-0.052	-0.041	-0.011	-0.001	•008	-0.032	-0.046	0.396
0.408							•002			0.408
0.426	•172	•147	•151	•158	•185	•203	•202	•171	•148	0.426
0.451	•142	•129	•133	•143	•172	•179	•188		•123	0.451
0.476	•116	•109	•117	•134	•168	•176	•181	•148	•108	0.476
0.501	•092	•094	•112	•128	•157	•163	•170	•147	•095	0.501
0.530	-0.092	-0.091	-0.084	-0.075	-0.045	-0.039	-0.036	-0.056	-0.089	0.530
0.548							•022			0.548
0.566							•006			0.566
0.584	-0.038	-0.071	-0.059	-0.048	-0.021	-0.015	-0.005	-0.029	-0.073	0.584
0.602							•002			0.602
0.620							•001			0.620
0.638	-0.029	-0.059		-0.033	•002				-0.057	0.638
0.655							•004			0.655
0.673							•003			0.673
0.691	-0.016	-0.055	-0.051	-0.040	-0.023	-0.010	•004		-0.048	0.691
0.709							•000			0.709
0.727							•000			0.727
0.745	•001	-0.052		-0.037	-0.008		•007		-0.040	0.745
0.763							•003			0.763
0.781							•005			0.781
0.799							•001			0.799
0.817	•007	-0.040	-0.040	-0.034	-0.008	•003	•011	-0.015	-0.022	0.817
0.834							•005			0.834
0.852							•014			0.852
0.870							•014			0.870
0.888							•023			0.888
0.906							•014			0.906
0.921	-0.149						•023			0.921
0.924	•011	-0.023			-0.031	•003		•020		0.924
0.947							•026			0.947
0.964							•018	-0.000	-0.016	0.964
0.981	•046	-0.010	-0.019			•011	•016			0.981
1.000	-0.163						•018			1.000

Orifice station, $y, \text{in.}$	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, $y, \text{in.}$	Cp top	Cp side	
•010	•548	•585	•031	•706	•544	
•030	•789	1.260	•094	1.218	•952	
•050	1.039	1.661	•156	1.413	1.295	
•070	1.306	1.733	•219	1.420	1.546	
•090	1.555	1.724	•281	1.429	1.576	
•110	1.698	1.668	•344	1.457	1.545	
•130	1.716	1.593	•406	1.478	1.545	
•150	1.690	1.580	•469	1.495	1.545	
•170	1.644	1.557	•750	1.529		
•190	1.523	1.457	1.000	1.558		
			1.250	1.567		
			1.500	1.563		

TABLE 2. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 1.57$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.4557			-0.055
-0.037							1.537			-0.037
-0.018							1.492			-0.018
.002							1.404			.002
.021							1.350			.021
.041							1.303			.041
.060							1.251			.060
.075	.024	.058	.103	.155	.239	.267	.282	.196		.075
.085	.051	.101	.138	.210	.312	.335	.332	.241	.013	.085
.096	.162									.096
.126		.100	.142	.194	.294	.317	.335	.230	.086	.126
.143							.050			.143
.154	-0.180	-0.164	-0.140	-0.103	-0.043	-0.020	-0.008	-0.087		.154
.171							.002			.171
.189							.005			.189
.206	-0.088	-0.143	-0.114	-0.085	-0.011	.008	.015	-0.051		.206
.223							.025			.223
.241							.022			.241
.258	-0.037	-0.145	-0.117	-0.080	-0.006	.021	.021	-0.045	-0.146	.258
.275							.019			.275
.293							.018			.293
.310	-0.012	-0.145	-0.128	-0.081	-0.010	.015	.017	-0.047	-0.143	.310
.327							.024			.327
.344							.021			.344
.362	-0.002	-0.132	-0.126	-0.092	-0.017	.010	.014	-0.056	-0.119	.362
.379							.019			.379
.396	-0.001	-0.112	-0.120	-0.092	-0.009	.016	.020	-0.058	-0.101	.396
.408							.030			.408
.426	.253	.085	.077	.096	.184	.226	.18	.138	.095	.426
.451	.155	.074	.077	.095	.173	.199	.231		.074	.451
.476	.113	.053	.072	.100	.174	.200	.212	.134	.060	.476
.501	.085	.041	.073	.103	.168	.190	.202	.141	.052	.501
.530	-0.110	-0.129	-0.115	-0.090	-0.030	-0.015	-0.007	-0.054	-0.125	.530
.548							.009			.548
.566							.024			.566
.584	-0.055	-0.115	-0.093	-0.068	-0.006	.012	.025	-0.030	-0.115	.584
.602							.030			.602
.620							.041			.620
.638	-0.038	-0.105		-0.052	-0.007		.035			.638
.655							.031			.655
.673							.033			.673
.691	-0.024	-0.113	-0.098	-0.071	-0.015	.011	.029			.691
.709							.031			.709
.727							.027			.727
.745	-0.016	-0.112		-0.068	-0.002		.031			.745
.763							.011			.763
.781							.021			.781
.799							.035			.799
.817	-0.004	-0.098	-0.097	-0.072	-0.003	.022	.025			.817
.834							.036			.834
.852							.040			.852
.870							.036			.870
.888							.039			.888
.906										.906
.921	-0.208									.921
.924	-0.008	-0.071		-0.069	.000		.043			.924
.947							.040			.947
.964							.046			.964
.981	.035	-0.054	-0.071		.010	.029	.037	.013	-0.052	.981
1.000	-0.215									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.473	.530	.031	.882	.451
.030	.672	1.216	.094	1.643	1.068
.050	.876	1.650	.156	1.555	1.527
.070	1.089	1.711	.219	1.514	1.630
.090	1.310	1.700	.281	1.521	1.508
.110	1.502	1.622	.344	1.498	1.489
.130	1.637	1.547	.406	1.484	1.507
.150	1.657	1.539	.469	1.471	1.507
.170	1.619	1.517	.750	1.415	
.190	1.494	1.418	1.000	1.433	
			1.250	1.478	
			1.500	1.313	

TABLE 3. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 2.98$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.740			-0.055
-0.037							.088			-0.037
-0.018							.051			-0.018
.002							.028			.002
.029							.012			.029
.041							.000			.041
.060							.008			.060
.075	.238	.054	.025	.011	-.004	-.002	.000	-.001	.062	.075
.085	.260	.077	.042	.027	.012	.018	.014	.019		.085
.096	.275								.062	.096
.126		.077	.047	.024	.019	.021	.024	.019	.070	.126
.143							-.067			.143
.154		.055	-.050	-.066	-.075	-.076	-.072	-.078	-.052	.154
.171							-.073			.171
.189							-.068			.189
.206		.055	-.053	-.069	-.076	-.065	-.060	-.053	-.053	.206
.223							-.053			.223
.241							-.046			.241
.258		.053	-.058	-.073	-.080	-.053	-.045	-.041	-.055	.258
.275							-.035			.275
.293							-.032			.293
.310		.052	-.063	-.079	-.082	-.046	-.037	-.029	-.059	.310
.327							-.026			.327
.344							-.023			.344
.362		.047	-.069	-.085	-.070	-.045	-.034	-.020	-.051	.362
.379							-.019			.379
.396		.043	-.073	-.089	-.061	-.047	-.034	-.016	-.052	.396
.408							-.016			.408
.426		.168	-.027	-.036	-.014	-.009	.020	.074	-.010	.426
.451		.172	-.019	-.042	-.003	-.002	.022	.080	-.002	.451
.476		.174	-.018	-.041	.001	-.009	.020	.075	.002	.476
.501		.168	-.039	.004	-.015	.018	.060	.060	-.016	.501
.530		.041	-.074	-.077	-.046	-.061		-.035	-.037	.530
.548							-.036			.548
.566							-.036			.566
.584		.045	-.077	-.060	-.047	-.072	-.041	-.037	-.048	.584
.602							-.037			.602
.620							-.034			.620
.638		.044	-.077		-.053	-.063		-.030		.638
.655							-.028			.655
.673							-.027			.673
.691		.040	-.080	-.056		-.047	-.037		-.044	.691
.709							-.027			.709
.727							-.026			.727
.745		.043	-.071		-.048	-.040		-.025		.745
.763							-.025			.763
.781							-.027			.781
.799							-.027			.799
.817		.042	-.081	-.050	-.041	-.047	-.035		-.043	.817
.834							-.024			.834
.870							-.026			.870
.888							-.026			.888
.921		-.117						-.025		.921
.924		.040	-.064		-.043			-.025		.924
.947								-.023		.947
.964								-.023		.964
.981		.057	-.052	-.041		-.045	-.040		-.039	.981
.984						-.047			-.047	.984
.998								-.024		.998
1.000		-.117								1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.164	1.358	.031	1.762	.102
.030	2.858	1.997	.094	1.756	.437
.050	3.545	2.639	.156	1.940	.976
.070	2.854	1.991	.219	1.966	1.170
.090	2.764	1.947	.281	1.951	1.310
.110	2.764	1.973	.344	1.935	1.330
.130	2.781	2.002	.406	1.928	1.324
.150	2.793	2.018	.469	1.944	1.296
.170	2.803	2.023	.750	1.923	
.190	2.703	1.944	1.000	1.930	
			1.250	1.930	
			1.500	1.799	

TABLE 3. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 2.98$ - Continued
(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-.055							1.761			-.055
-.037							.131			-.037
-.018							.092			-.018
.002							.060			.002
.029							.042			.029
.041							.021			.041
.060							.012			.060
.075	.167	.067	.051	.035	.021	.020	.021	.030		.075
.085	.186	.092	.071	.062	.042	.046	.040	.048	.082	.085
.096	.202									.096
.126		.095	.078	.061	.050	.046	.048	.051	.091	.126
.143							.060			.143
.154	.015	-.043	-.054	-.060	-.065		.065	-.063	-.040	.154
.171							.063			.171
.189							.058			.189
.206	.016	-.040	-.050	-.055	-.053	-.051	.054	-.055	-.040	.206
.223							.048			.223
.241							.043			.241
.258	.018	-.040	-.048	-.051	-.042	-.038	-.038	-.047	-.036	.258
.275							.033			.275
.293							.030			.293
.310	.017	-.042	-.049	-.049	-.032	-.027	-.026	-.040	-.040	.310
.327							.023			.327
.344							.020			.344
.362	.015	-.044	-.049	-.043	-.025	-.020	-.018	-.031	-.042	.362
.379							.016			.379
.396	.011	-.047	-.050	-.038	-.022	-.016	-.014	-.026	-.044	.396
.408							.013			.408
.426	.122	.017	.019	.031	.041	.050	.061	.037	.023	.426
.451	.127	.027	.028	.039	.055	.061	.075	.047	.032	.451
.476	.124	.027	.028	.036	.050	.054	.067	.048	.033	.476
.501	.123		.027	.033	.044	.048	.059		.034	.501
.530	.008	-.047	-.046	-.038	-.022		.024	-.026	-.042	.530
.548							.029			.548
.566							.027			.566
.584	.012	-.045	-.042	-.035	-.031	-.030	-.027	-.026	-.041	.584
.602							.026			.602
.620							.024			.620
.638	.012	-.043		-.032	-.027		.021		-.039	.638
.655							.019			.655
.673							.017			.673
.691	.009	-.042	-.037		-.024	-.021	-.015	-.024	-.039	.691
.709							.013			.709
.727							.011			.727
.745	.011	-.034		-.026	-.021		.010		-.039	.745
.763							.012			.763
.781							.010			.781
.799							.007			.799
.817	.012	-.039	-.031	-.021	-.017	-.013	-.007	-.019	-.038	.817
.834							.008			.834
.870							.006			.870
.888							.005			.888
.921	-.111		-.035		-.019					.921
.924	.011						.004			.924
.947							.006			.947
.964							.004			.964
.981	.022	-.032	-.023			-.017	-.011		-.033	.981
.984							.005		-.039	.984
.998										.998
1.000	-.112									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.953	.377	.031	.601	.296
.030	2.532	1.620	.094	1.371	.760
.050	3.377	2.601	.156	1.639	1.024
.070	2.693	2.307	.219	1.770	1.294
.090	2.580	2.143	.281	1.793	1.467
.110	2.592	2.083	.344	1.785	1.513
.130	2.609	2.077	.406	1.786	1.534
.150	2.609	2.088	.469	1.795	1.544
.170	2.605	2.090	.750	1.789	
.190	2.502	2.012	1.000	1.798	
			1.250	1.806	
			1.500	1.667	

TABLE 3. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT M = 2.98 - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-.055							1.760			-.055
-.037							.177			-.037
-.018							.135			-.018
.002		.154					.095	.119		.002
.029							.064			.029
.041							.041			.041
.060		.068	.059	.050	.044	.042	.040			.060
.075	.119	.076	.070	.058	.051	.051	.051	.057		.075
.085	.136	.098	.088	.085	.072	.070	.069	.077	.094	.085
.096	.149									.096
.126		.106	.099	.086	.078	.074	.073	.083	.100	.126
.143							.057			.143
.154	-.012	-.041	-.044	-.049	-.055		.055	-.051	-.032	.154
.171							.051			.171
.189							.047			.189
.206	-.010	-.033	-.037	-.040	-.041	-.042	-.044	-.040	-.031	.206
.223							.040			.223
.241							.036			.241
.258	-.007	-.028	-.031	-.032	-.033	-.031	-.034	-.032	-.023	.258
.275							.030			.275
.293							.028			.293
.310	-.004	-.025	-.028	-.026	-.025	-.024	-.025	-.025	-.022	.310
.327							.022			.327
.344							.020			.344
.362	-.003	-.022	-.024	-.022	-.019	-.018	-.017	-.019	-.019	.362
.379							.016			.379
.396	-.006	-.021	-.022	-.020	-.015	-.014	-.014	-.017	-.018	.396
.408							.013			.408
.426	.094	.055	.058	.057	.060	.061	.063	.058	.061	.426
.451	.097	.067	.066	.067	.073	.074	.077	.072	.071	.451
.476	.094	.064	.062	.062	.067	.068	.070	.069	.068	.476
.501	.091	.059	.058	.061	.061	.061	.063	.066	.066	.501
.530	-.010	-.025	-.025	-.024	-.020		.018	-.019	-.021	.530
.548							.022			.548
.566							.021			.566
.584	-.007	-.025	-.025	-.024	-.025	-.024	-.020	-.021	-.019	.584
.602							.019			.602
.620							.018			.620
.638	-.006	-.022			-.020	-.019			-.017	.638
.655							.016			.655
.673							.014			.673
.691	-.005	-.019	-.019			-.016			-.015	.691
.709							.010			.709
.727							.009			.727
.745	-.005	-.010			-.014	-.012			-.015	.745
.763							.008			.763
.781							.007			.781
.799							.008			.799
.817	-.002	-.013	-.013	-.010	-.008	-.008	-.004	-.011	-.013	.817
.834							.004			.834
.870							.002			.870
.888							.001			.888
.921	-.106						.001			.921
.924	-.000	-.011			-.010		.003			.924
.947							.002			.947
.964										.964
.981	.008	-.009	-.009			-.006	-.005	-.004	-.006	.981
.984									-.009	.984
.998									-.014	.998
1.000	-.108									1.000

Orifice station, y , in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side	
.010	.799	.304	.031	.453	.335	
.030	2.268	1.667	.094	.961	.652	
.050	3.186	2.681	.156	1.197	.824	
.070	2.526	2.384	.219	1.508	1.013	
.090	2.420	2.203	.281	1.660	1.219	
.110	2.429	2.154	.344	1.689	1.405	
.130	2.448	2.133	.406	1.699	1.561	
.150	2.452	2.123	.469	1.714	1.622	
.170	2.440	2.118	.750	1.717		
.190	2.335	2.039	1.000	1.730		
			1.250	1.741		
			1.500	1.613		

TABLE 3. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 2.98$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-.055							1.759			-.055
-.037							.176			-.037
-.018							.145			-.018
.002							.110			.002
.029							.066			.029
.041							.046			.041
.060							.070			.060
.075	.081	.070	.070	.064	.071	.070	.070			.075
.085	.098	.102	.100	.106	.106	.110	.081	.085		.085
.096	.123						.105	.105		.096
.126		.106	.106	.106	.108	.106	.108	.111	.106	.126
.143							.040			.143
.154							.038	.038		.154
.171							.036			.171
.189							.032			.189
.206							.030	.029		.206
.223							.027			.223
.241							.026			.241
.258							.024	.024		.258
.275							.023			.275
.293							.021			.293
.310							.017			.310
.327							.016			.327
.344							.015			.344
.362							.013			.362
.379							.014			.379
.396							.013			.396
.408							.011			.408
.426	.078	.067	.072	.069	.068	.072	.071	.071		.426
.451	.085	.080	.079	.080	.082	.081	.082	.081		.451
.476	.079	.075	.074	.075	.078	.077	.078	.079		.476
.501	.076		.070	.070	.072	.073	.072			.501
.530	.015		.018	.018	.015		.016	.015		.530
.548							.017			.548
.566							.015			.566
.584							.015			.584
.602							.013			.602
.620							.013			.620
.638							.011			.638
.655							.011			.655
.673							.011			.673
.691							.011			.691
.709							.011			.709
.727							.008			.727
.745							.008			.745
.763							.007			.763
.781							.008			.781
.799							.007			.799
.817							.003			.817
.834							.005			.834
.870							.006			.870
.888							.006			.888
.921							.002			.921
.924							.003			.924
.947							.001			.947
.964							.002			.964
.981							.002			.981
.984							.007			.984
.998							.003			.998
1.000										1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	.795	.300	.031	.446	.332
.030	2.269	1.658	.094	.957	.651
.050	3.181	2.678	.156	1.190	.820
.070	2.520	2.374	.219	1.512	1.017
.090	2.415	2.202	.281	1.662	1.212
.110	2.425	2.152	.344	1.693	1.407
.130	2.441	2.127	.406	1.708	1.554
.150	2.447	2.121	.469	1.713	1.622
.170	2.435	2.113	.750	1.710	
.190	2.332	2.036	1.000	1.726	
			1.250	1.734	
			1.500	1.812	

TABLE 3. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 2.98$ - Continued
(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-0.055							1.758			-0.055
-0.037							.301			-0.037
-0.018							.247			-0.018
.002							.215			.002
.029							.167			.029
.041							.130			.041
.060							.109			.060
.075	.054						.118			.075
.085	.072						.122			.085
.096	.088						.146			.096
.126							.152			.126
.143							.018			.143
.154	-0.046	-0.039	-0.032	-0.029	-0.020		.016	-0.025	-0.032	.154
.171							.014			.171
.189							.012			.189
.206	-0.041	-0.032	-0.026	-0.023	-0.013	-0.010	-0.010	-0.018	-0.030	.206
.223							.008			.223
.241							.007			.241
.258	-0.030	-0.027	-0.023	-0.019	-0.010	-0.007	-0.006	-0.015	-0.023	.258
.275							.006			.275
.293							.006			.293
.310	-0.023	-0.024	-0.021	-0.017	-0.009	-0.007	-0.003	-0.012	-0.023	.310
.327							.003			.327
.344							.005			.344
.362	-0.015	-0.022	-0.018	-0.014	-0.008	-0.006	-0.003	-0.009	-0.021	.362
.379							.004			.379
.396	-0.011	-0.021	-0.019	-0.015	-0.008	-0.006	-0.003	-0.010	-0.018	.396
.408							.004			.408
.426	.070	.054	.065	.068	.080	.086	.089	.077	.058	.426
.451	.081	.067	.070	.077	.093	.093	.096	.086	.070	.451
.476	.073	.064	.066	.074	.087	.092	.093	.079	.067	.476
.501	.065		.064	.069	.085	.088	.091		.065	.501
.530	-0.013	-0.023	-0.022	-0.019	-0.011		.007	-0.013	-0.021	.530
.548							.008			.548
.566							.005			.566
.584	-0.020	-0.023	-0.020	-0.018	-0.011	-0.009	-0.005	-0.010	-0.020	.584
.602							.004			.602
.620							.004			.620
.638	-0.017	-0.019		-0.014	-0.008		.003		-0.016	.638
.655							.002			.655
.673							.003			.673
.691	-0.013	-0.018	-0.016		-0.009	-0.008		-0.008	-0.016	.691
.709							.003			.709
.727							.002			.727
.745	-0.009	-0.009			-0.011	-0.006			-0.014	.745
.763							.001			.763
.781							.002			.781
.799							.005			.799
.817	-0.005	-0.012	-0.011	-0.010	-0.007	-0.006	-0.004	-0.010	-0.012	.817
.834							.003			.834
.870							.000			.870
.888							.001			.888
.921	-0.099						.003			.921
.924	-0.002	-0.012			-0.008		.001			.924
.947							.002			.947
.964							.002			.964
.981	.004	-0.009	-0.009			-0.003	.000	-0.002	-0.010	.981
.984							.000		-0.015	.984
.998							.003			.998
1.000	-0.102									1.000

Orifice station, y , in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side	
.010	.319	.480	.031	.289	.328	
.030	1.589	1.882	.094	.534	.662	
.050	2.120	2.315	.156	.636	.853	
.070	2.240	2.310	.219	.746	1.054	
.090	2.226	2.315	.281	.828	1.273	
.110	2.223	2.294	.344	.906	1.451	
.130	2.219	2.257	.406	.977	1.578	
.150	2.202	2.191	.469	1.040	1.646	
.170	2.177	2.131	.570	1.362		
.190	2.074	2.017	1.000	1.622		
			1.250	1.706		
			1.500	1.577		

TABLE 3. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 2.98$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-.055							1.763			-.055
-.037							.369			-.037
-.018							.313			-.018
.002							.271	.235		.002
.029							.225			.029
.041							.183			.041
.060							.160			.060
.075	.021	.066	.088	.112	.151	.164	.173	.139		.075
.085	.037	.092	.110	.144	.185	.204	.201	.164	.082	.085
.096	.072									.096
.126		.093	.117	.139	.187	.199	.206	.180	.091	.126
.143							.008			.143
.154	-.058	-.044	-.033	-.019	.001		.012	-.002	-.040	.154
.171							.013			.171
.189							.014			.189
.206	-.052	-.041	-.030	-.016	.005	.012	.015	-.004	-.040	.206
.223							.018			.223
.241							.018			.241
.258	-.037	-.040	-.031	-.017	.007	.015	.017	-.004	-.035	.258
.275							.017			.275
.293							.016			.293
.310	-.026	-.042	-.033	-.019	.005	.012	.015	-.004	-.041	.310
.327							.015			.327
.344							.014			.344
.362	-.017	-.045	-.035	-.021	.003	.011	.014	-.006	-.043	.362
.379							.013			.379
.396	-.013	-.046	-.038	-.024	.002	.010	.013	-.008	-.045	.396
.408							.013			.408
.426	.065	.016	.037	.056	.100	.117	.121	.083	.017	.426
.451	.075	.026	.041	.063	.109	.118	.126	.089	.030	.451
.476	.068	.026	.038	.061	.105	.119	.125	.092	.030	.476
.501	.059		.038	.059	.103	.116	.124		.031	.501
.530	-.023	-.045	-.039	-.027	-.001		.010	-.010	-.043	.530
.548							.013			.548
.566							.012			.566
.584	-.027	-.044	-.037	-.027	-.001	.005	.012	-.008	-.041	.584
.602							.011			.602
.620							.013			.620
.638	-.021	-.041		-.024	-.001		.013			.638
.655							.012			.655
.673							.008			.673
.691	-.017	-.041	-.036		-.004	.005		-.009	-.039	.691
.709							.007			.709
.727							.008			.727
.745	-.013	-.032		-.027	-.004		.011		-.038	.745
.763							.010			.763
.781							.010			.781
.799							.015			.799
.817	-.009	-.037	-.035	-.025	.000	.010	.013	-.011	-.037	.817
.834							.015			.834
.870							.012			.870
.888							.013			.888
.921	-.107						.011			.921
.924	-.007	-.035			-.027		.012			.924
.947							.013			.947
.964							.013			.964
.981	.003	-.033	-.035			.009	.012	-.012	-.032	.981
.984							.012		-.037	.984
.998										.998
1.000	-.109									1.000

Orifice station, $y_{in.}$	Nose rake		Base rake		
	Cp _{top}	Cp _{side}	Orifice station, $y_{in.}$	Cp _{top}	Cp _{side}
.010	.256	.503	.031	.525	.286
.030	.479	2.157	.094	1.223	.778
.050	.809	2.538	.156	1.314	1.021
.070	1.536	2.292	.219	1.389	1.300
.090	2.238	2.164	.281	1.428	1.449
.110	2.244	2.104	.344	1.460	1.502
.130	2.059	2.083	.406	1.487	1.518
.150	1.984	2.098	.469	1.516	1.527
.170	1.973	2.082	.750	1.587	
.190	1.890	1.998	1.000	1.660	
			1.250	1.687	
			1.500	1.569	

TABLE 3. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE I WITH FIXED TRANSITION AT $M = 2.98$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-.055							1.757			-.055
-.037							.459			-.037
-.018							.400			-.018
.002	.140						.364	.303		.002
.029							.307			.029
.041							.268			.041
.060							.229			.060
.075	-.001	.050	.086	.123	.201	.220	.246	.180		.075
.085	.011	.076	.112	.168	.209	.232	.277	.204	.069	.085
.096	.038				.247	.278				.096
.126		.075	.117	.162	.249	.271	.282	.217	.068	.126
.143							.049			.143
.154	-.068	-.051	-.030	-.006	.035		.058	.020	-.053	.154
.171							.058			.171
.189							.058			.189
.206	-.060	-.054	-.033	-.008	.039	.052	.058	.022	-.053	.206
.223							.058			.223
.241							.057			.241
.258	-.040	-.059	-.038	-.011	.037	.052	.056	.010	-.053	.258
.275							.054			.275
.293							.053			.293
.310	-.028	-.065	-.043	-.017	.030	.047	.055	.007	-.065	.310
.327							.053			.327
.344							.051			.344
.362	-.020	-.072	-.049	-.021	.028	.043	.049	.007	-.070	.362
.379							.048			.379
.396	-.016	-.075	-.054	-.025	.026	.041	.049	.004	-.073	.396
.408							.047			.408
.426	.073	-.030	.012	.054	.136	.166	.174	.103	-.028	.426
.451	.083	-.023	.014	.058	.143	.164	.177	.108	-.017	.451
.476	.078	-.025	.013	.056	.140	.167	.178	.110	-.017	.476
.501	.063			.014	.055	.139	.162	.177	-.019	.501
.530	-.031	-.076	-.054	-.029	.022		.044	.003	-.074	.530
.548							.047			.548
.566							.048			.566
.584	-.035	-.078	-.056	-.029	.021	.036	.044	.003	-.076	.584
.602							.047			.602
.620							.045			.620
.638	-.029	-.078		-.029	.020		.047		-.078	.638
.655							.047			.655
.673							.046			.673
.691	-.029	-.081	-.059		.016	.036		.001	-.081	.691
.709							.045			.709
.727							.046			.727
.745	-.026	-.073		-.032	.020		.046		-.082	.745
.763							.045			.763
.781							.040			.781
.799							.040			.799
.817	-.026	-.080	-.061	-.035	.019	.037	.046	-.005	-.081	.817
.834							.041			.834
.870							.044			.870
.888							.042			.888
.921	-.116						.041			.921
.924	-.026	-.061		-.040			.035			.924
.947							.040			.947
.964							.043			.964
.981	-.013	-.050	-.069		.015	.035		-.008	-.050	.981
.984							.043		-.062	.984
.998										.998
1.000	-.118									1.000

Orifice station, y,in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side	
.010	.218	.364	.031	.551	.101	
.030	.367	1.897	.094	1.372	.434	
.050	.4545	2.661	.156	1.441	.994	
.070	.748	2.029	.219	1.453	1.173	
.090	1.003	1.963	.281	1.450	1.296	
.110	1.212	1.983	.344	1.452	1.304	
.130	1.365	2.005	.406	1.456	1.265	
.150	1.530	2.017	.469	1.453	1.223	
.170	1.675	2.010	.750	1.448		
.190	1.707	1.938	1.000	1.466		
			1.250	1.521		
			1.500	1.471		

TABLE 4. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH NATURAL TRANSITION AT $M = 1.57$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-.002							1.502			-.002
.008							.634			.008
.018							.410			.018
.029							.177			.029
.039							-.018			.039
.050							-.129			.050
.060							-.137			.060
.075	.259	.028	-.013	-.042	-.070	-.077	-.077	-.067		.075
.085	.320	.075	.029	-.000	-.038	-.041	-.036	-.053	.035	.085
.096	.391									.096
.126		.090	.040	.017	.038	.047	.038	.025	.077	.126
.143							-.212			.143
.154	-.009	-.178	-.206	-.211	-.200	-.184	-.170	-.216		.154
.171							-.146			.171
.189							-.114			.189
.206	.022	-.159	-.172	-.174	-.120	-.099	-.089	-.135		.206
.223							-.064			.223
.241							-.048			.241
.258	.021	-.154	-.162	-.136	-.068	-.043	-.036	-.084	-.158	.258
.275							-.027			.275
.293							-.015			.293
.310	.019	-.154	-.143	-.089	-.050	-.025	-.010	-.058	-.152	.310
.327							-.008			.327
.344							-.006			.344
.362	.016	-.139	-.106	-.067	-.056	-.024	-.002	-.051	-.128	.362
.379							-.003			.379
.396	.012	-.120	-.095	-.053	-.057	-.021	-.006	-.043	-.110	.396
.408							-.001			.408
.426	.229	.068	.089	.079	.056	.185	.229	.085	.090	.426
.451	.216	.062	.057	.069	.052	.109	.150		.066	.451
.476	.208	.046	.037	.058	.036	.093	.113	.074	.052	.476
.501	.197	.036	.031	.043	.015	.066	.081	.048	.042	.501
.530	-.003	-.133	-.139	-.115	-.105	-.134	-.114	-.083	-.132	.530
.548							-.122			.548
.566							-.083			.566
.584	.025	-.121	-.106	-.091	-.096	-.084	-.061	-.078	-.120	.584
.602							-.053			.602
.620							-.046			.620
.638	.032	-.104		-.086	-.055				-.094	.638
.655							-.031			.655
.673							-.025			.673
.691	.030	-.128	-.113	-.074	-.049	-.049	-.024	-.036	-.100	.691
.709							-.025			.709
.727							-.022			.727
.745	.027	-.121		-.057	-.040		-.020			.745
.763							-.019			.763
.781							-.029			.781
.799							-.022			.799
.817	.044	-.099	-.068	-.041	-.037	-.033	-.009	-.028	-.075	.817
.834							-.016			.834
.852							-.012			.852
.870							-.006			.870
.888							-.007			.888
.906							-.004			.906
.921	-.210						-.000			.921
.924	.050	-.075		-.034	-.034		-.000		-.077	.924
.947							-.000			.947
.964							-.033			.964
.981	.067	-.060	-.021		-.013	-.021	-.021	-.002	-.048	.981
1.000	-.217									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.392	.991	.031	.859	.423
.030	1.532	1.352	.094	1.575	1.058
.050	1.574	1.444	.156	1.641	1.528
.070	1.594	1.485	.219	1.573	1.621
.090	1.615	1.489	.281	1.584	1.504
.110	1.630	1.490	.344	1.572	1.477
.130	1.625	1.494	.406	1.574	1.496
.150	1.627	1.490	.469	1.576	1.486
.170	1.610	1.472	.750	1.569	
.190	1.544	1.383	1.000	1.571	
			1.250	1.573	
			1.500	1.582	

TABLE 4. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE II WITH NATURAL TRANSITION AT $M = 1.57$ - Continued(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
-0.002							1.543			-0.002
.008							.729			.008
.018							.502			.018
.029							.245			.029
.039							.037			.039
.050							.079			.050
.060							.083			.060
.075	.173	.049	.022	.000	.018	.026	.032	-.010		.075
.085	.229	.100	.092	.061	.019	.018	.014	.015	.117	.085
.096	.294									.096
.126		.126	.099	.084	.086	.081	.062	.087	.115	.126
.143							.206			.143
.154	-.064	-.146	-.156	-.162	-.180	-.176	.163	-.172		.154
.171							.142			.171
.189							.112			.189
.206	-.024	-.108	-.118	-.127	-.097	-.087	-.084	-.105		.206
.223							.060			.223
.241							.049			.241
.258	-.017	-.086	-.085	-.075	-.051	-.039	-.040	-.056	-.085	.258
.275							.032			.275
.293							.024			.293
.310	-.010	-.065	-.065	-.051	-.030	-.018	-.016	-.032	-.068	.310
.327							.010			.327
.344							.005			.344
.362	-.009	-.058	-.047	-.031	-.019	-.010	-.006	-.017	-.055	.362
.379							.007			.379
.396	-.008	-.049	-.044	-.037	-.018	-.009	.001	-.022	-.047	.396
.408							.014			.408
.426	.201	.141	.143	.140	.136	.155	.169	.136	.145	.426
.451	.186	.123	.117	.116	.120	.127	.140	.120	.120	.451
.476	.178	.105	.095	.095	.097	.103	.113	.102	.106	.476
.501	.165	.093	.085	.077	.069	.082	.097	.076	.093	.501
.530	-.031	-.091	-.100	-.103	-.079	-.088	-.098	-.074	-.094	.530
.548							.080			.548
.566							.057			.566
.584	-.004	-.072	-.066	-.066	-.063	-.061	-.051	-.061	-.073	.584
.602							.043			.602
.620							.034			.620
.638	-.004	-.057			-.051	-.039				.638
.655							.016			.655
.673							.019			.673
.691	-.002	-.063	-.056	-.044	-.031	-.027	-.007	-.023	-.048	.691
.709							.013			.709
.727							.011			.727
.745	-.007	-.053			-.035	-.021			-.041	.745
.763							.008			.763
.781							.004			.781
.799							.013			.799
.817	.018	-.042	-.034	-.022	-.014	-.007	.005	-.010	-.027	.817
.834							.000			.834
.852							.004			.852
.870							.009			.870
.888							.011			.888
.906							.010			.906
.921	-.155						.015			.921
.924	.015	-.022			-.016	-.003				.924
.947							.014			.947
.964							.073			.964
.981	.062	-.015	-.006			.027	.037			.981
1.000	-.162						.049	.014	-.014	1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.304	1.039	.031	.743	.525
.030	1.481	1.363	.094	1.291	.931
.050	1.551	1.453	.156	1.590	1.274
.070	1.578	1.486	.219	1.580	1.516
.090	1.597	1.491	.281	1.556	1.552
.110	1.606	1.495	.344	1.561	1.517
.130	1.610	1.496	.406	1.562	1.517
.150	1.606	1.492	.469	1.566	1.522
.170	1.587	1.479	.750	1.555	
.190	1.520	1.388	1.000	1.561	
			1.250	1.557	
			1.500	1.549	

TABLE 4. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	130	165	180	225	270	
-•002							1.558			-•002
•008							•810			•008
•018							•584			•018
•029							•308	•310		•029
•039							•092			•039
•050							•031			•050
•060							•042			•060
•075	•115	•059	•042	•033	•017	•014	•006	•024		•075
•085	•170	•112	•129	•085	•060	•065	•049	•060	•152	•085
•096	•229									•096
•126		•135	•125	•116	•116	•115	•096	•121	•131	•126
•143							•187			•143
•154	-•099	-•130	-•136	-•142	-•153	-•152	•144	-•151		•154
•171							•126			•171
•189							•105			•189
•206	-•052	-•082	-•085	-•100	-•087	-•087	•087	-•088		•206
•223							•071			•223
•241							•058			•241
•258	-•035	-•052	-•056	-•060	-•050	-•045	•047	-•051	-•054	•258
•275							•035			•275
•293							•025			•293
•310	-•023	-•039	-•037	-•029	-•023	-•017	•018	-•024	-•041	•310
•327							•016			•327
•344							•015			•344
•362	-•014	-•023	-•020	-•018	-•019	-•013	•011	-•016	-•024	•362
•379							•011			•379
•396	-•016	-•021	-•022	-•020	-•012	-•007	•000	-•014	-•021	•396
•408							•018			•408
•426	•191	•171	•172	•170	•163	•167	•166	•182	•168	•426
•451	•169	•151	•147	•143	•138	•136	•140		•145	•451
•476	•153	•129	•120	•118	•117	•116	•118	•124	•132	•476
•501	•140	•113	•110	•103	•098	•097	•102	•107	•112	•501
•530	-•052	-•073	-•077	-•083	-•080	-•081	•079	-•074	-•077	•530
•548							•063			•548
•566							•047			•566
•584	-•018	-•049	-•050	-•051	-•045	-•050	•047	-•045	-•051	•584
•602							•043			•602
•620							•034			•620
•638	-•015	-•028		-•032	-•028				-•027	•638
•655							•023			•655
•673							•021			•673
•691	-•010	-•034	-•034	-•033	-•031	-•026	•014	-•016	-•025	•691
•709							•008			•709
•727							•012			•727
•745	-•009	-•028		-•020	-•012		•005		-•017	•745
•763							•008			•763
•781							•013			•781
•799							•004			•799
•817	•009	-•013	-•014	-•012	-•006	-•003	•000	-•003	•000	•817
•834							•003			•834
•852							•003			•852
•870							•004			•870
•888							•008			•888
•906							•009			•906
•921	-•091	-•019	-•000		-•007	•003		•012		•921
•924							•012			•924
•947							•009			•947
•964							•013			•964
•981	-•046	-•096	•012	•005		•008	•009	•002	•006	•981
1.000							•008			1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
•010	1.227	1.056	•031	4.637	•533
•030	1.447	1.371	•094	1.016	•793
•050	1.533	1.459	•156	1.308	1.021
•070	1.566	1.485	•219	1.498	1.237
•090	1.579	1.497	•281	1.529	1.403
•110	1.591	1.501	•344	1.541	1.491
•130	1.593	1.504	•406	1.541	1.522
•150	1.588	1.498	•469	1.548	1.533
•170	1.571	1.486	•750	1.542	
•190	1.498	1.395	1.000	1.553	
			1.250	1.553	
			1.500	1.553	

TABLE 4. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH NATURAL TRANSITION AT M = 1.57 - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-.002							1.553			-.002
.008							.889			.008
.018							.651			.018
.029							.366			.029
.039							.152			.039
.050							.018			.050
.060							.008			.060
.075	.060	.021	.016	.016	.010	.011	.008			.075
.085		.114	.061	.063	.056	.061	.054			.085
.096		.159	.113	.118	.109	.116	.096	.107	.163	.096
.126			.141	.143	.144	.153	.156	.138	.141	.126
.143								.166		.143
.154	-.128	-.123	-.127	-.127	-.130	-.132	-.124	-.136		.154
.171							.110			.171
.189							.089			.189
.206	-.075	-.076	-.075	-.084	-.074	-.074	-.074	-.074		.206
.223							.058			.223
.241							.052			.241
.258	-.051	-.041	-.044	-.044	-.047	-.046	-.050	-.042	-.041	.258
.275							.046			.275
.293							.036			.293
.310	-.025	-.032	-.033	-.026	-.027	-.021	-.021	-.026	-.033	.310
.327							.017			.327
.344							.017			.344
.362	-.017	-.015	-.011	-.009	-.019	-.016	-.015	-.014	-.014	.362
.379							.016			.379
.396	-.017	-.010	-.014	-.016	-.014	-.013	-.006	-.014	-.010	.396
.408							.012			.408
.426	.178	.176	.179	.179	.177	.181	.177	.177	.175	.426
.451	.154	.157	.156	.158	.155	.150	.154			.451
.476	.135	.136	.134	.132	.131	.130	.131	.134	.141	.476
.501	.117	.119	.121	.117	.118	.116	.118	.119	.123	.501
.530	-.068	-.066	-.068	-.072	-.070	-.071	-.069	-.068	-.069	.530
.548							.055			.548
.566							.043			.566
.584	-.027	-.041	-.040	-.042	-.041	-.042	-.036	-.039	-.043	.584
.602							.034			.602
.620							.032			.620
.638	-.026	-.021		-.022	-.022					.638
.655							.021			.655
.673							.021			.673
.691	-.017	-.026	-.025	-.024	-.032	-.028	-.015	-.015	-.017	.691
.709							.014			.709
.727							.010			.727
.745	-.013	-.019		-.015	-.014		.008		-.009	.745
.763							.009			.763
.781							.017			.781
.799							.008			.799
.817	.003	-.007	-.007	-.007	-.007	-.004	-.001	.006	.817	.817
.834							.006			.834
.852							.001			.852
.870							.002			.870
.888							.006			.888
.906							.007			.906
.921	-.074									.921
.924	.013	.007		-.005	.004		.009		-.002	.924
.947							.007			.947
.964							.012			.964
.981	.048	.025	.011		.011	.008	.006	.007	.009	.981
1.000	-.076						1.500	1.535		1.000

Orifice station, $y, \text{in.}$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y, \text{in.}$	Cp top	Cp side
.010	1.122	1.041	.031	.502	.512
.030	1.402	1.363	.094	.735	.715
.050	1.511	1.458	.156	.915	.868
.070	1.544	1.480	.219	1.086	1.033
.090	1.556	1.492	.281	1.225	1.178
.110	1.563	1.501	.344	1.353	1.305
.130	1.561	1.501	.406	1.443	1.411
.150	1.558	1.494	.469	1.493	1.477
.170	1.536	1.482	.750	1.513	
.190	1.465	1.388	1.000	1.521	
			1.250	1.528	
			1.500	1.535	

TABLE 4.- PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{T}$	C_p for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
-.002							1.556			-.002
.008							.978			.008
.018							.737			.018
.029							.441			.029
.039							.222			.039
.050							.077			.050
.060							.064			.060
.075	.010	.060	.075	.088	.103	.117	.109	.093		.075
.085	.072	.111	.129	.142	.157	.171	.149	.147	.146	.085
.096	.167									.096
.126										.126
.143										.143
.154	-.149	-.129	-.124	-.117	-.107	-.104	-.096	-.118		.154
.171										.171
.189										.189
.206	-.084	-.084	-.073	-.078	-.059	-.055	-.055	-.067		.206
.223										.223
.241										.241
.258	-.049	-.050	-.050	-.047	-.037	-.032	-.036	-.041	-.051	.258
.275										.275
.293										.293
.310	-.022	-.039	-.042	-.035	-.029	-.023	-.022	-.034	-.041	.310
.327										.327
.344										.344
.362	-.012	-.025	-.021	-.016	-.019	-.013	-.012	-.019	-.023	.362
.379										.379
.396	-.011	-.018	-.022	-.023	-.016	-.014	-.006	-.021	-.018	.396
.408										.408
.426	.167	.171	.175	.177	.181	.190	.186	.177	.166	.426
.451	.141	.151	.154	.159	.171	.168	.173	.145	.145	.451
.476	.120	.130	.133	.137	.147	.148	.150	.143	.132	.476
.501	.100	.113	.121	.122	.136	.136	.140	.131	.114	.501
.530	-.076	-.072	-.069	-.070	-.059	-.056	-.052	-.061	-.077	.530
.548										.548
.566										.566
.584	-.041	-.047	-.042	-.040	-.031	-.031	-.023	-.034	-.050	.584
.602										.602
.620										.620
.638	-.029	-.027			-.023	-.016				.638
.655										.655
.673										.673
.691	-.015	-.034	-.029	-.027	-.031	-.024				.691
.709										.709
.727										.727
.745	-.010	-.025			-.018	-.011				.745
.763										.763
.781										.781
.799										.799
.817	.003	-.014	-.012	-.012	-.007	-.002				.817
.834										.834
.852										.852
.870										.870
.888										.888
.906										.906
.921	-.084									.921
.924	.006	.001			-.010	.003				.924
.947										.947
.964										.964
.981	.061	.016	.003			.011	.010	.010	.005	.981
1.000	-.091									1.000

Orifice station, y , in.	Nose rake		Base rake		
	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side
.010	.716	1.037	.031	.457	.534
.030	1.207	1.364	.094	.642	.783
.050	1.493	1.453	.156	.763	.994
.070	1.575	1.485	.219	.867	1.206
.090	1.598	1.492	.281	.953	1.371
.110	1.594	1.500	.344	1.044	1.469
.130	1.570	1.501	.406	1.127	1.513
.150	1.546	1.496	.469	1.195	1.527
.170	1.511	1.481	.570	1.413	
.190	1.419	1.393	1.000	1.487	
			1.250	1.504	
			1.500	1.516	

TABLE 4.- PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
-•002							1.556			-•002
•008							1.064			•008
•018							•814			•018
•029		•362					•520	•458		•029
•039							•299			•039
•050							•142			•050
•060							•128			•060
•075	-•033	•011	•036	•067	•112	•124	•173	•127		•075
•085		•029	•102	•136	•165	•210	•233	•211	•189	•085
•096		•126							•103	•096
•126			•118	•151	•181	•237	•255	•244	•196	•126
•143								•103		•143
•154	-•169	-•141	-•127	-•108	-•079	-•068	-•059	-•103		•154
•171								•051		•171
•189							•037			•189
•206	-•087	-•104	-•089	-•081	-•040	-•029	-•026	-•061		•206
•223							•016			•223
•241							•017			•241
•258	-•041	-•086	-•081	-•058	-•026	-•013	-•015	-•044	-•085	•258
•275							•017			•275
•293							•014			•293
•310	-•017	-•069	-•067	-•051	-•025	-•011	-•007	-•044	-•069	•310
•327							•013			•327
•344							•013			•344
•362	-•006	-•061	-•059	-•046	-•024	-•012	-•008	-•038	-•056	•362
•379							•004			•379
•396	-•008	-•049	-•050	-•041	-•013	-•004	-•004	-•033	-•046	•396
•408							•005			•408
•426	•169	•141	•148	•155	•180	•200	•198	•166	•143	•426
•451	•139	•122	•129	•142	•171	•178	•186	•120	•120	•451
•476	•113	•104	•112	•129	•166	•177	•183	•148	•106	•476
•501	•093	•093	•111	•122	•152	•159	•167	•142	•092	•501
•530	-•094	-•090	-•082	-•073	-•044	-•039	-•034	-•058	-•095	•530
•548							•018			•548
•566							•006			•566
•584	-•049	-•070	-•059	-•047	-•020	-•014	-•006	-•033	-•072	•584
•602							•002			•602
•620							•007			•620
•638	-•030	-•056		-•030	-•003				•053	•638
•655							•004			•655
•673							•001			•673
•691	-•019	-•062	-•054	-•040	-•026	-•013	-•003	-•023	-•049	•691
•709							•001			•709
•727							•003			•727
•745	-•001	-•054		-•040	-•004		•008		-•041	•745
•763							•008			•763
•781							-•006			•781
•799							•001			•799
•817	•008	-•039	-•037	-•032	-•008	-•004	•013	-•015	-•026	•817
•834							•005			•834
•852							•014			•852
•870	-						•015			•870
•888							•018			•888
•906							•018			•906
•921	-•147									•921
•924	•017	-•021		-•027		•004		•022	-•033	•924
•947								•021		•947
•964								•026		•964
•981	•070	-•013	-•020			•014	•017	•019	-•002	-•019
1.000	-•162						1.250	1.502		1.000

Orifice station, y.in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y.in.	Cp top	Cp side
•010	•565	1.004	•031	•694	•537
•030	1.037	1.351	•094	1.147	•939
•050	1.426	1.445	•156	1.374	1.278
•070	1.551	1.483	•219	1.429	1.520
•090	1.579	1.495	•281	1.431	1.547
•110	1.574	1.496	•344	1.446	1.521
•130	1.550	1.500	•406	1.453	1.524
•150	1.522	1.492	•469	1.463	1.531
•170	1.482	1.481	•750	1.474	
•190	1.388	1.387	1.000	1.492	
			1.250	1.502	
			1.500	1.509	

TABLE 4. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH NATURAL TRANSITION AT $M = 1.57$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-.002							1.534			-.002
.008							1.157			.008
.018							.912			.018
.029							.621	.512		.029
.039							.404			.039
.050							.226			.050
.060										.060
.075	-.073	.034	.082	.134	.221	.260	.257	.171		.075
.085	-.009	.074	.134	.189	.279	.316	.295	.241	.030	.085
.096	.072									.096
.126		.085	.140	.194	.297	.327	.320	.223	.074	.126
.143							-.056			.143
.154	-.174	-.171	-.138	-.100	-.040	-.019	-.006	-.082		.154
.171							-.001			.171
.189							.011			.189
.206	-.088	-.151	-.122	-.090	-.010	.010	.019	-.049		.206
.223							.026			.223
.241							.021			.241
.258	-.036	-.146	-.122	-.085	-.003	.021	.021	-.047	-.147	.258
.275							.017			.275
.293							.016			.293
.310	-.011	-.151	-.129	-.083	-.014	.010	.020	-.054	-.147	.310
.327							.010			.327
.344							.012			.344
.362	-.001	-.133	-.132	-.095	-.017	.009	.020	-.056	-.119	.362
.379							.018			.379
.396	-.003	-.110	-.116	-.094	-.013	.011	.025	-.062	-.102	.396
.408							.013			.408
.426	.226	.074	.070	.090	.176	.219	.227	.132	.090	.426
.451	.149	.067	.072	.094	.172	.197	.213		.074	.451
.476	.111	.050	.066	.097	.173	.198	.209	.141	.058	.476
.501	.083	.042	.068	.093	.164	.186	.200	.135	.048	.501
.530	-.112	-.127	-.112	-.088	-.029	-.013	-.004	-.055	-.130	.530
.548							.010			.548
.566							.023			.566
.584	-.057	-.115	-.093	-.067	-.004	.013	.024	-.034	-.115	.584
.602							.038			.602
.620							.027			.620
.638	-.038	-.099			-.059	.005				.638
.655							.034			.655
.673							.029			.673
.691	-.023	-.120	-.098	-.071	-.016	.011	.032	-.031	-.095	.691
.709							.033			.709
.727							.028			.727
.745	-.012	-.113			-.072	.001				.745
.763							.032			.763
.781							.025			.781
.799							.013			.799
.817	-.002	-.094	-.096	-.073	-.003	.024	.036	-.028	-.068	.817
.834							.028			.834
.852							.041			.852
.870							.034			.870
.888							.037			.888
.906							.038			.906
.921	-.200	.002	-.069		-.066	.000				.921
.924							.042			.924
.947							.041			.947
.964							.047			.964
.981	.046	-.052	-.070			.013	.031	.039	-.014	.981
1.000	-.209									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.473	.935	.031	.878	.445
.030	.894	1.313	.094	1.560	1.060
.050	1.338	1.431	.156	1.509	1.515
.070	1.517	1.475	.219	1.456	1.624
.090	1.556	1.485	.281	1.466	1.511
.110	1.549	1.490	.344	1.448	1.491
.130	1.515	1.496	.406	1.438	1.503
.150	1.484	1.486	.469	1.427	1.506
.170	1.436	1.470	.750	1.402	
.190	1.345	1.380	1.000	1.426	
			1.250	1.469	
			1.500	1.496	

TABLE 5. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH FIXED TRANSITION AT $M = 1.57$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-.002										-.002
.008										.008
.018										.018
.029		.337								.029
.039										.039
.050										.050
.060										.060
.075	.266	.029	-.014	-.087	-.133	-.125				.075
.085	.332	.076	.059	-.013	-.072	-.075	-.092	-.033		.085
.096	.491				-.022	-.021	-.036	-.044	.039	.096
.126		.087	.039	.025	.038	.048	.042	.029	.075	.126
.143										.143
.154	-.008	-.178	-.200	-.208	-.194	-.182	-.173	-.215		.154
.171										.171
.189										.189
.206	.023	-.157	-.169	-.171	-.118	-.099	-.089	-.133		.206
.223										.223
.241										.241
.258	.023	-.152	-.161	-.133	-.066	-.044	-.035	-.082	-.156	.258
.275										.275
.293										.293
.310	.022	-.154	-.139	-.085	-.050	-.028	-.012	-.057	-.151	.310
.327										.327
.344										.344
.362	.017	-.136	-.103	-.064	-.054	-.027	-.002	-.048	-.126	.362
.379										.379
.396	.015	-.119	-.096	-.048	-.049	-.023	-.009	-.039	-.110	.396
.408										.408
.426	.229	.068	.092	.081	.059	.148	.239	.087	.089	.426
.451	.215	.065	.059	.071	.057	.102	.150		.067	.451
.476	.208	.046	.038	.059	.041	.086	.112	.077	.054	.476
.501	.198	.036	.032	.044	.019	.062	.082	.047	.044	.501
.530	-.001	-.133	-.138	-.114	-.101	-.133	-.114	-.082	-.130	.530
.548										.548
.566										.566
.584	.027	-.120	-.107	-.089	-.094	-.081	-.059	-.076	-.119	.584
.602										.602
.620										.620
.638										.638
.655	.032	-.108		-.083	-.052					.655
.673										.673
.691	.032	-.125	-.110	-.071	-.048	-.050	-.024	-.036	-.097	.691
.709										.709
.727										.727
.745	.030	-.119		-.055	-.038					.745
.763										.763
.781										.781
.799										.799
.817	.045	-.097	-.066	-.040	-.035	-.029	-.005	-.028	-.073	.817
.834										.834
.852										.852
.870										.870
.888										.888
.906										.906
.921	-.211									.921
.924	.050	-.075		-.031	-.030					.924
.947										.947
.964										.964
.981	.037	-.053	-.009		-.009	-.011				.981
1.000	-.218									1.000

Orifice station, y , in.	Nose rake			Base rake		
	Cp _{top}	Cp _{side}	Orifice station, y , in.	Cp _{top}	Cp _{side}	
.010	1.173	1.108	.031	1.929	1.622	
.030	1.456	1.508	.094	1.583	1.048	
.050	1.562	1.566	.156	1.632	1.517	
.070	1.597	1.489	.219	1.578	1.618	
.090	1.618	1.451	.281	1.576	1.512	
.110	1.628	1.450	.344	1.572	1.479	
.130	1.632	1.454	.406	1.575	1.496	
.150	1.631	1.444	.469	1.580	1.487	
.170	1.611	1.429	.750	1.571		
.190	1.542	1.346	1.000	1.573		
			1.250	1.564		
			1.500	1.568		

TABLE 5. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-•002							1.540			-•002
•008							.721			•008
•018							.496			•018
•029							.238			•029
•039							.033			•039
•050							.077			•050
•060							.090			•060
•075	•183	•046	•019	•000	•023	•025	•043	-•006		•075
•085	•246	•096	•100	•061	•029	•027	•008	•039	•121	•085
•096	•378									•096
•126		•123	•096	•083	•075	•076	•060	•080	•115	•126
•143							-•207			•143
•154	-•063	-•147	-•157	-•168	-•176	-•171	-•165	-•176		•154
•171							.142			•171
•189							.110			•189
•206	-•024	-•114	-•118	-•127	-•098	-•087	.085	-•105		•206
•223							.062			•223
•241							.052			•241
•258	-•015	-•088	-•088	-•078	-•053	-•041	-•041	-•058	-•087	•258
•275							.033			•275
•293							.026			•293
•310	-•009	-•069	-•068	-•054	-•031	-•019	.016	-•034	-•072	•310
•327							.007			•327
•344							.009			•344
•362	-•009	-•061	-•049	-•032	-•022	-•013	-•008	-•021	-•058	•362
•379							-•008			•379
•396	-•007	-•052	-•047	-•040	-•017	-•010	.000	-•023	-•050	•396
•408							.022			•408
•426	•202	•137	•142	•138	•128	•147	.166	•131	.142	•426
•451	•188	•121	•115	•113	•115	•122	.136	.117	.117	•451
•476	•180	•103	•092	•091	•094	•099	.111	.100	.104	•476
•501	•169	•090	•081	•074	•065	•081	.101	.073	.090	•501
•530	-•030	-•094	-•103	-•105	-•081	-•088	.104	-•077	-•096	•530
•548							.080			•548
•566							.057			•566
•584	-•002	-•074	-•070	-•067	-•064	-•063	.051	-•062	-•075	•584
•602							.043			•602
•620							.035			•620
•638		-•009	-•060		-•053	-•040			-•059	•638
•655							-•017			•655
•673							.019			•673
•691		-•003	-•065	-•058	-•045	-•031	-•010	-•023	-•051	•691
•709							.015			•709
•727							.012			•727
•745		-•007	-•056		-•037	-•022			-•043	•745
•763							.008			•763
•781							.005			•781
•799							.014			•799
•817		-•018	-•046	-•036	-•023	-•015	-•004	-•012	-•031	•817
•834							.000			•834
•852							.003			•852
•870							.009			•870
•888							.010			•888
•906							.009			•906
•921	-•158									•921
•924	•009	-•025			-•017	-•004			-•035	•924
•947							.049			•947
•964							.063			•964
•981		-•032	-•016	-•002			.040		-•016	•981
1.000	-•166						.018		-•016	1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
•010	1.069	1.104	•031	•799	•522
•030	1.406	1.500	•094	1.316	•941
•050	1.549	1.568	•156	1.601	1.291
•070	1.597	1.504	•219	1.581	1.526
•090	1.615	1.451	•281	1.565	1.559
•110	1.630	1.456	•344	1.560	1.520
•130	1.622	1.457	•406	1.563	1.520
•150	1.612	1.452	•469	1.569	1.525
•170	1.593	1.441	•750	1.564	
•190	1.513	1.348	1.000	1.562	
			1.250	1.562	
			1.500	1.554	

TABLE 5. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE II WITH FIXED TRANSITION AT $M = 1.57$ - Continued(c) $\alpha = -3.0^\circ$

Model station, x	Cp for meridian angle, θ , deg -									Model station, y
	0	90	105	120	150	165	180	225	270	
-002							1.559			-002
.008							.808			.008
.018							.582			.018
.029	.372						.300	.306		.029
.039							.090			.039
.050							.025			.050
.060	.018	.004	-.012	-.027	-.027	-.045				.060
.075	.120	.057	.043	.031	.015	.017	-.000	.021	.155	.075
.085	.181	.111	.129	.097	.071	.071	.048	.097		.085
.096	.297									.096
.126		.137	.126	.116	.114	.115	.097	.116	.130	.126
.143							-.187			.143
.154	-.099	-.130	-.137	-.143	-.152	-.153	-.146	-.150		.154
.171							-.128			.171
.189							-.106			.189
.206	-.053	-.085	-.087	-.100	-.090	-.089	-.089	-.090		.206
.223							-.070			.223
.241							-.057			.241
.258	-.036	-.055	-.060	-.059	-.050	-.043	-.045	-.050	-.055	.258
.275							-.033			.275
.293							-.027			.293
.310	-.026	-.038	-.037	-.028	-.026	-.019	-.020	-.025	-.039	.310
.327							-.018			.327
.344							-.016			.344
.362	-.014	-.025	-.022	-.021	-.019	-.013	-.012	-.016	-.025	.362
.379							-.010			.379
.396	-.017	-.023	-.023	-.021	-.011	-.004	-.003	-.014	-.020	.396
.408							-.022			.408
.426	.189	.168	.169	.166	.157	.161	.160	.178	.166	.426
.451	.170	.150	.144	.138	.134	.134	.137		.145	.451
.476	.154	.126	.118	.117	.115	.114	.117	.122	.130	.476
.501	.139	.114	.109	.101	.096	.095	.099	.107	.111	.501
.530	-.053	-.074	-.079	-.084	-.080	-.081	-.079	-.072	-.076	.530
.548							-.062			.548
.566							-.046			.566
.584	-.024	-.050	-.051	-.050	-.049	-.054	-.050	-.038	-.051	.584
.602							-.042			.602
.620							-.034			.620
.638	-.016	-.029			-.033	-.028				.638
.655							-.023			.655
.673							-.021			.673
.691	-.011	-.036	-.035	-.033	-.030	-.025	-.013	-.017	-.025	.691
.709							-.012			.709
.727							-.008			.727
.745	-.003	-.028			-.021	-.013				.745
.763							-.010			.763
.781							-.009			.781
.799							-.013			.799
.817	.008	-.015	-.015	-.012	-.007	-.004	-.000	-.003	-.000	.817
.834							-.003			.834
.852							-.002			.852
.870							-.004			.870
.888							-.007			.888
.906							-.009			.906
.921	-.094									.921
.924	.017	-.000			-.008	.002				.924
.947										.947
.964										.964
.981	.023	.011	.004				.006	.006	.007	.981
1.000	-.098									1.000

Orifice station, y,in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side	
.010	.932	1.122	.031	.667	.531	
.030	1.353	1.504	.094	1.013	.793	
.050	1.537	1.571	.156	1.310	1.013	
.070	1.593	1.512	.219	1.493	1.231	
.090	1.613	1.456	.281	1.533	1.396	
.110	1.622	1.461	.344	1.540	1.481	
.130	1.610	1.465	.406	1.543	1.515	
.150	1.595	1.458	.469	1.548	1.529	
.170	1.571	1.444	.570	1.543		
.190	1.486	1.358	1.000	1.550		
			1.250	1.553		
			1.500	1.554		

TABLE 5. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
-.002							1.555			-.002
.008							.892			.008
.018							.657			.018
.029							.366			.029
.039							.148			.039
.050							.024			.050
.060							.020			.060
.075	.061	.061	.060	.062	.057	.066	.048	.058		.075
.085	.121	.112	.151	.129	.116	.120	.096	.147	.161	.085
.096	.225									.096
.126		.141	.143	.142	.152	.157	.138	.145	.137	.126
.143							.167			.143
.154	-.127	-.123	-.127	-.127	-.131	-.133	-.126	-.133		.154
.171							.111			.171
.189							.090			.189
.206	-.074	-.076	-.074	-.083	-.074	-.074	-.075	-.075		.206
.223							.059			.223
.241							.053			.241
.258	-.051	-.042	-.044	-.044	-.047	-.046	-.050	-.043	-.041	.258
.275							.043			.275
.293							.036			.293
.310	-.025	-.031	-.031	-.025	-.026	-.021	-.021	-.026	-.032	.310
.327							.017			.327
.344							.018			.344
.362	-.017	-.015	-.010	-.009	-.019	-.016	-.015	-.014	-.013	.362
.379							.017			.379
.396	-.017	-.010	-.014	-.016	-.014	-.013	-.006	-.014	-.010	.396
.408							.012			.408
.426	.177	.176	.179	.177	.176	.180	.177	.175	.174	.426
.451	.154	.157	.152	.158	.154	.149	.152			.451
.476	.135	.136	.134	.132	.131	.130	.131	.133	.141	.476
.501	.117	.119	.121	.117	.119	.115	.117	.119	.123	.501
.530	-.067	-.066	-.067	-.071	-.069	-.071	-.069	-.068	-.068	.530
.548							.055			.548
.566							.043			.566
.584	-.027	-.041	-.039	-.041	-.040	-.041	-.035	-.039	-.042	.584
.602							.034			.602
.620							.032			.620
.638	-.026	-.020		-.022	-.021					.638
.655							.020			.655
.673							.021			.673
.691	-.016	-.026	-.024	-.024	-.032	-.028	-.015	-.015	-.017	.691
.709							.014			.709
.727							.010			.727
.745	-.013	-.018		-.014	-.013		.008			.745
.763							.009			.763
.781							.017			.781
.799							.008			.799
.817	.004	-.006	-.007	-.006	-.007	-.003	-.001	-.001	.006	.817
.834							.006			.834
.852							.001			.852
.870							.002			.870
.888							.006			.888
.906							.007			.906
.921	-.073	.013	.007		-.005	.005				.921
.924							.009			.924
.947							.007			.947
.964							.012			.964
.981	.033	-.076	.026	.011		.011	.009	.007	.009	.981
1.000							1.000	1.000	1.000	1.000

Nose rake			Base rake		
Orifice station, y,in.	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.829	1.141	.031	.513	.512
.030	1.296	1.504	.094	.725	.715
.050	1.524	1.569	.156	.902	.873
.070	1.585	1.503	.219	1.066	1.029
.090	1.606	1.452	.281	1.202	1.174
.110	1.611	1.460	.344	1.328	1.299
.130	1.588	1.460	.406	1.421	1.403
.150	1.570	1.452	.469	1.478	1.474
.170	1.537	1.442	.750	1.517	
.190	1.452	1.354	1.000	1.526	
				1.250	1.532
				1.500	1.534

TABLE 5.- PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
-0.002							1.553			-0.002
.008							.974			.008
.018							.730			.018
.029		.369					.438			.029
.039							.220			.039
.050							.082			.050
.060							.059			.060
.075	.012	.021	.035	.048	.068	.074		.405		.075
.085	.072	.062	.072	.089	.104	.121	.103			.085
.096	.176	.110	.164	.154	.162	.173	.150	.177	.149	.096
.126		.136	.152	.162	.191	.202	.188	.171	.130	.126
.143							.138			.143
.154	-1.150	-1.129	-1.123	-1.115	-1.106	-1.104	-1.097	-1.118		.154
.171							.086			.171
.189							.069			.189
.206	-1.085	-1.085	-1.074	-1.077	-1.059	-1.056	-1.055	-1.067		.206
.223							.042			.223
.241							.040			.241
.258	-1.050	-1.051	-1.049	-1.047	-1.037	-1.033	-1.037	-1.042	-1.051	.258
.275							.039			.275
.293							.032			.293
.310	-1.023	-1.039	-1.041	-1.033	-1.028	-1.023	-1.023	-1.033	-1.040	.310
.327							.016			.327
.344							.014			.344
.362	-1.013	-1.026	-1.020	-1.015	-1.018	-1.014	-1.013	-1.019	-1.022	.362
.379							.015			.379
.396	-1.011	-1.018	-1.022	-1.022	-1.015	-1.014	-1.006	-1.020	-1.018	.396
.408							.014			.408
.426	.164	.168	.175	.177	.181	.189	.186	.177	.166	.426
.451	.139	.150	.153	.160	.171	.167	.172			.451
.476	.119	.129	.133	.138	.148	.148	.150	.144	.131	.476
.501	.099	.112	.121	.123	.136	.137	.140	.131	.115	.501
.530	-1.075	-1.072	-1.070	-1.069	-1.057	-1.056	-1.053	-1.060	-1.076	.530
.548							.040			.548
.566							.029			.566
.584	-1.041	-1.048	-1.042	-1.040	-1.031	-1.030	-1.024	-1.034	-1.050	.584
.602							.021			.602
.620							.022			.620
.638	-1.029	-1.028			-1.023	-1.015				.638
.655							.012			.655
.673							.014			.673
.691	-1.016	-1.035	-1.029	-1.026	-1.030	-1.023	-1.010	-1.017	-1.024	.691
.709							.008			.709
.727							.009			.727
.745	-1.011	-1.026			-1.017	-1.009				.745
.763							.003			.763
.781							.008			.781
.799							.017			.799
.817	.003	-1.015	-1.013	-1.011	-1.006	-1.001				.817
.834							.002			.834
.852							.003			.852
.870							.005			.870
.888							.006			.888
.906							.008			.906
.921	-1.084						.009			.921
.924	.004	.001			-1.009	.003				.924
.947							.013			.947
.964							.012			.964
.981	.049		.016	.002			.016			.981
1.000	-1.091						.010			1.000

Orifice station, y , in.	Nose rake			Base rake		
	Cp top	Cp side	Cp top	Orifice station, y , in.	Cp top	Cp side
.010	.651	1.120	.031	.458	.530	
.030	1.083	1.507	.094	.635	.779	
.050	1.435	1.573	.156	.752	.987	
.070	1.561	1.505	.219	.855	1.199	
.090	1.593	1.456	.281	.944	1.364	
.110	1.596	1.458	.344	1.033	1.462	
.130	1.576	1.459	.406	1.112	1.512	
.150	1.557	1.452	.469	1.174	1.526	
.170	1.515	1.441	.750	1.391		
.190	1.427	1.352	1.000	1.487		
				1.250	1.502	
				1.500	1.514	

TABLE 5. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
-0.002										-0.002
.008										.008
.018										.018
.029										.029
.039										.039
.050										.050
.060										.060
.075	-0.034	.017	.044	.071	.119	.132	.123			.075
.085	.026	.050	.078	.111	.155	.181	.165	.131		.085
.096	.132	.097	.166	.174	.214	.234	.211	.204	.110	.096
.126		.117	.150	.177	.235	.253	.244	.194	.109	.126
.143								.104		.143
.154	-0.170	-0.142	-0.128	-0.108	-0.079	-0.069	-0.062	-0.103		.154
.171								.052		.171
.189								.038		.189
.206	-0.088	-0.105	-0.090	-0.082	-0.041	-0.031	-0.029	-0.062		.206
.223								.017		.223
.241								.018		.241
.258	-0.041	-0.087	-0.080	-0.059	-0.026	-0.013	-0.017	-0.045	-0.086	.258
.275								.019		.275
.293								.016		.293
.310	-0.018	-0.069	-0.067	-0.051	-0.026	-0.013	-0.009	-0.045	-0.069	.310
.327								.014		.327
.344								.013		.344
.362	-0.006	-0.061	-0.057	-0.045	-0.024	-0.011	-0.007	-0.037	-0.055	.362
.379								.006		.379
.396	-0.008	-0.048	-0.049	-0.041	-0.015	-0.005	-0.002	-0.034	-0.046	.396
.408								.007		.408
.426	.164	.141	.146	.153	.179	.198	.197	.164	.143	.426
.451	.135	.122	.129	.141	.171	.178	.185		.120	.451
.476	.112	.103	.111	.129	.168	.179	.184	.148	.106	.476
.501	.092	.093	.110	.118	.151	.158	.166	.139	.093	.501
.530	-0.090	-0.091	-0.082	-0.073	-0.044	-0.038	-0.035	-0.058	-0.097	.530
.548								.018		.548
.566								.007		.566
.584	-0.049	-0.072	-0.059	-0.047	-0.021	-0.015	-0.007	-0.033	-0.073	.584
.602								.002		.602
.620								.007		.620
.638	-0.030	-0.055		-0.029	-0.007					.638
.655								.004		.655
.673								.000		.673
.691	-0.015	-0.062	-0.055	-0.040	-0.027	-0.013		.024	-0.050	.691
.709								.003		.709
.727								.003		.727
.745	-0.001	-0.054		-0.039	-0.006			.004	-0.041	.745
.763								.010		.763
.781								.008		.781
.799	.007	-0.041	-0.038	-0.032	-0.008	.004		.013		.799
.817								.004		.817
.834								.014		.834
.852								.015		.852
.870								.014		.870
.888								.021		.888
.906								.020		.906
.921	-0.145	.017	-0.021		-0.028	.002		.021		.921
.924								.020		.924
.947								.026		.947
.964								.018		.964
.981	.016	-0.012	-0.021		.012	.016		.003	-0.019	.981
1.000	-0.159									1.000

Orifice station, $y, \text{in.}$	Nose rake		Base rake		C_p side
	C_p top	C_p side	Orifice station, $y, \text{in.}$	C_p top	
.010	.533	1.123	.031	.706	.535
.030	.912	1.498	.094	1.121	.931
.050	1.326	1.570	.156	1.337	1.269
.070	1.522	1.505	.219	1.398	1.514
.090	1.571	1.451	.281	1.407	1.543
.110	1.575	1.454	.344	1.416	1.518
.130	1.551	1.456	.406	1.429	1.520
.150	1.534	1.448	.469	1.437	1.527
.170	1.485	1.438	.750	1.456	
.190	1.394	1.348	1.000	1.482	
			1.250	1.498	
			1.500	1.504	

TABLE 5. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE II WITH FIXED TRANSITION AT $M = 1.57$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
-0.002										-0.002
.008										.008
.018										.018
.029		.330								.029
.039										.039
.050										.050
.060										.060
.075	-.078	.006	.048	.099	.184	.209	1.532			.075
.085	-.014	.034	.084	.136	.225	.263	.158			.085
.096	.094	.077	.166	.198	.284	.318	.911			.096
.126		.083	.139	.192	.295	.325	.611	.519		.126
.143							.397			.143
.154	-.180	-.170	-.138	-.101	-.041	-.019	.224			.154
.171							.207			.171
.189							.027			.189
.206	-.089	-.155	-.121	-.089	-.012	.010	.017	-.050		.206
.223							.026			.223
.241							.021			.241
.258	-.036	-.148	-.125	-.084	-.003	.022	.020	-.048	-.149	.258
.275							.016			.275
.293							.016			.293
.310	-.014	-.150	-.129	-.086	-.017	.009	.018	-.057	-.146	.310
.327							.017			.327
.344							.015			.344
.362	-.002	-.132	-.131	-.094	-.014	.013	.021	-.055	-.117	.362
.379							.017			.379
.396	-.003	-.109	-.118	-.096	-.015	.010	.022	-.063	-.101	.396
.408							.012			.408
.426	.236	.075	.070	.090	.177	.221	.227	.132	.090	.426
.451	.149	.069	.073	.094	.172	.200	.216	.073	.451	
.476	.106	.049	.070	.096	.166	.192	.203	.135	.060	.476
.501	.092	.040	.066	.094	.164	.187	.200	.134	.046	.501
.530	-.116	-.129	-.112	-.089	-.030	-.013	-.003	-.054	-.131	.530
.548							.011			.548
.566							.023			.566
.584	-.055	-.116	-.094	-.068	-.002	.015	.027	-.033	-.115	.584
.602							.030			.602
.620							.027			.620
.638	-.037	-.102		-.060	.005					.638
.655							.035			.655
.673							.031			.673
.691	-.026	-.121	-.099	-.073	-.014	.011	.033	-.031	-.097	.691
.709							.032			.709
.727							.035			.727
.745	-.012	-.114		-.067	-.006		.026			.745
.763							.025			.763
.781							.014			.781
.799							.023			.799
.817	-.002	-.095	-.097	-.074	-.002	.022	.036	-.030	-.070	.817
.834							.032			.834
.852							.035			.852
.870							.038			.870
.888							.039			.888
.906							.038			.906
.921	-.194						.043			.921
.924	.002	-.066			-.066	.001	.041			.924
.947							.047			.947
.964							.037			.964
.981	-.008	-.052	-.068			.014	.031	-.012	-.049	.981
1.000	-.203						.050			1.000

Orifice station, $y_{in.}$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y_{in.}$	Cp top	Cp side
.010	.448	1.072	.031	.955	.441
.030	.778	1.474	.094	1.561	1.053
.050	1.198	1.560	.156	1.464	1.512
.070	1.462	1.479	.219	1.440	1.624
.090	1.544	1.439	.281	1.429	1.504
.110	1.552	1.441	.344	1.418	1.486
.130	1.529	1.447	.406	1.407	1.499
.150	1.507	1.438	.469	1.399	1.505
.170	1.460	1.425	.750	1.374	
.190	1.361	1.338	1.000	1.397	
			1.250	1.448	
			1.500	1.437	

TABLE 6. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 1.57$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{y}{L}$	Cp for meridian angle, θ , deg -									Model station, $\frac{y}{L}$
	0	90	105	120	150	165	180	225	270	
.039							1.509			.039
.040							1.360			.040
.041							1.090			.041
.044							.631			.044
.049							.049			.049
.055							.039			.055
.060	.800	.530	.483	.429	.137	.095	.128	.230		.060
.075	.137	-.074	-.116	-.151	-.153	-.153	-.153	-.153		.075
.085	.274	-.015	-.066	-.114	-.133	-.131	-.098	-.136	-.032	.085
.096	.504									.096
.126		.093	.037	.006	-.006	-.019	-.029	-.011	.078	.126
.143							.152			.143
.154	-.008	-.177	-.203	-.215	-.213	-.206	-.152	-.153		.154
.171							.152			.171
.189							.122			.189
.206	.027	-.157	-.171	-.179	-.128	-.102	-.092	-.149		.206
.223							.066			.223
.241							.049			.241
.258	.022	-.152	-.167	-.143	-.067	-.042	-.037	-.087	-.153	.258
.275							.024			.275
.293							.015			.293
.310	.019	-.159	-.144	-.068	-.048	-.023	-.009	-.056	-.153	.310
.327							.005			.327
.344							.002			.344
.362	.015	-.140	-.103	-.063	-.051	-.020	.001	-.051	-.135	.362
.379							.002			.379
.396	.012	-.118	-.092	-.044	-.048	-.015	.013	-.031	-.114	.396
.408							.008			.408
.426	.226	.072	.092	.082	.061	.170	.243	.079	.085	.426
.451	.211	.065	.060	.074	.053	.111	.151	.064	.451	
.476	.202	.047	.041	.063	.035	.098	.115	.075	.049	.476
.501	.193	.038	.035	.046	.011	.070	.082	.044	.040	.501
.530	-.004	-.131	-.137	-.109	-.100	-.130	-.108	-.087	-.133	.530
.548							.122			.548
.566							.064			.566
.584	.025	-.118	-.104	-.087	-.090	-.067	-.053	-.082	-.123	.584
.602							.046			.602
.620							.036			.620
.638	.031	-.105		-.078	-.046		-.024		-.103	.638
.655							.023			.655
.673							.021			.673
.691	.030	-.123	-.103	-.064	-.043	-.046	-.022	-.041	-.106	.691
.709							.018			.709
.727							.018			.727
.745	.026	-.113		-.048	-.033		-.018		-.101	.745
.763							.018			.763
.781							.028			.781
.799							.020			.799
.817	.043	-.090	-.057	-.035	-.030	-.025	-.007	-.032	-.079	.817
.834							.019			.834
.852							.008			.852
.870							.009			.870
.888							.004			.888
.906							.003			.906
.921	-.195						.025			.921
.924	.049	-.065			-.025	-.020	.038			.924
.947							.046			.947
.964							.034			.964
.981	.058	-.038	.007		.004	.009	.003	-.050		.981
1.000	-.202									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.266		.031	.873	.429
.030	1.478		.094	1.568	1.020
.050	1.535		.156	1.637	1.490
.070	1.556		.219	1.569	1.615
.090	1.569		.281	1.578	1.547
.110	1.577		.344	1.568	1.485
.130	1.574		.406	1.568	1.502
.150	1.566		.469	1.575	1.494
.170	1.549		.750	1.568	
.190	1.461		1.000	1.569	
			1.250	1.562	
			1.500	1.571	

TABLE 6. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.539			.039
.040							1.423			.040
.041							1.178			.041
.044							.736			.044
.049							.141			.049
.055							.169			.055
.060	.730	.568	.597	.545	.241	.190	.255	.280		.060
.075	.054	-.059	-.085	-.107	-.093	-.109	-.138	-.118		.075
.085	.144	-.003	-.033	-.063	-.069	-.067	-.048	-.059	-.015	.085
.096	.270									.096
.126		.114	.083	.057	.034	.022	.014	.038	.100	.126
.143							-.170			.143
.154	-.064	-.148	-.163	-.176	-.191	-.191	-.170	-.170		.154
.171							-.160			.171
.189							.122			.189
.206	-.022	-.114	-.122	-.133	-.105	-.094	-.091	-.115		.206
.223							-.066			.223
.241							-.054			.241
.258	-.017	-.090	-.092	-.082	-.055	-.043	-.044	-.062	-.093	.258
.275							-.033			.275
.293							-.026			.293
.310	-.011	-.071	-.070	-.056	-.031	-.018	-.017	-.037	-.075	.310
.327							-.009			.327
.344							-.008			.344
.362	-.010	-.062	-.050	-.033	-.020	-.011	-.008	-.023	-.060	.362
.379							-.008			.379
.396	-.009	-.052	-.047	-.040	-.016	-.008	-.000	-.022	-.052	.396
.408							.029			.408
.426	.197	.137	.140	.137	.133	.150	.163	.131	.140	.426
.452	.184	.120	.115	.113	.116	.122	.133	.117	.117	.451
.476	.177	.102	.092	.091	.094	.100	.109	.102	.102	.476
.501	.166	.090	.081	.074	.066	.083	.098	.069	.089	.501
.530	-.031	-.093	-.101	-.104	-.081	-.088	-.108	-.077	-.096	.530
.548							-.074			.548
.566							-.055			.566
.584	-.002	-.072	-.068	-.068	-.061	-.058	-.050	-.061	-.077	.584
.602							-.041			.602
.620							-.033			.620
.638	.006	-.059		-.052	-.038				-.059	.638
.655							-.015			.655
.673							-.019			.673
.691	.003	-.064	-.056	-.044	-.029	-.028	-.009	-.023	-.051	.691
.709							-.014			.709
.727							-.010			.727
.745	.007	-.054		-.035	-.020				-.044	.745
.763							-.007			.763
.781							-.003			.781
.799							-.012			.799
.817	.018	-.044	-.034	-.022	-.013	-.007	-.005	-.013	-.030	.817
.834							-.000			.834
.852							-.004			.852
.870							.011			.870
.888							.011			.888
.906							.010			.906
.921	-.146						.015			.921
.924	.012	-.022			-.014	-.000	.078			.924
.947							.064			.947
.964							.047			.964
.981	.052	-.009	.009			.035	.039	.023	-.013	.981
1.000	-.155						1.500	1.549		1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp _{top}	Cp _{side}	Orifice station, y , in.	Cp _{top}	Cp _{side}
.010	1.250		.031	.751	.518
.030	1.463		.094	1.288	.920
.050	1.523		.156	1.577	1.258
.070	1.537		.219	1.565	1.499
.090	1.541		.281	1.550	1.548
.110	1.543		.344	1.550	1.511
.130	1.543		.406	1.552	1.510
.150	1.534		.469	1.556	1.513
.170	1.515		.750	1.555	
.190	1.429		1.000	1.556	
			1.250	1.556	
			1.500	1.549	

TABLE 6. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 1.57$ - Continued(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.551			.039
.040							1.468			.040
.041							1.251			.041
.044							.824			.044
.049							.232			.049
.055							.270			.055
.060	.702	.700	.671	.574	.356	.367	.389	.366		.060
.075	-.003	-.046	-.058	-.068	-.057	-.077	-.076	-.073		.075
.085	.086	.018	.006	-.006	-.019	-.014	-.023	-.011	.002	.085
.096	.239									.096
.126		.124	.106	.092	.077	.072	.066	.075	.111	.126
.143							.178			.143
.154	-.104	-.134	-.144	-.154	-.164	-.166	.164	-.165		.154
.171							.141			.171
.189							.116			.189
.206	-.055	-.088	-.091	-.103	-.096	-.095	-.096	-.097		.206
.223							.073			.223
.241							.059			.241
.258	-.037	-.057	-.063	-.061	-.051	-.045	-.048	-.052	-.058	.258
.275							.035			.275
.293							.028			.293
.310	-.028	-.038	-.037	-.029	-.026	-.021	-.021	-.026	-.041	.310
.327							.018			.327
.344							.016			.344
.362	-.016	-.025	-.024	-.021	-.018	-.014	-.012	-.017	-.027	.362
.379							-.010			.379
.396	-.018	-.023	-.022	-.021	-.008	-.002	.005	-.014	-.022	.396
.408							.020			.408
.426	.182	.168	.168	.166	.158	.161	.160	.162	.166	.426
.451	.168	.149	.143	.137	.135	.133	.135		.146	.451
.476	.153	.125	.118	.116	.115	.114	.116	.117	.129	.476
.501	.138	.113	.108	.100	.096	.095	.098	.100	.110	.501
.530	-.052	-.073	-.077	-.083	-.079	-.081	-.080	-.075	-.076	.530
.548							.061			.548
.566							.047			.566
.584	-.025	-.049	-.050	-.048	-.049	-.052	-.048	-.041	-.052	.584
.602							.041			.602
.620							.034			.620
.638	-.016	-.029		-.032	-.027				-.028	.638
.655							.022			.655
.673							.020			.673
.691	-.011	-.035	-.033	-.032	-.028	-.022	-.010	-.017	-.025	.691
.709							.013			.709
.727							.007			.727
.745	-.001	-.027		-.021	-.013		.010		-.018	.745
.763							.008			.763
.781							.011			.781
.799							.003			.799
.817	.008	-.015	-.014	-.011	-.006	-.002			-.001	.817
.834							.000			.834
.852							.002			.852
.870							.004			.870
.888							.008			.888
.906							.008			.906
.921	-.084	.017	.000		-.008	.003			-.011	.921
.924							.011			.924
.947							.008			.947
.964							.012			.964
.981	.041	.014	.006		.007	.009	.007	.002	.005	.981
1.000	-.091									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	1.043		.031	.637	.521
.030	1.413		.094	.996	.776
.050	1.523		.156	1.281	.988
.070	1.554		.219	1.459	1.201
.090	1.565		.281	1.512	1.365
.110	1.534		.344	1.528	1.461
.130	1.525		.406	1.527	1.498
.150	1.500		.469	1.536	1.511
.170	1.469		.750	1.531	
.190	1.382		1.000	1.539	
			1.250	1.540	
			1.500	1.543	

TABLE 6. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
.039							1.558			.039
.040							1.505			.040
.041							1.317			.041
.044							.911			.044
.049							.322			.049
.055							.420			.055
.060	.563	.595	.682	.662	.656	.593	.576	.565		.060
.075	-.043	-.034	-.036	-.039	-.029	-.041	-.045	-.047		.075
.085	.029	.033	.032	.027	.026	.030	.012	.026	.021	.085
.096	.153									.096
.126		.127	.125	.121	.123	.122	.118	.111	.114	.126
.143							.177			.143
.154	-.137	-.131	-.134	-.137	-.140	-.139	-.136	-.145		.154
.171							.117			.171
.189							.096			.189
.206	-.078	-.081	-.078	-.087	-.078	-.077	-.079	-.080		.206
.223							.062			.223
.241							.055			.241
.258	-.054	-.044	-.047	-.046	-.049	-.048	-.053	-.045	-.044	.258
.275							.045			.275
.293							.038			.293
.310	-.026	-.032	-.032	-.027	-.028	-.022	-.022	-.027	-.033	.310
.327							.018			.327
.344							.019			.344
.362	-.017	-.015	-.011	-.010	-.020	-.017	-.016	-.015	-.014	.362
.379							.018			.379
.396	-.017	-.011	-.015	-.016	-.014	-.013	-.006	-.014	-.011	.396
.408							.012			.408
.426	.170	.174	.177	.176	.175	.179	.175	.174	.174	.426
.451	.149	.155	.154	.156	.153	.148	.151		.152	.451
.476	.131	.134	.132	.130	.131	.129	.130	.133	.138	.476
.501	.114	.118	.120	.116	.118	.115	.117	.118	.120	.501
.530	-.066	-.065	-.067	-.072	-.068	-.070	-.068	-.067	-.069	.530
.548							.055			.548
.566							.042			.566
.584	-.028	-.041	-.039	-.041	-.040	-.041	-.034	-.039	-.042	.584
.602							.034			.602
.620							.032			.620
.638	-.026	-.020			-.022	-.021			-.018	.638
.655							.020			.655
.673							.021			.673
.691	-.016	-.025	-.024	-.024	-.031	-.027	-.015	-.014	-.016	.691
.709							.014			.709
.727							.010			.727
.745	-.013	-.018			-.014	-.013			-.008	.745
.763							.008			.763
.781							.009			.781
.799							.017			.799
.817	.003	-.006	-.006	-.006	-.006	-.003	-.000	-.000	.007	.817
.834							.006			.834
.852							.002			.852
.870							.003			.870
.888							.006			.888
.906							.008			.906
.921	-.067									.921
.924	.013	.008			-.005	.005				.924
.947							.010			.947
.964							.008			.964
.981	.047	.029	.012			.011	.010	.007	.010	.981
1.000	-.069						.008			1.000

Orifice station, $y, \text{in.}$	Nose rake		Base rake		
	Cp _{top}	Cp _{side}	Orifice station, $y, \text{in.}$	Cp _{top}	Cp _{side}
.010	.884		.031	.476	.501
.030	1.359		.094	.671	.690
.050	1.505		.156	.824	.844
.070	1.540		.219	.975	.996
.090	1.546		.281	1.100	1.138
.110	1.495		.344	1.229	1.266
.130	1.480		.406	1.332	1.370
.150	1.452		.469	1.409	1.441
.170	1.425		.750	1.488	
.190	1.334		1.000	1.506	
			1.250	1.517	
			1.500	1.527	

TABLE 6. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.556			.039
.040							1.530			.040
.041							1.372			.041
.044							.992			.044
.049							.420			.049
.055							.619			.055
.060	.361	.703	.698	.677	.694	.684	.669	.674		.060
.075	-.069	-.043	-.035	-.026	.000	-.006	-.007	-.028		.075
.085	-.016	.026	.037	.044	.070	.077	.065	.050	.011	.085
.096	.082									.096
.126		.124	.136	.146	.172	.176	.174	.147	.111	.126
.143							.145			.143
.154	-.168	-.135	-.130	-.123	-.113	-.108	-.103	-.125		.154
.171							.089			.171
.189							.071			.189
.206	-.092	-.088	-.076	-.081	-.061	-.056	-.057	-.070		.206
.223							.044			.223
.241							.041			.241
.258	-.052	-.054	-.052	-.049	-.038	-.034	-.041	-.044	-.055	.258
.275							.042			.275
.293							.033			.293
.310	-.023	-.041	-.043	-.035	-.030	-.024	-.024	-.035	-.043	.310
.327							.017			.327
.344							.017			.344
.362	-.012	-.026	-.021	-.017	-.020	-.015	-.015	-.022	-.024	.362
.379							.017			.379
.396	-.011	-.019	-.023	-.024	-.017	-.014	-.008	-.022	-.019	.396
.408							.016			.408
.426	.163	.167	.173	.175	.179	.188	.183	.175	.166	.426
.451	.138	.148	.151	.157	.167	.164	.168		.144	.451
.476	.118	.128	.131	.135	.146	.147	.148	.142	.131	.476
.501	.098	.111	.120	.122	.136	.136	.138	.131	.113	.501
.530	-.075	-.072	-.069	-.070	-.058	-.055	-.053	-.060	-.075	.530
.548							.040			.548
.566							.028			.566
.584	-.038	-.047	-.042	-.040	-.031	-.028	-.020	-.033	-.051	.584
.602							.023			.602
.620							.022			.620
.638	-.029	-.027		-.023	-.015				-.026	.638
.655							.012			.655
.673							.014			.673
.691	-.015	-.034	-.029	-.027	-.030	-.023	-.010	-.017	-.024	.691
.709							.006			.709
.727							.011			.727
.745	-.010	-.025		-.019	-.008		.004		.016	.745
.763							.008			.763
.781							.016			.781
.799							.005			.799
.817	.003	-.014	-.012	-.012	-.006	-.001	.003	-.004	-.002	.817
.834							.003			.834
.852							.006			.852
.870							.005			.870
.888							.010			.888
.906							.009			.906
.921	-.075	-.004		.002			.013			.921
.924							.012			.924
.947							.017			.947
.964							.011			.964
.981	.059	.019	.003		.011	.011	.011	.005	.001	.981
1.000	-.082									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.885		.031	.425	.520
.030	1.352		.094	.580	.762
.050	1.491		.156	.681	.965
.070	1.518		.219	.762	1.172
.090	1.500		.281	.832	1.339
.110	1.454		.344	.909	1.443
.130	1.441		.406	.979	1.488
.150	1.421		.469	1.048	1.507
.170	1.394		.750	1.307	
.190	1.304		1.000	1.435	
			1.250	1.454	
			1.500	1.468	

TABLE 6. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.545			.039
.040							1.549			.040
.041							1.430			.041
.044							1.091			.044
.049							.611			.049
.055							.745			.055
.060	.208	.627	.650	.651	.706	.713	.724	.664		.060
.075	-.122	-.058	-.036	-.006	.044	.047	.048	.004		.075
.085	-.039	.000	.033	.065	.120	.145	.141	.087	-.012	.085
.096	.034									.096
.126		.112	.144	.174	.229	.238	.239	.191	.099	.126
.143							-.107			.143
.154	-.194	-.146	-.132	-.111	-.081	-.070	-.062	-.105		.154
.171							-.050			.171
.189							-.037			.189
.206	-.095	-.106	-.091	-.083	-.039	-.029	-.027	-.061		.206
.223							-.017			.223
.241							-.018			.241
.258	-.043	-.089	-.081	-.058	-.025	-.013	-.019	-.046	-.092	.258
.275							-.020			.275
.293							-.017			.293
.310	-.018	-.072	-.069	-.054	-.026	-.014	-.011	-.047	-.073	.310
.327							-.015			.327
.344							-.014			.344
.362	-.006	-.062	-.058	-.047	-.024	-.011	-.006	-.040	-.058	.362
.379							-.008			.379
.396	-.007	-.048	-.050	-.042	-.016	-.007	-.000	-.036	-.048	.396
.408							-.010			.408
.426	.164	.139	.144	.151	.177	.195	.193	.161	.139	.426
.451	.135	.121	.128	.140	.170	.175	.182		.119	.451
.476	.112	.102	.110	.128	.168	.177	.180	.147	.105	.476
.501	.091	.093	.108	.117	.150	.158	.163	.136	.092	.501
.530	-.094	-.091	-.081	-.073	-.043	-.038	-.035	-.058	-.098	.530
.548							-.018			.548
.566							-.007			.566
.584	-.046	-.072	-.059	-.046	-.020	-.014	-.007	-.033	-.075	.584
.602							-.002			.602
.620							-.004			.620
.638	-.028	-.054		-.028	-.007				-.052	.638
.655							-.004			.655
.673							-.000			.673
.691	-.013	-.061	-.055	-.040	-.027	-.012	-.003	-.023	-.051	.691
.709							-.000			.709
.727							-.005			.727
.745	-.000	-.053		-.037	-.007		-.002		-.043	.745
.763							-.009			.763
.781							-.008			.781
.799							-.001			.799
.817	.008	-.040	-.036	-.031	-.007	.005	.012	-.016	-.029	.817
.834							-.004			.834
.852							-.014			.852
.870							-.016			.870
.888							-.014			.888
.906							-.022			.906
.921	-.133									.921
.924	.017	-.020							-.035	.924
.947										.947
.964										.964
.981	.077	-.009	-.020						-.021	.981
1.000	-.147									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.635		.031	.678	.530
.030	1.247		.094	1.098	.918
.050	1.478		.156	1.304	1.247
.070	1.520		.219	1.367	1.496
.090	1.506		.281	1.381	1.536
.110	1.425		.344	1.391	1.508
.130	1.399		.406	1.401	1.511
.150	1.372		.469	1.414	1.514
.170	1.348		.520	1.425	
.190	1.259		1.000	1.441	
			1.250	1.451	
			1.500	1.456	

TABLE 6. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 1.57$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.039							1.517				.039
.040							1.557				.040
.041							1.482				.041
.044							1.198				.044
.049							.793				.049
.055							.847				.055
.060	.102	.540	.617	.666	.766	.783	.795	.702			.060
.075	-.149	-.072	-.034	.020	.108	.123	.127	.049			.075
.085	-.096	-.015	.040	.105	.202	.239	.245	.149	-.024		.085
.096	-.031										.096
.126		.091	.147	.202	.298	.319	.323	.238	.079		.126
.143							-.057				.143
.154	-.208	-.171	-.138	-.099	-.040	-.019	-.007	-.079			.154
.171							.001				.171
.189							.014				.189
.206	-.094	-.150	-.121	-.088	-.008	.015	.020	-.046			.206
.223							.026				.223
.241							.021				.241
.258	-.036	-.145	-.120	-.082	-.003	.022	.017	-.049	-.148		.258
.275							.015				.275
.293							.013				.293
.310	-.010	-.154	-.129	-.084	-.015	.010	.018	-.057	-.148		.310
.327							.010				.327
.344							.009				.344
.362	.001	-.132	-.134	-.095	-.018	.009	.018	-.059	-.126		.362
.379							.015				.379
.396	.002	-.107	-.117	-.096	-.015	.009	.022	-.067	-.107		.396
.408							.010				.408
.426	.239	.079	.072	.088	.173	.216	.221	.124	.085		.426
.451	.146	.070	.075	.094	.168	.194	.207		.071		.451
.476	.112	.052	.068	.098	.169	.193	.202	.136	.054		.476
.501	.085	.044	.068	.093	.161	.183	.195	.130	.043		.501
.530	-.110	-.125	-.110	-.087	-.030	-.014	-.007	-.056	-.133		.530
.548							.008				.548
.566							.020				.566
.584	-.051	-.113	-.092	-.066	-.005	.012	.023	-.037	-.119		.584
.602							.023				.602
.620							.025				.620
.638	-.032	-.097		-.059	.004						.638
.655							.032				.655
.673							.027				.673
.691	-.020	-.116	-.097	-.070	-.016	.010	.030	-.033	-.103		.691
.709							.031				.709
.727							.026				.727
.745	-.012	-.106		-.071	.000						.745
.763							.029				.763
.781							.024				.781
.799							.012				.799
.817	-.004	-.084	-.090	-.070	-.004	.023	.035	-.032	-.074		.817
.834							.027				.834
.852							.040				.852
.870							.032				.870
.888							.035				.888
.906							.036				.906
.921	-.183	-.002	-.059		-.062	.002		.041			.921
.924							.040				.924
.947							.047				.947
.964							.047				.964
.981	.045	-.038	-.062			.014	.032	.038	-.017	-.057	.981
1.000	-.192										1.000

Nose rake			Base rake		
Orifice station, y,in.	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.494		.031	.856	.451
.030	1.120		.094	1.459	1.030
.050	1.432		.156	1.467	1.483
.070	1.491		.219	1.411	1.613
.090	1.459		.281	1.424	1.523
.110	1.366		.344	1.416	1.490
.130	1.335		.406	1.416	1.499
.150	1.312		.469	1.420	1.506
.170	1.289		.750	1.414	
.190	1.204		1.000	1.422	
			1.250	1.429	
			1.500	1.443	

TABLE 7. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT M = 2.29
(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.638			.039
.040							1.480			.040
.041							1.202			.041
.044							.754			.044
.049							.206			.049
.055							.140			.055
.060	.791	.449	.380	.301	.201	.175	.172	.237		.060
.075	.226	.061	.024	-.007	-.049	-.063	-.070	-.036		.075
.085	.255	.067	.034	-.002	-.036	-.049	-.055	-.021	.061	.085
.096	.324									.096
.126		.075	.039	.008	-.015	-.017	-.014	-.006	.072	.126
.143							.135			.143
.154	.037	-.076	-.097	-.112	-.125		.120	-.123	-.079	.154
.171							.106			.171
.189							.097			.189
.206	.041	-.084	-.101	-.113	-.097	-.086	-.083	-.106	-.084	.206
.223							.064			.223
.241							.053			.241
.258	.037	-.088	-.109	-.118	-.070	-.051	-.045	-.098	-.090	.258
.275							.037			.275
.293							.033			.293
.310	.035	-.102	-.123	-.121	-.053	-.035	-.028	-.068	-.103	.310
.327							.024			.327
.344							.021			.344
.362	.032	-.112	-.135	-.082	-.050	-.032	-.018	-.059	-.114	.362
.379							.016			.379
.396	.029	-.120	-.130	-.071	-.055	-.028	-.011	-.058	-.121	.396
.408							.011			.408
.426	.171	-.053	-.007	.005	.014	.074	.094	.010	-.054	.426
.451	.169	-.050	-.004	.014	.006	.058	.094	.018	-.053	.451
.476	.168	-.048	-.002	.016	-.003	.048	.066	.016	-.050	.476
.501	.168	.000	.015	-.012	.039	.050	-.054	-.060	-.104	.501
.530	.022	-.107	-.087	-.062	-.095					.530
.548							.053			.548
.566							.052			.566
.584	.027	-.091	-.072	-.058	-.090	-.051	-.044	-.063	-.090	.584
.602							.037			.602
.620							.033			.620
.638	.025	-.088		-.061	-.077		.027		-.089	.638
.655							.025			.655
.673							.025			.673
.691							.025			.691
.709							.025			.709
.727							.024			.727
.745	.026	-.090		-.051	-.048		.024		-.091	.745
.763							.024			.763
.781							.028			.781
.799							.027			.799
.817	.025	-.098	-.056	-.044	-.047	-.037	-.022	-.040	-.096	.817
.834							.025			.834
.870							.025			.870
.888							.027			.888
.921	-.166						.025			.921
.924	.024	-.074			-.042		.026			.924
.947							.026			.947
.964							.026			.964
.981	.059	-.060	-.038			-.042	-.040	-.038	-.066	.981
.984							.026		-.079	.984
.998							.029			.998
1.000	-.168									1.000

Nose rake			Base rake		
Orifice station, y/l	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	1.273		.031	.727	.261
.030	1.537		.094	1.709	.815
.050	1.612		.156	1.716	1.308
.070	1.648		.219	1.754	1.395
.090	1.670		.281	1.750	1.440
.110	1.693		.344	1.753	1.454
.130	1.723		.406	1.750	1.471
.150	1.734		.469	1.744	1.481
.170	1.743		.750	1.751	
.190	1.668		1.000	1.750	
			1.250	1.736	
			1.500	1.720	

TABLE 7. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.666			.039
.040							1.546			.040
.041							1.295			.041
.044							.861			.044
.049							.282			.049
.055							.221			.055
.060	.648	.442	.403	.351	.284	.265	.266	.306		.060
.075	.155	.062	.039	.016	-.009	-.020	-.026	-.005		.075
.085	.175	.071	.050	.028	.002	-.008	-.011	.011	.066	.085
.096	.237									.096
.126		.084	.062	.043	.023	.019	.018	.031	.083	.126
.143							-.118			.143
.154	-.007	-.070	-.081	-.092	-.101		-.105	-.099	-.071	.154
.171							-.094			.171
.189							-.085			.189
.206	-.003	-.066	-.073	-.080	-.078	-.076	-.075	-.079	-.066	.206
.223							-.066			.223
.241							-.057			.241
.258	.000	-.063	-.072	-.072	-.055	-.047	-.050	-.063	-.061	.258
.275							-.041			.275
.293							-.036			.293
.310	.000	-.061	-.066	-.061	-.040	-.032	-.029	-.044	-.060	.310
.327							-.026			.327
.344							-.022			.344
.362	-.002	-.064	-.063	-.053	-.029	-.021	-.019	-.035	-.061	.362
.379							-.018			.379
.396	-.004	-.064	-.061	-.048	-.025	-.017	-.012	-.033	-.060	.396
.408							-.013			.408
.426	.122	.033	.044	.056	.074	.083	.084	.067	.038	.426
.451	.122	.039	.044	.053	.068	.071	.076	.060	.042	.451
.476	.119	.042	.042	.048	.056	.059	.062	.053	.043	.476
.501	.117	.039	.039	.042	.046	.047	.050		.041	.501
.530	-.014	-.059	-.061	-.057	-.050		-.050	-.051	-.061	.530
.548							-.046			.548
.566							-.038			.566
.584	-.005	-.054	-.055	-.051	-.044	-.039	-.035	-.045	-.053	.584
.602							-.031			.602
.620							-.028			.620
.638	-.004	-.051		-.045	-.032		-.025		-.051	.638
.655							-.023			.655
.673							-.022			.673
.691	-.003	-.053	-.050		-.028	-.026	-.017		-.051	.691
.709							-.016			.709
.727							-.015			.727
.745	-.000	-.050		-.033	-.024		-.015		-.048	.745
.763							-.015			.763
.781							-.018			.781
.799							-.015			.799
.817	-.000	-.046	-.037	-.028	-.022	-.018	-.011	-.021	-.039	.817
.834							-.012			.834
.870							-.009			.870
.888							-.009			.888
.921	-.152	-.008	-.040		-.025					.921
.924							-.006			.924
.947							-.007			.947
.964							-.005			.964
.981	.026	-.036	-.027		-.018	-.013	-.007	-.017	-.031	.981
.984					-.020				-.038	.984
.998							-.010			.998
1.000	-.153									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.180		.031	.577	.369
.030	1.471		.094	1.285	.809
.050	1.520		.156	1.467	1.104
.070	1.559		.219	1.601	1.280
.090	1.579		.281	1.599	1.411
.110	1.597		.344	1.613	1.453
.130	1.610		.406	1.626	1.468
.150	1.624		.469	1.631	1.497
.170	1.624		.750	1.646	
.190	1.548		1.000	1.659	
			1.250	1.675	
			1.500	1.675	

TABLE 7. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.678			.039
.040							1.602			.040
.041							1.375			.041
.044							.950			.044
.049							.368			.049
.055							.310			.055
.060	.532	.440	.427	.397	.363	.351	.355	.372		.060
.075	.102	.063	.050	.039	.029	.020	.017	.026		.075
.085	.117	.073	.063	.052	.039	.033	.031	.043	.069	.085
.096	.173									.096
.126		.089	.078	.069	.058	.056	.055	.062	.089	.126
.143							.098			.143
.154	-0.041	-0.065	-0.072	-0.076	-0.083		-0.085	-0.081	-0.067	.154
.171							.077			.171
.189							.072			.189
.206	-0.029	-0.054	-0.058	-0.060	-0.064	-0.064	-0.065	-0.062	-0.053	.206
.223							.058			.223
.241							.053			.241
.258	-0.025	-0.044	-0.048	-0.050	-0.048	-0.046	-0.048	-0.047	-0.041	.258
.275							.044			.275
.293							.041			.293
.310	-0.020	-0.040	-0.042	-0.039	-0.036	-0.035	-0.034	-0.037	-0.036	.310
.327							.031			.327
.344							.028			.344
.362	-0.017	-0.033	-0.031	-0.031	-0.029	-0.025	-0.024	-0.023	-0.027	.362
.379							.023			.379
.396	-0.019	-0.031	-0.030	-0.027	-0.022	-0.020	-0.018	-0.022	-0.027	.396
.408							.019			.408
.426	.099	.078	.085	.081	.087	.087	.083	.087	.082	.426
.451	.098	.076	.075	.075	.079	.078	.078	.078	.077	.451
.476	.094	.071	.068	.069	.069	.069	.069	.071	.073	.476
.501	.091	.065	.065	.062	.060	.057	.058	.067	.067	.501
.530	-0.029	-0.041	-0.044	-0.045	-0.045		-0.039	-0.044	-0.040	.530
.548							.035			.548
.566							.031			.566
.584	-0.024	-0.036	-0.038	-0.038	-0.033	-0.033	-0.030	-0.032	-0.034	.584
.602							.028			.602
.620							.026			.620
.638	-0.019	-0.033			-0.027	-0.026			-0.031	.638
.655							.024			.655
.673							.021			.673
.691	-0.016	-0.030	-0.027			-0.023	-0.023			.691
.709							.019			.709
.727							.018			.727
.745	-0.013	-0.021			-0.022	-0.020			-0.019	.745
.763							.017			.763
.781							.016			.781
.799							.017			.799
.817	-0.010	-0.021	-0.021	-0.020	-0.017	-0.014			-0.014	.817
.834							.015			.834
.870							.011			.870
.888							.008			.888
.921	-0.132						.007			.921
.924	-0.006	-0.019			-0.017					.924
.947							.005			.947
.964							.006			.964
.981	.020	-0.016	-0.014			-0.007			-0.009	.981
.984							.005			.984
.998							.012			.998
1.000	-0.136						.016			1.000

Orifice station, $y, \text{in.}$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y, \text{in.}$	Cp top	Cp side
.010	1.102		.031	.445	.347
.030	1.407		.094	.869	.600
.050	1.458		.156	1.108	.794
.070	1.480		.219	1.323	.948
.090	1.509		.281	1.476	1.090
.110	1.522		.344	1.533	1.235
.130	1.529		.406	1.567	1.342
.150	1.524		.469	1.581	1.424
.170	1.521		.750	1.584	
.190	1.444		1.000	1.578	
			1.250	1.605	
			1.500	1.615	

TABLE 7. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.683			.039
.040							1.634			.040
.041							1.444			.041
.044							1.034			.044
.049							1.448			.049
.055							1.395			.055
.060	.432	.439	.450	.444	.445	.439	.448	.438		.060
.075	.055	.062	.061	.063	.068	.063	.062	.058		.075
.085	.071	.074	.075	.076	.078	.077	.076	.076	.071	.085
.096	.121									.096
.126		.091	.092	.092	.094	.095	.095	.092	.092	.126
.143							.074			.143
.154	-.066	-.065	-.064	-.064	-.063		-.062	-.064	-.064	.154
.171							-.058			.171
.189							-.055			.189
.206	-.048	-.050	-.050	-.050	-.049	-.048	-.050	-.049	-.049	.206
.223							-.046			.223
.241							-.044			.241
.258	-.038	-.040	-.039	-.038	-.036	-.038	-.040	-.037	-.035	.258
.275							-.037			.275
.293							-.036			.293
.310	-.030	-.032	-.033	-.032	-.032	-.032	-.031	-.027	-.028	.310
.327							-.029			.327
.344							-.028			.344
.362	-.024	-.027	-.028	-.027	-.027	-.025	-.025	-.020	-.022	.362
.379							-.024			.379
.396	-.022	-.024	-.025	-.024	-.022	-.022	-.019	-.019	-.020	.396
.408							-.021			.408
.426	.090	.088	.095	.091	.088	.087	.085	.093	.093	.426
.451	.085	.088	.088	.087	.086	.083	.085	.086	.091	.451
.476	.080	.080	.080	.079	.080	.080	.081	.083	.083	.476
.501	.075		.075	.074	.074	.074	.075		.075	.501
.530	-.034	-.035	-.036	-.035	-.036		-.036	-.033	-.034	.530
.548							-.035			.548
.566							-.030			.566
.584	-.029	-.030	-.030	-.031	-.032	-.031	-.027	-.030	-.028	.584
.602							-.025			.602
.620							-.023			.620
.638	-.024	-.027		-.024	-.021		-.020		-.025	.638
.655							-.019			.655
.673							-.018			.673
.691	-.020	-.024	-.021		-.022	-.019	-.013		-.020	.691
.709							-.014			.709
.727							-.015			.727
.745	-.016	-.016		-.017	-.018		-.014		-.012	.745
.763							-.014			.763
.781							-.016			.781
.799							-.014			.799
.817	-.011	-.014	-.014	-.017	-.016	-.015	-.010	-.010	-.006	.817
.834							-.012			.834
.870							-.009			.870
.888							-.008			.888
.921	-.121						-.005			.921
.924	-.008	-.013		-.017			-.007			.924
.947							-.005			.947
.964							-.008			.964
.981	.010	-.010	-.010		-.009	-.008	-.004	-.003		.981
.984					-.012		-.012		-.009	.984
.998										.998
1.000	-.125									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	1.008		.031	.291	.316
.030	1.334		.094	.508	.515
.050	1.387		.156	.647	.688
.070	1.405		.219	.778	.846
.090	1.421		.281	.885	.980
.110	1.438		.344	1.000	1.113
.130	1.447		.406	1.103	1.237
.150	1.445		.469	1.205	1.305
.170	1.432		.750	1.423	
.190	1.360		1.000	1.496	
			1.250	1.544	
			1.500	1.572	

TABLE 7. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.039							1.675				.039
.040							1.651				.040
.041							1.502				.041
.044							1.107				.044
.049							.533				.049
.055							.494				.055
.060	.340	.439	.474	.492	.531	.536	.568	.508			.060
.075	.011	.061	.072	.085	.109	.106	.109	.089			.075
.085	.026	.072	.086	.099	.118	.123	.123	.108	.069		.085
.096	.071										.096
.126		.087	.100	.112	.133	.139	.141	.123	.089		.126
.143											.143
.154	-.089	-.066	-.060	-.053	-.040		-.049	-.036	-.047	-.067	.154
.171											.171
.189											.189
.206	-.063	-.055	-.050	-.042	-.031	-.030	-.029	-.038	-.054		.206
.223											.223
.241											.241
.258	-.047	-.046	-.042	-.038	-.026	-.024	-.026	-.031	-.042		.258
.275											.275
.293											.293
.310	-.033	-.039	-.036	-.031	-.024	-.022	-.020	-.025	-.038		.310
.327											.327
.344											.344
.362	-.023	-.037	-.032	-.029	-.021	-.019	-.019	-.022	-.030		.362
.379											.379
.396	-.018	-.034	-.034	-.030	-.021	-.019	-.015	-.023	-.028		.396
.408											.408
.426	.084	.075	.080	.082	.090	.099	.098	.089	.078		.426
.451	.079	.073	.075	.080	.094	.096	.100	.088	.075		.451
.476	.070	.069	.071	.077	.089	.093	.095	.086	.071		.476
.501	.064		.069	.073	.084	.086	.088		.066		.501
.530	-.037	-.042	-.041	-.039	-.035						.530
.548											.548
.566											.566
.584	-.033	-.038	-.037	-.032	-.025	-.024	-.018	-.024	-.037		.584
.602											.602
.620											.620
.638	-.027	-.032		-.023	-.018				-.028		.638
.655											.655
.673											.673
.691											.691
.709											.709
.727											.727
.745	-.016	-.023		-.024	-.017				-.022		.745
.763											.763
.781											.781
.799											.799
.817	-.010	-.024	-.024	-.022	-.015	-.011	-.006	-.013	-.017		.817
.834											.834
.870											.870
.888											.888
.921	-.128										.921
.924	-.007										.924
.947											.947
.964											.964
.981	.008	-.018	-.018		-.009	-.007					.981
.984											.984
.998											.998
1.000	-.133										1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.891		.031	.230	.360
.030	1.247		.094	.372	.647
.050	1.321		.156	.464	.867
.070	1.337		.219	.560	1.029
.090	1.350		.281	.642	1.188
.110	1.361		.344	.720	1.331
.130	1.361		.406	.789	1.427
.150	1.361		.469	.852	1.466
.170	1.347		.750	1.075	
.190	1.277		1.000	1.229	
			1.250	1.290	
			1.500	1.345	

TABLE 7. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT M = 2.29 - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -								Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	
.039							1.670		.039
.040							1.681		.040
.041							1.553		.041
.044							1.203		.044
.049							.615		.049
.055							.615		.055
.060	.252	.441	.504	.550	.633	.649	.667	.592	.060
.075	-.031	.060	.084	.110	.157	.161	.165	.127	.075
.085	-.014	.070	.096	.125	.168	.181	.184	.148	.085
.096	.024							.069	.096
.126		.082	.109	.136	.183	.197	.202	.160	.126
.143							-.012		.143
.154	-.106	-.071	-.056	-.039	-.012		-.000	-.027	.154
.171							-.000		.171
.189							-.000		.189
.206	-.072	-.066	-.053	-.038	-.008	-.003	-.000	-.024	.206
.223							.004		.223
.241							.003		.241
.258	-.049	-.064	-.052	-.034	-.006	.001	.002	-.021	.258
.275							.001		.275
.293							-.000		.293
.310	-.028	-.066	-.054	-.038	-.010	-.000	.003	-.022	.310
.327							.002		.327
.344							.001		.344
.362	-.018	-.070	-.059	-.042	-.013	-.002	.001	-.024	.362
.379							.000		.379
.396	-.015	-.069	-.062	-.046	-.013	-.003	.002	-.027	.396
.408							-.000		.408
.426	.083	.027	.041	.058	.102	.121	.126	.086	.426
.451	.075	.033	.040	.060	.106	.117	.124	.085	.451
.476	.062	.036	.039	.057	.098	.110	.115	.082	.476
.501	.053		.043	.054	.095	.110	.119	.087	.501
.530	-.046	-.064	-.061	-.052	-.021		-.006	-.034	.530
.548							-.002		.548
.566							-.000		.566
.584	-.036	-.057	-.048	-.039	-.016	-.008	-.002	-.026	.584
.602							.000		.602
.620							.000		.620
.638	-.029	-.055		-.034	-.013		-.001		.638
.655							-.001		.655
.673							-.001		.673
.691	-.023	-.055	-.050		-.018	-.008	-.000	-.022	.691
.709							-.000		.709
.727							-.000		.727
.745	-.017	-.051		-.038	-.010		.002		.745
.763							.001		.763
.781							-.001		.781
.799							-.000		.799
.817	-.011	-.051	-.048	-.039	-.010	-.000	.006	-.019	.817
.834							.002		.834
.870							.010		.870
.888							.008		.888
.921	-.149						.007		.921
.924	-.008						.005		.924
.947							.008		.947
.964							.004		.964
.981	.013	-.041	-.044		-.010			-.014	.981
.984					-.012				.984
.998									.998
1.000	-.150						.003		1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.736		.031	.474	.363
.030	1.135		.094	.982	.848
.050	1.235		.156	1.015	1.143
.070	1.251		.219	1.061	1.315
.090	1.261		.281	1.083	1.420
.110	1.267		.344	1.098	1.449
.130	1.266		.406	1.110	1.468
.150	1.257		.469	1.124	1.480
.170	1.243		.750	1.176	
.190	1.180		1.000	1.203	
				1.250	1.220
				1.500	1.220

TABLE 7. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.29$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.652			.039
.040							1.687			.040
.041							1.616			.041
.044							1.308			.044
.049							.762			.049
.055							.796			.055
.060	.167	.451	.546	.624	.755	.785	.809	.692		.060
.075	-.072	.058	.095	.139	.217	.232	.239	.171		.075
.085	-.050	.067	.107	.158	.235	.259	.265	.201	.067	.085
.096	-.022									.096
.126		.075	.116	.162	.247	.274	.279	.207	.075	.126
.143										.143
.154	-.120	-.079	-.053	-.025	.022		.046	-.001	-.080	.154
.171										.171
.189										.189
.206	-.078	-.084	-.057	-.029	.026	.041	.047	.000	-.084	.206
.223										.223
.241										.241
.258	-.040	-.093	-.066	-.034	.023	.040	.043	-.004	-.088	.258
.275										.275
.293										.293
.310	-.026	-.107	-.078	-.044	.014	.033	.041	-.009	-.103	.310
.327										.327
.344										.344
.362	-.016	-.120	-.089	-.051	.009	.030	.038	-.014	-.116	.362
.379										.379
.396	-.012	-.126	-.096	-.058	.007	.027	.039	-.020	-.122	.396
.408										.408
.426	.089	-.061	-.010	.038	.132	.168	.178	.097	-.058	.426
.451	.096	-.058	-.014	.037	.136	.162	.176	.094	-.057	.451
.476	.068	-.056	-.015	.037	.134	.163	.176	.095	-.054	.476
.501	.051								-.050	.501
.530	-.051	-.114	-.100	-.066	.001			.029	-.119	.530
.548										.548
.566										.566
.584	-.045	-.1	-.098	-.064	.001	.020	.031	-.025	-.103	.584
.602										.602
.620										.620
.638	-.029	-.093		-.063	.000					.638
.655										.655
.673										.673
.691	-.027	-.098	-.105		-.005	.017				.691
.709										.709
.727										.727
.745	-.026	-.096		-.071	.000					.745
.763										.763
.781										.781
.799										.799
.817	-.026	-.101	-.091	-.073	-.002	.020	.031	-.030	-.097	.817
.834										.834
.870										.870
.888										.888
.921	-.165									.921
.924	-.028	-.072		-.069						.924
.947										.947
.964										.964
.981	-.006	-.059	-.101		-.002	.017	.026	-.022	-.069	.981
.984										.984
.998										.998
1.000	-.169									1.000

Orifice station, $y, \text{in.}$	Nose rake			Base rake		
	Cp top	Cp side	Cp bottom	Orifice station, $y, \text{in.}$	Cp top	Cp side
.010	.440			.031	.532	.253
.030	.980			.094	1.189	.810
.050	1.130			.156	1.101	1.350
.070	1.154			.219	1.118	1.4410
.090	1.162			.281	1.121	1.4453
.110	1.163			.344	1.130	1.4463
.130	1.159			.406	1.140	1.4473
.150	1.162			.469	1.145	1.4476
.170	1.149			.750	1.177	
.190	1.090			1.000	1.220	
				1.250	1.257	
				1.500	1.290	

TABLE 8. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.98$ (a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.698			.039
.040							1.541			.040
.041							1.255			.041
.044							.788			.044
.049							.255			.049
.055							.175			.055
.060	.748	.417	.361	.296	.218	.197	.193	.254		.060
.075	.256	.092	.060	.029	-.002	-.012	-.017	.011		.075
.085	.262	.092	.061	.033	.001	-.008	-.010	.017	.092	.085
.096	.315									.096
.126		.079	.051	.027	.004	.000	.000	.013	.082	.126
.143		.061	-.035	-.051	-.063	-.074		-.076		.143
.154								-.070	-.072	-.035
.171								-.065		.171
.189								-.061		.189
.206	.055	-.046	-.060	-.067	-.062	-.057	-.055	-.066	-.046	.206
.223								-.048		.223
.241								-.041		.241
.258	.047	-.053	-.069	-.074	-.059	-.040	-.033	-.068	-.049	.258
.275								-.027		.275
.293								-.024		.293
.310	.045	-.060	-.077	-.083	-.050	-.026	-.022	-.068	-.062	.310
.327								-.018		.327
.344								-.016		.344
.362	.041	-.068	-.083	-.085	-.030	-.020	-.014	-.058	-.069	.362
.379								-.013		.379
.396	.039	-.073	-.089	-.075	-.032	-.020	-.010	-.058	-.073	.396
.408								-.012		.408
.426	.154	-.029	-.045	-.027	.016	.036	.050	-.021	-.031	.426
.451	.158	-.024	-.051	-.006	.009	.034	.054	-.012	-.023	.451
.476	.165	-.022	-.049	-.001	.003	.026	.042	-.005	-.020	.476
.501	.169	-.047	-.001	-.002	.019	.031		-.019		.501
.530	.046	-.075	-.080	-.053	-.054		-.037	-.040	-.074	.530
.548								-.036		.548
.566								-.032		.566
.584	.045	-.077	-.073	-.057	-.058	-.042	-.033	-.045	-.076	.584
.602								-.028		.602
.620								-.027		.620
.638	.042	-.078		-.052	-.051				-.080	.638
.655								-.024		.655
.673								-.022		.673
.691	.041	-.081	-.060		-.045	-.035			-.051	.691
.709								-.019		.709
.727								-.017		.727
.745	.042	-.071		-.046	-.047			-.018		.745
.763								-.019		.763
.781								-.021		.781
.799								-.021		.799
.817	.041	-.074	-.044	-.044	-.053	-.036		-.018		.817
.834								-.020		.834
.870								-.019		.870
.888								-.019		.888
.921	-.113							-.019		.921
.924	.040	-.054			-.041			-.021		.924
.947								-.020		.947
.964								-.020		.964
.981	.066	-.044	-.037			-.047		-.020		.981
.984								-.044	-.045	.984
.998								-.021	-.055	.998
1.000	-.114									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	1.237		.031	.702	.109
.030	1.554		.094	1.675	.390
.050	1.639		.156	1.888	.907
.070	1.675		.219	1.942	1.147
.090	1.716		.281	1.946	1.320
.110	1.762		.344	1.938	1.369
.130	1.795		.406	1.927	1.403
.150	1.829		.469	1.938	1.399
.170	1.847		.750	1.935	
.190	1.790		1.000	1.928	
			1.250	1.930	
			1.500	1.921	

TABLE 8. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
*039							1.736			*039
*040							1.616			*040
*041							1.357			*041
*044							.898			*044
*049							.324			*049
*055							.249			*055
*060	.614	.420	.387	.341	.288	.272	.272	.313		*060
*075	.186	.093	.071	.052	.031	.022	.017	.038		*075
*085	.191	.093	.074	.055	.032	.024	.023	.045	.095	*085
*096	.239									*096
*126		.085	.067	.051	.033	.029	.028	.042	.088	*126
*143							-.070			*143
*154	.022	-.032	-.042	-.051	-.059	-.062	-.055	-.032		*154
*171							-.058			*171
*189							-.053			*189
*206	.014	-.037	-.045	-.050	-.049	-.047	-.049	-.049	-.035	*206
*223							-.044			*223
*241							-.041			*241
*258	.013	-.041	-.047	-.048	-.042	-.039	-.038	-.043	-.035	*258
*275							-.034			*275
*293							-.032			*293
*310	.011	-.042	-.049	-.049	-.035	-.028	-.028	-.043	-.040	*310
*327							-.024			*327
*344							-.023			*344
*362	.009	-.044	-.049	-.046	-.022	-.018	-.019	-.032	-.041	*362
*379							-.016			*379
*396	.008	-.047	-.051	-.043	-.019	-.014	-.013	-.028	-.043	*396
*408							-.012			*408
*426	.103	.009	.012	.020	.046	.049	.050	.038	.013	*426
*451	.113	.016	.014	.025	.047	.049	.053	.038	.021	*451
*476	.110	.020	.018	.026	.039	.042	.045	.036	.024	*476
*501	.113		.020	.025	.034	.035	.039		.026	*501
*530	.010	-.046	-.044	-.039	-.030		-.023	-.030	-.041	*530
*548							-.022			*548
*566							-.020			*566
*584	.011	-.046	-.045	-.038	-.028	-.023	-.019	-.028	-.040	*584
*602							-.017			*602
*620							-.015			*620
*638	.010	-.045		-.032	-.023		-.014		-.039	*638
*655							-.013			*655
*673							-.012			*673
*691	.007	-.045	-.039		-.020	-.018		-.022	-.041	*691
*709							-.012			*709
*727							-.010			*727
*745	.010	-.034		-.024	-.018		-.010		-.040	*745
*763							-.009			*763
*781							-.010			*781
*799							-.009			*799
*817	.011	-.038	-.029	-.020	-.017	-.013	-.007	-.017	-.039	*817
*834							-.007			*834
*870							-.006			*870
*888							-.005			*888
*921	-.108						-.004			*921
*924	.011	-.035		-.019			-.005			*924
*947							-.003			*947
*964							-.003			*964
*981	.031	-.032	-.021		-.017	-.012	-.005	-.013	-.033	*981
*984					-.017				-.039	*984
*998							-.005			*998
1.000	-.108									1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	1.127		.031	.536	.274
.030	1.462		.094	1.246	.697
.050	1.521		.156	1.462	.946
.070	1.558		.219	1.663	1.160
.090	1.588		.281	1.697	1.313
.110	1.616		.344	1.719	1.373
.130	1.637		.406	1.727	1.415
.150	1.658		.469	1.739	1.441
.170	1.659		.750	1.751	
.190	1.593		1.000	1.779	
			1.250	1.789	
			1.500	1.775	

TABLE 8. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{t}$	C_p for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.743			.039
.040							1.658			.040
.041							1.429			.041
.044							1.994			.044
.049							1.404			.049
.055							1.320			.055
.060	.510	.419	.409	.383	.356	.347	.349	.370		.060
.075	.132	.092	.082	.071	.063	.057	.054	.065		.075
.085	.137	.094	.084	.075	.064	.059	.059	.071	.095	.085
.096	.179									.096
.126		.088	.079	.071	.062	.059	.058	.068	.092	.126
.143							.052			.143
.154	-.006	-.030	-.035	-.039	-.044		-.047	-.042	-.030	.154
.171							.044			.171
.189							-.042			.189
.206	-.009	-.030	-.033	-.035	-.038	-.038	-.039	-.037	-.028	.206
.223							.037			.223
.241							.035			.241
.258	-.010	-.028	-.031	-.032	-.032	-.032	-.033	-.031	-.023	.258
.275							.031			.275
.293							.028			.293
.310	-.007	-.025	-.028	-.029	-.027	-.026	-.026	-.026	-.025	.310
.327							.024			.327
.344							.023			.344
.362	-.006	-.024	-.025	-.025	-.023	-.021	-.021	-.024	-.022	.362
.379							.020			.379
.396	-.007	-.022	-.023	-.022	-.018	-.017	-.019	-.019	-.020	.396
.408							.017			.408
.426	.072	.050	.056	.053	.053	.053	.051	.055	.053	.426
.451	.078	.052	.053	.054	.057	.053	.053	.055	.055	.451
.476	.082	.053	.051	.051	.050	.050	.051	.053	.056	.476
.501	.081		.047	.048	.048	.047	.048		.057	.501
.530	-.007	-.021	-.024	-.023	-.021		-.016	-.020	-.018	.530
.548							.016			.548
.566							.014			.566
.584	-.006	-.023	-.022	-.021	-.018	-.017	-.017	-.019		.584
.602							.013			.602
.620							.013			.620
.638	-.005	-.021		.019	-.015		-.012		-.017	.638
.655							.012			.655
.673							.012			.673
.691	-.005	-.018	-.018		-.015	-.013	-.010		-.016	.691
.709							.010			.709
.727							.010			.727
.745	-.004	-.008		-.013	-.010		-.009		-.015	.745
.763							.009			.763
.781							.009			.781
.799							.007			.799
.817	-.001	-.014	-.011	-.010	-.008	-.007	-.005	-.011	-.014	.817
.834							.005			.834
.870							.003			.870
.888							.002			.888
.921	-.096									.921
.924	.000		-.010		-.009		-.003			.924
.947							-.004			.947
.964							-.003			.964
.981	.013	-.008	-.008		-.005	-.004	-.005	-.005	-.010	.981
.984							-.005		-.014	.984
.998							-.010			.998
1.000	-.099									1.000

Orifice station, y,in.	Nose rake		Base rake		
	C_p top	C_p side	Orifice station, y,in.	C_p top	C_p side
.010	1.000		.031	.359	.264
.030	1.345		.094	.760	.500
.050	1.417		.156	.957	.655
.070	1.443		.219	1.226	.790
.090	1.465		.281	1.422	.954
.110	1.484		.344	1.514	1.091
.130	1.501		.406	1.535	1.243
.150	1.501		.469	1.558	1.338
.170	1.496		.750	1.575	
.190	1.429		1.000	1.618	
			1.250	1.648	
			1.500	1.672	

TABLE 8. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.754			.039
.040							1.706			.040
.041							1.495			.041
.044							1.089			.044
.049							.479			.049
.055							.408			.055
.060	.422	.419	.428	.423	.425	.422	.428	.427		.060
.075	.089	.091	.091	.092	.096	.094	.093	.093		.075
.085	.094	.093	.094	.095	.097	.096	.097	.099	.096	.085
.096	.131									.096
.126		.089	.089	.090	.091	.092	.092	.093	.093	.126
.143							-.034			.143
.154	-.030	-.031	-.029	-.028	-.028	-.028	-.028	-.028	-.028	.154
.171							-.028			.171
.189							-.028			.189
.206	-.027	-.028	-.028	-.028	-.027	-.028	-.027	-.025	-.026	.206
.223							-.026			.223
.241							-.026			.241
.258	-.026	-.025	-.025	-.025	-.025	-.024	-.025	-.022	-.019	.258
.275							-.024			.275
.293							-.023			.293
.310	-.021	-.022	-.022	-.022	-.023	-.023	-.023	-.020	-.020	.310
.327							-.021			.327
.344							-.020			.344
.362	-.017	-.018	-.019	-.018	-.019	-.019	-.019	-.016	-.016	.362
.379							-.018			.379
.396	-.016	-.004	-.007	-.011	-.016	-.017	-.014	-.012	-.003	.396
.408							-.014			.408
.426	.057	.021	.031	.036	.059	.056	.056	.052	.022	.426
.451	.062	.059	.063	.060	.062	.060	.062	.060	.064	.451
.476	.060	.063	.061	.060	.060	.060	.062	.063	.067	.476
.501	.060						.058	.057	.061	.501
.530	-.013	-.016	-.017	-.016	-.016	-.016	-.013	-.012	-.012	.530
.548							-.013			.548
.566							-.012			.566
.584	-.014	-.016	-.015	-.016	-.015	-.016	-.013	-.011	-.012	.584
.602							-.012			.602
.620							-.013			.620
.638	-.013	-.016			-.014	-.015	-.012			.638
.655							-.011			.655
.673							-.011			.673
.691							-.011			.691
.709							-.010			.709
.727							-.009			.727
.745	-.010	-.003			-.010	-.011	-.008			.745
.763							-.007			.763
.781							-.008			.781
.799							-.008			.799
.817	-.008	-.007	-.007	-.006	-.007	-.007	-.005	-.007	-.007	.817
.834							-.006			.834
.870							-.006			.870
.888							-.006			.888
.921	-.088						-.003			.921
.924	-.004	-.004			-.008		-.004			.924
.947							-.001			.947
.964							-.001			.964
.981	.006	-.004	-.005				-.003	-.003	-.007	.981
.984							-.006			.984
.998										.998
1.000	-.091									1.000

Nose rake			Base rake		
Orifice station, y, in.	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.883		.031	.235	.236
.030	1.271		.094	.416	.401
.050	1.339		.156	.525	.538
.070	1.352		.219	.607	.649
.090	1.370		.281	.701	.764
.110	1.388		.344	.803	.885
.130	1.385		.406	.916	1.021
.150	1.385		.469	1.011	1.107
.170	1.377		.760	1.332	
.190	1.307		1.000	1.451	
				1.250	1.548
				1.500	1.596

TABLE 8. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.98$ - Continued
(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.757			.039
.040							1.725			.040
.041							1.563			.041
.044							1.167			.044
.049							.570			.049
.055							.493			.055
.060	.344	.418	.447	.463	.499	.505	.515	.486		.060
.075	.051	.091	.101	.112	.134	.135	.136	.123		.075
.085	.055	.093	.104	.116	.133	.137	.139	.129	.096	.085
.096	.089									.096
.126		.088	.097	.109	.126	.131	.133	.121	.092	.126
.143							-.011			.143
.154	-.048	-.030	-.025	-.019	-.010		-.005	-.014	-.029	.154
.171							-.006			.171
.189							-.007			.189
.206	-.039	-.029	-.025	-.020	-.012	-.010	-.008	-.016	-.029	.206
.223							-.008			.223
.241							-.009			.241
.258	-.033	-.028	-.025	-.021	-.012	-.011	-.009	-.016	-.024	.258
.275							-.010			.275
.293							-.012			.293
.310	-.026	-.027	-.024	-.021	-.012	-.009	-.010	-.013	-.026	.310
.327							-.009			.327
.344							-.008			.344
.362	-.020	-.026	-.022	-.017	-.013	-.008	-.007	-.013	-.024	.362
.379							-.007			.379
.396	-.017	-.023	-.021	-.018	-.013	-.009	-.006	-.013	-.023	.396
.408							-.008			.408
.426	.049	.048	.059	.055	.065	.070	.070	.062	.050	.426
.451	.054	.052	.055	.060	.074	.075	.078	.068	.054	.451
.476	.051	.050	.054	.061	.074	.079	.081	.072	.057	.476
.501	.049	.054	.060	.060	.074	.079	.081	.071	.055	.501
.530	-.017	-.022	-.020	-.017	-.010		-.005	-.011	-.019	.530
.548							-.007			.548
.566							-.006			.566
.584	-.015	-.024	-.020	-.019	-.014	-.009	-.007	-.012	-.019	.584
.602							-.005			.602
.620							-.005			.620
.638	-.013	-.020		-.015	-.010		-.004			.638
.655							-.004			.655
.673							-.004			.673
.691	-.012	-.019	-.017		-.011	-.007		-.010	-.017	.691
.709							-.004			.709
.727							-.004			.727
.745	-.010	-.011		-.013	-.009		-.003		-.016	.745
.763							-.004			.763
.781							-.006			.781
.799							-.008			.799
.817	-.007	-.014	-.013	-.012	-.009	-.006	-.004	-.010	-.014	.817
.834							-.004			.834
.870							-.000			.870
.888							-.001			.888
.921	-.093						.000			.921
.924	-.005	-.014		-.011			.000			.924
.947							.001			.947
.964										.964
.981	.006	-.010	-.011				-.000		-.010	.981
.984									-.015	.984
.998							-.002			.998
1.000	-.095									1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	.732		.031	.184	.270
.030	1.156		.094	.303	.522
.050	1.243		.156	.387	.687
.070	1.258		.219	.450	.842
.090	1.272		.281	.493	.994
.110	1.276		.344	.534	1.163
.130	1.276		.406	.582	1.296
.150	1.268		.469	.631	1.384
.170	1.254		.750	.810	
.190	1.193		1.000	.983	
			1.250	1.087	
			1.500	1.170	

TABLE 8. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{y}{c}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{c}$
	0	90	105	120	150	165	180	225	270	
.039							1.740			.039
.040							1.744			.040
.041							1.624			.041
.044							1.247			.044
.049							.650			.049
.055							.596			.055
.060	.267	.418	.471	.511	.589	.605	.620	.559		.060
.075	.015	.091	.112	.136	.180	.186	.191	.158		.075
.085	.021	.092	.114	.140	.178	.189	.194	.165	.095	.085
.096	.050									.096
.126										.126
.143										.143
.154	-.061	-.031	-.020	-.006	.015					.154
.171										.171
.189										.189
.206	-.046	-.035	-.026	-.013	.007	.012	.014	-.003	-.036	.206
.223										.223
.241										.241
.258	-.036	-.040	-.030	-.017	.007	.013	.015	-.003	-.035	.258
.275										.275
.293										.293
.310	-.026	-.042	-.032	-.019	.003	.010	.013	-.005	-.042	.310
.327										.327
.344										.344
.362	-.015	-.044	-.036	-.021	.000	.008	.011	-.008	-.043	.362
.379										.379
.396	-.012	-.047	-.038	-.024	-.000	.007	.011	-.011	-.046	.396
.408										.408
.426	.049	.009	.032	.043	.084	.098	.103	.068	.011	.426
.451	.053	.015	.031	.053	.097	.107	.113	.077	.017	.451
.476	.045	.017	.032	.053	.095	.107	.113	.080	.021	.476
.501	.041		.032	.052	.096	.107	.114		.022	.501
.530	-.022	-.045	-.038	-.027	-.000					.530
.548										.548
.566										.566
.584	-.018	-.044	-.037	-.026	-.000	.006	.012	-.009	-.041	.584
.602										.602
.620										.620
.638	-.015	-.042		-.025	-.001					.638
.655										.655
.673										.673
.691	-.013	-.043	-.037		-.004	.004				.691
.709										.709
.727										.727
.745	-.011	-.032		-.027	-.004					.745
.763										.763
.781										.781
.799										.799
.817	-.009	-.039	-.036	-.025	.001	.010	.013	-.012	-.041	.817
.834										.834
.870										.870
.888										.888
.921	-.103									.921
.924	-.007	-.035		-.026						.924
.947										.947
.964										.964
.981	.007	-.032	-.034		.000	.009		.011	-.012	.981
.984					-.003					.984
.998										.998
1.000	-.105									1.000

Orifice station, $y, in.$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y, in.$	Cp top	Cp side
.010	.528		.031	.362	.264
.030	1.031		.094	.768	.704
.050	1.152		.156	.784	.940
.070	1.151		.219	.831	1.169
.090	1.162		.281	.843	1.303
.110	1.159		.344	.855	1.364
.130	1.158		.406	.862	1.389
.150	1.154		.469	.871	1.429
.170	1.135		.750	.902	
.190	1.077		1.000	.938	
			1.250	.971	
			1.500	.993	

TABLE 8. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 2.98$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.710			.039
.040							1.758			.040
.041							1.674			.041
.044							1.352			.044
.049							.781			.049
.055							.737			.055
.060	.191	.421	.501	.573	.708	.739	.760	.653		.060
.075	-.018	.092	.127	.167	.241	.257	.265	.204		.075
.085	-.011	.092	.128	.172	.241	.262	.269	.213	.096	.085
.096	.012									.096
.126		.081	.116	.157	.233	.253	.263	.199	.084	.126
.143										.143
.154	-.073	-.037	-.017	.006	.046					.154
.171										.171
.189										.189
.206	-.054	-.048	-.029	-.004	.040	.052	.058	.020	-.046	.206
.223										.223
.241										.241
.258	-.033	-.055	-.035	-.010	.035	.048	.052	.014	-.049	.258
.275										.275
.293										.293
.310	-.021	-.063	-.042	-.017	.028	.043	.049	.009	-.062	.310
.327										.327
.344										.344
.362	-.016	-.071	-.050	-.022	.024	.040	.046	.006	-.068	.362
.379										.379
.396	-.014	-.075	-.055	-.027	.023	.039	.047	.003	-.073	.396
.408										.408
.426	.044	-.034	.009	.044	.123	.153	.162	.095	-.030	.426
.451	.051	-.028	.007	.049	.132	.156	.167	.099	-.022	.451
.476	.042	-.027	.008	.051	.134	.160	.171	.105	-.020	.476
.501	.034		.011	.052	.135	.160	.173	.049	-.018	.501
.530	-.033	-.076	-.054	-.027	.025			.006	-.073	.530
.548										.548
.566										.566
.584	-.035	-.079	-.055	-.029	.021	.038	.046	.004	-.076	.584
.602										.602
.620										.620
.638	-.025	-.080		-.029	.020					.638
.655										.655
.673										.673
.691	-.021	-.084	-.061		.017	.035				.691
.709										.709
.727										.727
.745	-.020	-.073		-.033	.020					.745
.763										.763
.781										.781
.799										.799
.817	-.020	-.072	-.063	-.035	.019	.037	.045	-.003	-.080	.817
.834										.834
.870										.870
.888										.888
.921	-.115									.921
.924	-.021	-.050			-.039					.924
.947										.947
.964										.964
.981	-.004	-.044	-.070							.981
.984										.984
.998										.998
1.000	-.117									1.000

Nose rake			Base rake		
Orifice station, $y_{in.}$	C_p top	C_p side	Orifice station, $y_{in.}$	C_p top	C_p side
.010	.279		.031	.500	.099
.030	.838		.094	1.065	.380
.050	1.049		.156	.951	.924
.070	1.039		.219	.944	1.162
.090	1.040		.281	.945	1.314
.110	1.035		.344	.951	1.360
.130	1.035		.406	.956	1.375
.150	1.026		.469	.963	1.386
.170	1.018		.750	.976	
.190	.962		1.000	1.021	
			1.250	1.068	
			1.500	1.097	

TABLE 9. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 3.96$ (a) $\alpha = -10.1^\circ$

Model station, $\frac{X}{L}$	Cp for meridian angle, θ , deg -										Model station, $\frac{X}{L}$
	0	90	105	120	150	165	180	225	270		
*039							1.722				*039
*040							1.563				*040
*041							1.281				*041
*044							1.803				*044
*049							1.271				*049
*055							1.202				*055
*060	.713	.408	.356	.299	.231	.209	.206	.255			*060
*075	*.270	.114	.088	.063	.032	.024	.020	.041			*075
*085	*.261	.108	.083	.057	.031	.023	.021	.040	.105		*085
*096	*.298										*096
*126		.083	.060	.041	.022	.017	.017	.029	.082		*126
*143							-.033				*143
*154	.072	-.006	-.019	-.027	-.034		-.033	-.033	-.008		*154
*171							-.031				*171
*189							-.029				*189
*206	.058	-.020	-.028	-.033	-.029	-.028	-.028	-.032	-.019		*206
*223							-.025				*223
*241							-.023				*241
*258	.051	-.026	-.034	-.035	-.032	-.028	-.020	-.033	-.014		*258
*275							-.017				*275
*293							-.015				*293
*310	.046	-.030	-.038	-.034	-.037	-.025	-.013	-.035	-.031		*310
*327							-.011				*327
*344							-.011				*344
*362	.044	-.034	-.041	-.036	-.036	-.013	-.010	-.038	-.035		*362
*379							-.012				*379
*396	.043	-.036	-.042	-.037	-.033	-.016	-.011	-.037	-.037		*396
*408							-.011				*408
*426	.131	-.015	-.003	-.025	-.018	.025	.026	-.023	-.016		*426
*451	.146	-.004	-.023	-.018	-.005	.009	.032	-.013	-.020		*451
*476	.154	-.002	-.021	-.016	-.014	.008	.024	-.004	-.003		*476
*501	.160	-.021	-.016	-.009	.005	.018			.001		*501
*530	.056	-.035	-.040	-.037	-.030		-.021	-.026	-.035		*530
*548							-.024				*548
*566							-.024				*566
*584	.048	-.038	-.039	-.036	-.028	-.025	-.025	-.030	-.037		*584
*602							-.022				*602
*620							-.021				*620
*638	.046	-.039		-.032	-.025		-.023		-.039		*638
*655							-.023				*655
*673							-.022				*673
*691	.045	-.041	-.039		-.028	-.025		-.036	-.041		*691
*709							-.021				*709
*727							-.021				*727
*745	.044	-.020		-.034	-.031		-.023		-.042		*745
*763							-.022				*763
*781							-.022				*781
*799							-.022				*799
*817	.044	-.041	-.033	-.032	-.032	-.027	-.022	-.032	-.041		*817
*834							-.021				*834
*870							-.021				*870
*888							-.020				*888
*921	-.061						-.020				*921
*924	*.042	-.037		-.028			-.022				*924
*947							-.022				*947
*964							-.022				*964
*981	*.060	-.033	-.027		-.032	-.033	-.021	-.032	-.038		*981
*984							-.022		-.039		*984
*998											*998
1.000	-.062										1.000

Orifice station, y , in.	Nose rake		Base rake		Orifice station, y , in.	
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side	
*010	1.129		*031	*611	*025	
*030	1.500		*094	1.4576	*309	
*050	1.612		*156	1.988	*624	
*070	1.648		*219	2.131	*821	
*090	1.717		*281	2.111	*935	
*110	1.775		*344	2.137	1.009	
*130	1.815		*406	2.162	1.058	
*150	1.845		*469	2.166	1.089	
*170	1.883		*750	2.157		
*190	1.840		1.000	2.175		
			1.250	2.182		
			1.500	2.171		

TABLE 9. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.756			.039
.040							1.624			.040
.041							1.378			.041
.044							.906			.044
.049							.343			.049
.055							.247			.055
.060	.585	.406	.376	.335	.288	.273	.270	.307		.060
.075	.200	.113	.094	.077	.056	.049	.047	.063		.075
.085	.196	.107	.090	.074	.053	.048	.047	.061	.106	.085
.096	.232									.096
.126		.086	.071	.057	.043	.039	.037	.050	.086	.126
.143							.028			.143
.154	.038	-.005	-.013	-.020	-.027		-.027	-.024	-.005	.154
.171							.026			.171
.189							.025			.189
.206	.024	-.014	-.020	-.024	-.024	-.023	-.023	-.024	-.013	.206
.223							.022			.223
.241							.021			.241
.258	.021	-.019	-.023	-.026	-.023	-.020	-.020	-.023	-.007	.258
.275							.018			.275
.293							.017			.293
.310	.016	-.023	-.027	-.024	-.023	-.020	-.016	-.022	-.023	.310
.327							.015			.327
.344							.014			.344
.362	.014	-.025	-.028	-.025	-.024	-.019	-.013	-.024	-.025	.362
.379							.011			.379
.396	.014	-.027	-.029	-.025	-.025	-.016	-.005	-.025	-.027	.396
.408							.022			.408
.426	.076	.003	.022	-.005	-.002	.027	-.001	-.005	.002	.426
.451	.092	.013	.006	.001	.022	.029	.033	.012	.011	.451
.476	.095	.016	.007	.003	.018	.025	.026	.016	.016	.476
.501	.098	.008	.004	.018	.022	.022	.022	.018	.501	
.530	.021	-.025	-.030	-.028	-.015		.012	-.019	-.023	.530
.548							.014			.548
.566							.014			.566
.584	.018	-.030	-.034	-.029	-.019	-.016	-.012	-.019	-.027	.584
.602							.011			.602
.620							.010			.620
.638	.015	-.031		-.024	-.017		-.009		-.029	.638
.655							.009			.655
.673							.009			.673
.691	.014	-.033	-.033		-.016	-.013	-.009		-.031	.691
.709							.009			.709
.727							.009			.727
.745	.013	-.014		-.022	-.016		-.008		-.032	.745
.763							.008			.763
.781							.009			.781
.799							.009			.799
.817	.013	-.036	-.027	-.019	-.016	-.012	-.008	-.015	-.033	.817
.834							.008			.834
.870							.008			.870
.888							.008			.888
.921	-.062						-.008			.921
.924	.014	-.033		-.017			-.009			.924
.947							-.008			.947
.964							-.008			.964
.981	.024	-.029	-.020		-.017	-.013	-.009	-.014	-.029	.981
.984							-.008		-.032	.984
.998							-.011			.998
1.000	-.063									1.000

Orifice station, y/l	Nose rake		Base rake	
	Cp top	Cp side	Cp top	Cp side
.010	.985		.031	.401
.030	1.385		.094	1.013
.050	1.469		.156	1.293
.070	1.499		.219	1.575
.090	1.536		.281	1.677
.110	1.568		.344	1.707
.130	1.586		.406	1.753
.150	1.608		.469	1.771
.170	1.617		.750	1.818
.190	1.568		1.000	1.845
			1.250	1.845
			1.500	1.839

TABLE 9. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.782			.039
.040							1.675			.040
.041							1.452			.041
.044							1.020			.044
.049							.415			.049
.055							.321			.055
.060	.487	.407	.397	.373	.347	.337	.335	.358		.060
.075	.149	.111	.104	.094	.084	.079	.077	.086		.075
.085	.145	.107	.099	.091	.080	.076	.076	.085	.107	.085
.096	.177									.096
.126		.087	.079	.072	.064	.062	.061	.069	.089	.126
.143							-.019			.143
.154	.014	-.005	-.008	-.011	-.016		-.017	-.014	-.004	.154
.171							-.017			.171
.189							-.017			.189
.206	.006	-.009	-.012	-.014	-.017	-.017	-.017	-.016	-.008	.206
.223							-.017			.223
.241							-.017			.241
.258	.001	-.013	-.015	-.016	-.016	-.016	-.017	-.015	-.001	.258
.275							-.016			.275
.293							-.015			.293
.310	.000	-.014	-.016	-.016	-.016	-.015	-.014	-.015	-.013	.310
.327							-.014			.327
.344							-.013			.344
.362	-.000	-.015	-.015	-.014	-.013	-.013	-.012	-.012	-.012	.362
.379							-.011			.379
.396	-.000	-.010	-.009	-.008	-.009	-.008	-.007	-.006	-.008	.396
.408							-.005			.408
.426	.047	.012	.023	.000	.000	.009	.020	.003	.014	.426
.451	.057	.029	.023	.021	.030	.023	.024	.024	.032	.451
.476	.061	.033	.027	.025	.028	.027	.029	.030	.034	.476
.501	.065	.031	.028	.028	.028	.027	.029		.036	.501
.530	.006	-.011	-.014	-.014	-.011		-.008	-.009	-.009	.530
.548							-.009			.548
.566							-.008			.566
.584	-.000	-.016	-.015	-.015	-.012	-.011	-.008	-.009	-.012	.584
.602							-.006			.602
.620							-.005			.620
.638	-.000	-.016		-.014	-.010		-.005			.638
.655							-.005			.655
.673							-.005			.673
.691	-.001	-.016	-.016		-.011	-.008		-.009	-.012	.691
.709							-.005			.709
.727							-.006			.727
.745	-.001	*			-.012	-.010	-.005		-.011	.745
.763							-.006			.763
.781							-.006			.781
.799							-.006			.799
.817	-.001	-.015	-.014	-.010	-.008	-.007	-.006	-.007	-.010	.817
.834							-.005			.834
.870							-.004			.870
.888							-.004			.888
.921	-.061						-.003			.921
.924	-.000	-.014		-.010			-.005			.924
.947							-.003			.947
.964							-.003			.964
.981	.006	-.011	-.010		-.006	-.005	-.004	-.005	-.008	.981
.984					-.007				-.010	.984
.998							-.008			.998
1.000		-.063								1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.821		.031	.242	.142
.030	1.270		.094	.551	.314
.050	1.338		.156	.728	.419
.070	1.367		.219	.973	.531
.090	1.385		.281	1.188	.654
.110	1.414		.344	1.346	.772
.130	1.432		.406	1.411	.913
.150	1.429		.469	1.443	1.034
.170	1.421		.750	1.478	
.190	1.365		1.000	1.557	
			1.250	1.619	
			1.500	1.642	

TABLE 9. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.782			.039
.040							1.715			.040
.041							1.512			.041
.044							1.103			.044
.049							.495			.049
.055							.385			.055
.060	.411	.409	.415	.410	.410	.406	.408	.409		.060
.075	.111	.112	.112	.111	.114	.111	.112	.111		.075
.085	.108	.106	.109	.109	.110	.108	.109	.110	.108	.085
.096	.136									.096
.126		.088	.089	.088	.090	.090	.089	.090	.090	.126
.143							.004			.143
.154	-.002	-.003	-.004	-.003	-.004		-.003	-.005	-.003	.154
.171							.005			.171
.189							.007			.189
.206	-.007	-.008	-.008	-.008	-.008	-.008	-.008	-.008	-.007	.206
.223							.009			.223
.241							.009			.241
.258	-.010	-.009	-.010	-.010	-.009	-.009	-.009	-.010	.000	.258
.275							.009			.275
.293							.009			.293
.310	-.010	-.011	-.012	-.010	-.011	-.010	-.010	-.010	-.010	.310
.327							.010			.327
.344							.009			.344
.362	-.010	-.008	-.008	-.008	-.009	-.009	-.009	-.008	-.007	.362
.379							.009			.379
.396	-.008	.001	.002	.001	-.001	-.005	-.007	-.000	.002	.396
.408							.006			.408
.426	.036	.004	.024	.004	.013	.029	.039	.008	.004	.426
.451	.042	.027	.029	.030	.042	.037	.038	.034	.027	.451
.476	.040	.034	.034	.035	.039	.038	.037	.040	.035	.476
.501	.042	.036	.036	.039	.041	.040	.041	.040	.040	.501
.530	-.002	-.007	-.007	-.005	-.005		.005	-.004	-.004	.530
.548							.006			.548
.566							.005			.566
.584	-.007	-.007	-.007	-.007	-.007	-.008	-.006	-.004	-.004	.584
.602							.005			.602
.620							.006			.620
.638	-.008	-.009			-.006	-.009	-.007		-.005	.638
.655							.007			.655
.673							.007			.673
.691	-.007	-.010	-.010			-.008	-.009			.691
.709							.006			.709
.727							.006			.727
.745	-.006	.006			-.008	-.007			-.005	.745
.763							.006			.763
.781							.005			.781
.799							.005			.799
.817	-.005	-.007	-.007	-.007	-.006	-.005	-.005	-.005	-.003	.817
.834							.005			.834
.870							.004			.870
.888							.004			.888
.921	-.058									.921
.924	-.004	-.006			-.006					.924
.947										.947
.964										.964
.981	.000	-.004	-.005			-.004	-.003			.981
.984										.984
.998										.998
1.000		-.059								1.000

Orifice station, $y, \text{in.}$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y, \text{in.}$	Cp top	Cp side
.010	.647		.031	.129	.128
.030	1.177		.094	.272	.247
.050	1.239		.156	.339	.327
.070	1.264		.219	.419	.403
.090	1.286		.281	.484	.479
.110	1.291		.344	.566	.569
.130	1.286		.406	.649	.662
.150	1.277		.469	.716	.757
.170	1.262		.750	1.027	
.190	1.210		1.000	1.215	
			1.250	1.360	
			1.500	1.461	

TABLE 9. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
.039							1.787			.039
.040							1.754			.040
.041							1.579			.041
.044							1.199			.044
.049							.575			.049
.055							.479			.055
.060	.339	.407	.433	.446	.477	.480	.486	.462		.060
.075	.077	.110	.121	.128	.148	.149	.150	.136	.107	.075
.085	.076	.106	.116	.126	.141	.144	.146	.136		.085
.096	.100									.096
.126		.087	.096	.104	.117	.121	.122	.111	.089	.126
.143							.010			.143
.154	-.014	-.004	-.001	.001	.009		.011	.005	-.002	.154
.171							.009			.171
.189							.006			.189
.206	-.016	-.009	-.007	-.003	.002	.002	.003	.001	-.008	.206
.223							.002			.223
.241							.000			.241
.258	-.016	-.012	-.015	-.007	-.002	-.001	-.000	-.003	-.000	.258
.275							-.000			.275
.293							-.000			.293
.310	-.014	-.014	-.011	-.008	-.004	-.001	-.000	-.004	-.012	.310
.327							-.000			.327
.344							-.001			.344
.362	-.013	-.014	-.012	-.009	-.004	-.002	-.001	-.004	-.012	.362
.379							-.001			.379
.396	-.008	-.008	-.008	-.008	-.003	-.002	-.001	-.004	-.006	.396
.408							-.001			.408
.426	.025	.009	.033	.026	.043	.047	.046	.041	.011	.426
.451	.024	.028	.033	.041	.058	.055	.059	.050	.030	.451
.476	.028	.032	.038	.043	.056	.060	.064	.053	.034	.476
.501	.029	.040	.040	.045	.058	.062	.067		.036	.501
.530	-.007	-.010	-.008	-.005	.001		.006	.001	-.008	.530
.548							.002			.548
.566							.001			.566
.584							.000			.584
.602							.000			.602
.620							.000			.620
.638	-.008	-.015			-.008	-.005				.638
.655							.001			.655
.673							.001			.673
.691							.000			.691
.709							-.000			.709
.727							.000			.727
.745	-.007	.001			-.010	-.005				.745
.763							.000			.763
.781							.000			.781
.799							-.001			.799
.817	-.007	-.013	-.013	-.010	-.004	-.002	-.000	-.004	-.010	.817
.834							.000			.834
.870							.000			.870
.888							.000			.888
.921	-.058						.001			.921
.924	-.005	-.012			-.010					.924
.947										.947
.964										.964
.981	.000	-.010	-.011				.004			.981
.984							-.004			.984
.998										.998
1.000	-.060									1.000

Orifice station, y , in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side	
.010	.466		.031	.088	.153	
.030	1.052		.094	.158	.325	
.050	1.168		.156	.204	.439	
.070	1.161		.219	.238	.575	
.090	1.157		.281	.283	.701	
.110	1.166		.344	.310	.842	
.130	1.159		.406	.339	.987	
.150	1.150		.469	.370	1.101	
.170	1.134		.750	.520		
.190	1.089		1.000	.652		
			1.250	.788		
			1.500	.879		

TABLE 9. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.774			.039
.040							1.789			.040
.041							1.662			.041
.044							1.284			.044
.049							645			.049
.055							573			.055
.060	.272	.407	.454	.493	.563	.578	.588	.531		.060
.075	.048	.113	.135	.155	.194	.202	.203	.173		.075
.085	.046	.109	.129	.151	.187	.195	.200	.171	.109	.085
.096	.069									.096
.126		.087	.104	.121	.157	.165	.169	.140	.089	.126
.143							.039			.143
.154	-.024	-.005	.003	.013	.030		.039	.022	-.006	.154
.171							.031			.171
.189							.026			.189
.206	-.021	-.012	-.006	.002	.016	.021	.024	.009	-.012	.206
.223							.023			.223
.241							.022			.241
.258	-.018	-.018	-.012	-.003	.013	.019	.022	.005	-.007	.258
.275							.021			.275
.293							.019			.293
.310	-.016	-.021	-.015	-.005	.010	.016	.018	.002	-.021	.310
.327							.018			.327
.344							.016			.344
.362	-.013	-.024	-.017	-.008	.008	.014	.016	.000	-.025	.362
.379							.015			.379
.396	-.010	-.025	-.019	-.010	.007	.013	.016	.000	-.026	.396
.408							.015			.408
.426	.022	.004	.034	.034	.065	.075	.078	.052	.003	.426
.451	.031	.013	.027	.044	.085	.090	.094	.065	.014	.451
.476	.024	.017	.031	.047	.081	.091	.096	.069	.018	.476
.501	.023		.031	.046	.082	.093	.099		.020	.501
.530	-.010	-.023	-.015	-.007	.011		.020	.004	-.021	.530
.548							.018			.548
.566							.019			.566
.584	-.012	-.027	-.019	-.010	.008	.014	.017	.001	-.025	.584
.602							.017			.602
.620							.016			.620
.638	-.010	-.028		-.010	.006		.016		-.027	.638
.655							.016			.655
.673							.016			.673
.691	-.009	-.030	-.023		.005	.012		-.001	-.028	.691
.709							.016			.709
.727							.016			.727
.745	-.009	-.011		-.013	.005				-.028	.745
.763							.016			.763
.781							.014			.781
.799							.014			.799
.817	-.009	-.030	-.023	-.015	.005	.013	.016	-.001	-.029	.817
.834							.016			.834
.870							.015			.870
.888							.016			.888
.921	-.060						.014			.921
.924	-.008	-.030		-.017			.013			.924
.947							.013			.947
.964							.013			.964
.981	-.003	-.026	-.028		.003	.012	.014	-.003	-.028	.981
.984					.003				-.029	.984
.998							.015			.998
1.000	-.061									1.000

Orifice station, y , in.	Nose rake			Base rake		
	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side	Orifice station, y , in.
.010	.276		.031	.180	.099	
.030	.906		.094	.381	.403	
.050	1.067		.156	.437	.665	
.070	1.045		.219	.486	.790	
.090	1.056		.281	.500	.929	
.110	1.047		.344	.526	1.018	
.130	1.036		.406	.540	1.092	
.150	1.027		.469	.549	1.136	
.170	1.007		.750	.595		
.190	.949		1.000	.640		
			1.250	.681		
			1.500	.710		

TABLE 9. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 3.96$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{1}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{1}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.730			.039
.040							1.791			.040
.041							1.710			.041
.044							1.395			.044
.049							.790			.049
.055							.703			.055
.060	.203	.405	.477	.544	.669	.696	.715	.611		.060
.075	.019	.112	.146	.181	.251	.267	.275	.216		.075
.085	.020	.107	.139	.178	.240	.257	.264	.214	.109	.085
.096	.037									.096
.126		.082	.112	.148	.215	.234	.241	.184	.084	.126
.143							.076			.143
.154	-.033	-.006	.008	.026	.058		.074	.044	-.006	.154
.171							.064			.171
.189							.061			.189
.206	-.026	-.018	-.006	.011	.045	.056	.060	.030	-.018	.206
.223							.059			.223
.241							.056			.241
.258	-.020	-.026	-.012	.004	.038	.050	.054	.024	-.013	.258
.275							.052			.275
.293							.050			.293
.310	-.014	-.030	-.017	-.000	.033	.043	.049	.019	-.032	.310
.327							.048			.327
.344							.048			.344
.362	-.011	-.035	-.022	-.003	.029	.042	.047	.017	-.036	.362
.379							.046			.379
.396	-.011	-.037	-.024	-.007	.029	.041	.047	.014	-.037	.396
.408							.046			.408
.426	.018	-.013	.029	.041	.105	.128	.134	.079	-.013	.426
.451	.027	-.004	.021	.053	.123	.141	.150	.090	-.004	.451
.476	.023	-.003	.022	.057	.128	.151	.160	.099	-.001	.476
.501	.019		.025	.060	.132	.155	.165		-.000	.501
.530	.016	-.034	-.021	-.001	.040		.058	.022	-.034	.530
.548							.054			.548
.566							.053			.566
.584	-.025	-.038	-.024	-.008	.030	.043	.050	.017	-.039	.584
.602							.050			.602
.620							.047			.620
.638	-.021	-.039		-.007	.028		.049		-.039	.638
.655							.050			.655
.673							.049			.673
.691	-.021	-.042	-.028		.026	.040		.014	-.041	.691
.709							.049			.709
.727							.048			.727
.745	-.021	-.021		-.013	.026		.048		-.042	.745
.763							.047			.763
.781							.046			.781
.799							.046			.799
.817	-.020	-.042	-.030	-.013	.026	.040	.049	.012	-.042	.817
.834							.047			.834
.870							.047			.870
.888							.047			.888
.921	-.063						.046			.921
.924	-.020	-.039		-.015			.044			.924
.947							.044			.947
.964							.044			.964
.981	-.014	-.034	-.033		.024	.038	.043	.010	-.040	.981
.984					.024				-.041	.984
.998							.043			.998
1.000	-.063									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.124		.031	.308	.034
.030	.659		.094	.703	.352
.050	.972		.156	.641	.665
.070	.920		.219	.632	.820
.090	.932		.281	.629	.927
.110	.923		.344	.616	1.019
.130	.920		.406	.601	1.072
.150	.903		.469	.591	1.108
.170	.887		.750	.551	
.190	.831		1.000	.574	
			1.250	.654	
			1.500	.744	

TABLE 10. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 4.65$ (a) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.814			.039
.040							1.683			.040
.041							1.406			.041
.044							.932			.044
.049							.358			.049
.055							.259			.055
.060	.580	.419	.387	.347	.298	.280	.276	.303		.060
.075	.206	.128	.111	.089	.069	.061	.058	.071		.075
.085	.199	.120	.103	.086	.065	.059	.058	.069	.106	.085
.096	.230									.096
.126		.089	.077	.065	.049	.046	.043	.052	.084	.126
.143							-.011			.143
.154		.005	.000	-.006	-.011		-.011	-.010	.002	.154
.171							-.011			.171
.189							-.011			.189
.206	.031	-.001	-.008	-.010	-.012	-.011	-.011	-.011	-.003	.206
.223							-.010			.223
.241							-.010			.241
.258	.027	-.008	-.012	-.014	-.012	-.010	-.009	-.012	.011	.258
.275							-.007			.275
.293							-.007			.293
.310	.024	-.011	-.016	-.013	-.013	-.012	-.008	-.012	-.013	.310
.327							-.007			.327
.344							-.007			.344
.362	.020	-.015	-.016	-.012	-.012	-.012	-.006	-.011	-.015	.362
.379							-.006			.379
.396	.019	-.014	-.013	-.012	-.010	-.010	-.006	-.010	-.015	.396
.408							-.004			.408
.426	.063	.006	.034	-.009	-.007	-.000	.014	-.006	.002	.426
.451	.080	.016	.006	.001	.017	-.002	.019	.000	.009	.451
.476	.085	.018	.008	.001	.002	.006	.019	.006	.011	.476
.501	.095		.008	.001	.006	.010	.016		.013	.501
.530	.032	-.012	-.016	-.017	-.010		-.008	-.010	-.015	.530
.548							-.009			.548
.566							-.009			.566
.584	.024	-.017	-.019	-.017	-.012	-.010	-.008	-.013	-.019	.584
.602							-.007			.602
.620							-.006			.620
.638	.021	-.017			-.012	-.013			-.021	.638
.655							-.006			.655
.673							-.005			.673
.691	.020	-.020	-.019			-.014	-.010		-.015	.691
.709							-.006			.709
.727							-.006			.727
.745	.020	.012			-.017	-.014			-.021	.745
.763							-.005			.763
.781							-.007			.781
.799							-.006			.799
.817	.020	-.021	-.019		-.015	-.012	-.010		-.022	.817
.834							-.006			.834
.870							-.008			.870
.888							-.007			.888
.921	-.037									.921
.924	.020	-.021								.924
.947										.947
.964										.964
.981	.026	-.019	-.017				-.014		-.012	.981
.984							-.014		-.019	.984
.998									-.011	.998
1.000	-.037									1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	.883		.031	.353	.047
.030	1.383		.094	.869	.314
.050	1.450		.156	1.155	.507
.070	1.485		.219	1.469	.620
.090	1.528		.281	1.635	.759
.110	1.552		.344	1.708	.853
.130	1.579		.406	1.757	.932
.150	1.600		.469	1.791	.975
.170	1.602		.750	1.877	
.190	1.563		1.000	1.877	
			1.250	1.902	
			1.500	1.886	

TABLE 10. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 4.65$ - Continued

(b) $\alpha = -3.0^\circ$

Model station, $\frac{x}{L}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{L}$
	0	90	105	120	150	165	180	225	270	
.039							1.826			.039
.040							1.727			.040
.041							1.496			.041
.044							1.042			.044
.049							.427			.049
.055							.335			.055
.060	.485	.421	.408	.384	.356	.344	.342	.353		.060
.075	.155	.127	.119	.107	.096	.091	.088	.092		.075
.085	.150	.119	.111	.103	.091	.085	.085	.090	.107	.085
.096	.178									.096
.126		.093	.086	.080	.071	.067	.066	.068	.085	.126
.143							.005			.143
.154	.025	.007	.004	.001	-.003		-.005	-.003	.003	.154
.171							.006			.171
.189							.007			.189
.206	.015	.001	-.002	-.003	-.006	-.007	-.007	-.006	-.001	.206
.223							.007			.223
.241							.008			.241
.258	.011	-.003	-.005	-.008	-.008	-.007	-.008	-.008	.016	.258
.275							.006			.275
.293							.005			.293
.310	.007	-.007	-.009	-.008	-.008	-.008	-.006	-.007	-.007	.310
.327							.005			.327
.344							.002			.344
.362	.006	-.007	-.005	-.004	-.003	-.003	-.002	-.002	-.004	.362
.379							.001			.379
.396	.007	-.002	-.002	-.002	-.002	-.002	-.001	-.000	-.001	.396
.408							.000			.408
.426	.037	.009	.040	-.000	-.000	.002	.003	.000	.002	.426
.451	.050	.024	.018	.011	.021	.002	.008	.004	.017	.451
.476	.053	.027	.022	.016	.009	.010	.013	.012	.022	.476
.501	.060		.022	.016	.014	.013	.015	.024	.501	
.530	.017	-.004	-.008	-.009	-.009		.009	-.009	-.008	.530
.548							.008			.548
.566							.006			.566
.584	.007	-.008	-.010	-.012	-.009	-.008	-.007	-.008	-.010	.584
.602							.007			.602
.620							.006			.620
.638	.005	-.010			-.008	-.009			.012	.638
.655							.004			.655
.673							.004			.673
.691	.005	-.011	-.012			-.009		-.007	-.011	.691
.709							.004			.709
.727							.003			.727
.745	.006	.019			-.011	-.008			.011	.745
.763							.004			.763
.781							.005			.781
.799							.005			.799
.817	.006	-.011	-.012	-.009	-.006	-.006	-.004	-.006	-.011	.817
.834							.004			.834
.870							.004			.870
.888							.003			.888
.921	-.036									.921
.924	.006	-.012								.924
.947										.947
.964										.964
.981	.009	-.009	-.009							.981
.984										.984
.998										.998
1.000	-.036									1.000

Nose rake			Base rake		
Orifice station, y , in.	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side
.010	.687		.031	.199	.084
.030	1.277		.094	.436	.245
.050	1.328		.156	.597	.365
.070	1.351		.219	.775	.459
.090	1.372		.281	.957	.561
.110	1.395		.344	1.148	.694
.130	1.406		.406	1.266	.816
.150	1.395		.469	1.321	.938
.170	1.383		.750	1.358	
.190	1.344		1.000	1.471	
			1.250	1.556	
			1.500	1.609	

TABLE 10. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 4.65$ - Continued(c) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.828			.039
.040							1.761			.040
.041							1.576			.041
.044							1.136			.044
.049							.503			.049
.055							.399			.055
.060	.409	.422	.428	.421	.419	.412	.409	.403		.060
.075	.117	.127	.129	.125	.127	.122	.121	.115		.075
.085	.113	.120	.120	.120	.119	.117	.117	.112	.108	.085
.096	.138									.096
.126		.096	.096	.095	.095	.093	.091	.088	.086	.126
.143							.008			.143
.154	.009	.008	.007	.007	.006		.006	.004	.004	.154
.171							.004			.171
.189							.001			.189
.206	.001	.0	.000	.001	.001	.000	.001	-.000	-.000	.206
.223							-.000			.223
.241							-.001			.241
.258	-.001	-.002	-.002	-.002	-.002	-.002	-.002	-.003	.016	.258
.275							-.002			.275
.293							-.002			.293
.310	-.004	-.006	-.005	-.004	-.005	-.004	-.003	-.005	-.005	.310
.327							-.004			.327
.344							-.004			.344
.362	-.004	-.006	-.005	-.003	-.002	-.002	-.000	-.000	-.000	.362
.379							.002			.379
.396	-.001	.002	.003	.003	.003	.003	.004	.004	.003	.396
.408							.005			.408
.426	.028	.007	.044	.007	.006	.006	.007	.005	.004	.426
.451	.030	.022	.021	.021	.035	.019	.019	.015	.013	.451
.476	.030	.028	.028	.028	.027	.026	.026	.024	.023	.476
.501	.035		.029	.028	.027	.028	.030		.025	.501
.530	.004	-.003	-.002	-.004	-.002		-.003	-.004	-.004	.530
.548							-.002			.548
.566							-.002			.566
.584	-.003	-.005	-.004	-.006	-.004	-.003	-.003	-.005	-.004	.584
.602							-.003			.602
.620							-.004			.620
.638	-.003	-.007			-.002	-.005			-.004	.638
.655							-.004			.655
.673							-.004			.673
.691	-.004	-.007	-.006		-.006	-.005		-.007	-.004	.691
.709							-.004			.709
.727							-.005			.727
.745	-.002	.022			-.006	-.005			-.004	.745
.763							-.004			.763
.781							-.004			.781
.799							-.004			.799
.817	-.002	-.006	-.005	-.006	-.003	-.003	-.003	-.004	-.004	.817
.834							-.003			.834
.870							-.003			.870
.888							-.002			.888
.921	-.037									.921
.924	-.002	-.005			-.005					.924
.947										.947
.964										.964
.981	.000	-.003	-.004							.981
.984										.984
.998										.998
1.000	-.037									1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	.494		.031	.090	.136
.030	1.164		.094	.196	.291
.050	1.254		.156	.272	.381
.070	1.261		.219	.323	.454
.090	1.268		.281	.360	.544
.110	1.263		.344	.411	.609
.130	1.261		.406	.471	.673
.150	1.247		.469	.528	.752
.170	1.238		.750	.811	
.190	1.192		1.000	1.007	
			1.250	1.187	
			1.500	1.314	

TABLE 10. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 4.65$ - CONTINUED(d) $\alpha = 3.0^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270		
.039							1.850				.039
.040							1.814				.040
.041							1.632				.041
.044							1.235				.044
.049							.593				.049
.055							.480				.055
.060	.341	.421	.445	.457	.483	.484	.485	.453			.060
.075	.088	.125	.137	.143	.158	.160	.159	.142			.075
.085	.084	.120	.128	.138	.150	.152	.154	.137	.108		.085
.096	.103										.096
.126		.094	.101	.108	.118	.120	.120	.106	.086		.126
.143							.020				.143
.154	-.002	.007	.010	.013	.019		.020	.012	.004		.154
.171							.017				.171
.189							.014				.189
.206	-.005	.0	.002	.006	.010	.010	.011	.007	-.000		.206
.223							.009				.223
.241							.007				.241
.258	-.007	-.004	-.002	.000	.003	.005	.004	.001	.015		.258
.275							.004				.275
.293							.004				.293
.310	-.007	-.007	-.005	-.002	.001	.002	.003	-.000	-.007		.310
.327							.002				.327
.344							.002				.344
.362	-.005	-.007	-.006	-.004	.000	.001	.001	-.002	-.006		.362
.379							.001				.379
.396	.000	-.002	.032	-.002	.000	.001	.002	-.001	-.001		.396
.408							.009				.408
.426	.013	.008	.046	.016	.023	.025	.027	.020	.004		.426
.451	.016	.023	.028	.032	.054	.044	.047	.034	.019		.451
.476	.018	.027	.031	.037	.047	.050	.052	.040	.022		.476
.501	.025		.033	.039	.050	.054	.056		.023		.501
.530	.001	-.006	-.002	.000	.005		.008	-.000	-.009		.530
.548							.006				.548
.566							.004				.566
.584	-.005	-.010	-.006	-.004	-.001	-.000	.003	-.003	-.009		.584
.602							.002				.602
.620							.002				.620
.638	-.006	-.011			-.001	-.001			-.011		.638
.655							.002				.655
.673							.002				.673
.691	-.005	-.012	-.009		-.001	-.000		-.004	-.011		.691
.709							.002				.709
.727							.002				.727
.745	-.005	.018			-.006	-.002			-.011		.745
.763							.001				.763
.781							.002				.781
.799							.001				.799
.817	-.004	-.011	-.009	-.007	-.000	-.000	.002	-.003	-.009		.817
.834							.001				.834
.870							.002				.870
.888							.002				.888
.921	-.038						.001				.921
.924	-.003	-.010			-.007			-.000			.924
.947								-.000			.947
.964											.964
.981	-.000	-.006	-.008						-.004		.981
.984									-.008		.984
.998									-.009		.998
1.000	-.038								-.000		1.000

Orifice station, y, in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.332		.031	.056	.074
.030	1.005		.094	.120	.388
.050	1.162		.156	.146	.701
.070	1.143		.219	.194	.784
.090	1.134		.281	.210	.867
.110	1.143		.344	.240	.915
.130	1.129		.406	.268	.989
.150	1.118		.469	.288	1.046
.170	1.104		.750	.376	
.190	1.053		1.000	.457	
			1.250	.572	
			1.500	.669	

TABLE 10.- PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH NATURAL TRANSITION AT $M = 4.65$ - CONCLUDED

(e) $\alpha = 6.30$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.822			.039
.040							1.836			.040
.041							1.693			.041
.044							1.319			.044
.049							.676			.049
.055							.568			.055
.060	.278	.420	.465	.501	.568	.578	.583	.517		.060
.075	.060	.128	.149	.168	.205	.210	.211	.175	.110	.075
.085	.058	.121	.139	.162	.195	.201	.204	.172		.085
.096	.075									.096
.126		.093	.107	.125	.155	.162	.164	.132	.086	.126
.143							.048			.143
.154	-.008	.006	.015	.025	.041		.047	.029	.002	.154
.171							.039			.171
.189							.033			.189
.206	-.009	-.0	.004	.012	.024	.027	.030	.014	-.003	.206
.223							.027			.223
.241							.026			.241
.258	-.009	-.006	-.001	.004	.019	.024	.025	.009	.013	.258
.275							.024			.275
.293							.022			.293
.310	-.008	-.010	-.005	.002	.015	.019	.021	.005	-.013	.310
.327							.020			.327
.344							.019			.344
.362	-.006	-.012	-.008	-.001	.012	.016	.018	.002	-.015	.362
.379							.017			.379
.396	-.003	-.012	.001	-.002	.011	.015	.016	.002	-.012	.396
.408							.022			.408
.426	.016	.002	.046	.025	.049	.057	.060	.034	-.003	.426
.451	.017	.014	.025	.042	.082	.080	.083	.050	.007	.451
.476	.016	.018	.031	.048	.076	.085	.088	.058	.012	.476
.501	.021		.033	.047	.077	.089	.091		.014	.501
.530	-.001	-.011	-.006	.001	.018		.024	.005	-.014	.530
.548							.021			.548
.566							.022			.566
.584	-.008	-.016	-.009	-.001	.013	.018	.020	.001	-.017	.584
.602							.020			.602
.620							.018			.620
.638	-.006	-.016		.000	.012		.019		-.017	.638
.655							.018			.655
.673							.018			.673
.691	-.006	-.018	-.011		.010	.015		-.000	-.019	.691
.709							.018			.709
.727							.017			.727
.745	-.006	.013		-.005	.010		.017		-.019	.745
.763							.018			.763
.781							.017			.781
.799							.017			.799
.817	-.006	-.017	-.013	-.006	.010	.015		-.002	-.018	.817
.834							.018			.834
.870							.016			.870
.888							.017			.888
.921	-.037									.921
.924	-.006	-.017		-.007						.924
.947										.947
.964										.964
.981	-.003	-.015	-.014		.008	.015		-.002	-.017	.981
.984									-.017	.984
.998										.998
1.000	-.037									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.134		.031	.093	.001
.030	.798		.094	.229	.395
.050	1.031		.156	.275	.911
.070	1.017		.219	.330	.941
.090	1.010		.281	.342	.953
.110	1.001		.344	.365	.973
.130	.990		.406	.395	1.008
.150	.973		.469	.402	1.063
.170	.964		.750	.427	
.190	.909		1.000	.466	
			1.250	.503	
			1.500	.542	

TABLE 11. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 1.57$ (a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039										.039
.040										.040
.041										.041
.044										.044
.049										.049
.055										.055
.060										.060
.075	.555	.509	.415	.213	.149	.134	.331			.075
.075	.132	-.071	-.112	-.146	-.162	-.173	-.190	-.161		.075
.085	.273	-.005	-.052	-.101	-.129	-.122	-.110	-.118	-.017	.085
.096	.501									.096
.126		.093	.043	.004	-.008	-.016	-.026	-.009	.080	.126
.143							-.249			.143
.154	-.011	-.173	-.202	-.213	-.211	-.205	-.198	-.217		.154
.171							-.158			.171
.189							-.119			.189
.206	.025	-.155	-.169	-.176	-.123	-.100	-.091	-.144		.206
.223							-.066			.223
.241							-.048			.241
.258	.021	-.149	-.160	-.136	-.065	-.042	-.036	-.084	-.157	.258
.275							-.027			.275
.293							-.015			.293
.310	.018	-.150	-.139	-.087	-.041	-.019	-.008	-.053	-.156	.310
.327							-.006			.327
.344							-.003			.344
.362	.015	-.133	-.099	-.060	-.046	-.019	-.000	-.047	-.129	.362
.379							-.000			.379
.396	.009	-.112	-.087	-.042	-.039	-.016	.011	-.025	-.109	.396
.408							.008			.408
.426	.225	.079	.097	.087	.071	.158	.227	.083	.091	.426
.451	.212	.071	.065	.076	.060	.106	.146		.069	.451
.476	.204	.053	.045	.062	.042	.092	.113	.076	.053	.476
.501	.194	.044	.038	.046	.020	.070	.085	.046	.043	.501
.530	-.004	-.126	-.134	-.111	-.108	-.128	-.112	-.086	-.129	.530
.548							-.112			.548
.566							-.063			.566
.584	.025	-.115	-.100	-.085	-.086	-.066	-.053	-.080	-.120	.584
.602							-.044			.602
.620							-.037			.620
.638	.031	-.097		-.074	-.047				-.095	.638
.655							-.022			.655
.673							-.021			.673
.691	.030	-.118	-.099	-.062	-.040	-.041	-.017	-.038	-.102	.691
.709							-.020			.709
.727							-.018			.727
.745	.027	-.107		-.047	-.032		-.016		-.097	.745
.763							-.015			.763
.781							-.025			.781
.799							-.017			.799
.817	.043	-.085	-.055	-.032	-.028	-.025	-.007	-.029	-.074	.817
.834							-.013			.834
.852							-.010			.852
.870							-.002			.870
.888							-.003			.888
.906							-.000			.906
.921	-.187									.921
.924	.050	-.061		-.024	-.023				-.080	.924
.947										.947
.964										.964
.981	.055	-.041	-.001							.981
1.000	-.194									1.000

Orifice station, y.in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y.in.	Cp top	Cp side
.010	1.216		.031	.864	.438
.030	1.462		.094	1.551	1.026
.050	1.533		.156	1.635	1.491
.070	1.557		.219	1.571	1.616
.090	1.575		.281	1.577	1.534
.110	1.580		.344	1.568	1.489
.130	1.582		.406	1.568	1.502
.150	1.575		.469	1.575	1.494
.170	1.554		.750	1.568	
.190	1.468		1.000	1.567	
			1.250	1.570	
			1.500	1.577	

TABLE 11. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.542			.039
.040							1.425			.040
.041							1.185			.041
.044							.744			.044
.049							.194			.049
.055							.130			.055
.060	.738	.588	.552	.477	.346	.263	.244	.413		.060
.075	.053	-.058	-.082	-.101	-.113	-.122	-.127	-.118		.075
.085	.167	.009	-.018	-.047	-.069	-.060	-.066	-.063	-.000	.085
.096	.360									.096
.126		.113	.080	.058	.032	.024	.017	.036	.099	.126
.143							-.218			.143
.154	-.064	-.150	-.163	-.177	-.190	-.190	-.187	-.193		.154
.171							-.158			.171
.189							-.120			.189
.206	-.022	-.115	-.122	-.132	-.104	-.094	-.090	-.114		.206
.223							-.066			.223
.241							-.053			.241
.258	-.017	-.090	-.091	-.081	-.054	-.043	-.043	-.062	-.092	.258
.275							-.034			.275
.293							-.026			.293
.310	-.011	-.070	-.070	-.055	-.030	-.019	-.017	-.036	-.075	.310
.327							-.009			.327
.344							-.009			.344
.362	-.010	-.061	-.049	-.032	-.020	-.012	-.008	-.022	-.060	.362
.379							-.008			.379
.396	-.009	-.052	-.047	-.039	-.014	-.008	-.000	-.021	-.052	.396
.408							.037			.408
.426	.197	.137	.141	.139	.132	.146	.159	.132	.140	.426
.451	.184	.121	.116	.114	.116	.120	.131		.118	.451
.476	.177	.103	.093	.091	.094	.100	.108	.098	.104	.476
.501	.166	.090	.081	.074	.066	.081	.097	.069	.089	.501
.530	-.031	-.093	-.102	-.104	-.081	-.087	-.105	-.077	-.096	.530
.548							-.072			.548
.566							-.054			.566
.584	-.002	-.073	-.069	-.067	-.060	-.058	-.051	-.060	-.078	.584
.602							-.042			.602
.620							-.034			.620
.638	.007	-.060		-.051	-.037				-.059	.638
.655							-.016			.655
.673							-.019			.673
.691	.002	-.064	-.056	-.043	-.029	-.027	-.010	-.023	-.051	.691
.709							-.015			.709
.727							-.011			.727
.745	.007	-.053		-.035	-.020		-.007		-.044	.745
.763							-.004			.763
.781							-.012			.781
.799							-.003			.799
.817	.018	-.043	-.034	-.021	-.012	-.006	.005	-.012	-.030	.817
.834							-.000			.834
.852							.004			.852
.870							.011			.870
.888							.010			.888
.906							.010			.906
.921	-.145									.921
.924	.011	-.022		-.013	.000		.016		-.036	.924
.947							.076			.947
.964							.061			.964
.981	.052	-.007	.010		.034	.038	.046	.024	-.013	.981
1.000	-.154									1.000

Nose rake			Base rake		
Orifice station, y.in.	Cp top	Cp side	Orifice station, y.in.	Cp top	Cp side
.010	1.085		.031	.754	.523
.030	1.418		.094	1.294	.924
.050	1.520		.156	1.582	1.261
.070	1.556		.219	1.574	1.501
.090	1.569		.281	1.553	1.554
.110	1.572		.344	1.556	1.523
.130	1.565		.406	1.558	1.518
.150	1.552		.469	1.564	1.520
.170	1.527		.750	1.563	
.190	1.428		1.000	1.567	
			1.250	1.563	
			1.500	1.557	

TABLE 11. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.555			.039
.040							1.471			.040
.041							1.254			.041
.044							.833			.044
.049							.327			.049
.055							.207			.055
.060	.676	.613	.589	.538	.466	.401	.386	.497		.060
.075	-.000	-.044	-.054	-.067	-.068	-.077	-.081	-.083		.075
.085	.090	.026	.011	-.007	-.017	-.011	-.028	-.013	.013	.085
.096	.255									.096
.126		.123	.107	.091	.075	.071	.066	.074	.109	.126
.143							-.205			.143
.154	-.104	-.134	-.144	-.155	-.164	-.166	-.163	-.165		.154
.171							.140			.171
.189							-.114			.189
.206	-.056	-.088	-.091	-.103	-.093	-.093	-.093	-.095		.206
.223							.074			.223
.241							-.061			.241
.258	-.038	-.056	-.060	-.062	-.053	-.049	-.051	-.053	-.057	.258
.275							-.039			.275
.293							.027			.293
.310	-.025	-.041	-.040	-.032	-.024	-.019	-.019	-.025	-.043	.310
.327							-.018			.327
.344							.016			.344
.362	-.018	-.024	-.021	-.018	-.019	-.015	-.013	-.018	-.025	.362
.379							-.013			.379
.396	-.017	-.022	-.024	-.021	-.012	-.009	-.002	-.015	-.023	.396
.408							.029			.408
.426	.183	.167	.168	.165	.162	.162	.161	.164	.165	.426
.451	.169	.148	.144	.141	.135	.132	.135		.143	.451
.476	.150	.127	.118	.115	.114	.113	.113	.116	.130	.476
.501	.138	.111	.108	.101	.096	.095	.097	.099	.111	.501
.530	-.051	-.072	-.076	-.082	-.078	-.079	-.079	-.075	-.077	.530
.548							.065			.548
.566							-.049			.566
.584	-.013	-.049	-.049	-.051	-.042	-.044	-.043	-.047	-.051	.584
.602							-.043			.602
.620							-.035			.620
.638	-.015	-.028			-.031	-.027				.638
.655							-.024			.655
.673							-.022			.673
.691	-.011	-.034	-.033	-.033	-.030	-.025	-.015	-.017	-.024	.691
.709							-.010			.709
.727							.011			.727
.745	-.008	-.027			-.021	-.013			-.018	.745
.763							-.004			.763
.781							-.008			.781
.799							-.013			.799
.817	.009	-.012	-.013	-.012	-.006	-.002			.000	.817
.834							-.003	-.004		.834
.852							.002			.852
.870							.003			.870
.888							.009			.888
.906							.007			.906
.921	-.080									.921
.924	.020	-.001			-.006	.004				.924
.947								.007		.947
.964								.012		.964
.981	-.036	.016	.006				.009			.981
1.000		-.085						.007		1.000

Orifice station, $y, in.$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y, in.$	Cp top	Cp side
.010	.925		.031	.634	.521
.030	1.357		.094	.993	.779
.050	1.502		.156	1.270	.990
.070	1.541		.219	1.452	1.201
.090	1.557		.281	1.504	1.365
.110	1.549		.344	1.528	1.465
.130	1.532		.406	1.529	1.501
.150	1.518		.469	1.534	1.514
.170	1.482		.750	1.529	
.190	1.382		1.000	1.538	
			1.250	1.542	
			1.500	1.543	

TABLE 11. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 1.57$ - Continued(d) $\alpha = 0^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
.039										.039
.040										.040
.041										.041
.044										.044
.049										.049
.055										.055
.060	.597	.617	.618	.595	.593	.575	.580	.584		.060
.075	-.043	-.036	-.037	-.040	-.029	-.040	-.046	-.051		.075
.085	.031	.033	.031	.027	.027	.035	.016	.025	.021	.085
.096	.169									.096
.126		.127	.124	.121	.122	.121	.116	.111	.113	.126
.143							.177			.143
.154	-.137	-.131	-.134	-.137	-.140	-.140	.135	-.145		.154
.171							.116			.171
.189							.095			.189
.206	-.078	-.081	-.077	-.086	-.076	-.077	-.078	-.080		.206
.223							.061			.223
.241							.054			.241
.258	-.054	-.044	-.046	-.045	-.048	-.048	-.052	-.045	-.044	.258
.275							.045			.275
.293							.038			.293
.310	-.026	-.032	-.032	-.027	-.027	-.022	-.022	-.027	-.033	.310
.327							.018			.327
.344							.019			.344
.362	-.018	-.015	-.012	-.009	-.019	-.018	-.016	-.016	-.014	.362
.379							.018			.379
.396	-.017	-.011	-.015	-.016	-.013	-.013	-.007	-.014	-.012	.396
.408							.012			.408
.426	.170	.175	.177	.176	.175	.178	.175	.174	.173	.426
.451	.150	.155	.154	.156	.153	.147	.150		.151	.451
.476	.131	.134	.132	.130	.131	.130	.129	.132	.137	.476
.501	.115	.118	.120	.117	.118	.116	.116	.118	.119	.501
.530	-.066	-.065	-.066	-.071	-.068	-.068	-.069	-.067	-.069	.530
.548							.055			.548
.566							.043			.566
.584	-.028	-.040	-.038	-.040	-.039	-.039	-.035	-.038	-.042	.584
.602							.035			.602
.620							.033			.620
.638	-.026	-.019		-.021	-.021				-.018	.638
.655							.022			.655
.673							.021			.673
.691	-.017	-.025	-.024	-.024	-.030	-.026	-.016	-.014	-.016	.691
.709							.014			.709
.727							.010			.727
.745	-.012	-.018		-.013	-.012		.007		-.009	.745
.763							.009			.763
.781							.017			.781
.799							.007			.799
.817	.003	-.006	-.006	-.005	-.005	-.002	-.000	-.000	.006	.817
.834							.005			.834
.852							.002			.852
.870							.003			.870
.888							.006			.888
.906							.008			.906
.921	-.066						.010			.921
.924	.013	.008		-.004	.005		.008		-.001	.924
.947							.008			.947
.964							.012			.964
.981	.047	.029	.012			.010	.006		.010	.981
1.000	-.069						.008			1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.794		.031	.479	.501
.030	1.311		.094	.679	.692
.050	1.492		.156	.829	.842
.070	1.541		.219	.973	.998
.090	1.550		.281	1.098	1.139
.110	1.526		.344	1.226	1.265
.130	1.491		.406	1.330	1.368
.150	1.469		.469	1.408	1.442
.170	1.427		.750	1.488	
.190	1.335		1.000	1.505	
			1.250	1.519	
			1.500	1.529	

TABLE 11. ~ PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.556			.039
.040							1.531			.040
.041							1.374			.041
.044							.998			.044
.049							.445			.049
.055							.630			.055
.060	.428	.618	.635	.630	.656	.651	.651	.632		.060
.075	-.075	-.040	-.032	-.025	.004	-.003	-.009	-.028		.075
.085	-.013	.028	.039	.052	.070	.086	.067	.057	.015	.085
.096	.108									.096
.126		.123	.134	.144	.167	.172	.169	.144	.111	.126
.143							-.147			.143
.154	-.164	-.134	-.129	-.124	-.115	-.111	-.104	-.126		.154
.171							-.091			.171
.189							-.073			.189
.206	-.090	-.086	-.076	-.080	-.062	-.058	-.058	-.070		.206
.223							-.045			.223
.241							-.041			.241
.258	-.052	-.051	-.051	-.047	-.039	-.035	-.042			.258
.275							-.032			.275
.293							-.017			.293
.310	-.023	-.039	-.041	-.033	-.030	-.025	-.024	-.033	-.043	.310
.327							-.017			.327
.344							-.017			.344
.362	-.012	-.024	-.020	-.015	-.020	-.016	-.014	-.021	-.022	.362
.379							-.017			.379
.396	-.011	-.018	-.022	-.023	-.017	-.015	-.007	-.020	-.019	.396
.408							-.015			.408
.426	.162	.168	.174	.175	.179	.187	.183	.175	.167	.426
.451	.138	.149	.153	.158	.166	.162	.167		.145	.451
.476	.119	.129	.131	.135	.145	.146	.147	.141	.132	.476
.501	.099	.112	.121	.122	.135	.135	.137	.130	.114	.501
.530	-.073	-.070	-.068	-.069	-.058	-.056	-.054	-.061	-.074	.530
.548							-.041			.548
.566							-.029			.566
.584	-.037	-.045	-.041	-.040	-.030	-.028	-.021	-.033	-.050	.584
.602							-.025			.602
.620							-.023			.620
.638	-.027	-.026		-.023	-.015				-.024	.638
.655							-.013			.655
.673							-.015			.673
.691	-.014	-.032	-.028	-.026	-.030	-.023	-.011	-.016	-.023	.691
.709							-.008			.709
.727							-.010			.727
.745	-.010	-.023		-.018	-.008		-.004		-.015	.745
.763							-.008			.763
.781							-.016			.781
.799							-.005			.799
.817	.003	-.012	-.010	-.011	-.006	-.001	-.002	-.002	-.001	.817
.834							-.003			.834
.852							-.006			.852
.870							-.005			.870
.888							-.009			.888
.906							-.009			.906
.921	-.073	-.005	.003		-.009	.003		.013		.921
.924								.012		.924
.947								.017		.947
.964								.010		.964
.981	.060	.020	.005			.012	.011		.006	.981
1.000	-.079								.002	1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.670		.031	.429	.523
.030	1.255		.094	.579	.753
.050	1.480		.156	.673	.948
.070	1.527		.219	.748	1.150
.090	1.535		.281	.821	1.312
.110	1.501		.344	.890	1.425
.130	1.457		.406	.965	1.480
.150	1.431		.469	1.030	1.503
.170	1.392		.750	1.297	
.190	1.297		1.000	1.436	
			1.250	1.459	
			1.500	1.475	

TABLE 11. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 1.57$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.548			.039
.040							1.553			.040
.041							1.434			.041
.044							1.097			.044
.049							.600			.049
.055							.745			.055
.060	.278	.590	.635	.647	.711	.716	.719	.667		.060
.075	-.124	-.052	-.033	-.006	.048	.048	.048	.003		.075
.085	-.056	.016	.048	.082	.134	.164	.144	.105	.002	.085
.096	.044									.096
.126		.112	.142	.171	.226	.238	.239	.186	.099	.126
.143							-.107			.143
.154	-.194	-.147	-.131	-.111	-.080	-.070	-.062	-.103		.154
.171							-.050			.171
.189							-.036			.189
.206	-.094	-.106	-.091	-.082	-.039	-.029	-.026	-.061		.206
.223							-.016			.223
.241							-.017			.241
.258	-.043	-.088	-.081	-.058	-.025	-.013	-.018	-.045	-.091	.258
.275							-.019			.275
.293							-.016			.293
.310	-.018	-.071	-.068	-.053	-.026	-.014	-.010	-.047	-.073	.310
.327							-.015			.327
.344							-.014			.344
.362	-.005	-.062	-.058	-.046	-.025	-.013	-.008	-.039	-.057	.362
.379							-.007			.379
.396	-.006	-.048	-.050	-.041	-.015	-.007	-.001	-.035	-.048	.396
.408							-.009			.408
.426	.160	.139	.144	.152	.177	.195	.194	.162	.140	.426
.451	.133	.121	.128	.140	.170	.175	.182		.120	.451
.476	.110	.103	.110	.128	.157	.178	.181	.147	.106	.476
.501	.091	.092	.110	.118	.150	.158	.164	.138	.092	.501
.530	-.094	-.090	-.081	-.072	-.043	-.038	-.035	-.058	-.096	.530
.548							-.018			.548
.566							-.007			.566
.584	-.045	-.071	-.059	-.046	-.020	-.013	-.006	-.032	-.075	.584
.602							-.002			.602
.620							-.007			.620
.638	-.028	-.056		-.028	-.005				-.052	.638
.655							-.003			.655
.673							-.000			.673
.691	-.016	-.061	-.054	-.040	-.026	-.012	-.002	-.023	-.049	.691
.709							-.000			.709
.727							-.003			.727
.745	-.000	-.053		-.038	-.004		-.005		-.042	.745
.763							-.008			.763
.781							-.000			.781
.799							-.013			.799
.817	.008	-.039	-.037	-.030	-.007	.006	-.013	-.015	-.027	.817
.834							-.004			.834
.852							-.014			.852
.870							-.015			.870
.888							-.016			.888
.906							-.020			.906
.921	-.134									.921
.924	.017	-.019			-.026	.004	.022		-.034	.924
.947							.021			.947
.964							.026			.964
.981	.076	-.009	-.019			.014	.018	.019	-.002	-.020
1.000	-.147									1.000

Orifice station, y, in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.540		.031	.677	.532
.030	1.181		.094	1.084	.920
.050	1.459		.156	1.284	1.251
.070	1.516		.219	1.336	1.495
.090	1.510		.281	1.350	1.535
.110	1.463		.344	1.363	1.504
.130	1.412		.406	1.377	1.510
.150	1.380		.469	1.390	1.517
.170	1.346		.750	1.427	
.190	1.257		1.000	1.450	
			1.250	1.454	
			1.500	1.457	

TABLE 11. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 1.57$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.521			.039
.040							1.560			.040
.041							1.485			.041
.044							1.210			.044
.049							.786			.049
.055							.851			.055
.060	.171	.561	.631	.665	.770	.788	.794	.704		.060
.075	-.151	-.071	-.035	.021	.115	.125	.131	.050		.075
.085	-.103	.001	.057	.124	.224	.271	.253	.173	-.013	.085
.096	-.019									.096
.126		.091	.145	.200	.299	.322	.326	.235	.078	.126
.143							-.056			.143
.154	-.207	-.171	-.137	-.098	-.038	-.018	-.007	-.076		.154
.171							.003			.171
.189							.016			.189
.206	-.092	-.154	-.121	-.088	-.006	.016	.022	-.046		.206
.223							.029			.223
.241							.024			.241
.258	-.036	-.147	-.122	-.082	-.001	.023	.020	-.047	-.151	.258
.275							.017			.275
.293							.016			.293
.310	-.012	-.156	-.131	-.086	-.015	.009	.019	-.057	-.151	.310
.327							.013			.327
.344							.011			.344
.362	.001	-.136	-.137	-.096	-.016	.011	.020	-.058	-.127	.362
.379							.016			.379
.396	.002	-.110	-.121	-.099	-.015	.008	.022	-.068	-.109	.396
.408							.011			.408
.426	.230	.077	.068	.085	.174	.217	.223	.123	.084	.426
.451	.145	.069	.074	.092	.168	.195	.210	.070	.451	
.476	.111	.051	.068	.098	.166	.191	.200	.134	.053	.476
.501	.088	.041	.066	.093	.162	.185	.196	.131	.041	.501
.530	-.111	-.127	-.111	-.088	-.030	-.013	-.005	-.056	-.135	.530
.548							.009			.548
.566							.021			.566
.584	-.051	-.114	-.093	-.066	-.003	.014	.024	-.036	-.120	.584
.602							.033			.602
.620							.025			.620
.638	-.031	-.101		-.059	-.005		.033			.638
.655							.029			.655
.673										.673
.691	-.023	-.118	-.097	-.072	-.015	.011	-.031	-.032	-.106	.691
.709							.034			.709
.727							.029			.727
.745	-.012	-.107		-.069	-.002		.027			.745
.763							.025			.763
.781							.013			.781
.799							.023			.799
.817	-.004	-.087	-.091	-.071	-.003	.024	.035	-.033	-.076	.817
.834							.029			.834
.852							.036			.852
.870							.039			.870
.888							.036			.888
.906							.037			.906
.921	-.181									.921
.924	.000	-.060			-.062	.002			-.079	.924
.947							.042			.947
.964							.041			.964
.981	.042	-.040	-.063			.017	.035	-.017	-.058	.981
1.000	-.191						.040			1.000

Orifice station, $y_{in.}$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y_{in.}$	Cp top	Cp side
.010	.396		.031	.832	.450
.030	1.060		.094	1.440	1.036
.050	1.416		.156	1.459	1.490
.070	1.488		.219	1.410	1.618
.090	1.471		.281	1.423	1.526
.110	1.406		.344	1.412	1.493
.130	1.348		.406	1.414	1.503
.150	1.318		.469	1.420	1.508
.170	1.287		.520	1.408	
.190	1.199		1.000	1.415	
			1.250	1.427	
			1.500	1.446	

TABLE 12. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.29$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
.039							1.638			.039
.040							1.482			.040
.041							1.205			.041
.044							.749			.044
.049							.205			.049
.055							.124			.055
.060	.801	.470	.401	.307	.200	.185	.168	.241		.060
.075	.225	.062	.026	-.005	-.048	-.064	-.067	-.036		.075
.085	.259	.067	.034	.002	-.036	-.049	-.053	-.020	.062	.085
.096	.300									.096
.126		.079	.043	.014	-.009	-.012	-.010	-.002	.077	.126
.143							-.130			.143
.154	.038	-.076	-.098	-.112	-.123		.118	-.120	-.079	.154
.171							.104			.171
.189							.089			.189
.206	.041	-.082	-.100	-.111	-.092	-.081	-.076	-.103	-.083	.206
.223							.063			.223
.241							.055			.241
.258	.037	-.090	-.110	-.115	-.069	-.055	-.048	-.087	-.089	.258
.275							.040			.275
.293							.035			.293
.310	.036	-.102	-.122	-.111	-.055	-.040	-.029	-.065	-.102	.310
.327							.025			.327
.344							.021			.344
.362	.033	-.111	-.132	-.077	-.053	-.036	-.017	-.054	-.112	.362
.379							.017			.379
.396	.030	-.119	-.125	-.065	-.057	-.035	-.009	-.053	-.119	.396
.408							.010			.408
.426	.171	-.049	-.001	.009	.002	.084	.098	.014	-.052	.426
.451	.169	-.048	-.003	.014	.002	.046	.094	.022	-.049	.451
.476	.169	-.043	-.002	.018	-.006	.042	.068	.023	-.043	.476
.501	.167		-.000	.015	-.016	.034	.051		-.030	.501
.530	.024	-.104	-.088	-.060	-.095		.051	-.055	-.102	.530
.548							.054			.548
.566							.053			.566
.584	.027	-.094	-.070	-.056	-.089	-.052	-.041	-.058	-.091	.584
.602							.036			.602
.620							.032			.620
.638	.026	-.088		-.060	-.067		.027		-.087	.638
.655							.025			.655
.673							.025			.673
.691	.024	-.090	-.077		-.050	-.038		.051	-.088	.691
.709							.025			.709
.727							.025			.727
.745	.026	-.088		-.050	-.044		.024		-.088	.745
.763							.025			.763
.781							.028			.781
.799							.028			.799
.817	.025	-.094	-.056	-.043	-.045	-.040	-.025	-.038	-.090	.817
.834							.028			.834
.870							.025			.870
.888							.026			.888
.921	-.167							.024		.921
.924	.025	-.075			-.040			.025		.924
.947								.025		.947
.964								.026		.964
.981	.060	-.061	-.037			-.041	-.041		-.037	.981
.984						-.044			-.065	.984
.998									-.078	.998
1.000		-.168						-.029		1.000

Orifice station, y , in.	Nose rake		Base rake		
	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side
.010	1.218		.031	.727	.258
.030	1.607		.094	1.717	.818
.050	1.705		.156	1.714	1.311
.070	1.729		.219	1.763	1.392
.090	1.659		.281	1.761	1.450
.110	1.668		.344	1.756	1.461
.130	1.674		.406	1.756	1.473
.150	1.682		.469	1.754	1.480
.170	1.687		.750	1.749	
.190	1.616		1.000	1.747	
			1.250	1.725	
			1.500	1.714	

TABLE 12. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.29$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.663			.039
.040							1.540			.040
.041							1.293			.041
.044							1.846			.044
.049							1.286			.049
.055							1.216			.055
.060	.663	.453	.419	.356	.293	.273	.256	.311		.060
.075	.152	.060	.037	.014	-.010	-.022	-.026	-.006		.075
.085	.176	.069	.049	.026	-.000	-.008	-.012	.010	.064	.085
.096	.206									.096
.126		.086	.065	.046	.026	.021	.019	.032	.084	.126
.143							-.114			.143
.154	-.007	-.070	-.082	-.093	-.104		-.104	-.100	-.073	.154
.171							-.092			.171
.189							-.083			.189
.206	-.003	-.064	-.072	-.080	-.076	-.074	-.074	-.076	-.065	.206
.223							-.063			.223
.241							-.055			.241
.258	-.001	-.061	-.070	-.070	-.055	-.049	-.048	-.061	-.060	.258
.275							-.042			.275
.293							-.037			.293
.310	-.000	-.063	-.067	-.060	-.042	-.033	-.032	-.045	-.060	.310
.327							-.027			.327
.344							-.025			.344
.362	-.000	-.063	-.061	-.049	-.031	-.025	-.022	-.036	-.060	.362
.379							-.022			.379
.396	-.004	-.062	-.058	-.045	-.028	-.023	-.019	-.032	-.058	.396
.408							-.019			.408
.426	.120	.036	.047	.055	.057	.067	.075	.056	.039	.426
.451	.121	.043	.045	.048	.057	.062	.069	.054	.042	.451
.476	.117	.043	.041	.042	.050	.054	.060	.049	.044	.476
.501	.116		.037	.037	.041	.045	.052		.041	.501
.530	-.012	-.060	-.064	-.060	-.043		-.049	-.049	-.061	.530
.548							-.042			.548
.566							-.034			.566
.584	-.006	-.057	-.056	-.048	-.040	-.038	-.031	-.040	-.056	.584
.602							-.030			.602
.620							-.028			.620
.638	-.005	-.052			-.040	-.033			-.051	.638
.655							-.026			.655
.673							-.024			.673
.691	-.002	-.051	-.046			-.030	-.029		-.030	.691
.709							-.020			.709
.727							-.018			.727
.745	-.000	-.045			-.035	-.026			-.044	.745
.763							-.017			.763
.781							-.015			.781
.799							-.018			.799
.817	.001	-.045	-.039	-.029	-.022	-.020	-.011	-.021	-.040	.817
.834							-.012			.834
.870							-.009			.870
.888							-.009			.888
.921	-.151							-.008		.921
.924	.003	-.041			-.025			-.009		.924
.947								-.005		.947
.964								-.005		.964
.981	.033	-.036	-.026			-.016	-.008	-.016	-.033	.981
.984						-.020			-.040	.984
.998								-.012		.998
1.000		-.153								1.000

Orifice station, y , in.	Nose rake		Base rake		Orifice station, y , in.	C_p top	C_p side
	C_p top	C_p side	C_p top	C_p side			
.010	1.080		.031	.592	.366		
.030	1.565		.094	1.310	.816		
.050	1.669		.156	1.494	1.108		
.070	1.645		.219	1.626	1.290		
.090	1.544		.281	1.630	1.405		
.110	1.553		.344	1.638	1.445		
.130	1.563		.406	1.645	1.473		
.150	1.570		.469	1.651	1.488		
.170	1.565		.750	1.668			
.190	1.499		1.000	1.659			
			1.250	1.669			
			1.500	1.655			

TABLE 12. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.29$ - Continued(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.680			.039
.040							1.584			.040
.041							1.376			.041
.044							.944			.044
.049							.370			.049
.055							.286			.055
.060	.546	.453	.446	.405	.381	.362	.349	.380		.060
.075	.100	.062	.050	.040	.030	.020	.018	.027		.075
.085	.118	.072	.063	.053	.038	.035	.032	.043	.067	.085
.096	.143									.096
.126		.095	.084	.073	.063	.058	.057	.065	.093	.126
.143							.095			.143
.154	-.039	-.065	-.071	-.076	-.084		-.084	-.083	-.068	.154
.171							.077			.171
.189							.069			.189
.206	-.029	-.051	-.054	-.060	-.063	-.063	-.063	-.060	-.052	.206
.223							.056			.223
.241							.050			.241
.258	-.025	-.043	-.045	-.047	-.045	-.045	-.046	-.045	-.039	.258
.275							.041			.275
.293							.037			.293
.310	-.018	-.037	-.039	-.038	-.035	-.033	-.031	-.033	-.034	.310
.327							.028			.327
.344							.025			.344
.362	-.017	-.034	-.032	-.029	-.027	-.024	-.022	-.025	-.028	.362
.379							.021			.379
.396	-.018	-.030	-.030	-.027	-.022	-.020	-.017	-.022	-.026	.396
.408							.018			.408
.426	.099	.075	.078	.076	.077	.078	.077	.078	.077	.426
.451	.096	.074	.073	.072	.070	.069	.070	.072	.075	.451
.476	.094	.070	.066	.065	.063	.065	.066	.063	.072	.476
.501	.092		.062	.057	.060	.061	.063		.068	.501
.530	-.027	-.040	-.044	-.043	-.038		.038	-.038	-.040	.530
.548							.033			.548
.566							.030			.566
.584	-.022	-.037	-.035	-.032	-.032	-.032	-.029	-.029	-.033	.584
.602							.026			.602
.620							.023			.620
.638	-.018	-.029			-.026	-.023			-.026	.638
.655							.022			.655
.673							.022			.673
.691							.021			.691
.709	-.014	-.027	-.026			-.025	-.023		-.024	.709
.727							.019			.727
.745	-.012	-.020			-.022	-.019			-.018	.745
.763							.016			.763
.781							.015			.781
.799							.016			.799
.817	-.008	-.021	-.020	-.018	-.014	-.012	-.010	-.013	-.014	.817
.834							.010			.834
.870							.008			.870
.888							.006			.888
.921	-.129						.005			.921
.924	-.005	-.017			-.018		.006			.924
.947							.006			.947
.964							.003			.964
.981	.016	-.015	-.012			-.006	-.005	-.007	-.006	.981
.984						-.011			-.015	.984
.998										.998
1.000		-.133								1.000

Orifice station, y_1 in.	Nose rake		Base rake		
	Cp _{top}	Cp _{side}	Orifice station, y_1 in.	Cp _{top}	Cp _{side}
.010	.960		.031	.433	.345
.030	1.540		.094	.853	.600
.050	1.624		.156	1.076	.785
.070	1.559		.219	1.280	.944
.090	1.453		.281	1.450	1.084
.110	1.461		.344	1.520	1.225
.130	1.463		.406	1.551	1.329
.150	1.471		.469	1.574	1.402
.170	1.469		.570	1.598	
.190	1.402		1.000	1.594	
			1.250	1.619	
			1.500	1.626	

TABLE 12. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT M = 2.29 - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.684			.039
.040							1.630			.040
.041							1.438			.041
.044							1.028			.044
.049							4.76			.049
.055							3.75			.055
.060	.443	.449	.467	.449	.462	.440	.433	.445		.060
.075	.053	.059	.060	.063	.066	.061	.060	.056	.067	.075
.085	.068	.071	.074	.074	.075	.076	.074	.073		.085
.096	.091									.096
.126		.094	.095	.094	.098	.096	.095	.094	.095	.126
.143							.073			.143
.154	-.065	-.064	-.064	-.064	-.064		-.062	-.066	-.065	.154
.171							-.057			.171
.189							-.052			.189
.206	-.050	-.048	-.047	-.050	-.047	-.047	-.047	-.047	-.047	.206
.223							-.043			.223
.241							-.040			.241
.258	-.039	-.036	-.037	-.036	-.036	-.036	-.038	-.036	-.033	.258
.275							-.035			.275
.293							-.034			.293
.310	-.030	-.030	-.031	-.030	-.031	-.030	-.029	-.027	-.027	.310
.327							-.026			.327
.344							-.025			.344
.362	-.024	-.027	-.025	-.025	-.026	-.024	-.023	-.021	-.021	.362
.379							-.022			.379
.396	-.021	-.024	-.025	-.024	-.023	-.021	-.018	-.020	-.019	.396
.408							-.020			.408
.426	.087	.082	.086	.084	.084	.086	.086	.087	.086	.426
.451	.083	.080	.081	.082	.084	.083	.084	.083	.082	.451
.476	.078	.076	.075	.077	.079	.080	.079	.079	.077	.476
.501	.073		.072	.072	.074	.073	.074		.073	.501
.530	-.035	-.035	-.035	-.035	-.035		-.036	-.033	-.034	.530
.548							-.036			.548
.566							-.032			.566
.584	-.028	-.028	-.028	-.030	-.033	-.032	-.029	-.030	-.028	.584
.602							-.026			.602
.620							-.025			.620
.638	-.024	-.027		-.024	-.022		-.021			.638
.655							-.019			.655
.673							-.019			.673
.691	-.020	-.024	-.022		-.023	-.021		-.018	-.020	.691
.709							-.015			.709
.727							-.014			.727
.745	-.016	-.015		-.016	-.016		-.015		-.013	.745
.763							-.015			.763
.781							-.017			.781
.799							-.015			.799
.817	-.011	-.014	-.015	-.016	-.014	-.013	-.011	-.010	-.007	.817
.834							-.012			.834
.870							-.009			.870
.888							-.008			.888
.921	-.123						-.006			.921
.924	-.007	-.013		-.015			-.008			.924
.947							-.005			.947
.964							-.008			.964
.981	.012	-.008	-.009			-.007		-.004	-.004	.981
.984							-.011		-.009	.984
.998							-.013			.998
1.000	-.127									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.855		.031	.287	.313
.030	1.498		.094	.493	.501
.050	1.580		.156	.621	.644
.070	1.456		.219	.750	.769
.090	1.366		.281	.848	.875
.110	1.373		.344	.963	.989
.130	1.378		.406	1.072	1.113
.150	1.386		.469	1.178	1.215
.170	1.377		.570	1.417	
.190	1.315		1.000	1.484	
			1.250	1.537	
			1.500	1.569	

TABLE 12. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT M = 2.29 - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.039							1.677				.039
.040							1.650				.040
.041							1.495				.041
.044							1.115				.044
.049							.574				.049
.055							.457				.055
.060	.357	.452	.492	.502	.543	.537	.542	.527			.060
.075	.009	.059	.072	.085	.108	.106	.107	.089			.075
.085	.025	.072	.084	.097	.117	.123	.122	.107	.068		.085
.096	.048										.096
.126		.093	.104	.115	.138	.142	.142	.124	.093		.126
.143							.046				.143
.154	-.086	-.066	-.060	-.052	-.041		-.035	-.048	-.068		.154
.171							.033				.171
.189							.030				.189
.206	-.063	-.052	-.047	-.043	-.031	-.028	-.027	-.035	-.052		.206
.223							.025				.223
.241							.025				.241
.258	-.045	-.044	-.039	-.035	-.025	-.023	-.024	-.031	-.040		.258
.275							.023				.275
.293							.022				.293
.310	-.032	-.040	-.037	-.032	-.024	-.021	-.018	-.026	-.036		.310
.327							.018				.327
.344							.017				.344
.362	-.023	-.037	-.034	-.031	-.023	-.019	-.017	-.022	-.032		.362
.379							.017				.379
.396	-.020	-.033	-.034	-.030	-.021	-.018	-.014	-.023	-.029		.396
.408							.017				.408
.426	.079	.071	.077	.079	.090	.097	.097	.088	.073		.426
.451	.075	.071	.075	.080	.093	.095	.099	.087	.072		.451
.476	.067	.068	.070	.077	.090	.093	.096	.086	.070		.476
.501	.060		.069	.073	.085	.087	.090		.065		.501
.530	-.039	-.042	-.040	-.039	-.035		-.029	-.036	-.041		.530
.548							.023				.548
.566							.018				.566
.584	-.032	-.038	-.037	-.033	-.025	-.022	-.018	-.025	-.036		.584
.602							.015				.602
.620							.015				.620
.638	-.025	-.033		-.023	-.018		.013		-.029		.638
.655							.010				.655
.673							.010				.673
.691	-.021	-.030	-.028		-.019	-.017		-.017	-.027		.691
.709							.012				.709
.727							.011				.727
.745	-.017	-.023		-.023	-.017		.010		-.021		.745
.763							.009				.763
.781							.014				.781
.799							.011				.799
.817	-.010	-.024	-.023	-.022	-.015	-.010		-.014	-.016		.817
.834							.006				.834
.870							.009				.870
.888							.005				.888
.921	-.127						.003				.921
.924	-.008						.002				.924
.947							.000				.947
.964											.964
.981	.009	-.017	-.017		-.007		-.006	-.005	-.004	-.012	.981
.984					-.013					-.017	.984
.998											.998
1.000		-.132									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.760		.031	.213	.359
.030	1.432		.094	.333	.645
.050	1.522		.156	.428	.855
.070	1.369		.219	.518	1.012
.090	1.292		.281	.593	1.163
.110	1.287		.344	.661	1.309
.130	1.289		.406	.725	1.397
.150	1.295		.469	.785	1.454
.170	1.290		.750	1.005	
.190	1.229		1.000	1.199	
			1.250	1.285	
			1.500	1.345	

TABLE 12. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.29$ - Continued(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270		
.039											.039
.040											.040
.041											.041
.044											.044
.049											.049
.055											.055
.060	.264	.457	.527	.564	.647	.657	.673	.612			.060
.075	-.030	.060	.084	.110	.156	.160	.162	.125			.075
.085	-.016	.070	.095	.124	.167	.181	.182	.147	.068		.085
.096	-.007										.096
.126		.087	.113	.139	.187	.200	.202	.163	.088		.126
.143											.143
.154	-.104	-.071	-.056	-.040	-.013						.154
.171											.171
.189											.189
.206	-.071	-.065	-.051	-.038	-.009	-.002					.206
.223											.223
.241											.241
.258	-.045	-.063	-.052	-.036	-.006	.001	.001	-.021	-.059		.258
.275											.275
.293											.293
.310	-.029	-.067	-.056	-.039	-.010	-.001					.310
.327											.327
.344											.344
.379	-.019	-.070	-.059	-.042	-.013	-.003					.379
.396	-.017	-.067	-.062	-.046	-.013	-.004					.396
.408											.408
.426	.081	.030	.042	.057	.103	.121	.125	.085	.033		.426
.451	.072	.038	.043	.061	.106	.118	.126	.085	.039		.451
.476	.060	.040	.042	.058	.100	.113	.118	.084	.042		.476
.501	.052		.046	.056	.093	.106	.115		.040		.501
.530	-.050	-.062	-.060	-.052	-.022						.530
.548											.548
.566											.566
.584	-.035	-.057	-.049	-.038	-.015	-.007					.584
.602											.602
.620											.620
.638	-.027	-.055		-.034	-.010						.638
.655											.655
.673											.673
.691	-.021	-.055	-.050		-.017	-.006					.691
.709											.709
.727											.727
.745	-.016	-.050		-.039	-.011						.745
.763											.763
.781											.781
.799											.799
.817	-.010	-.050	-.049	-.039	-.010	.000	.006	-.019	-.045		.817
.834											.834
.870											.870
.868											.868
.921	-.148										.921
.924	-.009	-.044									.924
.947											.947
.964											.964
.981											.981
.984											.984
.998											.998
1.000	-.149										1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.648		.031	.456	.366
.030	1.347		.094	.980	.848
.050	1.445		.156	1.014	1.142
.070	1.285		.219	1.048	1.312
.090	1.207		.281	1.063	1.414
.110	1.195		.344	1.076	1.458
.130	1.193		.406	1.091	1.485
.150	1.195		.469	1.107	1.493
.170	1.198		.570	1.158	
.190	1.135		1.000	1.193	
			1.250	1.220	
			1.500	1.221	

TABLE 12. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.29$ - Concluded(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.639			.039
.040							1.684			.040
.041							1.604			.041
.044							1.294			.044
.049							.792			.049
.055							.746			.055
.060	.160	.473	.567	.632	.768	.790	.805	.706		.060
.075	-.073	.057	.094	.136	.215	.229	.235	.170		.075
.085	-.059	.065	.104	.155	.234	.258	.264	.198	.065	.085
.096	-.032									.096
.126		.076	.118	.166	.251	.272	.280	.209	.079	.126
.143							.031			.143
.154	-.120	-.080	-.055	-.026	.021		.045	-.002	-.079	.154
.171							.042			.171
.189							.045			.189
.206	-.074	-.086	-.059	-.031	.024	.039	.046	-.000	-.083	.206
.223							.047			.223
.241							.044			.241
.258	-.045	-.096	-.071	-.036	.021	.038	.042	-.004	-.089	.258
.275							.040			.275
.293							.038			.293
.310	-.028	-.110	-.080	-.046	.013	.032	.041	-.009	-.103	.310
.327							.039			.327
.344							.038			.344
.362	-.018	-.122	-.090	-.053	.008	.029	.038	-.015	-.115	.362
.379							.035			.379
.396	-.014	-.127	-.097	-.060	.006	.027	.038	-.020	-.122	.396
.408							.033			.408
.426	.095	-.063	-.011	.036	.132	.167	.177	.096	-.058	.426
.451	.091	-.058	-.015	.036	.134	.161	.176	.094	-.057	.451
.476	.065	-.051	-.017	.036	.132	.163	.176	.096	-.052	.476
.501	.050	-.014	.036	.132	.160		.174		-.044	.501
.530	-.052	-.108	-.101	-.067	.000		.029	-.028	-.109	.530
.548							.031			.548
.566							.033			.566
.584	-.044	-.097	-.099	-.065	.001	.020	.031	-.025	-.096	.584
.602							.032			.602
.620							.031			.620
.638	-.030	-.095		-.065	.001		.032		-.095	.638
.655							.032			.655
.673							.030			.673
.691	-.028	-.098	-.094		-.006	.016		-.029	-.097	.691
.709							.024			.709
.727							.027			.727
.745	-.027	-.099		-.075	-.002		.031		-.098	.745
.763							.032			.763
.781							.027			.781
.799							.026			.799
.817	-.029	-.102	-.094	-.064	-.002	.022	.034	-.028	-.095	.817
.834							.031			.834
.870							.031			.870
.888							.028			.888
.921	-.167									.921
.924	-.032	-.073		-.073			.030			.924
.947							.028			.947
.964							.030			.964
.981	-.008	-.060	-.102		-.000	.020	.027	-.024	-.070	.981
.984							.028		-.083	.984
.998										.998
1.000		-.171								1.000

Nose rake			Base rake		
Orifice station, y , in.	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side
.010	.390		.031	.506	.235
.030	1.122		.094	1.127	.787
.050	1.361		.156	1.062	1.336
.070	1.227		.219	1.101	1.402
.090	1.115		.281	1.108	1.434
.110	1.089		.344	1.115	1.458
.130	1.095		.406	1.123	1.461
.150	1.096		.469	1.132	1.454
.170	1.087		.750	1.168	
.190	1.031		1.000	1.207	
			1.250	1.256	
			1.500	1.289	

TABLE 13. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.98$ (a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.039							1.717				.039
.040							1.545				.040
.041							1.271				.041
.044							.791				.044
.049							.260				.049
.055							.177				.055
.060	.758	.426	.359	.300	.216	.200	.190	.252			.060
.075	.253	.090	.060	.031	-.000	-.014	-.015	.009			.075
.085	.263	.090	.062	.034	-.000	-.008	-.011	.016	.091		.085
.096	.278										.096
.126		.081	.052	.028	.005	.003	.002	.015	.083		.126
.143							.075				.143
.154	.061	-.034	-.050	-.063	-.072		-.071	-.070	-.034		.154
.171							.066				.171
.189							.057				.189
.206	.054	-.046	-.060	-.067	-.059	-.052	-.050	-.068	-.045		.206
.223							.042				.223
.241							.036				.241
.258	.049	-.052	-.067	-.074	-.047	-.036	-.032	-.067	-.048		.258
.275							.029				.275
.293							.026				.293
.310	.046	-.060	-.076	-.080	-.041	-.030	-.023	-.055	-.062		.310
.327							.022				.327
.344							.019				.344
.362	.043	-.067	-.083	-.067	-.042	-.029	-.017	-.050	-.069		.362
.379							.015				.379
.396	.041	-.072	-.086	-.058	-.045	-.030	-.013	-.051	-.073		.396
.408							.012				.408
.426	.155	-.029	-.027	-.013	-.010	.022	.057	-.013	-.029		.426
.451	.160	-.023	-.036	-.001	-.007	.019	.064	-.003	-.023		.451
.476	.165	-.024	-.035	.001	-.014	.017	.049	-.001	-.020		.476
.501	.168	-.033	-.002	.019	.015	.038			.019		.501
.530	.047	-.073	-.074	-.045	-.062	-.030	-.037	-.073			.530
.548							.031				.548
.566							.031				.566
.584	.045	-.076	-.063	-.048	-.063	-.037	-.030	-.044	-.075		.584
.602							.025				.602
.620							.025				.620
.638	.043	-.078		-.050	-.047				.080		.638
.655							.023				.655
.673							.022				.673
.691	.042	-.081	-.059		-.038	-.032		-.051	-.083		.691
.709							.021				.709
.727							.021				.727
.745	.042	-.071		-.044	-.038				.085		.745
.763							.021				.763
.781							.022				.781
.799							.025				.799
.817	.041	-.077	-.045	-.039	-.044	-.034		-.042	-.085		.817
.834							.024				.834
.870							.025				.870
.888							.024				.888
.921	-.111						.023				.921
.924	.040	-.051			-.041			.024			.924
.947							.023				.947
.964							.023				.964
.981	.068	-.042	-.035				-.043	-.025		-.045	.981
.984										-.056	.984
.998											.998
1.000		-.113									1.000

Nose rake			Base rake		
Orifice station, y,in.	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	1.261		.031	.716	.118
.030	1.697		.094	1.690	.378
.050	1.775		.156	1.902	.895
.070	1.718		.219	1.963	1.144
.090	1.692		.281	1.966	1.318
.110	1.711		.344	1.958	1.384
.130	1.739		.406	1.941	1.417
.150	1.760		.469	1.941	1.441
.170	1.793		.750	1.947	
.190	1.750		1.000	1.941	
			1.250	1.941	
			1.500	1.923	

TABLE 13. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT M = 2.98 - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.716			.039
.040							1.598			.040
.041							1.330			.041
.044							.884			.044
.049							.310			.049
.055							.244			.055
.060	.624	.422	.393	.346	.284	.277	.267	.318		.060
.075	.183	.091	.070	.052	.030	.020	.019	.037		.075
.085	.188	.091	.072	.054	.032	.025	.023	.044		.085
.096	.213									.096
.126		.086	.068	.052	.034	.030	.030	.042	.090	.126
.143							.064			.143
.154	.022	-.032	-.041	-.050	-.059		-.060	-.056	-.031	.154
.171							.057			.171
.189							.051			.189
.206	.013	-.035	-.042	-.047	-.047	-.047	-.046	-.048	-.033	.206
.223							.041			.223
.241							.037			.241
.258	.013	-.038	-.045	-.047	-.038	-.034	-.033	-.044	-.033	.258
.275							.029			.275
.293							.027			.293
.310	.011	-.040	-.046	-.044	-.031	-.026	-.023	-.037	-.040	.310
.327							.021			.327
.344							.019			.344
.362	.010	-.044	-.046	-.040	-.024	-.019	-.018	-.030	-.043	.362
.379							.016			.379
.396	.006	-.045	-.046	-.036	-.021	-.016	-.014	-.027	-.044	.396
.408							.013			.408
.426	.101	.011	.019	.026	.034	.041	.045	.031	.013	.426
.451	.110	.020	.021	.028	.039	.041	.047	.034	.022	.451
.476	.109	.024	.024	.027	.033	.036	.040	.033	.027	.476
.501	.110		.024	.025	.029	.031	.036		.029	.501
.530	.009	-.041	-.043	-.037	-.026		.027	-.029	-.038	.530
.548							.026			.548
.566							.021			.566
.584	.011	-.041	-.042	-.037	-.027	-.024	-.019	-.028	-.039	.584
.602							.018			.602
.620							.016			.620
.638	.010	-.041			-.031	-.021			-.038	.638
.655							.014			.655
.673							.014			.673
.691	.008	-.042	-.039			-.019	-.018			.691
.709							.012			.709
.727							.010			.727
.745	.010	-.032			-.023	-.017			-.040	.745
.763							.010			.763
.781							.011			.781
.799							.010			.799
.817	.011	-.038	-.029	-.020	-.016	-.015	-.007	-.017	-.039	.817
.834							.008			.834
.870							.006			.870
.888							.005			.888
.921	-.108						.005			.921
.924	.011	-.034			-.020		.006			.924
.947							.006			.947
.964							.004			.964
.981	.030	-.031	-.021			-.013	-.005		-.031	.981
.984									-.037	.984
.998										.998
1.000	-.107									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	1.085		.031	.535	.271
.030	1.595		.094	1.220	.692
.050	1.647		.156	1.452	.934
.070	1.587		.219	1.644	1.150
.090	1.554		.281	1.684	1.286
.110	1.556		.344	1.698	1.364
.130	1.565		.406	1.715	1.405
.150	1.579		.469	1.729	1.423
.170	1.595		.750	1.746	
.190	1.554		1.000	1.759	
			1.250	1.769	
			1.500	1.762	

TABLE 13. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.98$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
.039							1.754			.039
.040							1.646			.040
.041							1.423			.041
.044							.979			.044
.049							.395			.049
.055							.320			.055
.060	.520	.421	.410	.388	.355	.350	.342	.377		.060
.075	.131	.092	.081	.072	.062	.055	.055	.065		.075
.085	.135	.092	.083	.075	.064	.060	.059	.072	.094	.085
.096	.166									.096
.126		.090	.081	.071	.062	.059	.060	.068	.093	.126
.143							.049			.143
.154	-.006	-.029	-.034	-.038	-.045		-.046	-.042	-.028	.154
.171							.044			.171
.189							.040			.189
.206	-.007	-.026	-.031	-.034	-.035	-.037	-.038	-.035	-.026	.206
.223							.035			.223
.241							.033			.241
.258	-.011	-.026	-.029	-.030	-.030	-.030	-.031	-.030	-.021	.258
.275							.029			.275
.293							.027			.293
.310	-.007	-.025	-.027	-.026	-.025	-.024	-.024	-.025	-.024	.310
.327							.022			.327
.344							.020			.344
.362	-.008	-.022	-.022	-.021	-.020	-.019	-.019	-.021	-.021	.362
.379							.017			.379
.396	-.008	-.021	-.022	-.020	-.018	-.015	-.015	-.018	-.020	.396
.408							.015			.408
.426	.073	.045	.049	.045	.047	.048	.047	.047	.047	.426
.451	.078	.051	.048	.048	.051	.049	.052	.050	.053	.451
.476	.081	.051	.048	.046	.048	.048	.048	.050	.054	.476
.501	.081	.047	.045	.045	.045	.046	.047		.055	.501
.530	-.007	-.020	-.022	-.021	-.019		.015	-.018	-.018	.530
.548							.016			.548
.566							.015			.566
.584	-.006	-.022	-.021	-.020	-.018	-.017		.014	-.017	.584
.602							.013			.602
.620							.013			.620
.638	-.005	-.020			-.018	-.015		.012		.638
.655							.012			.655
.673							.012			.673
.691	-.005	-.018	-.017			-.014	-.014		-.015	.691
.709							.010			.709
.727							.010			.727
.745	-.004	-.007			-.012	-.011		.009		.745
.763							.009			.763
.781							.008			.781
.799							.007			.799
.817	-.003	-.011	-.011	-.010	-.008	-.007	-.006		-.010	.817
.834							.006			.834
.870							.005			.870
.888							.003			.888
.921	-.095						.002			.921
.924	-.000	-.010			-.011		.003			.924
.947							.002			.947
.964							.002			.964
.981	.014	-.010	-.009			-.004	-.002		-.005	.981
.984							.006		-.010	.984
.998							.008		-.014	.998
1.000	-.097									1.000

Nose rake			Base rake		
Orifice station, y , in.	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.961		.031	.360	.268
.030	1.487		.094	.748	.501
.050	1.529		.156	.935	.655
.070	1.481		.219	1.201	.782
.090	1.452		.281	1.406	.930
.110	1.453		.344	1.513	1.085
.130	1.437		.406	1.550	1.218
.150	1.449		.469	1.589	1.303
.170	1.452		.520	1.582	
.190	1.399		1.000	1.633	
			1.250	1.659	
			1.500	1.675	

TABLE 13. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.98$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.760			.039
.040							1.698			.040
.041							1.502			.041
.044							1.078			.044
.049							.488			.049
.055							.387			.055
.060	.424	.424	.438	.428	.437	.424	.420	.432		.060
.075	.089	.090	.091	.091	.095	.092	.093	.093		.075
.085	.093	.092	.094	.095	.096	.096	.097	.098	.095	.085
.096	.120									.096
.126		.091	.091	.091	.094	.092	.094	.094	.094	.126
.143							-.031			.143
.154	-.027	-.028	-.028	-.029	-.029		-.027	-.028	-.028	.154
.171							-.027			.171
.189							-.026			.189
.206	-.025	-.026	-.026	-.026	-.024	-.026	-.026	-.024	-.024	.206
.223							-.024			.223
.241							-.024			.241
.258	-.022	-.021	-.022	-.022	-.022	-.022	-.023	-.021	-.017	.258
.275							-.022			.275
.293							-.021			.293
.310	-.020	-.019	-.020	-.019	-.020	-.020	-.020	-.019	-.018	.310
.327							-.019			.327
.344							-.017			.344
.362	-.016	-.016	-.016	-.016	-.017	-.017	-.016	-.015	-.015	.362
.379							-.017			.379
.396	-.013	-.014	-.016	-.015	-.015	-.014	-.014	-.013	-.012	.396
.408							-.014			.408
.426	.058	.053	.059	.054	.054	.055	.054	.056	.056	.426
.451	.061	.057	.057	.058	.061	.059	.060	.060	.060	.451
.476	.061	.057	.057	.058	.059	.059	.060	.061	.061	.476
.501	.061			.057	.056	.058	.060	.060	.060	.501
.530	-.012	-.014	-.015	-.014	-.014	-.012	-.012	-.011	-.012	.530
.548							-.012			.548
.566							-.011			.566
.584	-.012	-.015	-.015	-.014	-.015	-.015	-.012	-.012	-.011	.584
.602							-.012			.602
.620							-.013			.620
.638	-.012	-.013			-.012	-.014				.638
.655							-.012			.655
.673							-.011			.673
.691	-.011	-.012	-.012		-.013	-.013				.691
.709							-.009			.709
.727							-.008			.727
.745	-.008	-.002			-.008	-.010				.745
.763							-.007			.763
.781							-.008			.781
.799							-.007			.799
.817	-.006	-.006	-.006	-.006	-.007	-.006	-.005	-.006	-.007	.817
.834							-.005			.834
.870							-.006			.870
.888							-.005			.888
.921	-.087						-.003			.921
.924	-.002	-.004			-.007		-.002			.924
.947							-.002			.947
.964							-.002			.964
.981	.007	-.003	-.004			-.002	-.003	-.002	-.004	.981
.984							-.006		-.008	.984
.998							-.006			.998
1.000	-.090									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.877		.031	.229	.233
.030	1.409		.094	.400	.397
.050	1.444		.156	.496	.507
.070	1.382		.219	.578	.605
.090	1.349		.281	.659	.687
.110	1.345		.344	.767	.796
.130	1.337		.406	.863	.915
.150	1.333		.469	.971	.986
.170	1.340		.530	1.319	
.190	1.281		1.000	1.416	
			1.250	1.530	
			1.500	1.587	

TABLE 13. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.98$ - Continued(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.750			.039
.040							1.711			.040
.041							1.557			.041
.044							1.160			.044
.049							.599			.049
.055							.459			.055
.060	.338	.423	.463	.472	.515	.506	.507	.495		.060
.075	.052	.090	.101	.113	.133	.134	.136	.123		.075
.085	.056	.092	.104	.115	.133	.136	.139	.128	.095	.085
.096	.083									.096
.126		.090	.099	.109	.129	.132	.135	.120	.094	.126
.143							-.009			.143
.154	-.045	-.029	-.025	-.019	-.010		-.005	-.014	-.029	.154
.171							-.007			.171
.189							-.007			.189
.206	-.038	-.029	-.024	-.019	-.010	-.008	-.008	-.013	-.027	.206
.223							-.007			.223
.241							-.009			.241
.258	-.031	-.027	-.024	-.021	-.012	-.009	-.009	-.015	-.021	.258
.275							-.009			.275
.293							-.008			.293
.310	-.025	-.026	-.023	-.019	-.012	-.008	-.007	-.013	-.025	.310
.327							-.007			.327
.344							-.007			.344
.362	-.019	-.024	-.021	-.018	-.012	-.009	-.008	-.013	-.022	.362
.379							-.009			.379
.396	-.016	-.022	-.022	-.019	-.011	-.009	-.008	-.013	-.020	.396
.408							-.008			.408
.426	.048	.042	.052	.052	.063	.069	.070	.059	.045	.426
.451	.051	.049	.052	.058	.073	.075	.077	.067	.052	.451
.476	.048	.049	.053	.059	.073	.078	.079	.071	.053	.476
.501	.046		.054	.059	.074	.078	.080	.054	.051	.501
.530	-.017	-.021	-.020	-.017	-.010		-.005	-.010	-.019	.530
.548							-.005			.548
.566							-.005			.566
.584	-.015	-.023	-.021	-.019	-.013	-.010	-.007	-.012	-.020	.584
.602							-.005			.602
.620							-.005			.620
.638	-.014	-.021			-.015	-.010			-.018	.638
.655							-.004			.655
.673							-.004			.673
.691							-.004			.691
.709							-.003			.709
.727							-.003			.727
.745	-.010	-.010			-.013	-.008			-.016	.745
.763							-.003			.763
.781							-.003			.781
.799							-.005			.799
.817	-.007	-.013	-.013	-.011	-.009	-.006	-.004	-.010	-.014	.817
.834							-.004			.834
.870							-.004			.870
.888							-.000			.888
.921	-.092						-.000			.921
.924	-.004	-.013			-.010		-.000			.924
.947							-.000			.947
.964							-.001			.964
.981	.005	-.010	-.010				-.000	-.004	-.010	.981
.984							-.000		-.016	.984
.998							-.001			.998
1.000	-.095									1.000

Nose rake			Base rake		
Orifice station, y,in.	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.543		.031	.173	.267
.030	1.215		.094	.295	.524
.050	1.360		.156	.362	.684
.070	1.290		.219	.419	.830
.090	1.246		.281	.468	1.001
.110	1.236		.344	.501	1.146
.130	1.222		.406	.554	1.282
.150	1.216		.469	.585	1.363
.170	1.211		.750	.758	
.190	1.172		1.000	.933	
			1.250	1.077	
			1.500	1.162	

TABLE 13. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.98$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.742			.039
.040							1.755			.040
.041							1.611			.041
.044							1.254			.044
.049							.702			.049
.055							.557			.055
.060	.264	.426	.485	.523	.601	.609	.614	.572		.060
.075	.017	.090	.114	.137	.179	.186	.190	.158		.075
.085	.021	.092	.114	.140	.178	.189	.194	.165	.096	.085
.096	.052									.096
.126		.087	.108	.130	.172	.182	.187	.154	.092	.126
.143							.021			.143
.154	-.059	-.033	-.021	-.007	.014		.024	.004	-.031	.154
.171							.020			.171
.189							.016			.189
.206	-.046	-.036	-.026	-.014	.006	.012	.015	-.001	-.034	.206
.223							.015			.223
.241							.015			.241
.258	-.034	-.040	-.030	-.017	.006	.012	.014	-.003	-.033	.258
.275							.013			.275
.293							.012			.293
.310	-.024	-.043	-.033	-.021	.001	.009	.013	-.005	-.041	.310
.327							.011			.327
.344							.011			.344
.362	-.017	-.046	-.036	-.023	-.000	.007	.010	-.007	-.044	.362
.379							.009			.379
.396	-.015	-.048	-.039	-.026	-.001	.006	.010	-.009	-.047	.396
.408							.009			.408
.426	.046	.005	.029	.042	.082	.097	.100	.069	.009	.426
.451	.048	.016	.030	.051	.095	.105	.113	.078	.019	.451
.476	.042	.019	.031	.052	.093	.106	.113	.080	.024	.476
.501	.037		.032	.051	.094	.107	.114		.026	.501
.530	-.026	-.043	-.038	-.027	-.000		.013	-.009	-.041	.530
.548							.013			.548
.566							.014			.566
.584	-.020	-.043	-.037	-.026	-.001	.006	.013	-.008	-.040	.584
.602							.013			.602
.620							.012			.620
.638	-.017	-.042		-.025	-.002		.013			.638
.655							.013			.655
.673							.012			.673
.691	-.014	-.044	-.037		-.004	.004		-.009	-.042	.691
.709							.011			.709
.727							.010			.727
.745	-.012	-.033		-.026	-.003		.009		-.041	.745
.763							.008			.763
.781							.007			.781
.799							.009			.799
.817	-.009	-.039	-.037	-.028	-.001	.009	.014	-.011	-.040	.817
.834							.013			.834
.870							.014			.870
.888							.014			.888
.921	-.104									.921
.924	-.008	-.034		-.026			.012			.924
.947							.011			.947
.964							.013			.964
.981	.005	-.033	-.035		-.000	.008	.011	-.011	-.034	.981
.984					-.002				-.039	.984
.998							.011			.998
1.000	-.106									1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	.370		.031	.369	.268
.030	1.037		.094	.822	.705
.050	1.209		.156	.807	.935
.070	1.183		.219	.838	1.166
.090	1.155		.281	.843	1.308
.110	1.155		.344	.857	1.357
.130	1.145		.406	.866	1.392
.150	1.132		.469	.875	1.427
.170	1.120		.530	.906	
.190	1.065		1.000	.952	
			1.250	.986	
			1.500	1.008	

TABLE 13. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 2.98$ - Concluded(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.705			.039
.040							1.749			.040
.041							1.666			.041
.044							1.357			.044
.049							.829			.049
.055							.701			.055
.060	.189	.430	.516	.585	.714	.739	.754	.661		.060
.075	-.017	.092	.126	.166	.239	.254	.261	.202		.075
.085	-.012	.092	.127	.170	.239	.260	.267	.213	.095	.085
.096	.015									.096
.126		.081	.118	.159	.235	.255	.264	.201	.086	.126
.143							.062			.143
.154	-.071	-.034	-.015	.007	.046		.055	.027	-.034	.154
.171							.057			.171
.189							.058			.189
.206	-.050	-.046	-.026	-.004	.040	.053	.058	.020	-.045	.206
.223							.057			.223
.241							.054			.241
.258	-.032	-.055	-.035	-.010	.035	.048	.052	.015	-.048	.258
.275							.050			.275
.293							.048			.293
.310	-.023	-.064	-.042	-.017	.028	.043	.049	.008	-.062	.310
.327							.048			.327
.344							.047			.344
.362	-.017	-.072	-.049	-.021	.024	.040	.046	.005	-.069	.362
.379							.045			.379
.396	-.014	-.075	-.053	-.026	.023	.039	.046	.003	-.073	.396
.408							.045			.408
.426	.054	-.034	.009	.044	.123	.152	.161	.094	-.030	.426
.451	.065	-.027	.008	.049	.132	.155	.167	.098	-.023	.451
.476	.050	-.027	.009	.051	.134	.161	.171	.105	-.020	.476
.501	.040		.011	.052	.135	.160	.173		-.018	.501
.530	-.028	-.077	-.053	-.026	.026		.049	.006	-.073	.530
.548							.048			.548
.566							.049			.566
.584	-.030	-.080	-.055	-.029	.021	.038	.046	.004	-.077	.584
.602							.046			.602
.620							.045			.620
.638	-.025	-.082		-.029	.020		.046		-.081	.638
.655							.046			.655
.673							.045			.673
.691	-.024	-.085	-.060		.017	.035		.001	-.084	.691
.709							.044			.709
.727							.045			.727
.745	-.024	-.074		-.033	.020		.046		-.087	.745
.763							.044			.763
.781							.041			.781
.799							.041			.799
.817	-.027	-.077	-.063	-.035	.019	.038	.046	-.003	-.084	.817
.834							.042			.834
.870							.045			.870
.888							.043			.888
.921	-.114						.042			.921
.924	-.025	-.049		-.039			.037			.924
.947							.042			.947
.964							.042			.964
.981	-.012	-.043	-.070			.017	.036		-.008	.981
.984						.016		.042		.984
.998										.998
1.000		-.116								1.000

Orifice station, $y_{in.}$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y_{in.}$	Cp top	Cp side
.010	.238		.031	.403	.100
.030	.852		.094	.905	.368
.050	1.076		.156	.830	.904
.070	1.080		.219	.854	1.167
.090	1.051		.281	.872	1.308
.110	1.038		.344	.873	1.367
.130	1.034		.406	.873	1.381
.150	1.022		.469	.875	
.170	1.005		.750	.893	
.190	.953		1.000	.918	
			1.250	.964	
			1.500	1.030	

TABLE 14. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT M = 3.96

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{z}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.733			.039
.040							1.565			.040
.041							1.275			.041
.044							.804			.044
.049							.283			.049
.055							.193			.055
.060	.720	.404	.253	.298	.228	.210	.203	.255		.060
.075	.266	.113	.086	.060	.031	.021	.018	.039		.075
.085	.263	.107	.081	.056	.029	.020	.019	.040	.104	.085
.096	.295									.096
.126		.080	.059	.040	.020	.016	.015	.026	.080	.126
.143							-.034			.143
.154		.073	-.005	-.019	-.027	-.034	-.033	-.032	-.006	.154
.171							-.031			.171
.189							-.029			.189
.206		.060	-.019	-.028	-.034	-.030	-.029	-.027	-.033	.206
.223							-.025			.223
.241							-.023			.241
.258		.052	-.015	-.034	-.039	-.033	-.029	-.019	-.036	.258
.275							-.016			.275
.293							-.014			.293
.310		.046	-.030	-.039	-.040	-.030	-.025	-.012	-.037	.310
.327							-.010			.327
.344							-.010			.344
.362		.045	-.035	-.041	-.039	-.030	-.020	-.010	-.038	.362
.379							-.010			.379
.396		.045	-.037	-.042	-.038	-.032	-.021	-.010	-.039	.396
.408							-.010			.408
.426		.133	-.013	-.000	-.022	-.018	.012	.023	-.022	.426
.451		.147	-.004	-.023	-.013	-.011	.006	.031	-.012	.451
.476		.156	-.003	-.022	-.011	-.020	.005	.027	-.006	.476
.501		.161		-.021	-.012	-.016	.004	.019	-.003	.501
.530		.057	-.036	-.041	-.037	-.030		.019	-.027	.530
.548								-.021		.548
.566								-.021		.566
.584		.049	-.039	-.040	-.039	-.029	-.025	-.023	-.034	.584
.602							-.020			.602
.620							-.021			.620
.638		.046	-.040		-.033	-.025		-.022		.638
.655							-.022			.655
.673							-.022			.673
.691		.046	-.042	-.039		-.028	-.025		-.038	.691
.709							-.021			.709
.727							-.021			.727
.745		.045	-.018		-.034	-.033		-.021		.745
.763							-.021			.763
.781							-.021			.781
.799							-.022			.799
.817		.045	-.041	-.034	-.032	-.034	-.027	-.022	-.034	.817
.834							-.023			.834
.870							-.022			.870
.888							-.022			.888
.921		-.060						-.022		.921
.924		.043	-.038		-.028			-.023		.924
.947								-.021		.947
.964								-.021		.964
.981		.061	-.033	-.027		-.034	-.033	-.023	-.033	.981
.984						-.035	-.033	-.023	-.041	.984
.998								-.024		.998
1.000		-.061								1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	1.130		.031	.618	.023
.030	1.557		.094	1.581	.298
.050	1.635		.156	2.001	.607
.070	1.669		.219	2.138	.828
.090	1.715		.281	2.122	.947
.110	1.756		.344	2.142	1.025
.130	1.801		.406	2.158	1.061
.150	1.818		.469	2.178	1.079
.170	1.848		.750	2.162	
.190	1.801		1.000	2.178	
			1.250	2.183	
			1.500	2.167	

TABLE 14. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 3.96$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270		
.039							1.775				.039
.040							1.636				.040
.041							1.385				.041
.044							.921				.044
.049							.369				.049
.055							.268				.055
.060	.577	.404	.374	.336	.287	.275	.267	.306			.060
.075	.194	.110	.094	.075	.055	.048	.045	.061			.075
.085	.191	.106	.089	.073	.052	.047	.045	.061	.104		.085
.096	.227										.096
.126		.084	.069	.056	.040	.037	.036	.047	.085		.126
.143							.029				.143
.154	.039	-.007	-.014	-.020	-.028		-.028	-.024	-.006		.154
.171							.027				.171
.189							.025				.189
.206	.026	-.014	-.020	-.024	-.025	-.026	-.024	-.026	-.012		.206
.223							.023				.223
.241							.022				.241
.258	.022	-.018	-.026	-.029	-.024	-.022	-.021	-.026	-.007		.258
.275							.018				.275
.293							.017				.293
.310	.020	-.023	-.029	-.030	-.023	-.019	-.015	-.025	-.021		.310
.327							.014				.327
.344							.013				.344
.362	.017	-.026	-.031	-.031	-.019	-.015	-.012	-.024	-.026		.362
.379							.010				.379
.396	.016	-.029	-.034	-.031	-.017	-.013	-.009	-.024	-.029		.396
.408							.009				.408
.426	.078	-.001	.014	-.002	.015	.019	.022	.007	-.001		.426
.451	.093	.008	.000	.007	.024	.024	.027	.014	.008		.451
.476	.094	.011	.003	.008	.016	.020	.024	.016	.012		.476
.501	.097		.005	.008	.013	.018	.020		.015		.501
.530	.023	-.026	-.031	-.026	-.017		.015	-.018	-.027		.530
.548							.016				.548
.566							.015				.566
.584	.018	-.032	-.033	-.030	-.020	-.017	-.015	-.020	-.030		.584
.602							.013				.602
.620							.012				.620
.638	.016	-.034		-.025	-.018		.011		-.031		.638
.655							.011				.655
.673							.011				.673
.691	.015	-.035	-.034		-.017	-.014	-.011		-.033		.691
.709							.011				.709
.727							.011				.727
.745	.015	-.012		-.023	-.018		.011		-.034		.745
.763							.011				.763
.781							.011				.781
.799							.011				.799
.817	.016	-.036	-.026	-.021	-.018	-.014	-.010	-.018	-.034		.817
.834							.011				.834
.870							.011				.870
.888							.010				.888
.921	-.061						.010				.921
.924	.014	-.033			-.018		.010				.924
.947							.010				.947
.964							.011				.964
.981	.026	-.029	-.020			-.019	-.015	-.010	-.015	-.031	.981
.984							.019			-.034	.984
.998								-.010			.998
1.000		-.062									1.000

Nose rake			Base rake		
Orifice station, y,in.	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.792		.031	.422	.116
.030	1.327		.094	1.029	.388
.050	1.492		.156	1.302	.659
.070	1.530		.219	1.578	.823
.090	1.555		.281	1.689	.959
.110	1.573		.344	1.728	1.071
.130	1.591		.406	1.768	1.132
.150	1.622		.469	1.790	1.183
.170	1.615		.750	1.842	
.190	1.573		1.000	1.868	
			1.250	1.855	
			1.500	1.853	

TABLE 14. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 3.96$ - Continued(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	130	145	165	180	225	
.039										.039
.040										.040
.041										.041
.044										.044
.049										.049
.055										.055
.060	.483	.405	.394	.371	.346	.340	.333	.356		.060
.075	.147	.109	.102	.091	.083	.077	.077	.085		.075
.085	.144	.105	.097	.089	.078	.075	.075	.085	.106	.085
.096	.175									.096
.126		.086	.079	.072	.063	.061	.060	.068	.089	.126
.143										.143
.154	.015	-.004	-.009	-.011	-.018					.154
.171										.171
.189										.189
.206	.007	-.010	-.013	-.015	-.018	-.018				.206
.223										.223
.241										.241
.258	.003	-.012	-.016	-.017	-.018	-.017				.258
.275										.275
.293										.293
.310	.002	-.014	-.018	-.017	-.016	-.015				.310
.327										.327
.344										.344
.362	.000	-.015	-.017	-.016	-.014	-.013				.362
.379										.379
.396	.000	-.015	-.017	-.016	-.013	-.012				.396
.408										.408
.426	.048	.019	.037	.021	.023	.024	.023	.024	.021	.426
.451	.058	.027	.027	.027	.035	.029	.030	.029	.029	.451
.476	.062	.031	.028	.027	.028	.028	.030	.029	.033	.476
.501	.065		.028	.027	.026	.026	.029		.035	.501
.530	.007	-.011	-.014	-.014	-.011					.530
.548										.548
.566										.566
.584	-.000	-.016	-.016	-.016	-.012	-.011				.584
.602										.602
.620										.620
.638	.000	-.018			-.014	-.011				.638
.655										.655
.673										.673
.691	.000	-.018	-.018		-.011	-.009				.691
.709										.709
.727										.727
.745	.000	.002			-.013	-.010				.745
.763										.763
.781										.781
.799										.799
.817	.001	-.016	-.014	-.011	-.009	-.008				.817
.834										.834
.870										.870
.888										.888
.921	-.060									.921
.924	.000	-.014			-.010					.924
.947										.947
.964										.964
.981	.007	-.012	-.010							.981
.984										.984
.998										.998
1.000	-.061									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.647		.031	.256	.149
.030	1.192		.094	.564	.309
.050	1.367		.156	.730	.423
.070	1.389		.219	.960	.546
.090	1.409		.281	1.176	.658
.110	1.429		.344	1.346	.772
.130	1.444		.406	1.416	.902
.150	1.451		.469	1.471	1.016
.170	1.444		.550	1.496	
.190	1.377		1.000	1.567	
			1.250	1.619	
			1.500	1.661	

TABLE 14. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 3.96$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
.039							1.774			.039
.040							1.729			.040
.041							1.519			.041
.044							1.092			.044
.049							1.494			.049
.055							1.395			.055
.060	.404	.404	.411	.405	.407	.405	.401	.404		.060
.075	.109	.109	.110	.109	.112	.110	.110	.107		.075
.085	.105	.105	.106	.105	.108	.106	.107	.109	.106	.085
.096	.134									.096
.126		.087	.087	.087	.089	.088	.088	.089	.090	.126
.143							-.003			.143
.154	-.002	-.004	-.005	-.004	-.005		-.003	-.004	-.004	.154
.171							-.006			.171
.189							-.008			.189
.206	-.008	-.009	-.009	-.009	-.009	-.009	-.009	-.009	-.008	.206
.223							-.009			.223
.241							-.009			.241
.258	-.009	-.009	-.010	-.010	-.010	-.010	-.010	-.009	.001	.258
.275							-.010			.275
.293							-.010			.293
.310	-.010	-.011	-.011	-.011	-.011	-.011	-.010	-.011	-.009	.310
.327							-.010			.327
.344							-.010			.344
.362	-.010	-.011	-.010	-.010	-.010	-.010	-.010	-.009	-.009	.362
.379							-.009			.379
.396	-.009	-.009	-.010	-.009	-.009	-.009	-.008	-.009	-.009	.396
.408							-.008			.408
.426	.031	.028	.044	.030	.030	.031	.030	.030	.029	.426
.451	.038	.035	.036	.036	.042	.037	.039	.038	.035	.451
.476	.039	.037	.037	.037	.037	.039	.040	.039	.039	.476
.501	.041		.038	.037	.039	.039	.041		.040	.501
.530	-.003	-.006	-.006	-.005	-.005		-.002	-.003	-.004	.530
.548							-.004			.548
.566							-.004			.566
.584	-.006	-.009	-.009	-.008	-.008	-.008	-.005	-.005	-.006	.584
.602							-.005			.602
.620							-.006			.620
.638	-.008	-.011		-.007	-.009		-.005			.638
.655							-.006			.655
.673							-.006			.673
.691	-.008	-.011	-.011		-.010	-.009	-.006	-.007	-.007	.691
.709							-.006			.709
.727							-.006			.727
.745	-.007	.007		-.009	-.009		-.006			.745
.763							-.005			.763
.781							-.006			.781
.799							-.006			.799
.817	-.006	-.009	-.008	-.008	-.006	-.007	-.005	-.006	-.005	.817
.834							-.005			.834
.870							-.004			.870
.888							-.004			.888
.921	-.058						-.005			.921
.924	-.005	-.008		-.007			-.004			.924
.947							-.004			.947
.964							-.004			.964
.981	-.000	-.004	-.006		-.006	-.005	-.005	-.003	-.002	.981
.984					-.007		-.005		-.004	.984
.998							-.007			.998
1.000	-.058									1.000

Orifice station, y, in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.491		.031	.138	.124
.030	1.091		.094	.256	.234
.050	1.257		.156	.322	.288
.070	1.279		.219	.393	.353
.090	1.290		.281	.462	.415
.110	1.300		.344	.534	.494
.130	1.295		.406	.614	.572
.150	1.291		.469	.691	.659
.170	1.279		.750	1.018	
.190	1.214		1.000	1.194	
			1.250	1.351	
			1.500	1.465	

TABLE 14. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 3.96$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	130	145	160	180	225	
.039							1.771			.039
.040							1.744			.040
.041							1.570			.041
.044							1.169			.044
.049							1.579			.049
.055							1.460			.055
.060	.335	.402	.426	.441	.476	.477	.477	.462		.060
.075	.076	.109	.119	.127	.144	.147	.148	.136		.075
.085	.076	.106	.114	.125	.139	.142	.145	.135	.107	.085
.096	.100									.096
.126		.086	.094	.103	.117	.120	.122	.113	.091	.126
.143							.011			.143
.154	-.015	-.005	-.002	.000	.007		.012	.003	-.003	.154
.171							.008			.171
.189							.006			.189
.206	-.015	-.009	-.006	-.002	.002	.003	.003	.000	-.009	.206
.223							.002			.223
.241							.001			.241
.258	-.015	-.011	-.009	-.007	-.001	-.000	.000	-.004	-.000	.258
.275							.000			.275
.293							.000			.293
.310	-.014	-.012	-.011	-.009	-.004	-.001	-.000	-.004	-.013	.310
.327							-.000			.327
.344							-.001			.344
.362	-.013	-.013	-.010	-.008	-.004	-.002	-.001	-.004	-.013	.362
.379							-.001			.379
.396	-.011	-.013	-.012	-.009	-.004	-.002	-.001	-.005	-.013	.396
.408							-.001			.408
.426	.024	.023	.044	.031	.040	.044	.044	.036	.021	.426
.451	.032	.030	.034	.039	.056	.054	.055	.046	.030	.451
.476	.030	.032	.036	.041	.054	.058	.061	.050	.033	.476
.501	.030		.039	.044	.057	.061	.064		.036	.501
.530	-.006	-.009	-.007	-.005	.001		.005	.000	-.007	.530
.548							.001			.548
.566							-.000			.566
.584	-.009	-.013	-.012	-.010	-.004	-.003	-.001	-.005	-.011	.584
.602							-.000			.602
.620							-.000			.620
.638	-.007	-.015			-.009	-.005			-.012	.638
.655							-.000			.655
.673							-.000			.673
.691	-.007	-.014	-.012		-.005	-.003		-.005	-.012	.691
.709							-.001			.709
.727							-.000			.727
.745	-.007	.003			-.011	-.006			-.011	.745
.763							-.000			.763
.781							-.001			.781
.799							-.001			.799
.817	-.007	-.013	-.013	-.010	-.005	-.002	-.000	-.005	-.010	.817
.834							-.000			.834
.870							-.000			.870
.888							-.000			.888
.921	-.057							.000		.921
.924	-.005	-.012			.043			-.000		.924
.947								-.000		.947
.964								-.000		.964
.981	.000	-.010	-.011			-.004	-.002	-.000	-.003	.981
.984								-.000	-.010	.984
.998								-.001		.998
1.000	-.059									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.402		.031	.091	.147
.030	.993		.094	.147	.319
.050	1.172		.156	.194	.451
.070	1.161		.219	.245	.554
.090	1.169		.281	.279	.681
.110	1.170		.344	.310	.822
.130	1.170		.406	.352	.952
.150	1.154		.469	.390	1.051
.170	1.134		.750	.503	
.190	1.082		1.000	.628	
			1.250	.785	
			1.500	.874	

TABLE 14. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 3.96$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{X}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{X}{T}$
	0	90	105	120	150	165	180	225	270	
.039							1.763			.039
.040							1.781			.040
.041							1.643			.041
.044							1.269			.044
.049							.703			.049
.055							.550			.055
.060	.271	.401	.448	.484	.562	.573	.576	.530		.060
.075	.048	.110	.130	.149	.188	.196	.200	.170		.075
.085	.048	.106	.125	.147	.183	.190	.195	.169	.108	.085
.096	.069									.096
.126		.086	.101	.118	.152	.162	.165	.137	.089	.126
.143							.039			.143
.154	-.025	-.006	.001	.012	.030		.038	.022	-.005	.154
.171							.030			.171
.189							.025			.189
.206	-.022	-.013	-.006	.001	.016	.019	.022	.010	-.011	.206
.223							.021			.223
.241							.020	.007	-.005	.241
.258	-.019	-.016	-.011	-.002	.013	.018				.258
.275							.019			.275
.293							.018			.293
.310	-.015	-.020	-.014	-.005	.008	.014	.017	.004	-.020	.310
.327							.016			.327
.344							.015			.344
.362	-.013	-.024	-.017	-.008	.006	.012	.014	.000	-.024	.362
.379							.014			.379
.396	-.011	-.026	-.021	-.010	.005	.011	.014	.000	-.026	.396
.408							.013			.408
.426	.020	.003	.031	.030	.062	.072	.075	.049	.002	.426
.451	.027	.011	.024	.042	.082	.087	.094	.062	.011	.451
.476	.023	.013	.028	.046	.080	.090	.097	.068	.015	.476
.501	.022		.029	.046	.081	.093	.100		.017	.501
.530	-.011	-.023	-.015	-.007	.011		.022	.005	-.023	.530
.548							.019			.548
.566							.020			.566
.584	-.014	-.028	-.021	-.011	.007	.013	.018	.002	-.027	.584
.602							.018			.602
.620							.017			.620
.638	-.011	-.028		-.010	.004		.017		-.027	.638
.655							.016			.655
.673							.016			.673
.691	-.011	-.030	-.023		.002	.009		-.001	-.029	.691
.709							.014			.709
.727							.015			.727
.745	-.010	-.010		-.015	.002		.016		-.029	.745
.763							.016			.763
.781							.014			.781
.799							.015			.799
.817	-.010	-.030	-.025	-.015	.004	.011	.016	-.002	-.028	.817
.834							.016			.834
.870							.016			.870
.888							.016			.888
.921	-.060						.017			.921
.924	-.009	-.028		-.006			.015			.924
.947							.014			.947
.964							.014			.964
.981	-.004	-.026	-.028		.002	.010	.014	-.001	-.028	.981
.984							.014	-.028	-.028	.984
.998							.014	-.028	-.028	.998
1.000	-.062									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.305		.031	.220	.106
.030	.863		.094	.495	.405
.050	1.071		.156	.531	.637
.070	1.053		.219	.542	.785
.090	1.053		.281	.553	.939
.110	1.053		.344	.560	1.053
.130	1.047		.406	.566	1.115
.150	1.035		.469	.575	1.156
.170	1.002		.750	.606	
.190	.972		1.000	.647	
			1.250	.693	
			1.500	.707	

TABLE 14. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 3.96$ - Concluded(g) $\alpha = 10.1^\circ$

Model station, $\frac{X}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{X}{T}$
	C	90	105	120	150	165	180	225	270	
.039							1.756			.039
.040							1.805			.040
.041							1.713			.041
.044							1.403			.044
.049							1.836			.049
.055							1.684			.055
.060	.201	.402	.470	.541	.677	.697	.709	.613		.060
.075	.017	.110	.144	.180	.250	.265	.272	.214		.075
.085	.019	.107	.137	.178	.240	.257	.265	.215	.107	.085
.096	.038									.096
.126		.082	.111	.146	.213	.231	.237	.182	.082	.126
.143							.038			.143
.154	-.024	-.006	.001	.012	.030		.037	.044	-.006	.154
.171							.031			.171
.189							.025			.189
.206	-.022	-.012	-.007	.003	.016	.019	.023	.030	-.018	.206
.223							.021			.223
.241							.022			.241
.258	-.019	-.016	-.011	-.002	.014	.018	.020	.025	-.010	.258
.275							.019			.275
.293							.018			.293
.310	-.017	-.020	-.014	-.005	.009	.014	.017	.019	-.031	.310
.327							.016			.327
.344							.016			.344
.362	-.014	-.025	-.017	-.009	.007	.012	.015	.016	-.036	.362
.379							.014			.379
.396	-.011	-.026	-.020	-.010	.006	.012	.014	.014	-.038	.396
.408							.014			.408
.426	.020	.003	.032	.029	.061	.072	.076	.078	-.015	.426
.451	.027	.010	.024	.041	.081	.089	.150	.089	-.005	.451
.476	.023	.015	.027	.046	.080	.090	.159	.097	-.003	.476
.501	.022		.028	.045	.081	.093	.163		-.001	.501
.530	-.010	-.023	-.016	-.008	.012		.058	.021	-.035	.530
.548							.053			.548
.566							.052			.566
.584	-.015	-.027	-.020	-.011	.006	.013	.050	.016	-.038	.584
.602							.050			.602
.620							.049			.620
.638	-.011	-.028		-.011	.004		.049		-.040	.638
.655							.049			.655
.673							.048			.673
.691	-.010	-.030	-.024		.003	.011		.013	-.041	.691
.709							.048			.709
.727							.049			.727
.745	-.010	-.010		-.015	.003		.049		-.043	.745
.763							.049			.763
.781							.047			.781
.799							.047			.799
.817	-.009	-.030	-.025	-.015	.003	.011	.049	.012	-.042	.817
.834							.047			.834
.870							.048			.870
.888							.046			.888
.921	-.060						.047			.921
.924	-.009	-.029		-.006						.924
.947							.044			.947
.964							.044			.964
.981	-.004	-.025	-.028		.003	.011	.044	.010	-.041	.981
.984							.044		-.041	.984
.998							.045			.998
1.000	-.061									1.000

Nose rake			Base rake		
Orifice station, y, in.	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.137		.031	.284	.036
.030	.628		.094	.640	.347
.050	.947		.156	.601	.659
.070	.941		.219	.599	.813
.090	.934		.281	.599	.934
.110	.934		.344	.593	1.024
.130	.925		.406	.590	1.081
.150	.916		.469	.582	1.106
.170	.899		.750	.555	
.190	.849		1.000	.573	
			1.250	.639	
			1.500	.713	

TABLE 15. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 4.65$ (a) $\alpha = -6.3^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.039							1.804			.039
.040							.661			.040
.041							.398			.041
.044							.923			.044
.049							.356			.049
.055							.270			.055
.060	.577	.418	.386	.346	.296	.283	.275	.306		.060
.075	.204	.124	.111	.090	.070	.061	.058	.070		.075
.085	.197	.119	.103	.087	.066	.058	.057	.068	.107	.085
.096	.230									.096
.126		.092	.079	.064	.049	.045	.043	.052	.084	.126
.143							.008			.143
.154	.047	.007	.001	-.003	-.009		-.009	-.006	.004	.154
.171							.008			.171
.189							.008			.189
.206	.033	-.001	-.006	-.007	-.008	-.008	-.008	-.008	-.002	.206
.223							.007			.223
.241							.007			.241
.258	.028	-.006	-.007	-.013	-.009	-.009	-.008	-.010	.012	.258
.275							.006			.275
.293							.005			.293
.310	.023	-.009	-.012	-.011	-.011	-.009	-.007	-.011	-.012	.310
.327							.006			.327
.344							.006			.344
.362	.020	-.012	-.013	-.013	-.011	-.011	-.006	-.012	-.014	.362
.379							.005			.379
.396	.022	-.011	-.014	-.011	-.009	-.009	-.004	-.011	-.015	.396
.408							.003			.408
.426	.065	.009	.038	-.003	.000	.004	.018	.002	.001	.426
.451	.082	.014	.008	.003	.022	.007	.024	.008	.008	.451
.476	.088	.018	.010	.005	.008	.012	.021	.011	.011	.476
.501	.092		.010	.007	.011	.012	.019		.011	.501
.530	.033	-.009	-.011	-.011	-.008		.007	-.009	-.014	.530
.548							.007			.548
.566							.006			.566
.584	.024	-.014	-.015	-.013	-.009	-.010	-.007	-.010	-.017	.584
.602							.006			.602
.620							.006			.620
.638	.021	-.014		-.008	-.011		.005		-.017	.638
.655							.005			.655
.673							.006			.673
.691	.021	-.014	-.015		-.011	-.009		-.011	-.018	.691
.709							.005			.709
.727							.005			.727
.745	.020	.017		-.013	-.011		.005		.019	.745
.763							.005			.763
.781							.005			.781
.799							.005			.799
.817	.021	-.017	-.014	-.011	-.009	-.008	-.005	-.009	-.019	.817
.834							.005			.834
.870							.005			.870
.888							.005			.888
.921	-.034	.021	-.017		-.009			.004		.921
.924							.005			.924
.947							.003			.947
.964							.005			.964
.981	.026	-.016	-.014			-.011	-.005	-.011	-.016	.981
.984							.003		-.017	.984
.998							.005		-.008	.998
1.000	-.035									1.000

Nose rake			Base rake		
Orifice station, $y/in.$	Cp top	Cp side	Orifice station, $y/in.$	Cp top	Cp side
.010	.828		.031	.365	.061
.030	1.364		.094	.865	.317
.050	1.467		.156	1.156	.508
.070	1.504		.219	1.451	.614
.090	1.541		.281	1.649	.762
.110	1.569		.344	1.709	.849
.130	1.601		.406	1.765	.946
.150	1.615		.469	1.797	.990
.170	1.610		.750	1.878	
.190	1.587		1.000	1.871	
			1.250	1.910	
			1.500	1.894	

TABLE 15. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 4.65$ - Continued

(b) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.834			.039
.040							1.720			.040
.041							1.492			.041
.044							1.046			.044
.049							1.448			.049
.055							1.310			.055
.060	.482	.420	.407	.384	.358	.349	.341	.354		.060
.075	.154	.126	.119	.106	.097	.091	.088	.094		.075
.085	.150	.119	.111	.102	.091	.086	.085	.089	.106	.085
.096	.176									.096
.126		.094	.087	.081	.070	.068	.066	.069	.086	.126
.143							.003			.143
.154	.024	.007	.004	.002	-.002		-.003	-.001	.005	.154
.171							-.004			.171
.189							-.004			.189
.206	.014	.001	-.002	-.003	-.005	-.005	-.005	-.004	.000	.206
.223							-.005			.223
.241							-.005			.241
.258	.009	-.002	-.004	-.006	-.007	-.007	-.006	-.006	.018	.258
.275							-.005			.275
.293							-.004			.293
.310	.007	-.006	-.007	-.008	-.007	-.007	-.006	-.007	-.006	.310
.327							-.005			.327
.344							-.004			.344
.362	.006	-.007	-.006	-.005	-.003	-.003	-.002	-.004	-.007	.362
.379							-.001			.379
.396	.007	-.002	-.003	-.002	-.002	-.002	-.000	-.001	-.004	.396
.408							0.000			.408
.426	.037	.009	.042	.001	.001	.002	.005	.003	.012	.426
.451	.049	.026	.019	.012	.024	.006	.013	.011	.022	.451
.476	.052	.028	.022	.018	.013	.012	.017	.018	.024	.476
.501	.058		.024	.019	.016	.016	.020		.026	.501
.530	.014	-.004	-.009	-.008	-.007		-.004	-.006	-.006	.530
.548							-.005			.548
.566							-.004			.566
.584	.006	-.007	-.009	-.010	-.007	-.007	-.003	-.006	-.010	.584
.602							-.003			.602
.620							-.003			.620
.638	.005	-.010		-.004	-.007		-.002		-.010	.638
.655							-.002			.655
.673							-.002			.673
.691	.006	-.010	-.012		-.007	-.005		-.007	-.011	.691
.709							-.002			.709
.727							-.002			.727
.745	.006	.023		-.009	-.007		-.002		-.011	.745
.763							-.002			.763
.781							-.002			.781
.799							-.002			.799
.817	.006	-.011	-.010	-.009	-.005	-.004	-.002	-.005	-.010	.817
.834							-.002			.834
.870							-.002			.870
.888							-.002			.888
.921	-.035									.921
.924	.006	-.011			-.007					.924
.947										.947
.964										.964
.981	.008	-.008	-.008			-.006	-.005		-.004	.981
.984							-.002		-.010	.984
.998							-.006			.998
1.000	-.036									1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	.527		.031	.206	.111
.030	1.148		.094	.428	.259
.050	1.370		.156	.599	.370
.070	1.379		.219	.774	.444
.090	1.404		.281	.970	.548
.110	1.400		.344	1.143	.663
.130	1.404		.406	1.266	.781
.150	1.407		.469	1.344	.899
.170	1.397		.5750	1.370	
.190	1.351		1.000	1.483	
			1.250	1.561	
			1.500	1.603	

TABLE 15. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 4.65$ - Continued

(c) $\alpha = 0^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	130	145	165	180	225	
.039								1.839		.039
.040								1.786		.040
.041								1.553		.041
.044								1.138		.044
.049								.515		.049
.055								.404		.055
.060	.408	.420	.426	.420	.417	.412	.408	.402		.060
.075	.119	.125	.129	.124	.125	.123	.121	.116		.075
.085	.114	.120	.120	.120	.119	.117	.116	.114	.109	.085
.096	.138									.096
.126		.095	.096	.096	.095	.094	.092	.089	.088	.126
.143								.008		.143
.154	.010	.009	.008	.009	.007		.006	.006	.005	.154
.171								.004		.171
.189								.001		.189
.206	.003	.002	.001	.003	.002	.000	.000	.001	.000	.206
.223								-.000		.223
.241								-.001		.241
.258	.000	-.0	.000	-.001	-.001	-.002	-.002	-.001	.019	.258
.275								-.002		.275
.293								-.003		.293
.310	-.001	-.003	-.003	-.003	-.004	-.004	-.003	-.003	-.003	.310
.327								-.004		.327
.344								-.004		.344
.362	-.001	-.005	-.003	-.005	-.005	-.005	-.004	-.004	-.005	.362
.379								-.004		.379
.396	.001	-.002	-.000	.001	.001	-.000	-.001	-.002	-.003	.396
.408								-.000		.408
.426	.026	.025	.050	.011	.009	.010	.019	.013	.016	.426
.451	.032	.028	.030	.027	.038	.026	.031	.029	.027	.451
.476	.032	.031	.032	.031	.031	.030	.033	.032	.029	.476
.501	.038		.033	.032	.031	.032	.033	.030	.030	.501
.530	.007	-.0	-.000	-.001	-.000		.000	-.001	-.003	.530
.548								-.000		.548
.566								-.000		.566
.584	-.000	-.003	-.003	-.003	-.002	-.003	-.000	-.002	-.003	.584
.602								-.001		.602
.620								-.001		.620
.638	-.001	-.004			.000	-.004			-.003	.638
.655								-.000		.655
.673								-.002		.673
.691	-.001	-.004	-.005		-.006	-.006		-.003	-.005	.691
.709								-.004		.709
.727								-.003		.727
.745	-.000	.028			-.005	-.006			-.005	.745
.763								-.002		.763
.781								-.002		.781
.799								-.002		.799
.817	-.000	-.005	-.005	-.005	-.003	-.003		-.001	-.003	.817
.834								-.001		.834
.870								-.001		.870
.888								-.000		.888
.921	-.034	.000	-.004					-.001		.921
.924								-.002		.924
.947								-.001		.947
.964								-.001		.964
.981	.001	-.002	-.003					-.001	-.002	.981
.984								-.001	-.003	.984
.998								-.004		.998
1.000	-.034									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.386		.031	.100	.107
.030	1.041		.094	.192	.211
.050	1.248		.156	.252	.280
.070	1.271		.219	.309	.358
.090	1.260		.281	.349	.411
.110	1.269		.344	.397	.485
.130	1.262		.406	.448	.556
.150	1.251		.469	.526	.660
.170	1.234		.750	.789	
.190	1.198		1.000	1.013	
			1.250	1.158	
			1.500	1.320	

TABLE 15. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 4.65$ - Continued

(d) $\alpha = 3.0^\circ$

Model station, $\frac{x}{L}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{L}$
	0	90	105	120	150	165	180	225	270	
.039							1.840			.039
.040							1.801			.040
.041							1.614			.041
.044							1.220			.044
.049							.605			.049
.055							.466			.055
.060	.340	.421	.443	.456	.483	.486	.482	.452		.060
.075	.089	.124	.135	.140	.158	.159	.158	.141	.108	.075
.085	.085	.119	.128	.136	.150	.153	.154	.137		.085
.096	.103									.096
.126		.092	.100	.106	.118	.122	.121	.110	.087	.126
.143								.022		.143
.154	.001	.007	.011	.015	.019		.021	.013	.004	.154
.171							.016			.171
.189							.015			.189
.206	-.002	.003	.004	.008	.011	.011	.013	.007	-.000	.206
.223							.010			.223
.241							.007			.241
.258	-.004	-.002	.000	.001	.004	.006	.006	.002	.017	.258
.275							.006			.275
.293							.005			.293
.310	-.005	-.004	-.003	-.000	.002	.003	.004	-.000	-.006	.310
.327							.004			.327
.344							.004			.344
.362	-.004	-.007	-.004	-.002	.001	.003	.003	-.001	-.007	.362
.379							.003			.379
.396	.001	-.003	-.004	-.002	.001	.002	.003	-.001	-.005	.396
.408							.003			.408
.426	.014	.014	.052	.026	.035	.036	.037	.027	.009	.426
.451	.021	.026	.031	.038	.056	.047	.047	.035	.021	.451
.476	.021	.029	.035	.040	.049	.051	.053	.041	.024	.476
.501	.027		.036	.041	.051	.055	.056		.025	.501
.530	.003	-.003	-.000	.003	.007		.010	.003	-.006	.530
.548							.004			.548
.566							.004			.566
.584	-.003	-.006	-.004	-.002	.001	.001	.003	-.003	-.009	.584
.602							.003			.602
.620							.003			.620
.638	-.003	-.008		.001	.001		.003		-.009	.638
.655							.003			.655
.673							.003			.673
.691	-.003	-.008	-.006		-.000	.001		-.005	-.009	.691
.709							.002			.709
.727							.002			.727
.745	-.003	.024		-.004	-.000		.002		-.008	.745
.763							.002			.763
.781							.002			.781
.799							.002			.799
.817	-.002	-.009	-.006	-.005	.000	.001	.002	-.003	-.008	.817
.834							.001			.834
.870							.001			.870
.888							.001			.888
.921	-.033						.001			.921
.924	-.001	-.008		-.004			.000			.924
.947							.000			.947
.964							.001			.964
.981	.001	-.006	-.007		-.000	.001	.001	-.002	-.006	.981
.984							.000		-.008	.984
.998							.000			.998
1.000	-.035									1.000

Nose rake			Base rake		
Orifice station, y, in.	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.249		.031	.065	.109
.030	.883		.094	.100	.277
.050	1.139		.156	.139	.386
.070	1.133		.219	.187	.473
.090	1.133		.281	.208	.589
.110	1.128		.344	.226	.724
.130	1.128		.406	.252	.844
.150	1.116		.469	.286	.948
.170	1.100		.750	.362	
.190	1.052		1.000	.471	
			1.250	.577	
			1.500	.674	

TABLE 15. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE III WITH FIXED TRANSITION AT $M = 4.65$ - Concluded(e) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.039							1.831			.039
.040							1.833			.040
.041							1.824			.041
.044							1.700			.044
.049							1.317			.049
.055							.713			.055
.060	.275	.418	.461	.497	.567	.575	.575	.513		.060
.075	.061	.124	.146	.164	.200	.208	.206	.173		.075
.085	.058	.120	.137	.160	.191	.198	.202	.169	.109	.085
.096	.075									.096
.126		.092	.106	.123	.153	.161	.164	.132	.086	.126
.143							.047			.143
.154	-.008	.007	.014	.026	.040		.046	.028	.003	.154
.171							.038			.171
.189							.032			.189
.206	-.008	.0	.004	.013	.024	.027	.028	.016	-.003	.206
.223							.026			.223
.241							.025			.241
.258	-.007	-.005	-.000	.005	.019	.022	.024	.010	.013	.258
.275							.023			.275
.293							.021			.293
.310	-.007	-.010	-.004	.002	.015	.019	.020	.006	-.013	.310
.327							.019			.327
.344							.018			.344
.362	-.007	.042	-.006	-.000	.012	.016	.017	.003	-.015	.362
.379							.016			.379
.396	-.003	-.011	-.009	-.002	.010	.015	.017	.001	-.015	.396
.408							.017			.408
.426	.017	.005	.049	.029	.051	.060	.063	.037	.002	.426
.451	.022	.014	.027	.042	.082	.081	.083	.051	.009	.451
.476	.019	.017	.031	.048	.078	.086	.088	.059	.013	.476
.501	.023		.033	.048	.078	.088	.090	.015	.015	.501
.530	-.000	-.010	-.005	.002	.018		.026	.005	-.013	.530
.548							.023			.548
.566							.023			.566
.584	-.007	-.014	-.009	-.002	.013	.018	.021	.002	-.015	.584
.602							.020			.602
.620							.019			.620
.638	-.006	-.015		.001	.011		.019		-.017	.638
.655							.018			.655
.673							.017			.673
.691	-.006	-.014	-.011		.010	.014		-.000	-.017	.691
.709							.017			.709
.727							.016			.727
.745	-.006	.018		-.005	.010		.016		-.018	.745
.763							.016			.763
.781							.016			.781
.799							.016			.799
.817	-.005	-.017	-.012	-.006	.010	.015	.018	-.002	-.018	.817
.834							.017			.834
.870							.016			.870
.888							.017			.888
.921	-.036						.016			.921
.924	-.005	-.017		-.005			.016			.924
.947							.014			.947
.964							.014			.964
.981	-.004	-.016	-.013		.008	.013	.015	-.002	-.015	.981
.998					.008		.015		-.015	.998
1.000	-.036									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.141		.031	.120	.369
.030	.651		.094	.277	.549
.050	1.029		.156	.346	.676
.070	1.029		.219	.381	.814
.090	1.022		.281	.388	.902
.110	1.015		.344	.399	.948
.130	1.003		.406	.413	1.001
.150	.990		.469	.423	1.005
.170	.967		.750	.448	
.190	.925		1.000	.473	
			1.250	.521	
			1.500	.558	

TABLE 16. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 1.57$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{d}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{d}$
	0	90	105	120	150	165	180	225	270	
.042							1.517			.042
.043							1.430			.043
.044							1.284			.044
.046							.994			.046
.049							.309			.049
.053							.534			.053
.060	-0.139		-0.332	-0.362	-0.381	-0.383	-0.381	-0.370		.060
.075	.186		-0.184	-0.249	-0.295	-0.303	-0.312	-0.282		.075
.085	.357	.050	-0.023	-0.100	-0.191	-0.208	-0.231	-0.174	-0.000	.085
.126		.106	.056	.021	-0.023	-0.024	-0.025	-0.009	.089	.126
.143							-0.234			.143
.154	-0.011	-0.171	-0.200	-0.221	-0.219	-0.200	-0.181	-0.240	-0.186	.154
.171							-0.143			.171
.189							-0.106			.189
.206	.022	-0.161	-0.186	-0.192	-0.112	-0.086	-0.078	-0.140	-0.163	.206
.223							-0.054			.223
.241							-0.041			.241
.258	.020	-0.166	-0.177	-0.128	-0.061	-0.037	-0.029	-0.080	-0.171	.258
.275							-0.022			.275
.293							-0.016			.293
.310	.018	-0.159	-0.129	-0.082	-0.046	-0.019	-0.007	-0.055	-0.157	.310
.327							-0.000			.327
.344							.000			.344
.362	.011	-0.129	-0.095	-0.055	-0.049	-0.015	.001	-0.045	-0.122	.362
.379							.000			.379
.396	.008	-0.111	-0.084	-0.043	-0.051	-0.013	.010	-0.037	-0.105	.396
.408							.004			.408
.426	.218	.080	.095	.083	.060	.171	.215	.086	.093	.426
.451	.205	.072	.060	.075	.052	.114	.144	.085	.070	.451
.476	.200	.048	.044	.063	.032	.095	.105	.072	.054	.476
.501	.194	.041	.036	.044	.014	.075	.084	.044	.044	.501
.530	-0.006	-0.129	-0.137	-0.114	-0.102	-0.131	-0.107	-0.081	-0.128	.530
.548							-0.123			.548
.566							-0.065			.566
.584	.026	-0.117	-0.105	-0.085	-0.088	-0.067	-0.056	-0.076	-0.118	.584
.602							-0.049			.602
.620							-0.039			.620
.638	.029	-0.103		-0.077	-0.049					.638
.655							-0.027			.655
.673							-0.023			.673
.691	.029	-0.116	-0.104	-0.067	-0.042	-0.043	-0.021	-0.037	-0.101	.691
.709							-0.018			.709
.727							-0.020			.727
.745	.034	-0.112		-0.050	-0.034		-0.019		-0.092	.745
.763							-0.018			.763
.781							-0.027			.781
.799							-0.020			.799
.817	.039	-0.091	-0.060	-0.037	-0.030	-0.027	-0.010	-0.027	-0.072	.817
.834							-0.016			.834
.852							-0.009			.852
.870							-0.009			.870
.888							-0.008			.888
.906							-0.002			.906
.921										.921
.924										.924
.947										.947
.964										.964
.981	.061	-0.042	.004		.009	.008	.035	.010	-0.045	.981
1.000		-0.184								1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.238	.386	.031	.876	.426
.030	.467	.883	.094	.563	.021
.050	.532	.038	.156	.638	.487
.070	.550	.090	.219	.574	.613
.090	.562	.109	.281	.579	.542
.110	.568	.133	.344	.568	.483
.130	.561	.146	.406	.570	.495
.150	.555	.159	.469	.575	.485
.170	.537	.157	.750	.566	
.190	.451	.107	.000	.569	
				.250	.562
				.500	.561

TABLE 16. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{y}{L}$	Cp for meridian angle, θ , deg -									Model station, $\frac{y}{L}$
	0	90	105	120	150	165	180	225	270	
.042							1.542			.042
.043							1.476			.043
.044							1.354			.044
.046							1.084			.046
.049							.559			.049
.053							.572			.053
.060	-.202		-.308	-.333	-.359	-.365	-.367	-.333		.060
.075	.045		-.172	-.221	-.258	-.263	-.260	-.252		.075
.085	.206	.030	-.013	-.069	-.150	-.143	-.156	-.122	.003	.085
.096	.419									.096
.126		.125	.091	.071	.033	.027	.025	.047	.106	.126
.143							.219			.143
.154	-.066	-.151	-.163	-.176	-.184	-.181	-.170	-.188	-.158	.154
.171							.142			.171
.189							.110			.189
.206	-.025	-.114	-.119	-.126	-.100	-.090	-.086	-.109	-.115	.206
.223							.066			.223
.241							.052			.241
.258	-.019	-.092	-.091	-.083	-.053	-.042	-.040	-.061	-.089	.258
.275							.030			.275
.293							.019			.293
.310	-.024	-.074	-.069	-.050	-.024	-.013	-.012	-.030	-.074	.310
.327							.010			.327
.344							.009			.344
.362	-.010	-.057	-.046	-.034	-.020	-.011	-.007	-.024	-.055	.362
.379							.007			.379
.396	-.014	-.048	-.046	-.034	-.013	-.007	.001	-.019	-.047	.396
.408							.032			.408
.426	.193	.139	.144	.141	.137	.155	.169	.135	.142	.426
.451	.179	.124	.117	.117	.116	.121	.132	.118	.118	.451
.476	.169	.103	.094	.092	.094	.099	.105	.097	.104	.476
.501	.161	.091	.081	.074	.064	.076	.086	.069	.092	.501
.530	-.034	-.092	-.101	-.104	-.082	-.089	-.103	-.081	-.093	.530
.548							.076			.548
.566							.057			.566
.584	-.001	-.072	-.071	-.067	-.061	-.057	-.050	-.060	-.071	.584
.602							.037			.602
.620							.037			.620
.638	-.000	-.052		-.044	-.038				-.055	.638
.655							.022			.655
.673							.021			.673
.691	-.000	-.056	-.054	-.046	-.032	-.030	-.014	-.025	-.048	.691
.709							.012			.709
.727							.009			.727
.745	-.000	-.054		-.036	-.018		.003		-.040	.745
.763							.006			.763
.781							.009			.781
.799							.007			.799
.817	.016	-.039	-.032	-.021	-.014	-.008	.000	-.011	-.024	.817
.834							.003			.834
.852							.005			.852
.870							.006			.870
.888							.009			.888
.906							.005			.906
.921	-.119									.921
.924	.027		-.028		-.017	-.003			-.037	.924
.947										.947
.964										.964
.981	.040		-.011	-.009		-.001	.001		-.006	.981
1.000		-.127							-.017	1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.117	.559	.031	.740	.513
.030	1.420	.954	.094	1.252	.907
.050	1.519	1.059	.156	1.543	1.239
.070	1.545	1.099	.219	1.548	1.480
.090	1.559	1.117	.281	1.533	1.535
.110	1.556	1.136	.344	1.541	1.496
.130	1.548	1.152	.406	1.545	1.494
.150	1.534	1.160	.469	1.546	1.504
.170	1.509	1.152	.750	1.546	
.190	1.414	1.087	1.000	1.550	
			1.250	1.548	
			1.500	1.539	

TABLE 16. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 1.57$ - Continued(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	130	145	180	225	270	
.042							1.552			.042
.043							1.511			.043
.044							1.403			.044
.046							1.158			.046
.049							.725			.049
.053							.617			.053
.060	-.246		-.289	-.297	-.310	-.334	-.340	-.312		.060
.075	-.050		-.147	-.182	-.212	-.219	-.199	-.215		.075
.085	.098	.017	-.001	-.021	-.078	-.072	-.094	-.060	.003	.085
.096	.421									.096
.126		.135	.119	.107	.081	.076	.076	.088	.116	.126
.143							.199			.143
.154	-.104	-.131	-.141	-.150	-.160	-.162	-.155	-.161	-.142	.154
.171							.133			.171
.189							.107			.189
.206	-.058	-.086	-.087	-.097	-.087	-.085	-.085	-.089	-.088	.206
.223							.063			.223
.241							.053			.241
.258	-.040	-.055	-.057	-.055	-.048	-.043	-.045	-.047	-.053	.258
.293							.033			.293
.310	-.029	-.037	-.038	-.034	-.029	-.023	-.023	-.028	-.039	.310
.327							.015			.327
.344							.011			.344
.362	-.018	-.028	-.024	-.017	-.013	-.011	-.009	-.014	-.026	.362
.379							.011			.379
.396	-.017	-.020	-.019	-.018	-.012	-.011	-.004	-.013	-.020	.396
.408							.013			.408
.426	.182	.168	.166	.162	.160	.164	.162	.165	.165	.426
.451	.163	.147	.143	.141	.140	.139	.142	.137	.140	.451
.476	.146	.126	.120	.119	.117	.116	.115	.124	.124	.476
.501	.134	.111	.110	.100	.094	.093	.095	.102	.112	.501
.530	-.055	-.072	-.079	-.084	-.079	-.081	-.083	-.076	-.073	.530
.548							.065			.548
.566							.053			.566
.584	-.017	-.048	-.049	-.051	-.051	-.052	-.048	-.045	-.049	.584
.602							.041			.602
.620							.030			.620
.638	-.017	-.028		-.031	-.021				-.030	.638
.655							.025			.655
.673							.023			.673
.691	-.012	-.030	-.028	-.030	-.029	-.025	-.017	-.020	-.024	.691
.709							.016			.709
.727							.012			.727
.745	-.010	-.027		-.023	-.015		-.009		-.015	.745
.763							.003			.763
.781							.013			.781
.799							.002			.799
.817	.006	-.015	-.012	-.011	-.007	-.004	-.002	-.001	.000	.817
.834							.005			.834
.852							.001			.852
.870							.003			.870
.888							.007			.888
.906							.003			.906
.921	-.082									.921
.924	.017	-.003								.924
.947										.947
.964										.964
.981	.036		.017	.008			.007	.007		.981
1.000	-.086									1.000

Nose rake			Base rake		
Orifice station, y,in.	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.963	.569	.031	.616	.510
.030	1.360	.955	.094	.955	.754
.050	1.506	1.060	.156	1.222	.962
.070	1.545	1.093	.219	1.413	1.162
.090	1.560	1.114	.281	1.478	1.329
.110	1.537	1.133	.344	1.492	1.428
.130	1.524	1.144	.406	1.491	1.467
.150	1.503	1.151	.469	1.500	1.481
.170	1.470	1.144	.750	1.515	
.190	1.375	1.071	1.000	1.538	
			1.250	1.563	
			1.500	1.573	

TABLE 16. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 1.57$ - Continued(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.556			.042
.043							1.532			.043
.044							1.444			.044
.046							1.229			.046
.049							.824			.049
.053							.665			.053
.060	-.275		-.277	-.295	-.288	-.307	-.310	-.275		.060
.075	-.135		-.111	-.138	-.143	-.148	-.129	-.151		.075
.085	.000	.007	.012	.009	.000	.010	-.012	-.008	.003	.085
.126		.138	.138	.136	.129	.128	.129	.127	.115	.126
.143							.172			.143
.154	-.135	-.127	-.129	-.131	-.136	-.136	-.131	-.142	-.136	.154
.171							.115			.171
.189							.094			.189
.206	-.077	-.076	-.074	-.081	-.076	-.076	-.077	-.077	-.079	.206
.223							.060			.223
.241							.052			.241
.258	-.047	-.043	-.045	-.046	-.044	-.041	-.045	-.043	-.042	.258
.275							.045			.275
.293							.042			.293
.310	-.029	-.025	-.028	-.027	-.031	-.028	-.028	-.029	-.028	.310
.327							.022			.327
.344							.016			.344
.362	-.016	-.019	-.017	-.013	-.014	-.013	-.013	-.014	-.017	.362
.379							.015			.379
.396	-.016	-.010	-.012	-.013	-.014	-.014	-.008	-.015	-.010	.396
.408							.013			.408
.426	.171	.175	.175	.174	.173	.175	.172	.173	.171	.426
.451	.147	.154	.154	.154	.154	.152	.155	.148	.150	.451
.476	.128	.134	.132	.135	.132	.130	.129	.136	.135	.476
.501	.112	.119	.121	.114	.114	.112	.113	.114	.121	.501
.530	-.070	-.069	-.068	-.072	-.069	-.071	-.072	-.074	-.066	.530
.548							.056			.548
.566							.044			.566
.584	-.041	-.040	-.039	-.041	-.044	-.046	-.041	-.037	-.041	.584
.602							.034			.602
.620							.026			.620
.638	-.026	-.025		-.018	-.020				-.026	.638
.655							.021			.655
.673							.022			.673
.691	-.017	-.023	-.021	-.021	-.030	-.027	-.018	-.019	-.017	.691
.709							.018			.709
.727							.014			.727
.745	-.003	-.018		-.016	-.011		.006		-.007	.745
.763							.010			.763
.781							.014			.781
.799							.009			.799
.817	.002	-.006	-.005	-.007	-.008	-.005	-.002	-.003	.006	.817
.834							.006			.834
.852							.000			.852
.870							.002			.870
.888							.006			.888
.906							.005			.906
.921	-.074									.921
.924	-.005	.004		-.005	.001		.007		-.003	.924
.947							.006			.947
.964							.010			.964
.981	.051	.030	.014		.009	.007	.005	.007	.006	.981
1.000	-.076						.500	.500		1.000

Nose rake			Base rake		
Orifice station, y,in.	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.817	.565	.031	.464	.497
.030	1.311	.957	.094	.660	.686
.050	1.491	1.059	.156	.810	.838
.070	1.536	1.097	.219	.954	.993
.090	1.541	1.112	.281	1.074	1.135
.110	1.495	1.130	.344	1.208	1.261
.130	1.471	1.161	.406	1.313	1.358
.150	1.437	1.150	.469	1.390	1.422
.170	1.413	1.142	.750	1.463	
.190	1.323	1.060	1.000	1.479	
			1.250	1.487	
			1.500	1.500	

TABLE 16. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
*042							1.555			*042
*043							1.543			*043
*044							1.484			*044
*046							1.292			*046
*049							.915			*049
*053							.713			*053
*060	-.298						.268	-.266		*060
*075	-.221						.091			*075
*085	-.083	.031					.056	.000		*085
*096	.047									*096
*126		.136	.149	.161	.178	.182	.187	.163	.113	*126
*143							.140			*143
*154	-.159	-.132	-.125	-.118	-.108	-.105	-.100	-.123	-.142	*154
*171							.087			*171
*189							.071			*189
*206	-.091	-.089	-.075	-.078	-.062	-.058	-.059	-.071	-.088	*206
*223							.045			*223
*241							.042			*241
*258	-.044	-.057	-.053	-.047	-.038	-.034	-.038	-.045	-.057	*258
*275							.038			*275
*293							.034			*293
*310	-.022	-.035	-.038	-.033	-.033	-.029	-.028	-.036	-.037	*310
*327							.025			*327
*344							.023			*344
*362	-.011	-.029	-.027	-.021	-.021	-.015	-.014	-.023	-.027	*362
*379							.015			*379
*396	-.010	-.018	-.020	-.019	-.014	-.013	-.006	-.020	-.020	*396
*408							.016			*408
*426		.161	.166	.171	.173	.177	.184	.174	.164	*426
*451		.137	.144	.148	.155	.164	.162	.153	.140	*451
*476		.116	.123	.126	.136	.154	.155	.156	.145	*476
*501		.096	.113	.122	.121	.132	.133	.137	.128	*501
*530	-.074	-.072	-.069	-.069	-.057	-.056	-.056	-.064	-.078	*530
*548							.041			*548
*566							.029			*566
*584	-.045	-.046	-.042	-.039	-.037	-.033	-.028	-.036	-.049	*584
*602							.023			*602
*620							.012			*620
*638	-.028	-.031			.018	-.012				*638
*655							.014			*655
*673							.015			*673
*691	-.012	-.033	-.030	-.027	-.029	-.024	-.011	-.017	-.022	*691
*709							.013			*709
*727							.009			*727
*745	-.004	-.026			.022	-.009				*745
*763							.001			*763
*781							.006			*781
*799							.013			*799
*817		.003	-.011	-.012	-.012	-.008	-.002			*817
*834							.001			*834
*852							.003			*852
*870							.004			*870
*888							.006			*888
*906							.009			*906
*921							.007			*921
*924										*924
*947										*947
*964										*964
*981										*981
1.000	-.095	.020	.002				.013	.011		1.000
									.006	
									.004	

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.667	.588	.031	.452	.508
.030	1.219	.968	.094	.604	.743
.050	1.456	1.063	.156	.706	.942
.070	1.518	1.100	.219	.787	1.148
.090	1.507	1.118	.281	.853	1.313
.110	1.433	1.133	.344	.931	1.411
.130	1.404	1.147	.406	1.004	1.462
.150	1.376	1.149	.469	1.068	1.475
.170	1.351	1.143	.570	1.342	
.190	1.269	1.043	1.000	1.453	
			1.250	1.469	
			1.500	1.479	

TABLE 16. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.542			.042
.043							1.556			.043
.044							1.513			.044
.046							1.359			.046
.049							1.000			.049
.053							.767			.053
.060	-.361		-.265	-.244	-.211	-.199	-.212	-.225		.060
.075	-.254		-.091	-.040	.026	.074	.054	-.024		.075
.085	-.153	.025	.088	.102	.194	.201	.188	.133	.001	.085
.096	-.045									.096
.126		.125	.154	.185	.233	.244	.253	.201	.104	.126
.143										.143
.154	-.177	-.145	-.127	-.108	-.078	-.067	-.061	-.102	-.155	.154
.171										.171
.189										.189
.206	-.082	-.114	-.096	-.084	-.042	-.031	-.028	-.064	-.115	.206
.223										.223
.241										.241
.258	-.037	-.090	-.084	-.065	-.029	-.018	-.019	-.050	-.091	.258
.275										.275
.293										.293
.310	-.015	-.071	-.071	-.056	-.029	-.017	-.013	-.049	-.072	.310
.327										.327
.344										.344
.362	-.008	-.059	-.059	-.049	-.030	-.020	-.016	-.043	-.055	.362
.379										.379
.396	-.007	-.049	-.052	-.044	-.017	-.008	-.009	-.036	-.048	.396
.408										.408
.426	.166	.139	.145	.153	.178	.195	.193	.162	.140	.426
.451	.134	.120	.126	.138	.167	.173	.180	.146	.118	.451
.476	.108	.101	.109	.127	.161	.171	.175	.144	.103	.476
.501	.096	.090	.106	.120	.153	.160	.167	.144	.088	.501
.530	-.102	-.088	-.082	-.074	-.046	-.039	-.036	-.058	-.093	.530
.548										.548
.566										.566
.584	-.048	-.067	-.059	-.047	-.020	-.014	-.007	-.032	-.072	.584
.602										.602
.620										.620
.638	-.029	-.058		-.032	.001					.638
.655										.655
.673										.673
.691	-.014	-.062	-.054	-.042	-.025	-.012	-.001	-.024	-.048	.691
.709										.709
.727										.727
.745	-.002	-.053		-.041	-.010					.745
.763										.763
.781										.781
.799										.799
.817	.007	-.035	-.036	-.031	-.009	.003	.010	-.016	-.024	.817
.834										.834
.852										.852
.870										.870
.888										.888
.906										.906
.921	-.134									.921
.924	.062	-.020								.924
.947										.947
.964										.964
.981	.057		.019	-.011			.014	.017	.018	.981
1.000	-.145									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.453	.629	.031	.681	.523
.030	1.127	.986	.094	1.074	.882
.050	1.442	1.064	.156	1.280	1.202
.070	1.497	1.100	.219	1.348	1.465
.090	1.462	1.116	.281	1.362	1.549
.110	1.353	1.138	.344	1.378	1.523
.130	1.318	1.151	.406	1.388	1.518
.150	1.293	1.157	.469	1.402	1.523
.170	1.278	1.150	.750	1.417	
.190	1.197	1.055	1.000	1.431	
			1.250	1.434	
			1.500	1.438	

TABLE 16. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 1.57$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270		
.042							1.520				.042
.043							1.560				.043
.044							1.542				.044
.046							1.420				.046
.049							1.094				.049
.053							.832				.053
.060	-.369			-.257	-.218	-.171	-.145	-.137	-.191		.060
.075	-.308			-.045	.032	.149	.207	.193	.099		.075
.085	-.237	.051		.123	.210	.310	.333	.326	.233	-.005	.085
.096	-.122										.096
.126		.098		.152	.209	.303	.325	.339	.239	.078	.126
.143											.143
.154	-.182	-.171	-.137	-.098	-.036	-.016					.154
.171											.171
.189											.189
.206	-.081	-.156	-.120	-.090	-.010	.013	.020				.206
.223											.223
.241											.241
.258	-.031	-.160	-.128	-.084	-.004	.019	.018	-.051	-.165		.258
.275											.275
.293											.293
.310	-.011	-.168	-.139	-.091	-.014	.009	.019	-.057	-.165		.310
.327											.327
.344											.344
.362	-.000	-.131	-.144	-.104	-.024	.002	.010	-.068	-.122		.362
.379											.379
.396	-.000	-.110	-.119	-.104	-.016	.007	.021	-.070	-.106		.396
.408											.408
.426	.219	.077	.070	.088	.169	.212	.219	.121	.088		.426
.451	.146	.065	.071	.092	.164	.189	.204	.120	.069		.451
.476	.109	.048	.064	.093	.168	.195	.205	.130	.055		.476
.501	.079	.042	.069	.095	.162	.184	.196	.134	.042		.501
.530	-.107	-.127	-.112	-.089	-.030	-.015	-.008	-.058	-.131		.530
.548											.548
.566											.566
.584	-.053	-.112	-.094	-.067	-.005	.012	.023	-.035	-.119		.584
.602											.602
.620											.620
.638	-.030	-.106			-.054	.004					.638
.655											.655
.673											.673
.691	-.020	-.120	-.104	-.070	-.017	.009	.031	-.033	-.101		.691
.709											.709
.727											.727
.745	-.012	-.112			-.071	-.002					.745
.763											.763
.781											.781
.799											.799
.817	-.000	-.092	-.093	-.073	-.004	.023	.036	-.033	-.073		.817
.834											.834
.852											.852
.870											.870
.888											.888
.906											.906
.921	-.179	.040	-.062		-.067	.001					.921
.924											.924
.947											.947
.964											.964
.981	.037		-.011	-.060			.012	.029			.981
1.000	-.189										1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.208	.481	.031	.884	.461
.030	.848	.918	.094	.469	.012
.050	1.311	1.045	.156	1.432	1.472
.070	1.426	1.079	.219	1.399	1.624
.090	1.407	1.097	.281	1.414	1.577
.110	1.289	1.119	.344	1.402	1.518
.130	1.229	1.137	.406	1.400	1.522
.150	1.196	1.146	.469	1.402	1.523
.170	1.182	1.150	.750	1.390	
.190	1.103	1.092	1.000	1.404	
			1.250	1.416	
			1.500	1.436	

TABLE 17. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT M = 2.29

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
.042							1.639			.042
.043							1.548			.043
.044							1.404			.044
.046							1.134			.046
.049							.328			.049
.053							.439			.053
.060	.057	-.072	-.094	-.109	-.116	-.135	-.141	-.115		.060
.075	.210	.014	-.019	-.052	-.088	-.092	-.091	-.074		.075
.085	.264	.066	.021	-.014	-.060	-.063	-.063	-.047	.047	.085
.096	.303									.096
.126		.085	.050	.023	-.009	-.018	-.022	.005	.077	.126
.143							-.136			.143
.154	.037	-.076	-.097	-.112	-.124		-.119	-.120	-.081	.154
.171							-.104			.171
.189							-.091			.189
.206	.040	-.079	-.098	-.109	-.095	-.081	-.077	-.103	-.083	.206
.223							-.064			.223
.241							-.054			.241
.258	.033	-.091	-.109	-.114	-.069	-.052	-.045	-.088	-.090	.258
.275							-.037			.275
.293							-.035			.293
.310	.033	-.103	-.122	-.110	-.055	-.042	-.030	-.064	-.104	.310
.327							-.027			.327
.344							-.024			.344
.362	.030	-.114	-.132	-.076	-.055	-.036	-.022	-.056	-.115	.362
.379							-.020			.379
.396	.027	-.121	-.118	-.067	-.061	-.034	-.014	-.055	-.122	.396
.408							-.016			.408
.426	.164	-.058	-.006	.004	-.002	.077	.088	.011	-.057	.426
.451	.167	-.053	-.005	.010	-.002	.052	.089	.016	-.051	.451
.476	.166	-.047	-.002	.013	-.012	.044	.059	.016	-.040	.476
.501	.166	-.001	.011	-.022	-.036	.043	.043	-.022	.501	
.530	.022	-.103	-.089	-.060	-.101		-.058	-.057	-.097	.530
.548							-.057			.548
.566							-.058			.566
.584	.026	-.095	-.074	-.060	-.090	-.053	-.046	-.061	-.090	.584
.602							-.040			.602
.620							-.035			.620
.638	.024	-.091		-.063	-.068		-.031		-.090	.638
.655							-.029			.655
.673							-.028			.673
.691	.022	-.094	-.082		-.052	-.040		-.050	-.090	.691
.709							-.029			.709
.727							-.029			.727
.745	.026	-.092		-.052	-.046		-.028		-.091	.745
.763							-.028			.763
.781							-.032			.781
.799							-.032			.799
.817	.024	-.099	-.058	-.046	-.047	-.041	-.026	-.038	-.092	.817
.834							-.029			.834
.870							-.028			.870
.888							-.029			.888
.921	-.166						-.026			.921
.924	.024	-.076		-.043			-.027			.924
.947							-.028			.947
.964							-.036	-.063	-.075	.964
.981	.045	-.063	-.041		-.043	-.043	-.028			.981
.998							-.030			.998
1.000	-.168						-.030			1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.198	.654	.031	.781	.257
.030	1.563	1.069	.094	1.711	.818
.050	1.663	1.150	.156	1.723	1.302
.070	1.657	1.170	.219	1.744	1.391
.090	1.624	1.184	.281	1.746	1.437
.110	1.615	1.195	.344	1.748	1.453
.130	1.620	1.207	.406	1.748	1.462
.150	1.615	1.207	.469	1.750	1.470
.170	1.599	1.153	.750	1.751	
.190	1.541	1.031	1.000	1.756	
			1.250	1.741	
			1.500	1.627	

TABLE 17. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{y}{L}$	Cp for meridian angle, θ , deg -									Model station, $\frac{y}{L}$
	0	90	105	120	150	165	180	225	270	
.042										.042
.043										.043
.044										.044
.046										.046
.049										.049
.053										.053
.060	-.000	-.068	-.082	-.091	-.094	-.096	-.105	-.093		.060
.075	.124	.014	-.006	-.028	-.050	-.056	-.056	-.034		.075
.085	.178	.065	.041	.015	-.015	-.020	-.021	-.008	.058	.085
.096	.237									.096
.126		.089	.072	.055	.025	.022	.017	.043	.087	.126
.143										.143
.154	-.006	-.068	-.080	-.090	-.103					.154
.171										.171
.189										.189
.206	.000	-.061	-.068	-.077	-.074	-.070	-.069	-.076	-.062	.206
.223										.223
.241										.241
.258	-.001	-.059	-.066	-.067	-.052	-.047	-.048	-.058	-.058	.258
.275										.275
.293										.293
.310	-.002	-.062	-.066	-.058	-.041	-.033	-.031	-.044	-.060	.310
.327										.327
.344										.344
.379										.379
.396	-.005	-.062	-.058	-.046	-.027	-.023	-.017	-.031	-.057	.396
.408										.408
.426	.117	.032	.046	.053	.061	.072	.079	.056	.039	.426
.451	.115	.037	.043	.047	.058	.062	.070	.053	.041	.451
.476	.115	.039	.039	.042	.047	.053	.057	.046	.041	.476
.501	.116	.036	.036	.038	.043	.047				.501
.530	-.010	-.059	-.060	-.059	-.047					.530
.548										.548
.566										.566
.584	-.006	-.054	-.053	-.046	-.043	-.042				.584
.602										.602
.620										.620
.638	-.004	-.051		-.042	-.036					.638
.655										.655
.673										.673
.691	-.002	-.052	-.050		-.031	-.030				.691
.709										.709
.727										.727
.745	-.001	-.048		-.035	-.026					.745
.763										.763
.781										.781
.799										.799
.817	.003	-.048	-.040	-.029	-.022	-.018	-.011	-.020	-.043	.817
.834										.834
.870										.870
.888										.888
.921	-.150	.001	-.042		-.025					.921
.924										.924
.947										.947
.964										.964
.981	.018	-.037	-.027							.981
.984										.984
.998										.998
1.000		-.151								1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.021	.758	.031	.616	.342
.030	1.483	1.104	.094	1.264	.766
.050	1.588	1.160	.156	1.478	1.071
.070	1.586	1.176	.219	1.599	1.242
.090	1.558	1.189	.281	1.615	1.384
.110	1.549	1.191	.344	1.625	1.440
.130	1.549	1.180	.406	1.634	1.461
.150	1.544	1.138	.469	1.639	1.480
.170	1.528	1.085	.750	1.655	
.190	1.462	1.075	1.000	1.656	
			1.250	1.663	
			1.500	1.559	

TABLE 17. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.29$ - Continued(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.042							1.674				.042
.043							1.635				.043
.044							1.527				.044
.046							1.280				.046
.049							.707				.049
.053							.758				.053
.060	-.038	-.065	-.072	-.074	-.080	-.081	-.083	-.073			.060
.075	.062	.014	.008	-.002	-.015	-.020	-.023	-.005			.075
.085	.111	.066	.058	.044	.025	.021	.019	.026	.070		.085
.096	.169										.096
.126		.096	.089	.076	.060	.059	.058	.070	.093		.126
.143							.096				.143
.154	-.039	-.063	-.071	-.076	-.082		.084	-.081	-.065		.154
.171							.076				.171
.189							.068				.189
.206	-.027	-.050	-.053	-.058	-.059	-.062	.061	-.059	-.050		.206
.223							.055				.223
.241							.050				.241
.258	-.025	-.041	-.045	-.046	-.043	-.043	.046	-.043	-.038		.258
.275							.040				.275
.293							.037				.293
.310	-.022	-.036	-.037	-.037	-.035	-.033	.032	-.032	-.034		.310
.327							.029				.327
.344							.027				.344
.362	-.018	-.035	-.033	-.032	-.028	-.026	.025	-.025	-.029		.362
.379							.023				.379
.396	-.019	-.032	-.032	-.029	-.024	-.022	.018	-.023	-.027		.396
.408							.019				.408
.426	.097	.073	.077	.076	.078	.081	.080	.079	.077		.426
.451	.092	.070	.070	.070	.074	.073	.073	.072	.072		.451
.476	.089	.065	.063	.064	.063	.064	.065	.065	.067		.476
.501	.086		.059	.057	.057	.057	.058		.063		.501
.530	-.027	-.041	-.044	-.044	-.043		.043	-.040	-.041		.530
.548							.038				.548
.566							.035				.566
.584	-.020	-.034	-.034	-.034	-.036	-.037	.035	-.032	-.032		.584
.602							.034				.602
.620							.029				.620
.638	-.018	-.028		-.030	-.029		.023		-.027		.638
.655							.020				.655
.673							.019				.673
.691	-.015	-.029	-.031		-.024	-.021	-.017		-.026		.691
.709							.016				.709
.727							.015				.727
.745	-.013	-.025		-.021	-.018		.012				.745
.763							.013				.763
.781							.013				.781
.799							.013				.799
.817	-.010	-.021	-.020	-.016	-.016	-.013	-.011		-.013		.817
.834							.012				.834
.870							.009				.870
.888							.007				.888
.921	-.128						.006				.921
.924	-.004	-.017		-.018			.007				.924
.947							.005				.947
.964							.007				.964
.981	.000	-.014	-.013		-.009		-.007		-.007		.981
.984					-.013		-.007		-.014		.984
.998							-.017				.998
1.000		-.131									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.898	.818	.031	.436	.319
.030	1.370	1.128	.094	.809	.546
.050	1.495	1.169	.156	1.024	.734
.070	1.497	1.187	.219	1.202	.876
.090	1.481	1.197	.281	1.342	1.011
.110	1.483	1.199	.344	1.408	1.139
.130	1.483	1.187	.406	1.442	1.249
.150	1.477	1.101	.469	1.464	1.309
.170	1.462	1.094	.750	1.530	
.190	1.403	1.082	1.000	1.563	
			1.250	1.598	
			1.500	1.513	

TABLE 17. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.29$ - Continued(d) $\alpha = 0^\circ$

Model station, $\frac{X}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{X}{T}$
	0	90	105	120	150	165	180	225	270	
.042							1.678			.042
.043							1.659			.043
.044							1.569			.044
.046							1.347			.046
.049							.896			.049
.053							.813			.053
.060	-.065	-.065	-.065	-.062	-.060	-.060	-.060	-.056		.060
.075	.013	.015	.017	.018	.019	.017	.016	.017	.074	.075
.085	.059	.066	.070	.067	.065	.065	.063	.062		.085
.096	.108								.096	
.126		.097	.096	.099	.095	.097	.099	.098	.095	.126
.143							.074			.143
.154	-.065	-.064	-.064	-.063	-.064		.064	-.064	-.065	.154
.171							.058			.171
.189							.053			.189
.206	-.050	-.048	-.047	-.049	-.047	-.048	-.048	-.047	-.046	.206
.223							.044			.223
.241							.041			.241
.258	-.039	-.037	-.036	-.036	-.036	-.036	-.039	-.035	-.033	.258
.275							.036			.275
.293							.035			.293
.310	-.031	-.032	-.032	-.031	-.032	-.030	-.031	-.027	-.027	.310
.327							.028			.327
.344							.028			.344
.362	-.027	-.029	-.026	-.027	-.028	-.027	-.027	-.022	-.022	.362
.379							.026			.379
.396	-.024	-.025	-.026	-.026	-.025	-.024	-.021	-.021	-.020	.396
.408							.023			.408
.426	.084	.080	.083	.082	.083	.084	.084	.086	.084	.426
.451	.079	.076	.077	.078	.082	.078	.081	.080	.078	.451
.476	.073	.071	.070	.072	.074	.074	.076	.076	.073	.476
.501	.067		.067	.067	.069	.069	.070		.068	.501
.530	-.038	-.037	-.037	-.038	-.037		.038	-.036	-.035	.530
.548							.035			.548
.566							.033			.566
.584	-.030	-.030	-.028	-.029	-.034	-.035	-.032	-.030	-.026	.584
.602							.026			.602
.620							.024			.620
.638	-.025	-.026		-.027	-.021		.020		-.023	.638
.655							.019			.655
.673							.019			.673
.691	-.021	-.025	-.024		-.023	-.023		-.019	-.020	.691
.709							.016			.709
.727							.014			.727
.745	-.018	-.017		-.018	-.016		.015		-.014	.745
.763							.015			.763
.781							.018			.781
.799							.016			.799
.817	-.014	-.016	-.016	-.015	-.016	-.015		-.011	-.007	.817
.834							.013			.834
.870							.009			.870
.888							.008			.888
.921	-.120	-.008	-.014		-.017			.007		.921
.924							.007			.924
.947							.005			.947
.964							.005			.964
.981	.003	-.009	-.012		-.010	-.007		-.005	-.004	.981
.984							.008		-.009	.984
.998							.013			.998
1.000		-.123								1.000

Orifice station, y , in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side	Orifice station, y , in.
.010	.784	.830	.031	.306	.302	
.030	1.251	1.127	.094	.504	.486	
.050	1.382	1.167	.156	.644	.626	
.070	1.389	1.188	.219	.757	.756	
.090	1.389	1.195	.281	.850	.863	
.110	1.394	1.200	.344	.955	.968	
.130	1.396	1.180	.406	1.057	1.081	
.150	1.394	1.089	.469	1.143	1.175	
.170	1.382	1.097	.750	1.323		
.190	1.320	1.080	1.000	1.381		
			1.250	1.421		
			1.500	1.365		

TABLE 17. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.682			.042
.043							1.680			.043
.044							1.612			.044
.046							1.420			.046
.049							1.017			.049
.053							.866			.053
.060	-.085	-.065	-.058	-.047	-.034	-.030	-.029	-.034		.060
.075	-.026	.015	.031	.042	.063	.067	.067	.049	.073	.075
.085	.014	.068	.084	.094	.113	.119	.120	.101		.085
.096	.069									.096
.126		.098	.110	.119	.137	.143	.147	.129	.094	.126
.143							.047			.143
.154	-.085	-.065	-.059	-.050	-.042		-.034	-.048	-.064	.154
.171							.032			.171
.189							.029			.189
.206	-.061	-.050	-.045	-.041	-.029	-.027	-.027	-.035	-.049	.206
.223							.024			.223
.241							.024			.241
.258	-.043	-.042	-.036	-.033	-.024	-.022		-.029	-.038	.258
.275							.022			.275
.293							.022			.293
.310	-.030	-.039	-.036	-.032	-.025	-.022	-.021	-.025	-.035	.310
.327							.021			.327
.344							.020			.344
.362	-.023	-.037	-.035	-.030	-.023	-.019	-.018	-.022	-.031	.362
.379							.019			.379
.396	-.020	-.033	-.034	-.031	-.022	-.019	-.015	-.023	-.028	.396
.408							.017			.408
.426	.079	.070	.076	.078	.088	.095	.096	.086	.074	.426
.451	.073	.068	.071	.076	.091	.091	.095	.083	.070	.451
.476	.064	.064	.067	.073	.086	.090	.092	.082	.067	.476
.501	.057		.066	.069	.082	.085	.087		.062	.501
.530	-.042	-.042	-.041	-.038	-.032		-.030	-.033	-.041	.530
.548							.025			.548
.566							.019			.566
.584	-.032	-.036	-.034	-.032	-.025	-.022	-.018	-.025	-.033	.584
.602							.016			.602
.620							.016			.620
.638	-.025	-.033		-.024	-.017		.014		-.028	.638
.655							.012			.655
.673							.010			.673
.691	-.019	-.031	-.028		-.019	-.016		-.017	-.026	.691
.709							.013			.709
.727							.012			.727
.745	-.016	-.023		-.023	-.017		.011		-.020	.745
.763							.010			.763
.781							.014			.781
.799							.013			.799
.817	-.011	-.024	-.022	-.021	-.015	-.011		-.013	-.016	.817
.834							.006			.834
.870							.009			.870
.888							.005			.888
.921	-.124						.003			.921
.924	-.008	-.020		-.022			.005			.924
.947							.000			.947
.964										.964
.981	.004	-.017	-.016		-.006	-.003	-.004	-.004	-.010	.981
.984					-.013				-.017	.984
.998							.007			.998
1.000		-.129								1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp _{top}	Cp _{side}	Orifice station, y , in.	Cp _{top}	Cp _{side}
.010	.563	.824	.031	.246	.335
.030	1.234	1.129	.094	.378	.592
.050	1.417	1.167	.156	.470	.805
.070	1.406	1.188	.219	.558	.958
.090	1.345	1.196	.281	.624	1.093
.110	1.319	1.198	.344	.688	1.222
.130	1.301	1.174	.406	.749	1.310
.150	1.295	1.091	.469	.801	1.348
.170	1.279	1.095	.750	1.019	
.190	1.216	1.077	1.000	1.180	
			1.250	1.237	
			1.500	1.183	

TABLE 17. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{T}$
	0	90	105	120	130	145	165	180	225	270	
.042							2.238				.042
.043							1.667				.043
.044							1.688				.044
.046							1.645				.046
.049							1.492				.049
.053							1.117				.053
.060	-.107	-.068	-.053	-.032	-.001	.006	.011	-.009			.060
.075	-.056	.015	.045	.070	.119	.129	.130	.091			.075
.085	-.022	.067	.098	.127	.172	.186	.190	.147	.058		.085
.096	.011										.096
.126		.091	.117	.141	.187	.197	.204	.165	.086		.126
.143							-.013				.143
.154	-.106	-.071	-.056	-.040	-.014		.000	-.028	-.067		.154
.171							.001				.171
.189							.002				.189
.206	-.068	-.063	-.049	-.036	-.006	.000	.001	-.022	-.062		.206
.223							.003				.223
.241							.001				.241
.258	-.044	-.061	-.050	-.036	-.007	.000	.000	-.022	-.059		.258
.275							-.000				.275
.293							-.001				.293
.310	-.028	-.067	-.056	-.040	-.012	-.001	.001	-.023	-.063		.310
.327							.000				.327
.344							.000				.344
.362	-.020	-.070	-.060	-.044	-.015	-.004	.000	-.025	-.064		.362
.379							-.000				.379
.396	-.018	-.067	-.063	-.047	-.015	-.005	.000	-.028	-.062		.396
.408							-.002				.408
.426	.080	.029	.036	.054	.098	.116	.121	.080	.032		.426
.451	.071	.034	.037	.054	.102	.114	.120	.081	.037		.451
.476	.058	.036	.039	.056	.099	.113	.118	.082	.039		.476
.501	.047		.043	.056	.092	.104	.111	.037	.051		.501
.530	-.051	-.061	-.058	-.050	-.022		-.007	-.035	-.060		.530
.548							-.003				.548
.566							-.000				.566
.584	-.039	-.059	-.050	-.041	-.015	-.007	-.002	-.024	-.056		.584
.602							-.001				.602
.620							.002				.620
.638	-.028	-.056			-.035	-.010					.638
.655							.000				.655
.673							-.001				.673
.691	-.023	-.056	-.050			-.018	-.007				.691
.709							.000				.709
.727							.001				.727
.745	-.017	-.052			-.040	-.013					.745
.763							.002				.763
.781							.000				.781
.799							-.001				.799
.817	-.011	-.052	-.049	-.039	-.011	.000	.005	-.019	-.044		.817
.834							.003				.834
.870							.006				.870
.888							.008				.888
.921	-.147	-.011									.921
.924											.924
.947											.947
.964											.964
.981	.000	-.039	-.044			-.010	.001	.004	-.013	-.035	.981
.998										-.042	.998
1.000											1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.500	.812	.031	.473	.351
.030	1.045	1.127	.094	.952	.819
.050	1.198	1.168	.156	1.047	1.137
.070	1.211	1.183	.219	1.084	1.301
.090	1.214	1.192	.281	1.092	1.424
.110	1.220	1.197	.344	1.104	1.461
.130	1.218	1.171	.406	1.108	1.481
.150	1.215	1.088	.469	1.112	2.234
.170	1.198	1.094	.750	1.133	
.190	1.151	1.071	1.000	1.158	
			1.250	1.181	
			1.500	1.107	

TABLE 17. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.29$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.640			.042
.043							1.681			.043
.044							1.667			.044
.046							1.545			.046
.049							1.217			.049
.053							.991			.053
.060	-.143	-.072	-.047	-.012	.045	.063	.072	.026		.060
.075	-.091	.013	.061	.106	.192	.213	.218	.148		.075
.085	-.074	.067	.116	.167	.246	.272	.280	.207	.044	.085
.096	-.038									.096
.126		.079	.124	.171	.246	.267	.281	.224	.078	.126
.143							.029			.143
.154	-.120	-.079	-.056	-.024	.022		.049	-.004	-.081	.154
.171							.048			.171
.189							.046			.189
.206	-.071	-.082	-.058	-.030	.025	.039	.047	-.000	-.082	.206
.223							.043			.223
.241							.038			.241
.258	-.043	-.096	-.070	-.038	.020	.037	.040	-.008	-.090	.258
.275							.036			.275
.293							.036			.293
.310	-.027	-.109	-.081	-.047	.012	.031	.039	-.013	-.105	.310
.327							.038			.327
.344							.037			.344
.362	-.018	-.122	-.092	-.054	.007	.028	.036	-.017	-.118	.362
.379							.034			.379
.396	-.016	-.128	-.098	-.061	.006	.026	.037	-.022	-.124	.396
.408							.032			.408
.426	.086	-.068	-.017	.031	.127	.161	.173	.091	-.062	.426
.451	.088	-.060	-.017	.034	.134	.160	.173	.091	-.058	.451
.476	.060	-.055	-.017	.034	.133	.162	.174	.094	-.051	.476
.501	.044		-.014	.034	.132	.160	.172		-.039	.501
.530	-.056	-.109	-.101	-.067	.000		.027	-.029	-.106	.530
.548							.029			.548
.566							.032			.566
.584	-.050	-.101	-.099	-.064	.001	.021	.029	-.026	-.098	.584
.602							.030			.602
.620							.027			.620
.638	-.033	-.097		-.064	-.000		.025		-.095	.638
.655							.024			.655
.673							.028			.673
.691	-.032	-.102	-.100		-.003	.020		-.029	-.097	.691
.709							.031			.709
.727							.032			.727
.745	-.031	-.099		-.071	-.000		.031		-.093	.745
.763							.032			.763
.781							.026			.781
.799							.022			.799
.817	-.030	-.104	-.095	-.068	-.003	.019	.030	-.031	-.093	.817
.834							.025			.834
.870							.029			.870
.888							.028			.888
.921	-.165						.030			.921
.924	-.030	-.076		-.071			.027			.924
.947							.031			.947
.964							.026			.964
.981	-.018	-.064	-.103		-.001	.019		-.024	-.065	.981
.984					-.004		.026		-.077	.984
.998							.028			.998
1.000	-.169									1.000

Orifice station, y/l	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side	
.010	.228	.784	.031	.576	.237	
.030	.751	1.239	.094	1.235	.816	
.050	1.084	1.201	.156	1.086	1.342	
.070	1.170	1.193	.219	1.094	1.395	
.090	1.152	1.199	.281	1.093	1.438	
.110	1.111	1.201	.344	1.086	1.447	
.130	1.092	1.193	.406	1.089	1.458	
.150	1.086	1.093	.469	1.086	1.462	
.170	1.071	1.093	.750	1.082		
.190	1.015	1.086	1.000	1.102		
			1.250	1.122		
			1.500	1.081		

TABLE 18. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.98$ (a) $\alpha = -10.1^\circ$

Model station, $\frac{y}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{y}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.708			.042
.043							1.623			.043
.044							1.480			.044
.046							1.206			.046
.049							.403			.049
.053							.456			.053
.060	.136	.012	-.006	-.020	-.042	-.055	-.057	-.024		.060
.075	.226	.051	.034	.011	-.019	-.028	-.030	-.003		.075
.085	.259	.094	.066	.034	-.000	-.008	-.010	.010	.079	.085
.096	.309									.096
.126		.083	.055	.035	.008	.005	.008	.021	.083	.126
.143							.076			.143
.154	.061	-.033	-.049	-.061	-.071		-.070	-.068	-.034	.154
.171							.065			.171
.189							.059			.189
.206	.054	-.042	-.056	-.064	-.058	-.051	-.050	-.064	-.041	.206
.223							.041			.223
.241							.034			.241
.258	.047	-.050	-.064	-.071	-.046	-.035	-.031	-.060	-.047	.258
.275							.027			.275
.293							.025			.293
.310	.043	-.060	-.073	-.077	-.038	-.029	-.023	-.051	-.062	.310
.327							.022			.327
.344							.020			.344
.362	.042	-.068	-.082	-.070	-.034	-.026	-.019	-.046	-.070	.362
.379							.018			.379
.396	.039	-.073	-.086	-.063	-.037	-.026	-.015	-.048	-.075	.396
.408							.016			.408
.426	.147	-.033	-.038	-.018	.002	.030	.046	-.010	-.034	.426
.451	.159	-.029	-.043	-.005	.004	.027	.054	-.007	-.028	.451
.476	.165	-.026	-.040	-.002	-.007	.020	.041	-.005	-.024	.476
.501	.168	-.037	-.000	-.012	.015	.030	.030	-.022	.501	
.530	.045	-.073	-.075	-.046	-.058		.038	-.037	-.073	.530
.548							.038			.548
.566							.036			.566
.584	.045	-.078	-.066	-.051	-.060	-.039	-.035	-.043	-.076	.584
.602							.027			.602
.620							.026			.620
.638	.044	-.078		-.050	-.049		.024		-.079	.638
.655							.023			.655
.673							.022			.673
.691	.042	-.082	-.061		-.041	-.035		-.047	-.083	.691
.709							.023			.709
.727							.023			.727
.745	.043	-.073		-.044	-.044		.023		-.084	.745
.763							.023			.763
.781							.026			.781
.799							.026			.799
.817	.042	-.076	-.046	-.042	-.049	-.036	-.024	-.040	-.079	.817
.834							.027			.834
.870							.024			.870
.888							.026			.888
.921	-.111	.040	-.053		-.041			.025		.921
.924							.026			.924
.947							.024			.947
.964							.024			.964
.981	.056	-.045	-.038			-.050	-.047		-.042	.981
.984						-.051			-.051	.984
.998								-.026		.998
1.000	-.111									1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	1.050	.820	.031	.744	.108
.030	1.463	1.179	.094	1.702	.390
.050	1.590	1.208	.156	1.907	.914
.070	1.613	1.220	.219	1.949	1.150
.090	1.633	1.229	.281	1.953	1.321
.110	1.651	1.217	.344	1.960	1.375
.130	1.661	1.150	.406	1.960	1.402
.150	1.669	1.118	.469	1.964	1.412
.170	1.663	1.139	.750	1.936	
.190	1.611	1.124	1.000	1.937	
			1.250	1.926	
			1.500	1.822	

TABLE 18. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.98$ - Continued(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
.042							1.742			.042
.043							1.671			.043
.044							1.545			.044
.046							1.286			.046
.049							.509			.049
.053							.607			.053
.060	.085	.018	.005	-.002	-.016	-.022	-.026	-.004		.060
.075	.157	.063	.047	.028	.009	.003	.002	.021		.075
.085	.188	.096	.081	.059	.032	.027	.026	.043	.095	.085
.096	.235									.096
.126		.089	.072	.056	.036	.033	.034	.044	.090	.126
.143							-.066			.143
.154	.020	-.030	-.039	-.049	-.057		-.059	-.053	-.028	.154
.171							-.055			.171
.189							-.049			.189
.206	.018	-.032	-.041	-.044	-.046	-.045	-.045	-.046	-.032	.206
.223							-.040			.223
.241							-.035			.241
.258	.013	-.037	-.043	-.045	-.037	-.030	-.031	-.041	-.032	.258
.275							-.028			.275
.293							-.027			.293
.310	.011	-.039	-.044	-.043	-.030	-.024	-.025	-.036	-.038	.310
.327							-.023			.327
.344							-.021			.344
.362	.009	-.042	-.045	-.040	-.024	-.020	-.019	-.029	-.040	.362
.379							-.017			.379
.396	.007	-.044	-.045	-.037	-.019	-.014	-.015	-.026	-.042	.396
.408							-.014			.408
.426	.098	.009	.016	.025	.040	.047	.049	.035	.015	.426
.451	.102	.015	.017	.026	.046	.043	.048	.035	.020	.451
.476	.107	.019	.019	.024	.034	.037	.040	.032	.023	.476
.501	.111		.020	.022	.029	.031	.033		.027	.501
.530	.014	-.041	-.042	-.037	-.029		-.029	-.031	-.038	.530
.548							-.028			.548
.566							-.025			.566
.584	.012	-.041	-.040	-.035	-.026	-.026	-.023	-.027	-.038	.584
.602							-.020			.602
.620							-.018			.620
.638	.011	-.042		-.032	-.023		-.015		-.039	.638
.655							-.014			.655
.673							-.013			.673
.691	.009	-.042	-.038		-.019	-.016		-.022	-.041	.691
.709							-.012			.709
.727							-.011			.727
.745	.011	-.035		-.024	-.017		-.011		-.041	.745
.763							-.010			.763
.781							-.012			.781
.799							-.011			.799
.817	.011	-.037	-.028	-.020	-.015	-.013	-.008	-.017	-.039	.817
.834							-.009			.834
.870							-.007			.870
.888							-.006			.888
.921	-.106	.012	-.033		-.018			-.006		.921
.924							-.007			.924
.947							-.005			.947
.964							-.005			.964
.981	.022	-.031	-.022			-.015	-.011	-.007	-.014	.981
.984						-.016			-.032	.984
.998									-.037	.998
1.000	-.105									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.912	.855	.031	.538	.251
.030	1.353	1.176	.094	1.203	.658
.050	1.489	1.201	.156	1.466	.902
.070	1.513	1.224	.219	1.653	1.110
.090	1.526	1.226	.281	1.693	1.293
.110	1.537	1.206	.344	1.711	1.374
.130	1.544	1.127	.406	1.725	1.410
.150	1.546	1.122	.469	1.739	1.438
.170	1.535	1.141	.750	1.761	
.190	1.481	1.128	1.000	1.774	
			1.250	1.781	
			1.500	1.676	

TABLE 18. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
.042							1.748			.042
.043							1.700			.043
.044							1.596			.044
.046							1.350			.046
.049							.662			.049
.053							.751			.053
.060	.049	.019	.013	.012	.007	.004	.001	.012		.060
.075	.107	.064	.059	.048	.040	.035	.035	.045		.075
.085	.135	.097	.093	.080	.065	.063	.061	.072	.100	.085
.096	.176									.096
.126		.091	.083	.075	.064	.060	.062	.068	.093	.126
.143							.051			.143
.154	-.008	-.028	-.031	-.036	-.042		.045	-.039	-.026	.154
.171							.042			.171
.189							.039			.189
.206	-.005	-.027	-.030	-.034	-.035	-.035	-.037	-.034	-.026	.206
.223							.035			.223
.241							.034			.241
.258	-.007	-.025	-.030	-.030	-.030	-.031	-.033	-.030	-.020	.258
.275							.032			.275
.293							.030			.293
.310	-.008	-.024	-.026	-.026	-.025	-.023	-.028	-.025	-.023	.310
.327							.024			.327
.344							.022			.344
.362	-.008	-.023	-.024	-.023	-.020	-.020	-.021	-.021	-.021	.362
.379							.019			.379
.396	-.009	-.021	-.021	-.022	-.017	-.016	-.018	-.019	-.020	.396
.408							.016			.408
.426	.068	.051	.053	.049	.052	.051	.049	.046	.049	.426
.451	.072	.049	.048	.048	.054	.049	.049	.048	.050	.451
.476	.073	.047	.045	.045	.045	.045	.045	.046	.051	.476
.501	.075		.043	.041	.041	.041	.042	.049	.051	.501
.530	-.007	-.024	-.026	-.026	-.024		.022	-.021	-.020	.530
.548							.021			.548
.566							.019			.566
.584	-.004	-.021	-.021	-.022	-.022	-.021	-.018	-.018	-.017	.584
.602							.017			.602
.620							.016			.620
.638	-.005	-.019		-.018	-.017					.638
.655							.014			.655
.673							.012			.673
.691	-.004	-.017	-.018		-.013	-.013		-.015	-.016	.691
.709							.010			.709
.727							.008			.727
.745	-.005	-.008		-.013	-.011		.008		-.016	.745
.763							.008			.763
.781							.009			.781
.799							.007			.799
.817	-.003	-.014	-.012	-.010	-.010	-.007	-.006	-.011	-.012	.817
.834							.006			.834
.870							.005			.870
.888							.004			.888
.921	-.092						.004			.921
.924	-.000	-.010		-.008			.006			.924
.947							.004			.947
.964							.004			.964
.981	.007	-.008	-.008		-.006	-.004	-.006	-.006	-.009	.981
.984									-.013	.984
.998					-.007					.998
1.000	-.095									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.763	.864	.031	.343	.121
.030	1.256	1.176	.094	.691	.451
.050	1.396	1.206	.156	.872	.595
.070	1.415	1.222	.219	1.082	.719
.090	1.425	1.230	.281	1.231	.847
.110	1.431	1.213	.344	1.318	.986
.130	1.433	1.128	.406	1.378	1.095
.150	1.435	1.120	.469	1.421	1.166
.170	1.422	1.141	.750	1.545	
.190	1.362	1.124	1.000	1.602	
			1.250	1.637	
			1.500	1.566	

TABLE 18. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.752			.042
.043							1.728			.043
.044							1.640			.044
.046							1.409			.046
.049							.867			.049
.053							.847			.053
.060	.022	.020	.019	.023	.024	.025	.024	.030		.060
.075	.068	.065	.068	.065	.070	.069	.069	.069		.075
.085	.094	.098	.102	.098	.097	.098	.096	.098	.102	.085
.096	.130									.096
.126		.092	.091	.092	.093	.093	.096	.093	.095	.126
.143							-.034			.143
.154	-.027	-.025	-.027	-.027	-.028		-.026	-.027	-.026	.154
.171							-.026			.171
.189							-.026			.189
.206	-.024	-.025	-.025	-.026	-.026		-.026	-.025	-.024	.206
.223							-.024			.223
.241							-.023			.241
.258	-.021	-.023	-.023	-.023	-.023		-.023	-.021	-.017	.258
.275							-.022			.275
.293							-.021			.293
.310	-.018	-.019	-.020	-.021	-.021	-.021	-.020	-.018	-.019	.310
.327							-.019			.327
.344							-.018			.344
.362	-.015	-.017	-.017	-.018	-.018	-.017	-.017	-.016	-.016	.362
.379							-.016			.379
.396	-.014	-.006	-.008	-.010	-.016	-.017	-.016	-.015	-.014	.396
.408							-.016			.408
.426	.054	.022	.031	.035	.058	.057	-.001	.056	.053	.426
.451	.058	.056	.057	.056	.061	.056	.056	.057	.056	.451
.476	.056	.058	.055	.055	.054	.055	.055	.057	.055	.476
.501	.055		.054	.053	.053	.052	.054	.054	.054	.501
.530	-.016	-.018	-.019	-.019	-.020		-.017	-.017	-.015	.530
.548							-.016			.548
.566							-.014			.566
.584	-.012	-.016	-.015	-.015	-.017	-.016	-.014	-.013	-.012	.584
.602							-.013			.602
.620							-.012			.620
.638	-.011	-.012		-.010	-.013		-.011		-.010	.638
.655							-.011			.655
.673							-.011			.673
.691	-.010	-.011	-.011		-.013	-.013		-.011	-.011	.691
.709							-.011			.709
.727							-.010			.727
.745	-.008	-.002		-.009	-.010		-.009		-.010	.745
.763							-.009			.763
.781							-.010			.781
.799							-.008			.799
.817	-.006	-.006	-.007	-.007	-.009	-.007	-.006	-.008	-.007	.817
.834							-.007			.834
.870							-.008			.870
.888							-.006			.888
.921	-.084						-.004			.921
.924	-.003	-.005					-.005			.924
.947							-.002			.947
.964							-.004			.964
.981	.002	-.004	-.005				-.004	-.005		.981
.984							-.004	-.008		.984
.998							-.007			.998
1.000	-.087									1.000

Nose rake			Base rake		
Orifice station, y , in.	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.624	.869	.031	.234	.223
.030	1.164	1.180	.094	.417	.389
.050	1.311	1.211	.156	.521	.515
.070	1.332	1.226	.219	.600	.626
.090	1.336	1.230	.281	.677	.730
.110	1.340	1.208	.344	.762	.850
.130	1.339	1.134	.406	.860	.954
.150	1.335	1.124	.469	.939	1.023
.170	1.314	1.141	.750	1.187	
.190	1.261	1.123	1.000	1.269	
			1.250	1.329	
			1.500	1.331	

TABLE 18. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.749			.042
.043							1.739			.043
.044							1.677			.044
.046							1.472			.046
.049							1.021			.049
.053							.915			.053
.060	.000	.020	.025	.036	.049	.052	.054	.050		.060
.075	.033	.065	.078	.085	.105	.108	.110	.096		.075
.085	.059	.097	.112	.118	.132	.139	.138	.127	.101	.085
.096	.091									.096
.126		.090	.100	.111	.126	.130	.135	.122	.093	.126
.143							.013			.143
.154							.006	-.013	-.025	.154
.171							.006			.171
.189							.007			.189
.206	-.043	-.028	-.022	-.018	-.010	-.007	-.007	-.013	-.025	.206
.223							.007			.223
.241							.008			.241
.258	-.030	-.026	-.023	-.018	-.010	-.008	-.008	-.013	-.019	.258
.275							.008			.275
.293							.009			.293
.310	-.024	-.024	-.023	-.020	-.012	-.010	-.009	-.013	-.023	.310
.327							.009			.327
.344							.009			.344
.379							.010			.379
.396	-.015	-.024	-.021	-.017	-.012	-.010	-.008	-.013	-.020	.396
.408							.009			.408
.426	.048	.045	.055	.054	.062	.065	.067	.061	.049	.426
.451	.051	.048	.051	.054	.070	.067	.071	.064	.050	.451
.476	.046	.046	.049	.053	.067	.069	.072	.065	.049	.476
.501	.043		.048	.051	.067	.071	.074		.048	.501
.530	-.021	-.025	-.023	-.019	-.010		.006	-.010	-.021	.530
.548							.005			.548
.566							.004			.566
.584	-.017	-.022	-.020	-.017	-.012	-.008	-.005	-.009	-.018	.584
.602							.005			.602
.620							.005			.620
.638	-.014	-.019			-.015	-.010				.638
.655							.004			.655
.673							.004			.673
.691	-.012	-.019	-.017		-.010	-.008		-.010	-.017	.691
.709							.004			.709
.727							.004			.727
.745	-.009	-.009			-.014	-.008			-.016	.745
.763							.004			.763
.781							.007			.781
.799							.008			.799
.817	-.006	-.013	-.014	-.013	-.009	-.006	-.004	-.010	-.014	.817
.834							.004			.834
.870							.001			.870
.888							.000			.888
.921	-.088						.000			.921
.924	-.005	-.012			-.010					.924
.947							.000			.947
.964							.000			.964
.981	.001	-.009	-.011			-.002	-.001	-.000	-.004	.981
.984						-.005			-.010	.984
.998									-.015	.998
1.000	-.091									1.000

Orifice station, y, in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side	Orifice station, y, in.
.010	.471	.870	.031	.191	.238	
.030	1.083	1.178	.094	.328	.461	
.050	1.233	1.205	.156	.416	.623	
.070	1.242	1.224	.219	.476	.747	
.090	1.247	1.226	.281	.525	.887	
.110	1.253	1.204	.344	.566	1.029	
.130	1.249	1.126	.406	.606	1.024	
.150	1.243	1.129	.469	.643	1.191	
.170	1.228	1.143	.750	.784		
.190	1.169	1.123	1.000	.940		
			1.250	1.022		
			1.500	.994		

TABLE 18. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.732			.042
.043							1.753			.043
.044							1.714			.044
.046							1.548			.046
.049							1.153			.049
.053							.986			.053
.060	-.029	.017	.030	.050	.079	.088	.091	.075		.060
.075	.000	.063	.088	.107	.148	.157	.161	.129		.075
.085	.023	.096	.121	.141	.178	.191	.191	.161	.093	.085
.096	.050									.096
.126		.087	.108	.129	.166	.175	.182	.156	.089	.126
.143							.014			.143
.154	-.060	-.031	-.021	-.006	.011		.022	.003	-.027	.154
.171							.021			.171
.189							.020			.189
.206	-.044	-.032	-.024	-.011	.009	.016	.019	.001	-.033	.206
.223							.018			.223
.241							.015			.241
.258	-.032	-.038	-.027	-.017	.006	.011	.013	-.003	-.030	.258
.275							.011			.275
.293							.011			.293
.310	-.023	-.040	-.032	-.020	.001	.007	.010	-.006	-.040	.310
.327							.010			.327
.344							.009			.344
.362	-.017	-.044	-.036	-.023	-.000	.006	.009	-.009	-.043	.362
.379							.008			.379
.396	-.015	-.047	-.038	-.027	-.001	.006	.010	-.011	-.044	.396
.408							.008			.408
.426	.045	.008	.026	.040	.077	.093	.097	.065	.013	.426
.451	.048	.013	.025	.045	.089	.097	.105	.071	.018	.451
.476	.040	.016	.026	.048	.091	.103	.110	.077	.021	.476
.501	.034		.030	.048	.093	.104	.113	.074		.501
.530	-.026	-.041	-.036	-.025	.001		.014	-.006	-.039	.530
.548							.014			.548
.566							.014			.566
.584	-.023	-.042	-.036	-.027	-.001	.005	.012	-.007	-.040	.584
.602							.012			.602
.620							.010			.620
.638	-.016	-.042			-.025	-.001				.638
.655							.012			.655
.673							.011			.673
.691	-.014	-.043	-.037		-.005	.003		-.009	-.043	.691
.709							.007			.709
.727							.007			.727
.745	-.012	-.034			-.028	-.005		.009	-.042	.745
.763							.011			.763
.781							.010			.781
.799							.010			.799
.817	-.008	-.040	-.036	-.026	-.000	.009	.013	-.010	-.040	.817
.834							.012			.834
.870							.013			.870
.888							.012			.888
.921	-.103									.921
.924	-.008	-.035			-.028					.924
.947										.947
.964										.964
.981	-.001	-.034	-.036			.001	.009	.013	-.010	.981
.984										.984
.998										.998
1.000		-.104								1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.259	.863	.031	.345	.243
.030	.924	1.179	.094	.720	.674
.050	1.135	1.207	.156	.787	.922
.070	1.142	1.224	.219	.845	1.139
.090	1.146	1.224	.281	.876	1.309
.110	1.148	1.196	.344	.891	1.372
.130	1.147	1.115	.406	.905	1.403
.150	1.143	1.130	.469	.908	1.425
.170	1.125	1.139	.750	.925	
.190	1.077	1.122	1.000	.938	
			1.250	.957	
			1.500	.917	

TABLE 18. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 2.98$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
.042							1.720			.042
.043							1.765			.043
.044							1.748			.044
.046							1.625			.046
.049							1.272			.049
.053							1.071			.053
.060	-.058	.014	.036	.068	.122	.138	.145	.106		.060
.075	-.018	.063	.102	.135	.209	.227	.231	.173		.075
.085	-.010	.098	.137	.174	.240	.263	.264	.209	.083	.085
.096	.000									.096
.126		.085	.119	.155	.222	.241	.254	.199	.082	.126
.143							.058			.143
.154	-.069	-.034	-.015	.006	.046		.066	.030	-.034	.154
.171							.064			.171
.189							.060			.189
.206	-.048	-.042	-.023	-.002	.040	.050	.057	.019	-.041	.206
.223							.056			.223
.241							.053			.241
.258	-.030	-.051	-.034	-.010	.034	.046	.051	.012	-.047	.258
.275							.048			.275
.293							.047			.293
.310	-.022	-.062	-.042	-.018	.027	.041	.047	.007	-.060	.310
.327							.046			.327
.344							.045			.344
.362	-.018	-.070	-.050	-.023	.023	.039	.045	.004	-.068	.362
.379							.043			.379
.396	-.016	-.074	-.054	-.027	.023	.038	.046	.001	-.072	.396
.408							.044			.408
.426	.045	-.035	.002	.038	.116	.144	.153	.087	-.033	.426
.451	.053	-.031	.004	.048	.135	.155	.168	.098	-.026	.451
.476	.042	-.028	.007	.050	.134	.160	.173	.106	-.022	.476
.501	.031		.009	.053	.134	.160	.174		-.020	.501
.530	-.033	-.072	-.052	-.026	.025		.048	.006	-.072	.530
.548							.049			.548
.566							.049			.566
.584	-.034	-.077	-.055	-.028	.022	.039	.047	.005	-.076	.584
.602							.048			.602
.620							.046			.620
.638	-.025	-.078		-.028	.021		.047		-.078	.638
.655							.047			.655
.673							.046			.673
.691	-.024	-.081	-.059		.016	.035		.001	-.082	.691
.709							.045			.709
.727							.047			.727
.745	-.025	-.071		-.033	.021		.046		-.083	.745
.763							.046			.763
.781							.041			.781
.799							.041			.799
.817	-.026	-.073	-.061	-.034	.020	.038	.046	-.003	-.074	.817
.834							.043			.834
.870							.044			.870
.888							.043			.888
.921	-.112						.044			.921
.924	-.026	-.052		-.039			.038			.924
.947							.039			.947
.964							.040			.964
.981	-.018	-.046	-.067		.015	.033	.040	-.009	-.041	.981
.984					.017				-.050	.984
.998							.042			.998
1.000	-.114									1.000

Orifice station, y, in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.023	.872	.031	.457	.108
.030	.239	1.192	.094	.301	.415
.050	.807	1.212	.156	.902	.963
.070	.937	1.223	.219	.883	1.173
.090	.994	1.224	.281	.888	1.331
.110	.995	1.201	.344	.883	1.358
.130	.988	1.119	.406	.879	1.365
.150	.990	1.126	.469	.882	1.372
.170	.983	1.142	.750	.865	
.190	.943	1.136	1.000	.874	
			1.250	.907	
			1.500	.890	

TABLE 19. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 3.96$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	135	150	165	180	225	
.042								1.751		.042
.043								1.653		.043
.044								1.514		.044
.046								1.241		.046
.049								.443		.049
.053								.470		.053
.060	.188	.069	.050	.037	.012	.004	.001	.025		.060
.075	.233	.096	.076	.055	.026	.017	.013	.036		.075
.085	.251	.115	.098	.067	.037	.029	.026	.048	.100	.085
.096	.290									.096
.126		.088	.067	.048	.029	.027	.026	.034	.084	.126
.143								.028		.143
.154	.070	-.005	-.017	-.023	-.029		-.028	-.029	-.005	.154
.171								.027		.171
.189								.026		.189
.206	.056	-.013	-.023	-.028	-.026	-.026	-.025	-.028	-.013	.206
.223								.024		.223
.241								.022		.241
.258	.048	-.021	-.028	-.032	-.026	-.024	-.020	-.031	-.011	.258
.275								.018		.275
.293								.015		.293
.310	.045	-.028	-.033	-.032	-.030	-.019	-.011	-.033	-.032	.310
.327								.009		.327
.344								.010		.344
.362	.042	-.031	-.036	-.033	-.033	-.014	-.010	-.034	-.034	.362
.379								.009		.379
.396	.041	-.033	-.036	-.033	-.031	-.014	-.009	-.034	-.037	.396
.408								.009		.408
.426	.122	-.015	-.007	-.023	-.014	.023	.023	-.015	-.017	.426
.451	.143	-.009	-.023	-.013	.005	.023	.028	-.011	-.009	.451
.476	.153	-.004	-.021	-.012	-.010	.015	.023	-.005	-.006	.476
.501	.160		.019	-.010	-.010	.011	.016		.004	.501
.530	.055							.022	-.023	.530
.548								.024		.548
.566								.023		.566
.584	.047	-.036	-.037	-.035	-.027	-.023	-.024	-.027	-.037	.584
.602								.023		.602
.620								.021		.620
.638	.045	-.036			-.032	-.024	-.020		-.039	.638
.655								.019		.655
.673								.019		.673
.691	.045	-.040	-.035			-.028	-.024	-.034	-.040	.691
.709								.019		.709
.727								.019		.727
.745	.045	-.020			-.032	-.031	-.020		-.039	.745
.763								.021		.763
.781								.020		.781
.799								.020		.799
.817	.044	-.038	-.032	-.030	-.032	-.025	-.020	-.030	-.039	.817
.834								.020		.834
.870								.019		.870
.888								.020		.888
.921	-.058							.020		.921
.924	.043	-.036						.022		.924
.947								.019		.947
.964								.020		.964
.981								.020		.981
.998								.022		.998
1.000	-.060									1.000

Orifice station, y/t	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y/t	Cp top	Cp side	
.010	.872	.808	.031	.640	.030	
.030	1.377	1.206	.094	1.583	.318	
.050	1.528	1.212	.156	1.986	.647	
.070	1.559	1.219	.219	2.158	.828	
.090	1.577	1.206	.281	2.178	.948	
.110	1.603	1.143	.344	2.167	1.027	
.130	1.615	1.107	.406	2.148	1.061	
.150	1.630	1.127	.469	2.158	1.094	
.170	1.628	1.150	.750	2.148		
.190	1.575	1.134	1.000	2.164		
			1.250	2.175		
			1.500	2.046		

TABLE 19. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.775			.042
.043							1.705			.043
.044							1.573			.044
.046							1.327			.046
.049							.545			.049
.053							.594			.053
.060	.140	.072	.059	.049	.032	.026	.023	.042		.060
.075	.175	.097	.085	.065	.048	.040	.038	.056		.075
.085	.191	.116	.108	.085	.062	.056	.054	.072	.109	.085
.096	.225									.096
.126		.090	.076	.061	.048	.046	.045	.052	.090	.126
.143							.027			.143
.154	.036	-.003	-.010	-.017	-.025		.026	-.021	.000	.154
.171							.024			.171
.189							.023			.189
.206	.031	-.010	-.017	-.019	-.023	-.020		-.020	-.009	.206
.223							.020			.223
.241							.018			.241
.258	.022	-.015	-.022	-.022	-.019	-.019	.016	-.020	-.002	.258
.275							.014			.275
.293							.014			.293
.310	.017	-.022	-.025	-.022	-.021	-.018	.012	-.020	-.019	.310
.327							.011			.327
.344							.010			.344
.362	.015	-.024	-.028	-.026	-.022	-.015	.010	-.019	-.024	.362
.379							.009			.379
.396	.012	-.026	-.030	-.028	-.019	-.012	.009	-.020	-.025	.396
.408							.009			.408
.426	.073	.003	.015	-.003	.016	.024	.024	.011	.004	.426
.451	.081	.009	.004	.004	.032	.027	.030	.019	.009	.451
.476	.089	.010	.003	.006	.018	.023	.024	.019	.009	.476
.501	.093		.003	.006	.015	.018	.021		.011	.501
.530	.026	-.025	-.030	-.026	-.018		.015	-.019	-.025	.530
.548							.017			.548
.566							.016			.566
.584	.018	-.029	-.030	-.028	-.022	-.019	.016	-.018	-.027	.584
.602							.014			.602
.620							.013			.620
.638	.016	-.032			-.025	-.017			-.029	.638
.655							.012			.655
.673							.011			.673
.691	.015	-.033	-.032			-.016	-.015		-.018	.691
.709							.009			.709
.727							.009			.727
.745	.015	-.014			-.025	-.016			-.032	.745
.763							.008			.763
.781							.009			.781
.799							.009			.799
.817	.015	-.036	-.027	-.022	-.016	-.014	.008	-.015	-.032	.817
.834							.009			.834
.870							.008			.870
.888							.008			.888
.921	-.061							.008		.921
.924	.015	-.033			-.018			.009		.924
.947								.009		.947
.964								.009		.964
.981	.019	-.029	-.021				-.013	-.009	-.013	.981
.984										.984
.998										.998
1.000	-.061									1.000

Orifice station, y , in.	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side	
.010	.719	.831	.031	.410	.093	
.030	1.271	1.202	.094	.970	.368	
.050	1.430	1.220	.156	1.283	.599	
.070	1.451	1.231	.219	1.576	.753	
.090	1.459	1.218	.281	1.679	.922	
.110	1.473	1.153	.344	1.723	1.043	
.130	1.479	1.117	.406	1.759	1.111	
.150	1.486	1.137	.469	1.783	1.157	
.170	1.470	1.159	.750	1.830		
.190	1.415	1.142	1.000	1.853		
			1.250	1.862		
			1.500	1.757		

TABLE 19. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.788			.042
.043							1.741			.043
.044							1.636			.044
.046							1.389			.046
.049							.667			.049
.053							.721			.053
.060	.102	.072	.065	.062	.054	.052	.049	.060		.060
.075	.133	.097	.092	.081	.073	.068	.067	.075		.075
.085	.146	.116	.116	.101	.089	.086	.085	.095	.110	.085
.096	.176									.096
.126		.090	.083	.076	.066	.064	.064	.069	.090	.126
.143							.018			.143
.154	.016	-.001	-.005	-.008	-.014		-.016	-.011	.000	.154
.171							.016			.171
.189							.015			.189
.206	.011	-.006	-.010	-.012	-.013	-.014	-.014	-.012	-.005	.206
.223							.014			.223
.241							.013			.241
.258	.006	-.009	-.012	-.013	-.013	-.014	-.013	-.013	.002	.258
.275							.012			.275
.293							.012			.293
.310	.001	-.012	-.014	-.014	-.013	-.013	-.012	-.012	-.009	.310
.327							.012			.327
.344							.010			.344
.362	.001	-.014	-.016	-.014	-.012	-.011	-.009	-.011	-.011	.362
.379							.007			.379
.396	.000	-.009	-.010	-.010	-.008	-.005	-.007	-.012	-.012	.396
.408							.004			.408
.426	.045	.012	.018	.002	.004	.012	.018	.024	.015	.426
.451	.052	.026	.021	.019	.034	.024	.025	.029	.029	.451
.476	.054	.028	.025	.023	.026	.025	.027	.027	.030	.476
.501	.057		.025	.023	.023	.023	.024		.032	.501
.530	.006	-.014	-.019	-.019	-.014		.013	-.011	-.013	.530
.548							.013			.548
.566							.012			.566
.584	.002	-.016	-.018	-.017	-.014	-.014	-.012	-.013	-.012	.584
.602							.010			.602
.620							.010			.620
.638	.001	-.016			-.014	-.013				.638
.655							.009			.655
.673							.008			.673
.691	.000	-.017	-.018		-.011	-.010				.691
.709							.008			.709
.727							.007			.727
.745	.000	.0			-.013	-.010				.745
.763							.006			.763
.781							.005			.781
.799							.006			.799
.817	.000	-.017	-.014	-.011	-.008	-.008	-.005	-.007	-.011	.817
.834							.006			.834
.870							.005			.870
.888							.005			.888
.921	-.059						.005			.921
.924	.000	-.014			-.010					.924
.947							.006			.947
.964							.005			.964
.981	.005	-.012	-.010		-.008	-.007	-.004	-.005	-.009	.981
.984							.005		-.011	.984
.998							.008			.998
1.000	-.059									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp _{top}	Cp _{side}	Orifice station, y , in.	Cp _{top}	Cp _{side}
.010	.540	.822	.031	.232	.133
.030	1.168	1.202	.094	.489	.274
.050	1.338	1.217	.156	.654	.383
.070	1.345	1.231	.219	.824	.477
.090	1.351	1.220	.281	.967	.569
.110	1.360	1.166	.344	1.059	.668
.130	1.352	1.123	.406	1.130	.781
.150	1.342	1.137	.469	1.200	.869
.170	1.322	1.148	.750	1.410	
.190	1.273	1.139	1.000	1.513	
			1.250	1.577	
			1.500	1.524	

TABLE 19.- PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.779			.042
.043							1.750			.043
.044							1.669			.044
.046							1.439			.046
.049							.814			.049
.053							.829			.053
.060	.077	.074	.072	.075	.076	.077	.076	.081		.066
.075	.102	.098	.102	.098	.102	.100	.101	.099		.075
.085	.115	.117	.125	.117	.115	.115	.114	.116	.110	.085
.096	.139									.096
.126	.090	.091	.091	.091	.091	.091	.092	.091	.092	.126
.143								.003		.143
.154	.001	-.001	-.002	-.001	-.003			-.001	-.002	-.000
.171										.171
.189										.189
.206	-.002	-.005	-.006	-.005	-.005	-.006	-.005	-.005	-.005	.206
.223										.223
.241										.241
.258	-.005	-.007	-.007	-.006	-.007	-.007	-.007	-.007	.002	.258
.275										.275
.293										.293
.310	-.006	-.008	-.008	-.007	-.008	-.008	-.008	-.007	-.007	.310
.327										.327
.344										.344
.362	-.007	-.008	-.009	-.008	-.008	-.008	-.008	-.008	-.008	.362
.379										.379
.396	-.007	-.0	.000	-.000	-.003	-.005	-.008	-.008	-.009	.396
.408										.408
.426	.032	.004	.018	.005	.013	.022	.035	.024	.028	.426
.451	.038	.024	.024	.026	.046	.034	.033	.035	.034	.451
.476	.035	.029	.029	.029	.033	.034	.033	.036	.035	.476
.501	.038	.029	.029	.032	.032	.034	.034	.037	.037	.501
.530	-.004	-.012	-.012	-.010	-.011					.530
.548										.548
.566										.566
.584	-.007	-.009	-.010	-.009	-.009	-.009	-.008	-.008	-.008	.584
.602										.602
.620										.620
.638	-.007	-.008		-.006	-.008					.638
.655										.655
.673										.673
.691	-.007	-.009	-.009		-.007	-.006				.691
.709										.709
.727										.727
.745	-.007	.005		-.008	-.007					.745
.763										.763
.781										.781
.799										.799
.817	-.006	-.008	-.007	-.008	-.005	-.006	-.006	-.005	-.003	.817
.834										.834
.870										.870
.888										.888
.921	-.057	-.004	-.007		-.008					.921
.924										.924
.947										.947
.964										.964
.981	-.000	-.005	-.006			-.005	-.005	-.004	-.001	.981
.984									-.004	.984
.998										.998
1.000	-.057									1.000

Orifice station, y_{in}	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y_{in}	Cp top	Cp side
.010	.397	.816	.031	.131	.131
.030	1.051	1.188	.094	.247	.247
.050	1.243	1.210	.156	.315	.326
.070	1.246	1.224	.219	.384	.409
.090	1.246	1.210	.281	.436	.478
.110	1.250	1.163	.344	.503	.545
.130	1.239	1.110	.406	.579	.632
.150	1.226	1.134	.469	.642	
.170	1.188	1.148	.750	.870	
.190	1.147	1.129	1.000	1.016	
			1.250	1.078	
			1.500	1.103	

TABLE 19. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 3.96$ - Continued(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	130	165	180	225	270	
.042							1.776			.042
.043							1.768			.043
.044							1.711			.044
.046							1.503			.046
.049							.992			.049
.053							.915			.053
.060	.050	.072	.077	.087	.099	.102	.103	.100		.060
.075	.069	.097	.109	.112	.130	.130	.132	.120		.075
.085	.085	.116	.131	.130	.141	.147	.145	.136	.108	.085
.096	.109									.096
.126		.089	.097	.106	.117	.121	.123	.115	.090	.126
.143							.011			.143
.154	-.012	-.003	-.000	.003	.008		.013	.005	-.000	.154
.171							.010			.171
.189							.009			.189
.206	-.010	-.006	-.005	-.001	.002	.003	.007	.001	-.006	.206
.223							.005			.223
.241							.004			.241
.258	-.014	-.009	-.006	-.004	.000	.001	.003	-.001	.002	.258
.275							.002			.275
.293							.002			.293
.310	-.014	-.011	-.009	-.006	-.003	-.001	.000	-.004	-.010	.310
.327							.001			.327
.344							.002			.344
.362	-.010	-.012	-.011	-.009	-.006	-.003	-.003	-.006	-.012	.362
.379							.003			.379
.396	-.008	-.006	-.006	-.005	-.003	-.003	-.003	-.004	-.010	.396
.408							.003			.408
.426	.027	.004	.021	.014	.026	.031	.037	.022	.009	.426
.451	.029	.023	.027	.034	.056	.047	.049	.041	.026	.451
.476	.025	.027	.030	.036	.047	.050	.050	.043	.030	.476
.501	.024		.031	.036	.046	.051	.052		.031	.501
.530	-.010	-.014	-.011	-.008	-.002		.000	-.004	-.012	.530
.548							.000			.548
.566							.001			.566
.584	-.012	-.013	-.011	-.009	-.004	-.004	-.001	-.003	-.012	.584
.602							.000			.602
.620							.001			.620
.638	-.010	-.012			-.008	-.005			-.010	.638
.655							.000			.655
.673							.001			.673
.691	-.009	-.014	-.014			-.005	-.005		-.011	.691
.709							.001			.709
.727							.001			.727
.745	-.007	.0			-.011	-.005			-.011	.745
.763							.001			.763
.781							.002			.781
.799							.002			.799
.817	-.006	-.014	-.014	-.011	-.005	-.004	-.001	-.006	-.010	.817
.834							.001			.834
.870							.001			.870
.888							.000			.888
.921	-.058						.000			.921
.924	-.005	-.012					.005			.924
.947							.001			.947
.964							.001			.964
.981	-.001	-.010	-.012			-.005	-.001	-.003	-.008	.981
.984							.005		-.010	.984
.998								-.002		.998
1.000	-.059									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.261	.815	.031	.088	.142
.030	.954	1.195	.094	.151	.308
.050	1.180	1.209	.156	.200	.448
.070	1.162	1.226	.219	.249	.556
.090	1.161	1.211	.281	.283	.683
.110	1.161	1.150	.344	.310	.786
.130	1.153	1.112	.406	.348	.875
.150	1.139	1.133	.469	.377	.929
.170	1.119	1.146	.750	.507	
.190	1.066	1.128	1.000	.647	
			1.250	.748	
			1.500	.730	

TABLE 19. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 3.96$ - Continued(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.769			.042
.043							1.784			.043
.044							1.746			.044
.046							1.559			.046
.049							1.139			.049
.053							.998			.053
.060	.025	.071	.085	.103	.131	.138	.140	.124		.060
.075	.040	.097	.118	.130	.166	.174	.176	.149		.075
.085	.054	.116	.140	.150	.180	.190	.191	.166	.109	.085
.096	.077									.096
.126		.089	.106	.123	.150	.159	.163	.143	.090	.126
.143							.034			.143
.154	-.022	-.003	.003	.013	.027		.036	.020	-.000	.154
.171							.032			.171
.189							.029			.189
.206	-.018	-.009	-.002	.006	.021	.025	.028	.014	-.009	.206
.223							.025			.223
.241							.023			.241
.258	-.015		-.008	.000	.014	.018	.020	.007	-.003	.258
.275							.018			.275
.293							.016			.293
.310	-.014	-.019	-.013	-.006	.008	.013	.015	.002	-.019	.310
.327							.014			.327
.344							.014			.344
.362	-.011	-.021	-.016	-.008	.005	.011	.013	-.000	-.023	.362
.379							.013			.379
.396	-.009	3.727	-.018	-.010	.005	.010	.012	-.001	-.025	.396
.408							.015			.408
.426	.021	-.0	.025	.025	.054	.063	.069	.041	-.001	.426
.451	.027	.008	.021	.035	.077	.075	.081	.053	.009	.451
.476	.023	.012	.023	.038	.074	.083		.058	.012	.476
.501	.021		.025	.042	.076	.089		.013	.013	.501
.530	-.013	-.023	-.016	-.006	.014		.023	.005	-.023	.530
.548							.020			.548
.566							.018			.566
.584	-.017	-.026	-.019	-.010	.007	.013	.016	.000	-.024	.584
.602							.016			.602
.620							.016			.620
.638	-.012	-.027			.010	.005				.638
.655							.016			.655
.673							.015			.673
.691	-.011	-.030	-.023		.002	.009		-.002	-.028	.691
.709							.014			.709
.727							.014			.727
.745	-.009	-.011			-.014	.002			-.028	.745
.763							.015			.763
.781							.015			.781
.799							.014			.799
.817	-.008	-.031	-.024	-.016	.004	.012	.015	-.002	-.029	.817
.834							.015			.834
.870							.015			.870
.888							.015			.888
.921	-.059	-.009	-.031		-.017					.921
.924										.924
.947										.947
.964										.964
.981	-.005	-.026	-.027	*		.003	.013	.014	-.002	.981
.998						.003			-.029	.998
1.000	-.061							.014	-.029	1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.120	.813	.031	.164	.075
.030	.741	1.192	.094	.328	.376
.050	1.092	1.212	.156	.408	.631
.070	1.067	1.223	.219	.450	.779
.090	1.058	1.206	.281	.481	.929
.110	1.058	1.143	.344	.499	1.023
.130	1.052	1.123	.406	.520	1.072
.150	1.047	1.138	.469	.529	1.123
.170	1.027	1.150	.750	.589	
.190	*.980	1.130	1.000	.627	
			1.250	.665	
			1.500	.653	

TABLE 19 - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 3.96$ - Concluded(g) $\alpha = 10.1^\circ$

Model station, $\frac{X}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{X}{T}$
	0	90	105	120	150	165	180	225	270	
.042							1.738			.042
.043							1.794			.043
.044							1.778			.044
.046							1.646			.046
.049							1.280			.049
.053							1.083			.053
.060	-.001	.069	.091	.122	.172	.187	.194	.156		.060
.075	.012	.096	.128	.153	.217	.232	.237	.186		.075
.085	.028	.115	.150	.176	.234	.252	.253	.205	.103	.085
.096	.043									.096
.126		.087	.115	.145	.200	.217	.224	.180	.087	.126
.143							.069			.143
.154	-.029	-.004	.007	.026	.057		.075	.045	-.004	.154
.171							.069			.171
.189							.064			.189
.206	-.024	-.013	-.001	.015	.047	.055	.060	.032	-.011	.206
.223							.056			.223
.241							.054			.241
.258	-.017	-.022	-.010	.005	.037	.048	.052	.022	-.011	.258
.275							.051			.275
.293							.049			.293
.310	-.010	-.029	-.017	-.001	.031	.043	.048	.017	-.030	.310
.327							.047			.327
.344							.046			.344
.362	-.010	-.034	-.022	-.004	.029	.041	.044	.015	-.037	.362
.379							.045			.379
.396	-.009	-.036	-.024	-.007	.028	.041	.044	.013	-.039	.396
.408							.047			.408
.426	.021	-.015	.019	.035	.095	.118	.126	.072	-.016	.426
.451	.026	-.010	.015	.047	.124	.141	.151	.088	-.010	.451
.476	.022	-.007	.019	.053	.127	.150	.161	.097	-.005	.476
.501	.017		.024	.057	.131	.153	.165		-.003	.501
.530	-.018	-.034	-.021	.000	.040		.058	.022	-.035	.530
.548							.054			.548
.566							.055			.566
.584	-.025	-.039	-.024	-.005	.034	.045	.052	.017	-.039	.584
.602							.051			.602
.620							.049			.620
.638	-.020	-.039		-.007	.030		.051		-.040	.638
.655							.050			.655
.673							.050			.673
.691	-.019	-.042	-.028		.027	.042		.014	-.043	.691
.709							.049			.709
.727							.049			.727
.745	-.019	-.020		-.010	.028		.050		-.043	.745
.763							.048			.763
.781							.047			.781
.799							.047			.799
.817	-.019	-.042	-.030	-.013	.028	.043	.051	.013	-.042	.817
.834							.049			.834
.870							.048			.870
.888							.047			.888
.921	-.063						.047			.921
.924	-.021	-.039		-.015			.043			.924
.947							.044			.947
.964							.044			.964
.981	-.019	-.034	-.033		.024	.040	.044	.010	-.040	.981
.984							.044		-.041	.984
.998							.044			.998
1.000	-.062									1.000

Orifice station, $y_{in.}$	Nose rake			Base rake		
	Cp top	Cp side	Orifice station, $y_{in.}$	Cp top	Cp side	
.010	.050	.815	.031	.280	.037	
.030	.269	1.197	.094	.672	.350	
.050	.841	1.217	.156	.598	.656	
.070	.964	1.217	.219	.580	.788	
.090	.938	1.199	.281	.584	.920	
.110	.933	1.121	.344	.575	1.016	
.130	.933	1.114	.406	.566	1.065	
.150	.926	1.132	.469	.560	1.109	
.170	.908	1.148	.750	.538		
.190	.882	1.134	1.000	.546		
			1.250	.580		
			1.500	.593		

TABLE 20. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 4.65$

(a) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.828			.042
.043							1.727			.043
.044							1.598			.044
.046							1.381			.046
.049							.572			.049
.053							.600			.053
.060	.159	.099	.085	.076	.059	.052	.050	.063		.060
.075	.181	.116	.105	.094	.071	.061	.058	.071		.075
.085	.191	.130	.129	.104	.081	.075	.070	.084	.111	.085
.096	.223									.096
.126		.098	.085	.074	.059	.056	.057	.059	.088	.126
.143							-.000			.143
.154	.041	.013	.009	.005	-.000		-.001	-.003	.009	.154
.171							-.001			.171
.189							-.001			.189
.206	.032	.006	.002	-.000	-.000	-.003	.001	-.003	.000	.206
.223							-.000			.223
.241							-.000			.241
.258	.023	.001	-.000	-.004	-.001	-.002	.000	-.005	.016	.258
.275							-.001			.275
.293							-.001			.293
.310	.020	-.003	-.008	-.002	-.002	-.002	-.000	-.006	-.008	.310
.327							-.000			.327
.344							-.000			.344
.362	.016	-.006	-.007	-.004	-.001	-.002	-.000	-.007	-.010	.362
.379							-.000			.379
.396	.016	-.003	-.002	-.001	-.000	-.001	-.000	-.005	-.011	.396
.408							-.001			.408
.426	.068	.014	.030	.002	.002	.007	.021	.002	.010	.426
.451	.067	.022	.012	.007	.035	.012	.024	.012	.013	.451
.476	.072	.021	.015	.011	.014	.020	.021	.017	.015	.476
.501	.077		.015	.013	.016	.019	.017		.013	.501
.530	.026	-.005	-.007	-.007	.001		-.005	-.007	-.009	.530
.548							-.009			.548
.566							-.007			.566
.584	.019	-.007	-.008	-.007	-.003	-.006	-.007	-.008	-.011	.584
.602							-.006			.602
.620							-.006			.620
.638	.017	-.009		-.001	-.005		-.006		-.013	.638
.655							-.004			.655
.673							-.005			.673
.691	.016	-.011	-.009		-.005	-.002	-.005	-.007	-.014	.691
.709							-.005			.709
.727							-.003			.727
.745	.016	.018		-.006	-.004		-.003		-.014	.745
.763							-.003			.763
.781							-.004			.781
.799							-.003			.799
.817	.016	-.012	-.009	-.007	-.002	-.001	-.003	-.006	-.013	.817
.834							-.004			.834
.870							-.002			.870
.888							-.002			.888
.921	-.032						-.003			.921
.924	.016	-.011		-.007			-.005			.924
.947							-.005			.947
.964							-.001			.964
.981	.020	-.009	-.009		-.004	-.009	-.003	-.008	-.012	.981
.984					-.004		-.003	-.008	-.013	.984
.998							-.006			.998
1.000	-.032									1.000

Orifice station, y.in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y.in.	Cp top	Cp side
.010	.588	.759	.031	.358	.060
.030	1.229	1.220	.094	.793	.265
.050	1.439	1.243	.156	1.123	.454
.070	1.450	1.236	.219	1.455	.554
.090	1.452	1.192	.281	1.611	.692
.110	1.443	1.127	.344	1.692	.814
.130	1.445	1.130	.406	1.750	.899
.150	1.445	1.146	.469	1.800	.971
.170	1.445	1.146	.750	1.865	
.190	1.392	1.153	1.000	1.904	
			1.250	1.923	
			1.500	1.777	

TABLE 20. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 4.65$ - Continued

(b) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.836			.042
.043							1.766			.043
.044							1.667			.044
.046							1.430			.046
.049							.685			.049
.053							.724			.053
.060	.122	.098	.090	.086	.078	.073	.070	.078		.060
.075	.143	.117	.114	.101	.091	.086	.083	.089		.075
.085	.150	.130	.136	.117	.103	.101	.098	.103	.109	.085
.096	.176									.096
.126		.096	.091	.084	.074	.071	.069	.071	.089	.126
.143							.002			.143
.154	.024	.012	.007	.005	.001		.000	.000	.010	.154
.171							.000			.171
.189							.000			.189
.206	.016	.004	.001	.001	-.000	-.002	-.000	-.000	.003	.206
.223							.000			.223
.241							.002			.241
.258	.013	.001	-.000	-.003	-.002	-.004	-.003	-.003	.019	.258
.275							.003			.275
.293							.003			.293
.310	.008	-.003	-.005	-.003	-.005	-.003	-.003	-.003	-.001	.310
.327							.002			.327
.344							.000			.344
.362	.005	-.005	-.005	-.003	-.002	-.000	.000	-.002	-.004	.362
.379							.001			.379
.396	.005	-.0	-.000	.001	.001	.000	.001	-.001	-.002	.396
.408							.002			.408
.426	.040	.010	.031	.004	.003	.003	.008	.021	.010	.426
.451	.044	.022	.018	.012	.039	.008	.015	.020	.022	.451
.476	.046	.024	.021	.018	.014	.014	.018	.019	.023	.476
.501	.049		.022	.018	.016	.016	.019		.024	.501
.530	.010	-.007	-.008	-.009	-.007		-.006	-.006	-.007	.530
.548							.006			.548
.566							.008			.566
.584	.007	-.009	-.008	-.009	-.007	-.009	-.008	-.007	-.009	.584
.602							.006			.602
.620							.006			.620
.638	.005	-.007		-.003	-.007					.638
.655							.005			.655
.673							.005			.673
.691	.004	-.008	-.008		-.007	-.006		-.005	-.009	.691
.709							.004			.709
.727							.004			.727
.745	.004	.018		-.009	-.006		.003		-.009	.745
.763							.003			.763
.781							.003			.781
.799							.003			.799
.817	.005	-.010	-.008	-.009	-.004	-.005	-.002	-.004	-.008	.817
.834							.003			.834
.870							.003			.870
.888							.002			.888
.921	-.035						-.003			.921
.924	.006	-.010		-.009			-.004			.924
.947							-.004			.947
.964							-.001			.964
.981	.006	-.006	-.008		-.005	-.007	-.003	-.003	-.007	.981
.984							-.006		-.008	.984
.998										.998
1.000	-.033									1.000

Orifice station, y.in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y.in.	Cp top	Cp side
.010	.402	.731	.031	.171	.081
.030	1.107	1.209	.094	.369	.210
.050	1.328	1.222	.156	.508	.309
.070	1.322	1.213	.219	.630	.390
.090	1.315	1.188	.281	.768	.473
.110	1.315	1.116	.344	.853	.561
.130	1.305	1.119	.406	.923	.664
.150	1.294	1.137	.469	.982	
.170	1.268	1.149	.750	1.227	
.190	1.229	1.130	1.000	1.372	
			1.250	1.499	
			1.500	1.462	

TABLE 20. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 4.65$ - Continued

(c) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.833			.042
.043							1.801			.043
.044							1.707			.044
.046							1.478			.046
.049							.833			.049
.053							.842			.053
.060	.095	.099	.097	.100	.100	.099	.097	.098		.060
.075	.114	.117	.122	.117	.119	.117	.115	.110	.111	.075
.085	.123	.131	.143	.129	.128	.126	.124	.122		.085
.096	.142									.096
.126		.095	.096	.097	.096	.095	.094	.089	.088	.126
.143							.010			.143
.154	.012	.012	.012	.011	.009		.008	.008	.009	.154
.171							.007			.171
.189							.005			.189
.206	.006	.005	.004	.005	.004	.003	.004	.004	.005	.206
.223							.002			.223
.241							.001			.241
.258	.003	.0	.002	.001	.000	.001	.001	.000	.021	.258
.275							.000			.275
.293							.000			.293
.310	.001	-.0	-.000	.000	-.000	-.001	-.000	-.000	-.000	.310
.327							-.000			.327
.344							-.000			.344
.362	-.000	-.003	-.003	-.001	-.000	-.002	.001	-.000	-.001	.362
.379							.003			.379
.396	.001	.004	.004	.005	.006	.003	.004	.004	-.001	.396
.408							.006			.408
.426	.028	.008	.034	.008	.008	.008	.007	.012	.025	.426
.451	.032	.021	.021	.017	.046	.017	.019	.022	.027	.451
.476	.029	.027	.027	.027	.026	.024	.025	.026	.026	.476
.501	.034		.025	.025	.025	.024	.026		.025	.501
.530	.004	-.003	-.005	-.006	-.006		-.004	-.003	-.005	.530
.548							.005			.548
.566							.005			.566
.584	-.001	-.006	-.005	-.006	-.005	-.005	-.003	-.003	-.007	.584
.602							.003			.602
.620							.004			.620
.638	-.001	-.006			-.001	-.006			-.006	.638
.655							.004			.655
.673							.003			.673
.691	-.001	-.006	-.005			-.007	-.003		-.005	.691
.709							.003			.709
.727							.003			.727
.745	-.001	.021			-.004	-.007			-.005	.745
.763							.003			.763
.781							.003			.781
.799							.003			.799
.817	-.001	-.005	-.005	-.004	-.003	-.001	-.003	-.003	-.004	.817
.834							.003			.834
.870							.000			.870
.888							-.003			.888
.921	-.035						-.003			.921
.924	-.001	-.005			-.004					.924
.947							-.003			.947
.964							-.001			.964
.981	.001	-.002	-.005			-.004	-.004	-.002	-.003	.981
.984									-.004	.984
.998										.998
1.000	-.034									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.277	.750	.031	.097	.118
.030	1.017	1.241	.094	.178	.240
.050	1.266	1.238	.156	.229	.339
.070	1.243	1.236	.219	.291	.418
.090	1.227	1.206	.281	.332	.471
.110	1.229	1.137	.344	.383	.531
.130	1.209	1.128	.406	.438	.593
.150	1.190	1.137	.469	.491	.641
.170	1.165	1.158	.750	.699	
.190	1.123	1.146	1.000	.812	
			1.250	.948	
			1.500	.923	

TABLE 20. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 4.65$ - Continued(d) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.042							1.844				.042
.043							1.828				.043
.044							1.736				.044
.046							1.547				.046
.049							.989				.049
.053							.941				.053
.060	.069	.098	.102	.113	.124	.127	.126	.117			.060
.075	.081	.116	.129	.132	.145	.146	.146	.130			.075
.085	.095	.130	.148	.143	.152	.155	.152	.140	.110		.085
.096	.116										.096
.126		.096	.104	.112	.121	.124	.124	.111	.088		.126
.143							.024				.143
.154	.000	.010	.014	.016	.020		.022	.016	.011		.154
.171							.019				.171
.189							.017				.189
.206	-.000	.004	.006	.010	.013	.013	.015	.009	.002		.206
.223							.013				.223
.241							.011				.241
.258	-.003	.0	.003	.004	.009	.010	.010	.005	.018		.258
.275							.008				.275
.293							.007				.293
.310	-.003	-.002	-.000	.003	.004	.005	.005	.001	-.003		.310
.327							.004				.327
.344							.003				.344
.362	-.003	-.005	-.003	-.002	.001	.002	.002	-.000	-.005		.362
.379							.002				.379
.396	.001	-.0	.020	.001	.001	.002	.003	.000	-.002		.396
.408							.008				.408
.426	.017	.008	.037	.017	.022	.024	.024	.018	.019		.426
.451	.020	.021	.025	.028	.058	.040	.040	.032	.020		.451
.476	.022	.023	.027	.034	.043	.042	.043	.035	.021		.476
.501	.021		.029	.033	.042	.044	.043		.022		.501
.530	-.000	-.006	-.003	-.002	.002		.003	-.002	-.009		.530
.548							.004				.548
.566							.004				.566
.584	-.010	-.007	-.005	-.003	.002	.002	.003	-.002	-.009		.584
.602							.002				.602
.620							.002				.620
.638	-.008	-.007			-.000	.001					.638
.655							.002				.655
.673							.001				.673
.691	-.006	-.007	-.006			-.000	-.000				.691
.709											.709
.727											.727
.745	-.006	.020			-.005	-.001					.745
.763											.763
.781											.781
.799											.799
.817	-.003	-.008	-.008	-.005	.000	-.000	-.000	-.004	-.010		.817
.834											.834
.870											.870
.888											.888
.921	-.035										.921
.924	-.002	-.008			-.006						.924
.947											.947
.964											.964
.981	-.000	-.007	-.008			-.001	.000				.981
.984											.984
.998											.998
1.000	-.037							-.004	-.009		1.000

Orifice station, y, in.	Nose rake		Base rake		Orifice station, y, in.	
	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side	
.010	.173	.745	.031	.065	.046	
.030	.851	1.236	.094	.120	.289	
.050	1.194	1.245	.156	.164	.593	
.070	1.160	1.226	.219	.187	.687	
.090	1.146	1.201	.281	.203	.724	
.110	1.139	1.107	.344	.226	.752	
.130	1.120	1.125	.406	.256	.798	
.150	1.104	1.139	.469	.277		
.170	1.077	1.157	.750	.397		
.190	1.017	1.125	1.000	.473		
			1.250	.586		
			1.500	.604		

TABLE 20. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE IV WITH NATURAL TRANSITION AT $M = 4.65$ - Concluded

(e) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.042							1.821			.042
.043							1.839			.043
.044							1.784			.044
.046							1.618			.046
.049							1.151			.049
.053							1.031			.053
.060	.043	.096	.109	.128	.155	.161	.162	.141		.060
.075	.055	.115	.136	.146	.177	.182	.184	.153		.075
.085	.066	.128	.156	.161	.187	.195	.194	.165	.111	.085
.096	.086									.096
.126		.095	.111	.127	.151	.158	.158	.136	.087	.126
.143							.041			.143
.154	-.006	.009	.015	.024	.036		.041	.026	.006	.154
.171							.037			.171
.189							.035			.189
.206	-.007	.001	.006	.015	.028	.029	.032	.017	-.001	.206
.223							.029			.223
.241							.026			.241
.258	-.007	-.002	.003	.007	.019	.021	.023	.010	.014	.258
.275							.021			.275
.293							.020			.293
.310	-.007	-.008	-.003	.002	.012	.017	.019	.005	-.011	.310
.327							.018			.327
.344							.016			.344
.362	-.006	-.012	-.007	-.001	.010	.013	.015	.001	-.014	.362
.379							.014			.379
.396	-.001	-.010	-.000	-.001	.009	.012	.013	.000	-.011	.396
.408							.013			.408
.426	.018	.004	.036	.021	.041	.051	.052	.028	.002	.426
.451	.021	.012	.023	.033	.079	.067	.069	.040	.007	.451
.476	.018	.015	.024	.035	.057	.071	.077	.046	.010	.476
.501	.020		.026	.039	.070	.075	.083		.010	.501
.530	-.003	-.011	-.006	.000	.017		.024	.004	-.014	.530
.548							.023			.548
.566							.020			.566
.584	-.012	-.013	-.009	-.002	.011	.015	.017	.000	-.016	.584
.602							.017			.602
.620							.014			.620
.638	-.010	-.014		-.000	.009		.016		-.017	.638
.655							.015			.655
.673							.015			.673
.691	-.008	-.016	-.011		.008	.013		-.003	-.017	.691
.709							.014			.709
.727							.015			.727
.745	-.007	.012		-.008	.008		.015		-.018	.745
.763							.015			.763
.781							.014			.781
.799							.015			.799
.817	-.007	-.017	-.013	-.008	.008	.013	.015	-.003	-.018	.817
.834							.015			.834
.870							.013			.870
.888							.014			.888
.921	-.036						.014			.921
.924	-.006	-.017		-.008			.014			.924
.947							.014			.947
.964							.012			.964
.981	-.005	-.014	-.013		.007	.014	.013	-.003	-.016	.981
.984					.006				-.016	.984
.998							.009			.998
1.000	-.036									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.086	.747	.031	.083	.012
.030	.554	1.213	.094	.192	.440
.050	1.093	1.236	.156	.247	.754
.070	1.049	1.224	.219	.314	.835
.090	1.037	1.180	.281	.328	.892
.110	1.040	1.118	.344	.353	.948
.130	1.019	1.120	.406	.381	.991
.150	1.014	1.127	.469	.399	1.040
.170	.984	1.153	.750	.436	
.190	.945	1.127	1.000	.466	
			1.250	.514	
			1.500	.540	

TABLE 21. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 1.57$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.059							1.526				.059
.060							1.460				.060
.061							1.362				.061
.063							1.221				.063
.064							1.092				.064
.075	-.353	-.408	-.407	-.407	-.407	-.408	-.407	-.408	-.408		.075
.085	.273	-.120	-.189	-.273	-.404	-.406	-.407	-.344	-.134		.085
.096	.548		-.034	-.121	-.255	-.278	-.309	-.210	.008		.096
.119	.402	.123	.063	.014		-.111	-.133	-.044	.111		.119
.143							-.257				.143
.154	-.011	-.180	-.215	-.237	-.242	-.219	-.195	-.250			.154
.171							-.147				.171
.189							-.110				.189
.206	.027	-.164	-.190	-.207	-.123	-.096	-.084	-.157			.206
.223							-.057				.223
.241							-.041				.241
.258	.023	-.171	-.194	-.144	-.063	-.037	-.028	-.084	-.179		.258
.275							-.019				.275
.293							-.011				.293
.310	.019	-.179	-.143	-.082	-.048	-.019	-.006	-.057	-.182		.310
.327							-.002				.327
.344							-.000				.344
.362	.015	-.136	-.099	-.061	-.057	-.015	.003	-.052	-.129		.362
.379							.003				.379
.396	.009	-.115	-.088	-.036	-.052	-.009	.014	-.027	-.110		.396
.408							.018				.408
.426	.220	.075	.090	.079	.050	.163	.213	.077	.087		.426
.451	.207	.065	.059	.075	.042	.113	.139		.063		.451
.476	.198	.047	.041	.061	.029	.105	.111	.072	.049		.476
.501	.191	.038	.034	.043	.007	.076	.079	.039	.039		.501
.530	-.003	-.130	-.138	-.112	-.112	-.131	-.108	-.089	-.132		.530
.548							-.107				.548
.566							-.063				.566
.584	.025	-.121	-.107	-.090	-.083	-.064	-.054	-.081	-.124		.584
.602							-.046				.602
.620							-.037				.620
.638	.032	-.106		-.075	-.045				-.103		.638
.655							-.026				.655
.673							-.025				.673
.691	.031	-.124	-.101	-.060	-.043	-.045			-.106		.691
.709							-.024				.709
.727							-.020				.727
.745	.026	-.111		-.047	-.033				-.098		.745
.763							-.019				.763
.781							-.027				.781
.799							-.019				.799
.817	.044	-.089	-.056	-.033	-.030	-.025			-.077		.817
.834							-.020				.834
.852							-.010				.852
.870							-.009				.870
.888							.016				.888
.906							.040				.906
.921	-.170	.051	-.063		-.009	.004			-.084		.921
.924							.051				.924
.947							.039				.947
.964							.040				.964
.981	.058	-.177	-.013	.025		.006	.004		-.034		.981
1.000							.034				1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.095		.031	.869	.437
.030	1.182		.094	1.567	1.004
.050	1.197		.156	1.637	1.477
.070	1.249		.219	1.569	1.622
.090	1.302		.281	1.579	1.591
.110	1.354		.344	1.569	1.514
.130	1.396		.406	1.568	1.520
.150	1.425		.469	1.573	1.513
.170	1.435		.750	1.566	
.190	1.392		1.000	1.566	
			1.250	1.568	
			1.500	1.577	

TABLE 21. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.542			.059
.060						1.493				.060
.061						1.411				.061
.063						1.289				.063
.064						1.157				.064
.075	-0.402	-0.424	-0.423	-0.423	-0.422	-0.422	-0.422	-0.422	-0.422	.075
.086	.099	-0.124	-0.186	-0.238	-0.367	-0.381	-0.367	-0.324	-0.155	.085
.096	.391	-0.033	-0.087	-0.176	-0.192	-0.214	-0.146	-0.028	-0.028	.096
.119	.311	.129	.089	.053	.033	.049	.008	.115	.115	.119
.143										.143
.154	-0.070	-0.161	-0.179	-0.194	-0.211	-0.205	-0.195	-0.209		.154
.171										.171
.189										.189
.206	-0.026	-0.124	-0.137	-0.146	-0.105	-0.091	-0.085	-0.117		.206
.223										.223
.241										.241
.258	-0.021	-0.101	-0.099	-0.082	-0.051	-0.039	-0.037	-0.059	-0.102	.258
.275										.275
.293										.293
.310	-0.016	-0.073	-0.070	-0.053	-0.029	-0.018	-0.016	-0.036	-0.076	.310
.327										.327
.344										.344
.362	-0.015	-0.063	-0.051	-0.033	-0.020	-0.012	-0.007	-0.022	-0.062	.362
.379										.379
.396	-0.015	-0.053	-0.047	-0.037	-0.016	-0.009	.000	-0.019	-0.052	.396
.408										.408
.426	.192	.136	.138	.132	.127	.148	.162	.125	.139	.426
.451	.181	.118	.111	.109	.112	.118	.131	.114	.114	.451
.476	.172	.099	.089	.088	.090	.097	.105	.095	.100	.476
.501	.162	.087	.078	.071	.058	.077	.092	.061	.087	.501
.530	.031	-.095	-.103	-.106	-.080	-.088	-.113	-.076	-.096	.530
.548										.548
.566										.566
.584	-0.003	-0.075	-0.068	-0.068	-0.062	-0.057	-0.050	-0.060	-0.077	.584
.602										.602
.620										.620
.638	.001	-0.060		-0.052	-0.038					.638
.655										.655
.673										.673
.691	.002	-0.065	-0.058	-0.045	-0.028	-0.026				.691
.709										.709
.727										.727
.745	.012	-0.054		-0.035	-0.021					.745
.763										.763
.781										.781
.799										.799
.817	.017	-0.042	-0.035	-0.023	-0.013	-0.007				.817
.834										.834
.852										.852
.870										.870
.888										.888
.906										.906
.921	-0.129	.025	-.023		-0.002	.030				.921
.924										.924
.947										.947
.964										.964
.981	.050		.008	.029		.036	.039	.047	.027	.981
1.000	-.140								-.002	1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp _{top}	Cp _{side}	Orifice station, y , in.	Cp _{top}	Cp _{side}
.010	1.058		.031	.739	.503
.030	1.244		.094	1.270	.684
.050	1.312		.156	1.564	1.217
.070	1.348		.219	1.557	1.472
.090	1.382		.281	1.535	1.554
.110	1.413		.344	1.545	1.521
.130	1.438		.406	1.544	1.516
.150	1.453		.469	1.551	1.512
.170	1.447		.520	1.547	
.190	1.389		1.000	1.553	
			1.250	1.550	
			1.500	1.544	

TABLE 21. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.550			.059
.060							1.515			.060
.061							1.447			.061
.063							1.334			.063
.064							1.208			.064
.075	-0.397	-0.408	-0.408	-0.409	-0.409	-0.410	-0.411	-0.409	-0.408	.075
.085	-0.032	-0.136	-0.171	-0.203	-0.271	-0.270	-0.278	-0.246	-0.170	.085
.096	.231		-0.017	-0.044	-0.095	-0.108	-0.118	-0.088	-0.044	.096
.119	.230		.126	.105	.086	.041	.030	.055	.109	.119
.143							-0.216			.143
.154	-0.111	-0.146	-0.155	-0.166	-0.176	-0.176	-0.172	-0.178		.154
.171							-0.144			.171
.189							-0.115			.189
.206	-0.060	-0.093	-0.096	-0.108	-0.094	-0.092	-0.091	-0.096		.206
.223							-0.070			.223
.241							-0.060			.241
.258	-0.043	-0.059	-0.061	-0.061	-0.051	-0.045	-0.048	-0.053	-0.059	.258
.275							-0.037			.275
.293							-0.025			.293
.310	-0.029	-0.045	-0.041	-0.031	-0.023	-0.017	-0.017	-0.025	-0.045	.310
.327							-0.017			.327
.344							-0.015			.344
.362	-0.020	-0.025	-0.020	-0.018	-0.019	-0.015	-0.013	-0.018	-0.025	.362
.379							-0.012			.379
.396	-0.019	-0.023	-0.024	-0.021	-0.012	-0.009	-0.002	-0.014	-0.023	.396
.408							-0.023			.408
.426	.184	.164	.165	.162	.161	.164	.162	.162	.162	.426
.451	.166	.145	.141	.138	.134	.131	.135		.141	.451
.476	.149	.124	.115	.112	.112	.112	.113	.115	.126	.476
.501	.136	.108	.105	.098	.094	.094	.096	.096	.108	.501
.530	-0.051	-0.073	-0.077	-0.082	-0.079	-0.081	-0.081	-0.076	-0.077	.530
.548							-0.066			.548
.566							-0.051			.566
.584	-0.011	-0.049	-0.049	-0.051	-0.043	-0.043	-0.042	-0.048	-0.050	.584
.602							-0.042			.602
.620							-0.034			.620
.638	-0.015	-0.029		-0.030	-0.026				-0.027	.638
.655							-0.023			.655
.673							-0.021			.673
.691	-0.011	-0.035	-0.033	-0.033	-0.029	-0.024	-0.014	-0.017	-0.023	.691
.709							-0.010			.709
.727							-0.010			.727
.745	-0.004	-0.027		-0.021	-0.012		-0.004		-0.017	.745
.763							-0.007			.763
.781							-0.012			.781
.799							-0.004			.799
.817	.008	-0.011	-0.011	-0.011	-0.005	-0.002	.000	-0.004	.000	.817
.834							-0.002			.834
.852							.003			.852
.870							.011			.870
.888							.007			.888
.906										.906
.921	-0.073	.019	-0.002		-0.006	.005		.013		.921
.924							.008		-0.016	.924
.947							.012			.947
.964							.007			.964
.981	.035		.020	.007		.008	.009	.007	.007	.981
1.000	-0.079									1.000

Orifice station, y.in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y.in.	Cp top	Cp side
.010	.957		.031	.608	.504
.030	1.271		.094	.953	.750
.050	1.375		.156	1.227	.958
.070	1.426		.219	1.416	1.163
.090	1.463		.281	1.466	1.329
.110	1.480		.344	1.481	1.425
.130	1.466		.406	1.487	1.460
.150	1.467		.469	1.494	1.466
.170	1.434		.750	1.505	
.190	1.348		1.000	1.519	
			1.250	1.523	
			1.500	1.530	

TABLE 21. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 1.57$ - Continued(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.555			.059
.060							1.537			.060
.061							1.480			.061
.063							1.382			.063
.064							1.260			.064
.075	-.398	-.406	-.405	-.406	-.407	-.408	-.412	-.407	-.406	.075
.085	-.146	-.141	-.145	-.145	-.164	-.153	-.167	-.161	-.174	.085
.096	.073		.003		-.007	-.013	-.011	-.019	-.050	.096
.119	.149	.124	.124	.125		.119	.110	.107	.106	.119
.143							.186			.143
.154	-.146	-.139	-.145	-.146	-.150	-.149	-.143	-.155		.154
.171							.123			.171
.189							.100			.189
.206	-.081	-.083	-.081	-.089	-.080	-.080	-.081	-.082		.206
.223							.063			.223
.241							.056			.241
.258	-.055	-.045	-.047	-.046	-.050	-.049	-.052	-.046	-.044	.258
.275							.045			.275
.293							.037			.293
.310	-.028	-.033	-.033	-.026	-.027	-.021	-.022	-.026	-.035	.310
.327							.019			.327
.344							.020			.344
.362	-.018	-.016	-.012	-.011	-.021	-.018	-.017	-.017	-.015	.362
.379							.018			.379
.396	-.018	-.012	-.016	-.017	-.014	-.014	-.007	-.014	-.012	.396
.408							.011			.408
.426	.172	.173	.176	.175	.173	.177	.175	.173	.172	.426
.451	.148	.153	.152	.154	.149	.144	.147		.149	.451
.476	.130	.132	.128	.127	.128	.127	.127	.130	.135	.476
.501	.114	.115	.118	.115	.115	.113	.114	.116	.116	.501
.530	-.066	-.066	-.067	-.072	-.069	-.069	-.070	-.068	-.070	.530
.548							.056			.548
.566							.042			.566
.584	-.026	-.040	-.038	-.041	-.038	-.037	-.032	-.038	-.043	.584
.602							.036			.602
.620							.031			.620
.638	-.025	-.018		-.022	-.021				-.017	.638
.655							.021			.655
.673							.021			.673
.691	-.015	-.025	-.024	-.024	-.030	-.027	-.015	-.013	-.016	.691
.709							.012			.709
.727							.011			.727
.745	-.012	-.017		-.013	-.012		-.005		-.009	.745
.763							.009			.763
.781							.016			.781
.799							.007			.799
.817	.003	-.005	-.007	-.006	-.005	-.003	-.000	-.000	.006	.817
.834							.005			.834
.852							.002			.852
.870							.004			.870
.888							.006			.888
.906							.009			.906
.921	-.064						.010			.921
.924	.014	.008		-.005	.005				-.001	.924
.947							.007			.947
.964							.012			.964
.981	.047		.031	.013		.011	.009	.006	.007	.981
1.000	-.066									1.000

Orifice station, $y_{in.}$	Nose rake			Base rake		
	Cp top	Cp side		Cp top	Cp side	
.010	.776			.031	.474	.484
.030	1.305			.094	.676	.670
.050	1.476			.156	.836	.819
.070	1.509			.219	.983	.968
.090	1.505			.281	1.110	1.106
.110	1.449			.344	1.235	1.226
.130	1.411			.406	1.326	1.326
.150	1.395			.469	1.379	1.396
.170	1.357			.750	1.434	
.190	1.290			1.000	1.459	
				1.250	1.473	
				1.500	1.489	

TABLE 21. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.553			.059
.060							1.549			.060
.061							1.511			.061
.063							1.425			.063
.064							1.306			.064
.075	-.389	-.397	-.396	-.396	-.397	-.398	-.400	-.398	-.397	.075
.085	-.310	-.138	-.113	-.084	-.064	-.043	-.050	-.082	-.177	.085
.096	-.079		.035	.061	.086	.092	.099	.058	-.046	.096
.119	.067	.127	.146	.165		.196	.191	.163	.107	.119
.143							.150			.143
.154	-.176	-.145	-.138	-.130	-.119	-.116	-.108	-.133		.154
.171							-.074			.171
.189							-.059	-.073		.189
.206	-.090	-.094	-.081	-.084	-.064	-.060	-.045			.206
.223							-.043			.223
.241							-.043	-.047	-.056	.241
.258	-.050	-.056	-.055	-.051	-.041	-.038	-.041			.258
.275							-.045			.275
.293							-.035			.293
.310	-.023	-.043	-.045	-.035	-.032	-.027	-.026	-.037	-.044	.310
.327							-.019			.327
.344							-.020			.344
.362	-.013	-.027	-.022	-.018	-.022	-.019	-.017	-.023	-.024	.362
.379							-.019			.379
.396	-.011	-.020	-.025	-.024	-.018	-.016	-.009	-.022	-.020	.396
.408							-.016			.408
.426	.161	.165	.170	.172	.176	.184	.182	.171	.163	.426
.451	.135	.145	.149	.155	.164	.160	.165		.141	.451
.476	.113	.125	.126	.132	.143	.145	.146	.139	.128	.476
.501	.095	.108	.118	.120	.134	.134	.137	.128	.110	.501
.530	-.077	-.072	-.070	-.070	-.057	-.054	-.054	-.060	-.076	.530
.548							-.040			.548
.566							-.028			.566
.584	-.040	-.047	-.042	-.039	-.030	-.025	-.019	-.033	-.050	.584
.602							-.024			.602
.620							-.022			.620
.638	-.028	-.027			-.023	-.014			-.024	.638
.655							-.013			.655
.673							-.014			.673
.691	-.015	-.034	-.030	-.027	-.029	-.023	-.010	-.016	-.023	.691
.709							-.007			.709
.727							-.010			.727
.745	-.010	-.025			-.020	-.008	-.004	-.016	-.016	.745
.763							-.008			.763
.781							-.015			.781
.799							-.005			.799
.817	.003	-.014	-.013	-.011	-.006	-.001	-.002	-.003	-.003	.817
.834							-.003			.834
.852							-.006			.852
.870							-.004			.870
.888							-.011			.888
.906							-.009			.906
.921	-.069									.921
.924	.007									.924
.947										.947
.964										.964
.981	.059									.981
1.000	-.077									1.000

Orifice station, y , in.	Nose rake		Base rake		C_p side
	C_p top	C_p side	Orifice station, y , in.	C_p top	
.010	.633		.031	.416	.504
.030	1.268		.094	.573	.734
.050	1.488		.156	.683	.929
.070	1.505		.219	.779	1.129
.090	1.393		.281	.855	1.291
.110	1.324		.344	.945	1.396
.130	1.308		.406	1.021	1.442
.150	1.299		.469	1.083	1.455
.170	1.277		.750	1.305	
.190	1.210		1.000	1.380	
			1.250	1.390	
			1.500	1.416	

TABLE 21. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 1.57$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.544			.059
.060							1.558			.060
.061							1.533			.061
.063							1.461			.063
.064							1.349			.064
.075	-.400	-.410	-.409	-.407	-.405	-.405	-.401	-.408	-.410	.075
.085	-.403	-.127	-.077	-.018	.046	.072	.088	.007	-.166	.085
.096	-.188		.076	.126	.185	.205	.215	.144	-.033	.096
.119	-.018	.130	.167	.205		.272	.270	.218	.107	.119
.143							-.111			.143
.154	-.202	-.155	-.137	-.117	-.086	-.077	-.067	-.111		.154
.171							-.054			.171
.189							-.039			.189
.206	-.089	-.116	-.098	-.086	-.042	-.032	-.029	-.065		.206
.223							-.019			.223
.241							-.019			.241
.258	-.039	-.097	-.090	-.065	-.029	-.017	-.019	-.051	-.098	.258
.275							-.022			.275
.293							-.020			.293
.310	-.017	-.073	-.073	-.058	-.031	-.019	-.015	-.053	-.074	.310
.327							-.021			.327
.344							-.019			.344
.362	-.006	-.061	-.059	-.049	-.036	-.017	-.012	-.043	-.057	.362
.379							-.012			.379
.396	-.007	-.048	-.049	-.043	-.019	-.012	-.002	-.037	-.048	.396
.408							-.013			.408
.426	.160	.138	.143	.149	.174	.190	.189	.159	.138	.426
.451	.130	.118	.127	.138	.168	.172	.180		.117	.451
.476	.105	.100	.108	.127	.165	.176	.179	.146	.103	.476
.501	.087	.091	.108	.116	.149	.157	.163	.136	.090	.501
.530	-.101	-.090	-.081	-.072	-.043	-.037	-.035	-.058	-.098	.530
.548							-.018			.548
.566							-.008			.566
.584	-.045	-.070	-.059	-.046	-.020	-.014	-.007	-.033	-.075	.584
.602							-.003			.602
.620							-.006			.620
.638	-.028	-.056		-.028	-.006				-.053	.638
.655							.003			.655
.673							-.001			.673
.691	-.017	-.061	-.055	-.041	-.026	-.012	-.002	-.023	-.049	.691
.709							-.000			.709
.727							.003			.727
.745	-.001	-.052		-.038	-.006		.004		-.042	.745
.763							.010			.763
.781							-.007			.781
.799							.000			.799
.817	.007	-.038	-.036	-.030	-.007	.005	.012	-.015	-.027	.817
.834							.004			.834
.852							.014			.852
.870							.015			.870
.888							.015			.888
.906							.021			.906
.921	-.118	.029	-.019		-.025	.004		.021		.921
.924							.020			.924
.947							.026			.947
.964							.019	-.002	-.013	.964
.981	-.074		.002	-.018		.014	.019			.981
1.000	-.132									1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	.450		.031	.643	.516
.030	1.143		.094	.970	.888
.050	1.442		.156	1.174	1.208
.070	1.389		.219	1.280	1.463
.090	1.215		.281	1.327	1.531
.110	1.200		.344	1.352	1.500
.130	1.183		.406	1.364	1.495
.150	1.182		.469	1.277	1.504
.170	1.152		.750	1.387	
.190	1.065		1.000	1.396	
			1.250	1.403	
			1.500	1.410	

TABLE 21. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 1.57$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.529			.059
.060							1.563			.060
.061							1.555			.061
.063							1.506			.063
.064							1.411			.064
.075	-.428	-.440	-.434	-.423	-.389	-.353	-.308	-.404	-.443	.075
.085	-.432	-.100	-.015	.089	.201	.244	.259	.141	-.141	.085
.096	-.295		.140	.212	.311	.338	.351	.252	.000	.096
.119	-.111	.122	.182	.243		.364	.366	.278	.102	.119
.143							-.059			.143
.154	-.205	-.178	-.142	-.101	-.040	-.022	-.010	-.080		.154
.171							.002			.171
.189							.015			.189
.206	-.086	-.159	-.123	-.088	-.007	.015	.021	-.046		.206
.223							.028			.223
.241							.023			.241
.258	-.032	-.165	-.130	-.084	-.002	.021	.021	-.049	-.172	.258
.275							.016			.275
.293							.014			.293
.310	-.008	-.178	-.141	-.093	-.016	.008	.017	-.060	-.179	.310
.327							.011			.327
.344							.011			.344
.362	.002	-.132	-.152	-.103	-.019	.009	.018	-.066	-.122	.362
.379							.013			.379
.396	.002	-.107	-.120	-.111	-.020	.005	.019	-.073	-.105	.396
.408							.007			.408
.426	.213	.079	.069	.083	.166	.211	.217	.112	.086	.426
.451	.135	.066	.074	.093	.161	.188	.205		.068	.451
.476	.106	.049	.067	.099	.163	.187	.196	.134	.052	.476
.501	.082	.041	.065	.093	.161	.183	.194	.131	.040	.501
.530	-.109	-.128	-.112	-.088	-.029	-.012	-.005	-.056	-.133	.530
.548							.009			.548
.566							.022			.566
.584	-.051	-.116	-.093	-.067	-.002	.015	.025	-.036	-.121	.584
.602							.033			.602
.620							.026			.620
.638	-.033	-.102		-.060	.006		.033		-.096	.638
.655							.029			.655
.673							.031			.673
.691	-.021	-.119	-.100	-.072	-.015	.011		-.033	-.104	.691
.709							.035			.709
.727							.029			.727
.745	-.014	-.107		-.071	-.001		.028		-.094	.745
.763							.025			.763
.781							.014			.781
.799							.023			.799
.817	-.005	-.086	-.090	-.070	-.003	.024	.035	-.032	-.074	.817
.834							.029			.834
.852							.036			.852
.870							.039			.870
.888							.036			.888
.906							.038			.906
.921	-.161						.043			.921
.924	.047	-.061		-.062	.003		.042		-.078	.924
.947							.042			.947
.964							.049			.964
.981	.046	-.030	-.064		.017	.036	.042	-.016	-.051	.981
1.000	-.171						1.500	1.401		1.000

Orifice station, y, in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.289		.031	.807	.457
.030	.979		.094	1.321	1.028
.050	1.276		.156	1.391	1.487
.070	1.136		.219	1.356	1.622
.090	1.022		.281	1.361	1.546
.110	.974		.344	1.357	1.500
.130	.900		.406	1.359	1.511
.150	.808		.469	1.363	1.512
.170	.741		.475	1.360	
.190	.714		1.000	1.377	
			1.250	1.392	
			1.500	1.401	

TABLE 22. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.29$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.638			.059
.060							1.563			.060
.061							1.462			.061
.063							1.338			.063
.064							1.213			.064
.075	-.096	-.163	-.166	-.182	-.176	-.175	-.174	-.179	-.163	.075
.085	.275	.022	-.024	-.073	-.139	-.149	-.153	-.119	.018	.085
.096	.356		.038	-.009	-.080	-.108	-.108	-.056	.074	.096
.119	.324	.114	.073	.037	-.019	-.036	-.046	.002	.112	.119
.143							-.142			.143
.154	.041	-.073	-.094	-.112	-.130		-.128	-.123	-.076	.154
.171							-.108			.171
.189							-.090			.189
.206	.041	-.076	-.095	-.108	-.097	-.082	-.075	-.107	-.077	.206
.223							-.062			.223
.241							-.053			.241
.258	.036	-.088	-.107	-.116	-.069	-.051	-.044	-.090	-.087	.258
.275							-.036			.275
.293							-.033			.293
.310	.035	-.102	-.122	-.113	-.053	-.037	-.028	-.065	-.101	.310
.327							-.025			.327
.344							-.022			.344
.362	.033	-.113	-.133	-.075	-.053	-.034	-.020	-.055	-.114	.362
.379							-.017			.379
.396	.028	-.121	-.121	-.065	-.058	-.031	-.012	-.054	-.121	.396
.408							-.013			.408
.426	.164	-.056	-.004	.004	.000	.082	.090	.009	-.058	.426
.451	.167	-.051	-.003	.010	-.001	.055	.087	.015	-.053	.451
.476	.167	-.045	-.002	.014	-.012	.046	.058	.015	-.043	.476
.501	.166	-.001	.011	.022	.036	.042	.042	-.028	.501	.501
.530	.022	-.102	-.088	-.059	-.100		-.055	-.058	-.099	.530
.566							-.057			.566
.584	.026	-.094	-.074	-.059	-.089	-.050	-.042	-.061	-.091	.584
.602							-.037			.602
.620							-.033			.620
.638	.025	-.091		-.063	-.068		-.029		-.091	.638
.655							-.027			.655
.673							-.027			.673
.691	.024	-.094	-.081		-.052	-.038		-.051	-.093	.691
.709							-.028			.709
.727							-.027			.727
.745	.026	-.093		-.048	-.046		-.027		-.093	.745
.763							-.027			.763
.781							-.031			.781
.799							-.031			.799
.817	.025	-.097	-.055	-.042	-.046	-.040	-.027	-.038	-.096	.817
.834							-.029			.834
.870							-.027			.870
.888							-.028			.888
.921	-.166	.024	-.072		-.040		-.026			.921
.924							-.028			.924
.947							-.029			.947
.964	*						-.028			.964
.981	.051	-.058	-.036		-.040	-.041		-.038	-.066	.981
.984					-.045				-.078	.984
.998							-.030			.998
1.000	-.168									1.000

Nose rake			Base rake		
Orifice station, y , in.	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	1.072		.031	.752	.249
.030	1.362		.094	1.699	.795
.050	1.455		.156	1.715	1.310
.070	1.489		.219	1.745	1.393
.090	1.515		.281	1.744	1.442
.110	1.501		.344	1.755	1.462
.130	1.519		.406	1.753	1.450
.150	1.531		.469	1.749	1.499
.170	1.521		.5750	1.752	
.190	1.474		1.000	1.748	
			1.250	1.722	
			1.500	1.708	

TABLE 22. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
.059							1.658			.059
.060							1.598			.060
.061							1.535			.061
.063							1.410			.063
.064							1.284			.064
.075	-0.128	-0.148	-0.149	-0.150	-0.149	-0.149	-0.149	-0.149	-0.147	.075
.085	.159	.013	-0.023	-0.055	-0.114	-0.121	-0.118	-0.092	.004	.085
.096	.237		.051	.017	-0.028	-0.047	-0.049	-0.016	.062	.096
.119	.245	.115	.087	.064	.028	.015	.010	.040	.111	.119
.143							-0.120			.143
.154	-0.006	-0.069	-0.082	-0.094	-0.107		-0.108	-0.102	-0.071	.154
.171							-0.094			.171
.189							-0.082			.189
.206	.000	-0.060	-0.069	-0.077	-0.075	-0.073	-0.071	-0.077	-0.060	.206
.223							-0.059			.223
.241							-0.051			.241
.258	-0.000	-0.059	-0.066	-0.068	-0.052	-0.046	-0.044	-0.059	-0.057	.258
.275							-0.038			.275
.293							-0.035			.293
.310	-0.002	-0.063	-0.067	-0.059	-0.039	-0.031	-0.030	-0.043	-0.060	.310
.327							-0.027			.327
.344							-0.024			.344
.362	-0.003	-0.065	-0.060	-0.048	-0.030	-0.025	-0.022	-0.034	-0.059	.362
.379							-0.021			.379
.396	-0.005	-0.062	-0.056	-0.043	-0.026	-0.020	-0.017	-0.032	-0.057	.396
.408							-0.017			.408
.426	.115	.033	.048	.054	.060	.071	.078	.057	.038	.426
.451	.114	.039	.042	.047	.058	.061	.068	.052	.040	.451
.476	.115	.040	.038	.041	.045	.051	.054	.045	.040	.476
.501	.113	.035	.035	.035	.036	.040	.044		.039	.501
.530	.011	-0.057	-0.060	-0.059	-0.049		-0.056	-0.051	-0.057	.530
.548							-0.052			.548
.566							-0.041			.566
.584	-0.005	-0.054	-0.054	-0.048	-0.042	-0.039	-0.035	-0.043	-0.053	.584
.602							-0.030			.602
.620							-0.026			.620
.638	-0.003	-0.052			-0.042	-0.030			-0.052	.638
.655							-0.023			.655
.673							-0.020			.673
.691	-0.001	-0.054	-0.048		-0.027	-0.024		-0.028	-0.051	.691
.709							-0.017			.709
.727							-0.015			.727
.745	.000	-0.049			-0.029	-0.023			-0.048	.745
.763							-0.015			.763
.781							-0.015			.781
.799							-0.017			.799
.817	.001	-0.044	-0.034	-0.025	-0.022	-0.018	-0.010	-0.020	-0.039	.817
.834							-0.013			.834
.870							-0.010			.870
.888							-0.009			.888
.921	-0.149	.006	-0.038		-0.024			-0.008		.921
.924								-0.009		.924
.947								-0.006		.947
.964								-0.008		.964
.981	.016	-0.034	-0.023			-0.012		-0.016	-0.032	.981
.984									-0.037	.984
.998								-0.011		.998
1.000	-0.149									1.000

Orifice station, y , in.	Nose rake		Orifice station, y , in.	Base rake	
	Cp top	Cp side		Cp top	Cp side
.010	.957		.031	.581	.334
.030	1.409		.094	1.233	.750
.050	1.514		.156	1.443	1.063
.070	1.532		.219	1.578	1.422
.090	1.485		.281	1.589	1.359
.110	1.466		.344	1.602	1.415
.130	1.473		.406	1.609	1.436
.150	1.477		.469	1.616	1.455
.170	1.468		.750	1.631	
.190	1.416		1.000	1.644	
			1.250	1.659	
			1.500	1.666	

TABLE 22. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
.059							1.668			.059
.060							1.641			.060
.061							1.572			.061
.063							1.463			.063
.064							1.343			.064
.075	-.137	-.142	-.143	-.144	-.145	-.145	-.146	-.144	-.143	.075
.085	.067	.005	-.010	-.025	-.055	-.059	-.054	-.042	-.000	.085
.096	.145	.057	.038	.019	.009	.011	.021	.049	.096	
.119	.182	.113	.099	.089	.070	.064	.062	.073	.110	.119
.143							.099			.143
.154	-.041	-.068	-.073	-.080	-.088		-.089	-.086	-.070	.154
.171							-.080			.171
.189							-.070			.189
.206	-.028	-.051	-.053	-.059	-.061	-.063	-.064	-.060	-.051	.206
.223							-.057			.223
.241							-.051			.241
.258	-.026	-.042	-.045	-.047	-.044	-.044	-.046	-.045	-.040	.258
.275							-.042			.275
.293							-.039			.293
.310	-.023	-.040	-.041	-.040	-.037	-.034	-.034	-.033	-.036	.310
.327							-.031			.327
.344							-.029			.344
.362	-.020	-.036	-.035	-.033	-.029	-.027	-.027	-.027	-.031	.362
.379							-.025			.379
.396	-.020	-.033	-.033	-.031	-.025	-.024	-.025	-.025	-.029	.396
.408							-.021			.408
.426	.093	.068	.073	.071	.074	.076	.076	.075	.072	.426
.451	.087	.066	.065	.065	.071	.068	.069	.068	.067	.451
.476	.085	.062	.059	.058	.059	.060	.061	.060	.063	.476
.501	.082	.056	.053	.053	.052		.053		.059	.501
.530	-.029	-.041	-.044	-.045	-.044		-.045	-.044	-.041	.530
.548							-.042			.548
.566							-.038			.566
.584	-.021	-.034	-.034	-.035	-.038	-.039	-.037	-.034	-.033	.584
.602							-.032			.602
.620							-.026			.620
.638	-.018	-.029		-.032	-.027		-.023		-.028	.638
.655							-.021			.655
.673							-.020			.673
.691	-.016	-.031	-.030		-.024	-.023		-.022	-.027	.691
.709							-.018			.709
.727							-.017			.727
.745	-.013	-.025		-.021	-.018		-.014		-.022	.745
.763							-.014			.763
.781							-.017			.781
.799							-.015			.799
.817	-.010	-.021	-.019	-.017	-.017	-.013	-.012	-.014	-.013	.817
.834							-.012			.834
.870							-.010			.870
.888							-.009			.888
.921	-.126									.921
.924	-.002	-.018		-.020			-.008			.924
.947							-.009			.947
.964							-.006			.964
.981	.001	-.014	-.013		-.010	-.007	-.009	-.010	-.008	.981
.984					-.014				-.015	.984
.998							-.019			.998
1.000	-.130									1.000

Nose rake			Base rake		
Orifice station, y , in.	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side
.010	.793		.031	.407	.299
.030	1.449		.094	.770	.518
.050	1.566		.156	.983	.694
.070	1.514		.219	1.158	.843
.090	1.389		.281	1.293	.962
.110	1.389		.344	1.353	1.078
.130	1.397		.406	1.387	1.183
.150	1.397		.469	1.414	1.248
.170	1.397		.5750	1.501	
.190	1.340		1.000	1.541	
			1.250	1.570	
			1.500	1.588	

TABLE 22. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.681			.059
.060							1.659			.060
.061							1.607			.061
.063							1.516			.063
.064							1.401			.064
.075	-0.145	-0.143	-0.143	-0.143	-0.142	-0.143	-0.143	-0.143	-0.144	.075
.085	-0.000	.004	.005	.007	.004	.005	.009	.003	-0.002	.085
.096	.076		.071	.073	.077	.071	.075	.066	.048	.096
.119	.125	.112	.112	.114	.115	.114	.114	.111	.109	.119
.143							-0.075			.143
.154	-0.068	-0.067	-0.067	-0.067	-0.068		-0.065	-0.069	-0.069	.154
.171							-0.059			.171
.189							-0.053			.189
.206	-0.048	-0.047	-0.047	-0.048	-0.047	-0.047	-0.049	-0.047	-0.046	.206
.223							-0.044			.223
.241							-0.042			.241
.258	-0.039	-0.037	-0.037	-0.037	-0.036	-0.037	-0.039	-0.036	-0.033	.258
.275							-0.037			.275
.293							-0.037			.293
.310	-0.031	-0.033	-0.033	-0.032	-0.032	-0.032	-0.033	-0.028	-0.028	.310
.327							-0.030			.327
.344							-0.029			.344
.362	-0.025	-0.029	-0.027	-0.028	-0.029	-0.028	-0.029	-0.023	-0.024	.362
.379							-0.027			.379
.396	-0.024	-0.026	-0.027	-0.026	-0.026	-0.024	-0.026	-0.022	-0.021	.396
.408							-0.023			.408
.426	.082	.078	.083	.080	.081	.083	.082	.083	.082	.426
.451	.075	.074	.074	.075	.079	.076	.077	.077	.076	.451
.476	.070	.068	.067	.069	.071	.071	.071	.072	.071	.476
.501	.066		.065	.064	.065	.064	.065	.065	.065	.501
.530	-0.040	-0.038	-0.038	-0.038	-0.039		-0.041	-0.037	-0.037	.530
.548							-0.037			.548
.566							-0.030			.566
.584	-0.029	-0.029	-0.029	-0.032	-0.031	-0.030	-0.028	-0.029	-0.027	.584
.602							-0.025			.602
.620							-0.023			.620
.638	-0.024	-0.027		-0.022	-0.022		-0.021		-0.024	.638
.655							-0.020			.655
.673							-0.018			.673
.691	-0.021	-0.023	-0.022		-0.022	-0.019		-0.018	-0.019	.691
.709							-0.016			.709
.727							-0.017			.727
.745	-0.017	-0.016		-0.016	-0.019		-0.017		-0.014	.745
.763							-0.017			.763
.781							-0.018			.781
.799							-0.016			.799
.817	-0.013	-0.016	-0.016	-0.016	-0.016	-0.016	-0.013	-0.011	-0.008	.817
.834							-0.013			.834
.870							-0.010			.870
.888							-0.008			.888
.921	-0.117									.921
.924	-0.008	-0.012		-0.017			-0.007			.924
.947							-0.008			.947
.964							-0.006			.964
.981	.006	-0.007	-0.010		-0.008	-0.008	-0.009	-0.006	-0.004	.981
.984					-0.012				-0.010	.984
.998							-0.013			.998
1.000	-0.121									1.000

Orifice station, y , in.	Nose rake			Base rake		
	Cp top	Cp side	Cp top	Cp side	Cp top	Cp side
.010	.723			.031	.297	.298
.030	1.442			.094	.478	.469
.050	1.555			.156	.606	.600
.070	1.427			.219	.722	.725
.090	1.337			.281	.802	.827
.110	1.322			.344	.895	.935
.130	1.319			.406	.987	1.030
.150	1.319			.469	1.065	1.117
.170	1.301			.750	1.243	
.190	1.252			1.000	1.312	
				1.250	1.379	
				1.500	1.437	

TABLE 22. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.29$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.678			.059
.060							1.676			.060
.061							1.635			.061
.063							1.563			.063
.064							1.444			.064
.075	-0.145	-0.141	-0.140	-0.137	-0.133	-0.133	-0.134	-0.138	-0.143	.075
.085	-0.068	.006	.023	.044	.066	.074	.083	.054	-0.001	.085
.096	.018	.093	.110	.139	.140	.146	.118	.051	.096	
.119	.071	.114	.126	.142	.163	.167	.169	.150	.111	.119
.143							-0.047			.143
.154	-0.090	-0.068	-0.061	-0.053	-0.043		-0.036	-0.050	-0.069	.154
.171							-0.032			.171
.189							-0.030			.189
.206	-0.062	-0.051	-0.045	-0.041	-0.029	-0.027	-0.027	-0.035	-0.049	.206
.223							-0.024			.223
.241							-0.024			.241
.258	-0.045	-0.045	-0.040	-0.035	-0.024	-0.023	-0.024	-0.030	-0.039	.258
.275							-0.024			.275
.293							-0.024			.293
.310	-0.033	-0.041	-0.039	-0.033	-0.025	-0.022	-0.020	-0.027	-0.036	.310
.327							-0.020			.327
.344							-0.020			.344
.362	-0.025	-0.038	-0.036	-0.033	-0.024	-0.020	-0.020	-0.023	-0.033	.362
.379							-0.020			.379
.396	-0.021	-0.034	-0.035	-0.033	-0.023	-0.020	-0.016	-0.024	-0.030	.396
.408							-0.018			.408
.426	.076	.067	.073	.075	.085	.091	.093	.083	.071	.426
.451	.070	.064	.068	.074	.086	.088	.090	.079	.067	.451
.476	.061	.061	.063	.070	.083	.087	.088	.079	.063	.476
.501	.053		.063	.067	.080	.083	.086		.059	.501
.530	-0.043	-0.043	-0.040	-0.039	-0.031		-0.030	-0.032	-0.041	.530
.548							-0.028			.548
.566							-0.022			.566
.584	-0.032	-0.037	-0.034	-0.033	-0.026	-0.022	-0.020	-0.026	-0.033	.584
.602							-0.017			.602
.620							-0.017			.620
.638	-0.024	-0.034		-0.024	-0.018		-0.014		-0.030	.638
.655							-0.014			.655
.673							-0.013			.673
.691	-0.020	-0.031	-0.029		-0.019	-0.014		-0.018	-0.027	.691
.709							-0.012			.709
.727							-0.013			.727
.745	-0.017	-0.024		-0.023	-0.017		-0.012		-0.021	.745
.763							-0.012			.763
.781							-0.015			.781
.799							-0.014			.799
.817	-0.011	-0.024	-0.025	-0.022	-0.016	-0.011		-0.014	-0.016	.817
.834							-0.010			.834
.870							-0.006			.870
.888							-0.006			.888
.921	-0.123						-0.003			.921
.924	-0.009	-0.020		-0.022			-0.005			.924
.947							-0.001			.947
.964							-0.002			.964
.981	.006	-0.017	-0.018		-0.007	-0.002		-0.006	-0.012	.981
.984					-0.014				-0.018	.984
.998							-0.007			.998
1.000		-0.128								1.000

Orifice station, y.in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y.in.	Cp top	Cp side
.010	.562		.031	.237	.326
.030	1.286		.094	.370	.563
.050	1.434		.156	.460	.766
.070	1.344		.219	.545	.927
.090	1.252		.281	.606	1.043
.110	1.227		.344	.669	1.169
.130	1.214		.406	.714	1.255
.150	1.209		.469	.763	1.297
.170	1.209		.750	.930	
.190	1.190		1.000	1.094	
			1.250	1.195	
			1.500	1.217	

TABLE 22. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.29$ - Continued(f) $\alpha = 6.3^\circ$

Model station, $\frac{t}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{t}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.663			.059
.060							1.677			.060
.061							1.649			.061
.063							1.595			.063
.064							1.493			.064
.075	-0.147	-0.146	-0.144	-0.139	-0.128	-0.125	-0.124	-0.135	-0.148	.075
.085	-0.137	-0.110	-0.049	-0.096	-0.145	-0.163	-0.175	-0.117	-0.001	.085
.096	-0.042		-0.118	-0.152	-0.208	-0.216	-0.226	-0.176	-0.060	.096
.119	-0.014		-0.113	-0.140	-0.170	-0.216	-0.228	-0.193	-0.111	.119
.143							-0.013			.143
.154	-0.111	-0.071	-0.056	-0.040	-0.013			-0.002	-0.028	-0.073
.171								-0.002		.171
.189								-0.002		.189
.206	-0.072	-0.062	-0.049	-0.035	-0.006	-0.000	-0.002	-0.019	-0.062	.206
.223								-0.003		.223
.241								-0.002		.241
.258	-0.045	-0.063	-0.051	-0.035	-0.007	-0.000	-0.000	-0.022	-0.059	.258
.275								-0.001		.275
.293								-0.001		.293
.310	-0.032	-0.069	-0.058	-0.041	-0.013	-0.003	-0.000	-0.023	-0.065	.310
.327							-0.000			.327
.344							-0.000			.344
.362	-0.023	-0.072	-0.062	-0.046	-0.015	-0.005	-0.001	-0.025	-0.066	.362
.379							-0.002			.379
.396	-0.021	-0.069	-0.064	-0.049	-0.017	-0.006	-0.000	-0.029	-0.063	.396
.408							-0.004			.408
.426	.075	.026	.035	.050	.095	.113	.117	.077	.030	.426
.451	.067	.031	.035	.053	.100	.110	.119	.077	.034	.451
.476	.054	.033	.037	.054	.098	.113	.117	.079	.037	.476
.501	.044		.042	.055	.095	.106	.112		.035	.501
.530	-0.055	-0.062	-0.057	-0.050	-0.027		-0.012	-0.037	-0.060	.530
.548							-0.006			.548
.566							-0.001			.566
.584	-0.038	-0.061	-0.053	-0.042	-0.016	-0.008	-0.003	-0.025	-0.058	.584
.602							-0.001			.602
.620							-0.002			.620
.638	-0.026	-0.057		-0.036	-0.012				-0.054	.638
.655							-0.000			.655
.673							-0.000			.673
.691	-0.022	-0.057	-0.051		-0.018	-0.008		-0.022	-0.054	.691
.709							-0.000			.709
.727							-0.000			.727
.745	-0.018	-0.054		-0.041	-0.013				-0.051	.745
.763							-0.000			.763
.781							-0.000			.781
.799							-0.001			.799
.817	-0.012	-0.052	-0.051	-0.040	-0.012	-0.000	-0.004	-0.022	-0.044	.817
.834							-0.002			.834
.870							-0.000			.870
.888							-0.000			.888
.921	-0.146	-0.012	-0.045		-0.044			.010		.921
.924							-0.004			.924
.947							.007			.947
.964							.004			.964
.981	.000	-0.038	-0.043		-0.009	-0.000	.003	-0.015	-0.036	.981
.984					-0.008		.002		-0.043	.984
.998							.002			.998
1.000	-0.150									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.434		.031	.419	.347
.030	1.034		.094	.829	.797
.050	1.214		.156	.947	1.123
.070	1.172		.219	.989	1.284
.090	1.123		.281	1.023	1.403
.110	1.094		.344	1.043	1.437
.130	1.075		.406	1.056	1.463
.150	1.062		.469	1.068	1.478
.170	1.039		.750	1.102	
.190	.983		1.000	1.128	
			1.250	1.140	
			1.500	1.153	

TABLE 22. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.29$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.635			.059
.060							1.673			.060
.061							1.674			.061
.063							1.629			.063
.064							1.543			.064
.075	-.168	-.158	-.151	-.128	-.108	-.083	-.073	-.118	-.163	.075
.085	-.160	.026	.084	.159	.249	.277	.295	.205	.014	.085
.096	-.108		.148	.202	.291	.308	.322	.246	.077	.096
.119	-.047	.112	.153	.203	.284	.306	.314	.245	.110	.119
.143							.034			.143
.154	-.130	-.076	-.050	-.022	.025		.051	.001	-.076	.154
.171							.050			.171
.189							.049			.189
.206	-.073	-.079	-.054	-.026	.026	.041	.048	.002	-.077	.206
.223							.048			.223
.241							.044			.241
.258	-.043	-.093	-.067	-.034	.022	.038	.042	-.004	-.087	.258
.275							.040			.275
.293							.038			.293
.310	-.029	-.108	-.080	-.045	.013	.033	.041	-.009	-.104	.310
.327							.039			.327
.344							.038			.344
.362	-.020	-.122	-.091	-.054	.007	.030	.038	-.015	-.117	.362
.379							.035			.379
.396	-.016	-.128	-.097	-.060	.007	.028	.038	-.020	-.124	.396
.408							.034			.408
.426	.085	-.069	-.016	.030	.127	.162	.173	.091	-.063	.426
.451	.084	-.062	-.018	.034	.135	.160	.175	.093	-.059	.451
.476	.056	-.054	-.018	.034	.134	.163	.175	.094	-.055	.476
.501	.043		.014	.034	.133	.161	.175		.046	.501
.530	-.055	-.111	-.101	-.067	.002		.029	-.027	-.111	.530
.548							.031			.548
.566							.033			.566
.584	-.044	-.102	-.100	-.065	.001	.021	.031	-.025	-.100	.584
.602							.031			.602
.620							.028			.620
.638	-.033	-.099		-.064	.000		.028		-.098	.638
.655							.025			.655
.673							.028			.673
.691	-.033	-.104	-.101		-.004	.020		-.029	-.102	.691
.709							.032			.709
.727							.034			.727
.745	-.032	-.102		-.072	.000		.032		-.100	.745
.763							.033			.763
.781							.028			.781
.799							.024			.799
.817	-.032	-.105	-.095	-.068	-.004	.020	.031	-.031	-.100	.817
.834							.027			.834
.870							.030			.870
.888							.028			.888
.921	-.165						.031			.921
.924	-.033	-.072		-.072			.028			.924
.947							.032			.947
.964							.032			.964
.981	-.019	-.060	-.103		-.001	.019	.027	-.023	-.069	.981
.984					-.006		.027		-.082	.984
.998							.027			.998
1.000	-.169									1.000

Orifice station, y, in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.221		.031	.501	.237
.030	.777		.094	1.068	.792
.050	.924		.156	1.005	1.334
.070	.902		.219	1.026	1.390
.090	.902		.281	1.031	1.433
.110	.885		.344	1.037	1.449
.130	.869		.406	1.043	1.460
.150	.856		.469	1.046	1.465
.170	.843		.520	1.053	
.190	.803		1.000	1.078	
			1.250	1.108	
			1.500	1.142	

TABLE 23. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.98$

(a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.711			.059
.060							1.631			.060
.061							1.538			.061
.063							1.414			.063
.064							1.281			.064
.075	.006	-0.057	-0.062	-0.076	-0.075	-0.075	-0.075	-0.076	-0.056	.075
.085	.291	.079	.031	-0.007	-0.043	-0.053	-0.057	-0.033	.079	.085
.096	.317		.077	.038	-0.012	-0.030	-0.031	.009	.113	.096
.119	.305	.120	.088	.060	.017	.005	.001	.036	.121	.119
.143							-0.077			.143
.154	.066	-0.027	-0.044	-0.057	-0.071		-0.072	-0.066	-0.029	.154
.171							-0.067			.171
.189							-0.057			.189
.206	.056	-0.035	-0.049	-0.059	-0.059	-0.050	-0.049	-0.062	-0.035	.206
.223							-0.042			.223
.241							-0.035			.241
.258	.047	-0.048	-0.060	-0.068	-0.044	-0.032	-0.030	-0.061	-0.043	.258
.275							-0.026			.275
.293							-0.023			.293
.310	.044	-0.058	-0.072	-0.074	-0.038	-0.027	-0.021	-0.051	-0.059	.310
.327							-0.020			.327
.344							-0.019			.344
.362	.042	-0.067	-0.082	-0.065	-0.037	-0.026	-0.018	-0.044	-0.068	.362
.379							-0.017			.379
.396	.038	-0.073	-0.086	-0.057	-0.040	-0.026	-0.016	-0.045	-0.072	.396
.408							-0.017			.408
.426	.145	-0.034	-0.025	-0.013	-0.005	.027	.048	-0.007	-0.033	.426
.451	.158	-0.029	-0.035	-0.004	.001	.023	.053	-0.002	-0.026	.451
.476	.164	-0.026	-0.033	.001	.012	.019	.038	-0.002	-0.023	.476
.501	.168		-0.031	-0.000	.017	.014	.026		-0.020	.501
.530	.047	-0.074	-0.074	-0.046	-0.062		-0.037	-0.008	-0.072	.530
.548							-0.036			.548
.566							-0.037			.566
.584	.046	-0.079	-0.068	-0.050	-0.060	-0.039	-0.033	-0.043	-0.076	.584
.602							-0.026			.602
.620							-0.026			.620
.638	.043	-0.081		-0.049	-0.047				-0.079	.638
.655							-0.025			.655
.673							-0.024			.673
.691	.042	-0.084	-0.064		-0.041	-0.034		-0.047	-0.083	.691
.709							-0.024			.709
.727							-0.023			.727
.745	.043	-0.076		-0.043	-0.042				-0.085	.745
.763							-0.024			.763
.781							-0.025			.781
.799							-0.028			.799
.817	.042	-0.080	-0.046	-0.041	-0.045	-0.037	-0.027	-0.040	-0.083	.817
.834							-0.028			.834
.870							-0.026			.870
.888							-0.026			.888
.921	-0.114						-0.026			.921
.924	.040	-0.054			-0.041			-0.027		.924
.947							-0.025			.947
.964							-0.026			.964
.981	.057	-0.044	-0.035			-0.048	-0.026		-0.044	.981
.984						-0.048			-0.056	.984
.998							-0.026			.998
1.000	-0.115									1.000

Nose rake			Base rake		
Orifice station, $y_{in.}$	C_p top	C_p side	Orifice station, $y_{in.}$	C_p top	C_p side
.010	1.020		.031	.733	.114
.030	1.470		.094	1.672	.381
.050	1.578		.156	1.891	.894
.070	1.590		.219	1.938	1.152
.090	1.534		.281	1.951	1.318
.110	1.525		.344	1.956	1.373
.130	1.541		.406	1.940	1.446
.150	1.549		.469	1.941	1.482
.170	1.555		.575	1.943	
.190	1.509		1.000	1.924	
			1.250	1.927	
			1.500	1.922	

TABLE 23. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.98$ - Continued(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
*059							1.729			*059
*060							1.676			*060
*061							1.608			*061
*063							1.486			*063
*064							1.359			*064
*075	-0.025	-0.053	-0.055	-0.059	-0.062	-0.061	-0.061	-0.060	-0.051	*075
*085	*019	*070	*039	*019	*017	*021	-0.023	*001	*071	*085
*096	*0232		*083	*058	*021	*012	*012	*038	*105	*096
*119	*0231	*120	*099	*080	*050	*043	*039	*064	*121	*119
*143							-0.063			*143
*154	*024	-0.028	-0.037	-0.046	-0.057		-0.059	-0.052	-0.028	*154
*171							-0.055			*171
*189							-0.049			*189
*206	*019	-0.028	-0.036	-0.042	-0.045	-0.043	-0.045	-0.044	-0.027	*206
*223							-0.040			*223
*241							-0.036			*241
*258	*012	-0.033	-0.039	-0.041	-0.036	-0.031	-0.032	-0.039	-0.028	*258
*275							-0.027			*275
*293							-0.025			*293
*310	*009	-0.038	-0.042	-0.040	-0.028	-0.024	-0.024	-0.034	-0.038	*310
*327							-0.022			*327
*344							-0.021			*344
*362	*008	-0.042	-0.043	-0.038	-0.024	-0.020	-0.019	-0.029	-0.041	*362
*379	*006	-0.044	-0.043	-0.036	-0.022	-0.017	-0.019	-0.026	-0.043	*379
*396							-0.016			*396
*408							-0.016			*408
*426	*091	*010	*022	*025	*033	*040	*044	*030	*013	*426
*451	*097	*015	*018	*025	*038	*039	*044	*031	*018	*451
*476	*102	*019	*018	*021	*030	*033	*036	*028	*022	*476
*501	*106		*018	*020	*024	*027	*030		*025	*501
*530	*012	-0.039	-0.041	-0.038	-0.030		-0.031	-0.032	-0.037	*530
*548							-0.031			*548
*566							-0.028			*566
*584	*011	-0.041	-0.041	-0.033	-0.028	-0.027	-0.026	-0.027	-0.037	*584
*602							-0.021			*602
*620							-0.019			*620
*638	*010	-0.041			-0.030	-0.022			-0.039	*638
*655							-0.016			*655
*673							-0.014			*673
*691	*008	-0.041	-0.038		-0.020	-0.017			-0.024	*691
*709							-0.012			*709
*727							-0.012			*727
*745	*008	-0.036			-0.023	-0.017			-0.042	*745
*763							-0.011			*763
*781							-0.010			*781
*799							-0.012			*799
*817	*010	-0.037	-0.028	-0.019	-0.015	-0.014			-0.018	*817
*834							-0.008		-0.041	*834
*870							-0.010			*870
*888							-0.008			*888
*921	-0.105									*921
*924	*012	-0.033			-0.019					*924
*947							-0.009			*947
*964							-0.010			*964
*981	*023	-0.032	-0.022			-0.014	-0.007			*981
*984							-0.008			*984
*998							-0.008			*998
1.000	-0.105									1.000

Nose rake			Base rake		
Orifice station, y,in.	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
*010	*890		*031	*520	*243
*030	1.518		*094	1.165	*639
*050	1.604		*156	1.424	*877
*070	1.536		*219	1.628	*1079
*090	1.437		*281	1.667	*1242
*110	1.448		*344	1.687	*1307
*130	1.465		*406	1.699	*1381
*150	1.463		*469	1.712	*1392
*170	1.469		*750	1.728	
*190	1.424		1.000	1.750	
			1.250	1.768	
			1.500	1.761	

TABLE 23. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.98$ - Continued(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.746			.059
.060							1.705			.060
.061							1.639			.061
.063							1.538			.063
.064							1.414			.064
.075	-.038	-.048	-.051	-.052	-.054	-.055	-.054	-.053	-.049	.075
.085	.124	.069	.056	.045	.025	.022	.022	.036	.071	.085
.096	.177		.092	.081	.063	.059	.060	.071	.102	.096
.119	.176	.119	.108	.100	.085	.081	.081	.091	.120	.119
.143							-.046			.143
.154	-.004	-.027	-.030	-.034	-.041		-.043	-.040	-.027	.154
.171							-.041			.171
.189							-.037			.189
.206	-.004	-.022	-.027	-.029	-.032	-.034	-.035	-.031	-.022	.206
.223							-.032			.223
.241							-.031			.241
.258	-.006	-.022	-.026	-.027	-.027	-.027	-.028	-.027	-.018	.258
.275							-.025			.275
.293							-.025			.293
.310	-.007	-.022	-.025	-.025	-.025	-.023	-.023	-.023	-.022	.310
.327							-.022			.327
.344							-.021			.344
.362	-.008	-.022	-.022	-.021	-.021	-.020	-.020	-.021	-.022	.362
.379							-.020			.379
.396	-.009	-.022	-.022	-.021	-.018	-.017	-.017	-.019	-.020	.396
.408							-.017			.408
.426	.066	.041	.049	.042	.046	.047	.047	.044	.044	.426
.451	.068	.045	.044	.043	.049	.047	.048	.046	.046	.451
.476	.069	.042	.041	.039	.042	.041	.041	.041	.046	.476
.501	.071		.039	.037	.038	.037	.037	.037	.045	.501
.530	-.006	-.024	-.025	-.026	-.025		-.025	-.024	-.021	.530
.548							-.024			.548
.566							-.021			.566
.584	-.004	-.021	-.021	-.021	-.020	-.019	-.019	-.018	-.017	.584
.602							-.016			.602
.620							-.014			.620
.638	-.005	-.019		-.017	-.014		-.012		-.017	.638
.655							-.011			.655
.673							-.010			.673
.691	-.005	-.018	-.017		-.012	-.011		-.015	-.017	.691
.709							-.010			.709
.727							-.009			.727
.745	-.004	-.011		-.013	-.010		-.009		-.016	.745
.763							-.009			.763
.781							-.009			.781
.799							-.009			.799
.817	-.003	-.012	-.012	-.009	-.008	-.008	-.006	-.011	-.013	.817
.834							-.006			.834
.870							-.005			.870
.888							-.004			.888
.921	-.090	-.001	-.009		-.009		-.005			.921
.924							-.007			.924
.947							-.005			.947
.964							-.005			.964
.981	.007	-.008	-.009		-.006	-.005	-.005	-.008	-.010	.981
.984							-.005		-.014	.984
.998							-.010			.998
1.000	-.092									1.000

Nose rake			Base rake		
Orifice station, y, in.	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.783		.031	.325	.225
.030	1.505		.094	.650	.412
.050	1.581		.156	.824	.548
.070	1.477		.219	1.025	.668
.090	1.393		.281	1.162	.777
.110	1.393		.344	1.266	.888
.130	1.388		.406	1.317	.993
.150	1.386		.469	1.365	
.170	1.375		.4750	1.505	
.190	1.326		1.000	1.566	
			1.250	1.608	
			1.500	1.636	

TABLE 23. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.98$ - Continued(d) $\alpha = 0^\circ$

Model station, $\frac{y}{t}$	Cp for meridian angle, θ , deg -										Model station, $\frac{y}{t}$
	0	90	105	120	150	165	180	225	270		
.059							1.754				.059
.060							1.725				.060
.061							1.674				.061
.063							1.588				.063
.064							1.476				.064
.075	-.047	-.047	-.047	-.047	-.046	-.047	-.047	-.047	-.047		.075
.085	.068	.069	.068	.071	.069	.070	.071	.072	.069		.085
.096	.129	.105	.105	.107	.106	.110	.106	.100	.096		.096
.119	.129	.119	.118	.120	.121	.121	.122	.121	.120		.119
.143							-.028				.143
.154	-.026	-.026	-.026	-.026	-.026		-.025	-.026	-.026		.154
.171							-.025				.171
.189							-.023				.189
.206	-.021	-.021	-.021	-.021	-.021	-.021	-.022	-.020	-.020		.206
.223							-.021				.223
.241							-.021				.241
.258	-.020	-.020	-.018	-.019	-.019	-.020	-.021	-.019	-.014		.258
.275							-.021				.275
.293							-.021				.293
.310	-.018	-.018	-.018	-.019	-.019	-.020	-.021	-.017	-.018		.310
.327							-.018				.327
.344							-.018				.344
.362	-.016	-.018	-.015	-.017	-.018	-.019	-.019	-.016	-.016		.362
.379							-.017				.379
.396	-.016	-.016	-.016	-.016	-.017	-.018	-.015	-.016	-.014		.396
.408							-.015				.408
.426	.054	.050	.057	.052	.051	.054	.053	.053	.052		.426
.451	.053	.051	.051	.052	.056	.053	.052	.054	.054		.451
.476	.051	.048	.048	.049	.051	.050	.051	.052	.051		.476
.501	.049		.047	.045	.049	.049	.049	.049	.049		.501
.530	-.017	-.020	-.021	-.020	-.019		-.018	-.017	-.018		.530
.548							-.016				.548
.566							-.013				.566
.584	-.014	-.016	-.014	-.014	-.015	-.016	-.013	-.012	-.012		.584
.602							-.012				.602
.620							-.011				.620
.638	-.012	-.012			-.012	-.013					.638
.655							-.010				.655
.673							-.011				.673
.691	-.012	-.012	-.011			-.013					.691
.709							-.011				.709
.727							-.010				.727
.745	-.010	-.005			-.010	-.010					.745
.763							-.009				.763
.781							-.008				.781
.799							-.009				.799
.817	-.007	-.007	-.007		-.006	-.008	-.007	-.008	-.007		.817
.834							-.008				.834
.870							-.008				.870
.888							-.007				.888
.921	-.082										.921
.924	-.005	-.005			-.007						.924
.947							-.003				.947
.964							-.004				.964
.981	.002	-.004	-.004			-.004	-.003	-.003	-.005		.981
.984						-.007			-.009		.984
.998							-.007				.998
1.000	-.083										1.000

Nose rake			Base rake		
Orifice station, y , in.	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side
.010	.652		.031	.231	.220
.030	1.328		.094	.403	.359
.050	1.441		.156	.506	.468
.070	1.384		.219	.578	.556
.090	1.339		.281	.647	.639
.110	1.327		.344	.719	.731
.130	1.320		.406	.788	.816
.150	1.317		.469	.865	.895
.170	1.285		.750	1.056	
.190	1.250		1.000	1.137	
			1.250	1.242	
			1.500	1.341	

TABLE 23. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
.059							1.739			.059
.060							1.743			.060
.061							1.710			.061
.063							1.639			.063
.064							1.523			.064
.075	-.055	-.048	-.045	-.042	-.037	-.036	-.035	-.040	-.047	.075
.085	.022	.070	.082	.099	.118	.125	.129	.112	.069	.085
.096	.071		.118	.132	.154	.158	.165	.144	.101	.096
.119	.084	.119	.129	.140	.157	.162	.164	.150	.119	.119
.143							-.008			.143
.154	-.043	-.027	-.022	-.018	-.010		-.002	-.012	-.027	.154
.171							-.003			.171
.189							-.002			.189
.206	-.034	-.024	-.020	-.015	-.006	-.004	-.003	-.010	-.022	.206
.223							-.003			.223
.241							-.004			.241
.258	-.028	-.023	-.020	-.016	-.009	-.007	-.006	-.012	-.018	.258
.275							-.007			.275
.293							-.007			.293
.310	-.024	-.024	-.022	-.018	-.013	-.010	-.008	-.013	-.024	.310
.327							-.008			.327
.344							-.008			.344
.362	-.020	-.024	-.022	-.018	-.013	-.010	-.009	-.013	-.022	.362
.379							-.009			.379
.396	-.018	-.024	-.022	-.020	-.013	-.010	-.009	-.014	-.021	.396
.408							-.009			.408
.426	.047	.041	.051	.049	.059	.063	.065	.056	.043	.426
.451	.046	.042	.045	.051	.066	.063	.066	.058	.045	.451
.476	.041	.041	.044	.050	.063	.066	.068	.061	.044	.476
.501	.038		.045	.050	.064	.068	.071		.045	.501
.530	-.024	-.025	-.024	-.019	-.010		-.005	-.011	-.022	.530
.548							-.005			.548
.566							-.004			.566
.584	-.020	-.022	-.020	-.017	-.011	-.008	-.006	-.010	-.018	.584
.602							-.006			.602
.620							-.007			.620
.638	-.014	-.021		-.016	-.011		-.005		-.018	.638
.655							-.006			.655
.673							-.006			.673
.691	-.012	-.020	-.019		-.011	-.008		-.011	-.019	.691
.709							-.006			.709
.727							-.004			.727
.745	-.010	-.012		-.015	-.008		-.004		-.017	.745
.763							-.005			.763
.781							-.006			.781
.799							-.008			.799
.817	-.007	-.014	-.014	-.014	-.009	-.008	-.005	-.011	-.014	.817
.834							-.005			.834
.870							-.001			.870
.888							-.000			.888
.921	-.089						-.000			.921
.924	-.006	-.012		-.012			-.000			.924
.947							-.000			.947
.964							-.000			.964
.981	.001	-.010	-.011			-.002	-.001		-.005	.981
.984						-.005		-.010	-.016	.984
.998							-.003			.998
1.000	-.091									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.496		.031	.192	.224
.030	1.162		.094	.302	.430
.050	1.294		.156	.383	.588
.070	1.247		.219	.454	.702
.090	1.216		.281	.496	.816
.110	1.216		.344	.539	.933
.130	1.214		.406	.564	1.027
.150	1.201		.469	.603	1.096
.170	1.186		.750	.744	
.190	1.154		1.000	.896	
			1.250	.992	
			1.500	1.025	

TABLE 23. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.98$ - Continued

(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.059							1.739				.059
.060							1.750				.060
.061							1.732				.061
.063							1.678				.063
.064							1.572				.064
.075	-.054	-.048	-.044	-.037	-.026	-.022	-.021	-.031	-.050		.075
.085	-.025	.071	.098	.136	.180	.195	.204	.162	.071		.085
.096	.013		.140	.167	.213	.220	.230	.190	.103		.096
.119	.038	.120	.140	.164	.200	.212	.215	.185	.121		.119
.143								.021			.143
.154	-.059	-.029	-.017	-.005	.015		.026	.005	-.028		.154
.171								.025			.171
.189								.023			.189
.206	-.043	-.029	-.019	-.007	.012	.018	.020	.004	-.027		.206
.223								.019			.223
.241								.016			.241
.258	-.031	-.034	-.026	-.014	.006	.012	.013	-.002	-.028		.258
.275								.012			.275
.293								.010			.293
.310	-.022	-.039	-.031	-.020	.000	.007	.011	-.006	-.038		.310
.327								.010			.327
.344								.009			.344
.362	-.017	-.045	-.036	-.023	-.002	.006	.009	-.009	-.042		.362
.379								.008			.379
.396	-.015	-.046	-.039	-.026	-.002	.005	.009	-.012	-.044		.396
.408								.008			.408
.426	.043	.007	.027	.039	.075	.089	.094	.062	.011		.426
.451	.043	.013	.023	.043	.087	.095	.101	.068	.016		.451
.476	.036	.016	.025	.046	.088	.101	.107	.074	.020		.476
.501	.030		.029	.047	.090	.103	.109		.024		.501
.530	-.030	-.041	-.035	-.024	.002			.015	-.038		.530
.548								.013			.548
.566								.013			.566
.584	-.025	-.042	-.036	-.026	-.002	.006	.011	-.009	-.040		.584
.602								.011			.602
.620								.010			.620
.638	-.016	-.043		-.025	-.002			.011			.638
.655								.011			.655
.673								.010			.673
.691	-.013	-.045	-.038		-.004	.004			-.010	-.043	.691
.709								.006			.709
.727								.006			.727
.745	-.011	-.038		-.028	-.004			.007		-.043	.745
.763								.010			.763
.781								.009			.781
.799								.010			.799
.817	-.009	-.041	-.038	-.027	.000	.010		.013	-.011	-.042	.817
.834								.012			.834
.870								.014			.870
.888								.011			.888
.921	-.104							.012			.921
.924	-.009	-.034			-.028			.011			.924
.947								.011			.947
.964								.013			.964
.981	-.001	-.033	-.036			.001	.010	.012	-.011	-.034	.981
.984								.012		-.039	.984
.998											.998
1.000		-.105									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.325		.031	.328	.232
.030	.886		.094	.671	.654
.050	1.061		.156	.749	.909
.070	1.059		.219	.795	1.106
.090	1.057		.281	.812	1.268
.110	1.047		.344	.827	1.350
.130	1.049		.406	.845	1.379
.150	1.041		.469	.859	1.396
.170	1.031		.750	.890	
.190	.997		1.000	.911	
.			1.250	.929	
.			1.500	.950	

TABLE 23. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 2.98$ - Concluded(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.708			.059
.060							1.748			.060
.061							1.750			.061
.063							1.708			.063
.064							1.621			.064
.075	-.069	-.054	-.048	-.033	-.005	.014	.022	-.014	-.057	.075
.085	-.054	.078	.126	.187	.264	.289	.305	.229	.072	.085
.096	-.035		.163	.208	.281	.297	.308	.246	.111	.096
.119	-.006		.120	.153	.192	.259	.279	.228	.119	.119
.143							.064			.143
.154	-.070	-.030	-.010	.012	.050		.071	.033	-.029	.154
.171							.067			.171
.189							.062			.189
.206	-.048	-.037	-.019	.001	.041	.053	.057	.023	-.036	.206
.223							.056			.223
.241							.052			.241
.258	-.030	-.050	-.032	-.009	.034	.047	.050	.013	-.042	.258
.275							.049			.275
.293							.046			.293
.310	-.020	-.061	-.041	-.017	.026	.041	.047	.008	-.059	.310
.327							.046			.327
.344							.045			.344
.379							.043			.379
.396	-.015	-.076	-.054	-.027	.022	.038	.045	.002	-.073	.396
.408							.043			.408
.426	.049	-.037	.005	.037	.114	.142	.151	.086	-.033	.426
.451	.055	-.032	.004	.047	.132	.153	.165	.097	-.027	.451
.476	.041	-.029	.007	.050	.133	.160	.170	.104	-.022	.476
.501	.030		.010	.052	.134	.160	.172		-.019	.501
.530	-.035	-.076	-.052	-.026	.025		.049	.007	-.073	.530
.548							.047			.548
.566							.049			.566
.584	-.033	-.081	-.056	-.029	.021	.038	.047	.004	-.077	.584
.602							.047			.602
.620							.045			.620
.638	-.027	-.083		-.029	.020		.046		-.080	.638
.655							.046			.655
.673							.045			.673
.691	-.025	-.086	-.061		.018	.036		.001	-.084	.691
.709							.045			.709
.727							.046			.727
.745	-.026	-.079		-.033	.020		.046		-.086	.745
.763							.045			.763
.781							.040			.781
.799							.040			.799
.817	-.028	-.077	-.063	-.034	.019	.038	.046	-.003	-.084	.817
.834							.041			.834
.870							.044			.870
.888							.041			.888
.921	-.115						.039			.921
.924	-.028	-.050		-.040			.037			.924
.947							.043			.947
.964							.043			.964
.981	-.018	-.045	-.071		.018	.035	.043	-.007	-.047	.981
.984					.015				-.057	.984
.998							.042			.998
1.000	-.117									1.000

Nose rake			Base rake		
Orifice station, y,in.	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.067		.031	.410	.103
.030	.614		.094	.890	.381
.050	.823		.156	.810	.924
.070	.858		.219	.823	1.146
.090	.858		.281	.827	1.311
.110	.865		.344	.831	1.357
.130	.869		.406	.824	1.373
.150	.860		.469	.823	1.382
.170	.858		.530	.837	
.190	.834		1.000		
			1.250	.884	
			1.500	.923	

TABLE 24. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 3.96$ (a) $\alpha = -10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.728			.059
.060							1.652			.060
.061							1.572			.061
.063							1.429			.063
.064							1.321			.064
.075	.075	.013	.007	-.004	-.009	-.010	-.009	-.007	.014	.075
.085	.300	.114	.080	.051	.015	.008	.006	.028	.109	.085
.096	.310	.103	.072	.031	.020	.019	.042	.132	.096	
.119	.274	.124	.101	.080	.046	.034	.029	.058	.123	.119
.143							-.031			.143
.154	.080	.0	-.010	-.018	-.028		-.029	-.024	.000	.154
.171							-.027			.171
.189							-.024			.189
.206	.062	-.007	-.018	-.024	-.026	-.023	-.021	-.025	-.008	.206
.223							-.018			.223
.241							-.015			.241
.258	.050	-.019	-.027	-.031	-.025	-.017	-.012	-.028	-.005	.258
.275							-.010			.275
.293							-.009			.293
.310	.044	-.027	-.033	-.035	-.024	-.012	-.009	-.029	-.027	.310
.327							-.008			.327
.344							-.008			.344
.362	.043	-.033	-.041	-.036	-.023	-.013	-.008	-.030	-.034	.362
.379							-.008			.379
.396	.041	-.037	-.039	-.036	-.024	-.015	-.008	-.032	-.037	.396
.408							-.008			.408
.426	.119	-.015	.004	-.017	-.003	.013	.023	-.016	-.017	.426
.451	.143	-.009	-.023	-.007	.016	.013	.028	-.010	-.010	.451
.476	.153	-.007	-.023	-.007	-.008	.010	.024	-.005	-.007	.476
.501	.160	-.023	-.008	-.011	.006	.017		-.004	.501	
.530	.058	-.035	-.042	-.033	-.032		-.020	-.022	-.034	.530
.548							-.022			.548
.566							-.022			.566
.584	.049	-.039	-.039	-.040	-.030	-.026	-.023	-.028	-.039	.584
.602							-.020			.602
.620							-.019			.620
.638	.046	-.041		-.030	-.027		-.018			.638
.655							-.017			.655
.673							-.018			.673
.691	.046	-.043	-.040		-.030	-.025	-.019			.691
.709							-.018			.709
.727							-.018			.727
.745	.046	-.024		-.036	-.034		-.019			.745
.763							-.019			.763
.781							-.020			.781
.799							-.019			.799
.817	.046	-.042	-.034	-.033	-.037	-.027	-.019			.817
.834							-.021			.834
.870							-.021			.870
.888							-.022			.888
.921	-.059						-.023			.921
.924	.044	-.038			-.029		-.024			.924
.947							-.022			.947
.964							-.031			.964
.981	.056	-.034	-.028			-.035	-.033	-.022		.981
.984							-.033			.984
.998							-.023			.998
1.000	-.059									1.000

Orifice station, y,in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y,in.	Cp top	Cp side
.010	.925		.031	.630	.029
.030	1.511		.094	1.567	.315
.050	1.627		.156	1.981	.639
.070	1.582		.219	2.153	.829
.090	1.525		.281	2.108	.946
.110	1.506		.344	2.121	1.015
.130	1.515		.406	2.150	1.050
.150	1.522		.469	2.135	1.080
.170	1.531		.750	2.141	
.190	1.495		1.000	2.162	
			1.250	2.173	
			1.500	2.166	

TABLE 24. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(b) $\alpha = -6.3^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.059							1.751				.059
.060							1.693				.060
.061							1.622				.061
.063							1.492				.063
.064							1.389				.064
.075	.044	.012	.007	.002	-.002	-.004	-.004	.000	.013	.075	
.085	.219	.113	.089	.070	.041	.036	.034	.051	.110	.085	
.096	.259	.111	.090	.063	.055	.052	.073	.128	.096		
.119	.204	.123	.108	.093	.071	.065	.063	.081	.123	.119	
.143							-.022				.143
.154	.041	.0	-.006	-.011	-.019		-.021	-.016	.000	.154	
.171							-.021				.171
.189							-.021				.189
.206	.032	-.004	-.011	-.015	-.018	-.019	-.019	-.018	-.004	.206	
.223							-.018				.223
.241							-.017				.241
.258	.022	-.010	-.015	-.018	-.016	-.014	-.016	-.016	.001	.258	
.275							-.013				.275
.293							-.013				.293
.310	.017	-.016	-.019	-.020	-.015	-.012	-.013	-.016	-.016	.310	
.327							-.013				.327
.344							-.012				.344
.362	.014	-.021	-.024	-.021	-.014	-.011	-.012	-.016	-.021	.362	
.379							-.012				.379
.396	.013	-.025	-.027	-.022	-.013	-.010	-.013	-.016	-.023	.396	
.408							-.009				.408
.426	.069	.004	.025	.008	.018	.020	.022	.014	.005	.426	
.451	.077	.009	.008	.012	.037	.024	.027	.018	.009	.451	
.476	.084	.011	.008	.010	.017	.020	.022	.016	.012	.476	
.501	.089		.008	.009	.013	.015	.017		.013	.501	
.530	.024	-.022	-.026	-.024	-.019		-.017	-.019	-.021	.530	
.548							-.019				.548
.566							-.019				.566
.584	.017	-.027	-.028	-.024	-.021	-.021	-.019	-.019	-.024	.584	
.602							-.017				.602
.620							-.015				.620
.638	.014	-.030		-.017	-.017		-.013		-.026	.638	
.655							-.011				.655
.673							-.010				.673
.691	.013	-.032	-.030		-.015	-.015		-.018	-.028	.691	
.709							-.009				.709
.727							-.009				.727
.745	.014	-.015		-.021	-.015		-.009		-.031	.745	
.763							-.009				.763
.781							-.009				.781
.799							-.009				.799
.817	.014	-.034	-.025	-.019	-.014	-.012	-.007	-.015	-.032	.817	
.834							-.008				.834
.870							-.007				.870
.888							-.007				.888
.921	-.059										.921
.924	.013	-.031		-.016			-.008				.924
.947							-.009				.947
.964							-.008				.964
.981	.019	-.026	-.017		-.017	-.014	-.008	-.014	-.029	.981	
.984							-.016		-.032	.984	
.998							-.010				.998
1.000	-.059										1.000

Orifice station, y , in.	Nose rake		Base rake		
	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side
.010	.790		.031	.385	.085
.030	1.471		.094	.917	.335
.050	1.563		.156	1.223	.554
.070	1.510		.219	1.512	.709
.090	1.463		.281	1.622	.863
.110	1.440		.344	1.659	.997
.130	1.431		.406	1.697	1.082
.150	1.431		.469	1.731	1.116
.170	1.382		.750	1.774	
.190	1.378		1.000	1.800	
			1.250	1.818	
			1.500	1.834	

TABLE 24. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(c) $\alpha = -3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.785			.059
.060							1.740			.060
.061							1.680			.061
.063							1.575			.063
.064							1.461			.064
.075	.024	.011	.008	.006	.004	.002	.003	.005	.011	.075
.085	.157	.111	.099	.091	.074	.070	.068	.081	.108	.085
.096	.211	.120	.111	.094	.098	.098	.090	.101	.126	.096
.119	.158	.124	.117	.110	.099	.095	.094	.102	.123	.119
.143							-.011			.143
.154	.019	.001	-.001	-.005	-.008		-.008	-.006	.001	.154
.171							-.010			.171
.189							-.011			.189
.206	.013	-.002	-.005	-.007	-.009	-.011	-.011	-.008	-.001	.206
.223							-.011			.223
.241							-.010			.241
.258	.008	-.004	-.006	-.009	-.009	-.011	-.010	-.009	.006	.258
.275							-.010			.275
.293							-.010			.293
.310	.003	-.007	-.009	-.010	-.009	-.011	-.010	-.010	-.008	.310
.327							-.010			.327
.344							-.010			.344
.362	.000	-.010	-.012	-.012	-.009	-.011	-.010	-.010	-.009	.362
.379							-.010			.379
.396	-.000	-.012	-.013	-.012	-.009	-.011	-.009	-.010	-.011	.396
.408							-.009			.408
.426	.045	.025	.043	.024	.026	.025	.026	.027	.024	.426
.451	.048	.029	.027	.027	.046	.030	.030	.028	.028	.451
.476	.046	.027	.025	.024	.026	.025	.026	.027	.028	.476
.501	.051	.024	.023	.023	.022	.023	.022	.022	.028	.501
.530	.002	-.012	-.015	-.016	-.016		-.013	-.014	-.011	.530
.548							-.016			.548
.566							-.015			.566
.584	.001	-.012	-.014	-.014	-.016	-.016	-.013	-.012	-.010	.584
.602							-.013			.602
.620							-.011			.620
.638	-.000	-.012			-.008	-.012				.638
.655							-.009			.655
.673							-.007			.673
.691	-.001	-.014	-.013		-.010	-.010				.691
.709							-.006			.709
.727							-.004			.727
.745	-.001	-.0			-.011	-.010				.745
.763							-.004			.763
.781							-.005			.781
.799							-.006			.799
.817	-.001	-.013	-.012	-.010	-.009	-.007	-.005	-.007	-.010	.817
.834							-.005			.834
.870							-.004			.870
.888							-.004			.888
.921	-.055						-.004			.921
.924	-.001	-.011			-.008					.924
.947							-.004			.947
.964							-.004			.964
.981	.002	-.008	-.008			-.008	-.004	-.005	-.008	.981
.984							-.009			.984
.998							-.008			.998
1.000	-.057						-.008			1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.691		.031	.221	.126
.030	1.326		.094	.434	.235
.050	1.456		.156	.571	.327
.070	1.438		.219	.729	.409
.090	1.405		.281	.844	.495
.110	1.391		.344	.955	.562
.130	1.380		.406	1.018	.644
.150	1.373		.469	1.090	.729
.170	1.351		.750	1.311	
.190	1.293		1.000	1.436	
			1.250	1.517	
			1.500	1.562	

TABLE 24. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(d) $\alpha = 0^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.059							1.765			.059
.060							1.774			.060
.061							1.702			.061
.063							1.624			.063
.064							1.508			.064
.075	.015	.015	.016	.016	.016	.016	.015	.015	.015	.075
.085	.111	.110	.109	.112	.111	.111	.111	.112	.110	.085
.096	.162	.129	.130	.131	.130	.131	.130	.127	.096	
.119	.121	.124	.123	.124	.125	.125	.125	.124	.123	.119
.143							-.000			.143
.154	.002	.002	.001	.001	.002		-.000	.001	.002	.154
.171							-.000			.171
.189							-.001			.189
.206	-.000	-.0	-.001	-.001	-.001	-.001	-.001	-.001	-.001	.206
.223							-.001			.223
.241							-.002			.241
.258	-.002	-.002	-.002	-.002	-.002	-.002	-.002	-.002	.009	.258
.275							-.002			.275
.293							-.003			.293
.310	-.005	-.005	-.004	-.004	-.006	-.005	-.006	-.004	-.004	.310
.327							-.006			.327
.344							-.006			.344
.362	-.007	-.007	-.007	-.007	-.007	-.007	-.007	-.006	-.006	.362
.379							-.007			.379
.396	-.007	-.008	-.009	-.008	-.008	-.007	-.006	-.007	-.008	.396
.408							-.006			.408
.426	.030	.028	.047	.029	.030	.029	.030	.031	.029	.426
.451	.035	.032	.032	.033	.030	.033	.036	.035	.034	.451
.476	.032	.030	.030	.030	.032	.031	.033	.033	.032	.476
.501	.032			.028	.029	.030	.030	.031	.031	.501
.530	-.008	-.011	-.012	-.011	-.011	-.011	-.009	-.009	-.011	.530
.548							-.009			.548
.566							-.008			.566
.584	-.007	-.010	-.009	-.010	-.010	-.009	-.007	-.006	-.008	.584
.602							-.006			.602
.620							-.006			.620
.638	-.006	-.008		-.003	-.008		-.004		-.006	.638
.655							-.004			.655
.673							-.005			.673
.691	-.006	-.009	-.009		-.008	-.007		-.005	-.005	.691
.709							-.006			.709
.727							-.005			.727
.745	-.006	.003		-.008	-.008		-.005		-.005	.745
.763							-.006			.763
.781							-.006			.781
.799							-.005			.799
.817	-.006	-.008	-.008	-.008	-.008	-.007	-.005	-.005	-.005	.817
.834							-.006			.834
.870							-.005			.870
.888							-.004			.888
.921	-.053						-.004			.921
.924	-.006	-.007		-.008			-.005			.924
.947							-.005			.947
.964							-.005			.964
.981	-.001	-.004	-.006		-.007	-.004	-.004	-.002	-.005	.981
.984					-.008		-.007		-.005	.984
.998							-.007			.998
1.000	-.055									1.000

Nose rake			Base rake		
Orifice station, y , in.	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.539		.031	.127	.107
.030	1.166		.094	.224	.210
.050	1.327		.156	.293	.262
.070	1.314		.219	.357	.326
.090	1.305		.281	.416	.375
.110	1.302		.344	.474	.436
.130	1.276		.406	.527	.509
.150	1.276		.469	.585	.574
.170	1.251		.750	.769	
.190	1.202		1.000	.826	
			1.250	.914	
			1.500	1.035	

TABLE 24. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 3.96$ - Continued

(e) $\alpha = 3.0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -										Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270		
.059							1.781				.059
.060							1.793				.060
.061							1.746				.061
.063							1.668				.063
.064							1.565				.064
.075	.004	.010	.013	.016	.024	.026	.027	.021	.010		.075
.085	.069	.110	.120	.133	.151	.156	.159	.144	.109		.085
.096	.116		.141	.151	.168	.171	.173	.159	.127		.096
.119	.089	.124	.132	.139	.152	.157	.158	.146	.123		.119
.143								.016			.143
.154	-.008	.002	.005	.009	.015			.016	.010	.001	.154
.171								.016			.171
.189								.014			.189
.206	-.010	-.001	.000	.005	.010	.011	.013	.007	-.001		.206
.223								.012			.223
.241								.009			.241
.258	-.010	-.004	-.001	.000	.005	.006	.007	.002	.005		.258
.275								.005			.275
.293								.003			.293
.310	-.010	-.009	-.006	-.003	-.000	.001	.002	-.001	-.008		.310
.327								.001			.327
.344								-.000			.344
.362	-.010	-.010	-.009	-.007	-.002	-.001	-.001	-.003	-.010		.362
.379								.001			.379
.396	-.010	-.011	-.010	-.008	-.003	-.001	-.001	-.006	-.012		.396
.408								.001			.408
.426	.024	.025	.046	.031	.039	.042	.043	.036	.024		.426
.451	.030	.028	.031	.034	.058	.046	.046	.040	.028		.451
.476	.025	.026	.029	.034	.042	.046	.047	.039	.028		.476
.501	.024		.029	.034	.044	.048	.049		.027		.501
.530	-.011	-.013	-.011	-.008	-.001			-.004	-.013		.530
.548								.000			.548
.566								.000			.566
.584	-.014	-.012	-.010	-.008	-.003	-.001	-.000	-.002	-.010		.584
.602								.000			.602
.620								.000			.620
.638	-.010	-.012			-.004	-.004				-.010	.638
.655								.000			.655
.673								.001			.673
.691	-.006	-.015	-.013		-.006	-.003			-.005	-.012	.691
.709								.001			.709
.727								.001			.727
.745	-.006	-.001			-.011	-.006				-.011	.745
.763								.002			.763
.781								.002			.781
.799								.002			.799
.817	-.006	-.014	-.013	-.011	-.005	-.003	-.001	-.006	-.010		.817
.834								.002			.834
.870								.001			.870
.888								.001			.888
.921	-.054							-.000			.921
.924	-.006	-.012			-.011			.001			.924
.947								.001			.947
.964								.001			.964
.981	-.000	-.009	-.010			-.004	-.002	-.000	-.004	-.008	.981
.984								.005		.010	.984
.998											.998
1.000	-.055										1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.396		.031	.098	.123
.030	1.017		.094	.143	.252
.050	1.198		.156	.190	.358
.070	1.205		.219	.239	.450
.090	1.203		.281	.284	.516
.110	1.203		.344	.304	.606
.130	1.192		.406	.335	.711
.150	1.183		.469	.369	.760
.170	1.158		.750	.494	
.190	1.109		1.000	.608	
			1.250	.711	
			1.500	.760	

TABLE 24. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 3.96$ - Continued(f) $\alpha = 6.3^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
.059							1.776			.059
.060							1.780			.060
.061							1.756			.061
.063							1.689			.063
.064							1.613			.064
.075	-.003	.013	.017	.026	.037	.044	.047	.034	.012	.075
.085	.036	.112	.135	.164	.203	.215	.221	.188	.110	.085
.096	.069	.153	.177	.212	.220	.226	.195	.128	.096	
.119	.056	.123	.139	.158	.187	.195	.199	.174	.123	.119
.143							.040			.143
.154	-.019	.001	.009	.019	.034		.041	.027	.002	.154
.171							.039			.171
.189							.036			.189
.206	-.016	-.003	.003	.012	.027	.030	.032	.020	-.002	.206
.223							.029			.223
.241							.026			.241
.258	-.013	-.009	-.003	.003	.018	.022	.022	.011	.002	.258
.275							.020			.275
.293							.018			.293
.310	-.010	-.015	-.010	-.002	.011	.015	.017	.003	-.015	.310
.327							.016			.327
.344							.014			.344
.362	-.009	-.020	-.015	-.008	.007	.012	.014	.002	-.020	.362
.379							.012			.379
.396	-.008	-.023	-.018	-.010	.006	.010	.011	-.000	-.023	.396
.408							.011			.408
.426	.022	.005	.036	.028	.057	.066	.069	.045	.005	.426
.451	.027	.010	.018	.033	.078	.072	.079	.051	.010	.451
.476	.022	.009	.021	.037	.071	.081	.086	.057	.012	.476
.501	.019		.024	.040	.074	.086	.092		.014	.501
.530	-.014	-.023	-.016	-.006	.013		.025	.007	-.021	.530
.548							.021			.548
.566							.020			.566
.584	-.018	-.026	-.019	-.011	.007	.013	.018	.002	-.024	.584
.602							.018			.602
.620							.016			.620
.638	-.013	-.029		-.008	.005		.017		-.026	.638
.655							.016			.655
.673							.016			.673
.691	-.009	-.032	-.025		.003	.009		-.001	-.029	.691
.709							.015			.709
.727							.015			.727
.745	-.008	-.015		-.016	.003		.015		-.030	.745
.763							.015			.763
.781							.014			.781
.799							.014			.799
.817	-.008	-.033	-.026	-.016	.005	.011	.015	-.001	-.030	.817
.834							.015			.834
.870							.015			.870
.888							.015			.888
.921	-.057						.017			.921
.924	-.008	-.030		-.018			.016			.924
.947							.016			.947
.964							.016			.964
.984							.016			.984
.998							.015			.998
1.000	-.059									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.161		.031	.206	.078
.030	.751		.094	.401	.349
.050	1.015		.156	.472	.590
.070	1.053		.219	.504	.736
.090	1.053		.281	.530	.892
.110	1.065		.344	.546	.991
.130	1.060		.406	.563	1.062
.150	1.056		.469	.568	1.103
.170	1.038		.750	.608	
.190	1.000		1.000	.637	
			1.250	.675	
			1.500	.707	

TABLE 24. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 3.96$ - Concluded

(g) $\alpha = 10.1^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.718			.059
.060							1.788			.060
.061							1.783			.061
.063							1.756			.063
.064							1.656			.064
.075	-.010	.014	.022	.038	.065	.085	.091	.056	.012	.075
.085	.012	.115	.153	.203	.275	.296	.307	.241	.109	.085
.096	.027		.172	.208	.267	.282	.292	.239	.132	.096
.119	.027	.122	.151	.180	.235	.251	.259	.209	.122	.119
.143							.080			.143
.154	-.025	.002	.016	.034	.067		.083	.050	.001	.154
.171							.077			.171
.189							.068			.189
.206	-.019	-.006	.004	.021	.052	.059	.063	.036	-.007	.206
.223							.059			.223
.241							.055			.241
.258	-.012	-.017	-.005	.009	.040	.049	.052	.025	-.004	.258
.275							.052			.275
.293							.048			.293
.310	-.007	-.026	-.014	.001	.032	.043	.047	.018	-.027	.310
.327							.046			.327
.344							.046			.344
.362	-.006	-.034	-.021	-.003	.029	.040	.045	.014	-.034	.362
.379							.045			.379
.396	-.008	-.036	-.024	-.006	.030	.040	.045	.012	-.036	.396
.408							.045			.408
.426	.022	-.015	.029	.035	.096	.116	.123	.070	-.016	.426
.451	.029	-.010	.012	.048	.126	.138	.146	.085	-.010	.451
.476	.023	-.008	.019	.055	.126	.147	.157	.094	-.007	.476
.501	.019		.023	.059	.130	.152	.162		-.005	.501
.530	-.018	-.034	-.018	.001	.041		.059	.023	-.035	.530
.548							.055			.548
.566							.052			.566
.584	-.023	-.037	-.024	-.006	.032	.045	.050	.017	-.039	.584
.602							.049			.602
.620							.048			.620
.638	-.019	-.040		-.002	.029		.049		-.041	.638
.655							.050			.655
.673							.049			.673
.691	-.018	-.042	-.028		.028	.042		.014	-.042	.691
.709							.048			.709
.727							.049			.727
.745	-.020	-.023		-.010	.029		.049		-.042	.745
.763							.049			.763
.781							.047			.781
.799							.047			.799
.817	-.020	-.041	-.030	-.012	.027	.042	.049	.012	-.042	.817
.834							.048			.834
.870							.048			.870
.888							.047			.888
.921	-.059						.046			.921
.924	-.023	-.038		-.014			.043			.924
.947							.045			.947
.964							.045			.964
.981	-.019	-.034	-.033		.025	.040	.045	.010	-.040	.981
.984							.045		-.041	.984
.998							.045			.998
1.000	-.060									1.000

Orifice station, y/l	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y/l	Cp top	Cp side
.010	.047		.031	.253	.032
.030	.318		.094	.575	.354
.050	.741		.156	.564	.667
.070	.908		.219	.553	.806
.090	.895		.281	.559	.917
.110	.895		.344	.559	1.018
.130	.895		.406	.557	1.065
.150	.895		.469	.550	1.103
.170	.881		.750	.514	
.190	.843		1.000	.515	
			1.250	.553	
			1.500	.609	

TABLE 25.- PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 4.65$ (a) $\alpha = -6.3^\circ$

Model station, $\frac{x}{T}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{T}$
	0	90	105	120	150	165	180	225	270	
.059							1.835			.059
.060							1.763			.060
.061							1.671			.061
.063							1.553			.063
.064							1.440			.064
.075	.073	.040	.034	.030	.024	.022	.021	.025	.037	.075
.085	.230	.138	.113	.098	.070	.062	.060	.072	.122	.085
.096	.276	.131	.113	.084	.075	.073	.087	.136	.096	
.119	.195	.135	.122	.108	.087	.080	.077	.090	.125	.119
.143							.002			.143
.154	.050	.017	.012	.007	.002		.000	.003	.013	.154
.171							.000			.171
.189							.000			.189
.206	.039	.011	.006	.002	.000	-.000	-.002	.001	.006	.206
.223							-.002			.223
.241							-.002			.241
.258	.029	.004	.002	-.001	-.000	-.000	-.002	-.000	.022	.258
.275							.002			.275
.293							.001			.293
.310	.021	-.0	-.000	-.004	-.001	-.000	.000	-.002	-.004	.310
.327							.001			.327
.344							.001			.344
.362	.016	-.004	-.006	-.006	-.001	-.000	.001	-.003	-.007	.362
.379							.001			.379
.396	.017	-.007	-.006	-.006	-.001	-.000	.001	-.004	-.010	.396
.408							.001			.408
.426	.057	.030	.050	.011	.020	.022	.022	.018	.010	.426
.451	.064	.019	.014	.015	.056	.024	.027	.022	.014	.451
.476	.070	.018	.014	.015	.020	.022	.023	.018	.012	.476
.501	.077			.014	.015	.017	.019	.020	.014	.501
.520	.029	-.006	-.007	-.006	-.005		-.003	-.006	-.008	.530
.548							.006			.548
.566							.006			.566
.584	.022	-.007	-.009	-.006	-.007	-.009	-.006	-.005	-.011	.584
.602							.006			.602
.620							-.005			.620
.638	.017	-.010			.001	-.006			-.013	.638
.655							-.005			.655
.673							-.002			.673
.691	.017	-.012	-.011		-.006	-.003		-.006	-.013	.691
.709							-.002			.709
.727							-.001			.727
.745	.017	.013			-.009	-.006			-.012	.745
.763							-.000			.763
.781							-.001			.781
.799							-.001			.799
.817	.017	-.013	-.011	-.008	-.004	-.003	-.001	-.004	-.015	.817
.834							-.000			.834
.870							-.001			.870
.888							-.000			.888
.921	-.031									.921
.924	.017	-.012			-.008					.924
.947										.947
.964										.964
.981	.022	-.011	-.009			-.008	-.006	-.001	-.007	.981
.984							-.006			.984
.998							-.005			.998
1.000	-.030									1.000

Orifice station, y , in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.737		.031	.350	.057
.030	1.410		.054	.786	.271
.050	1.560		.156	1.090	.449
.070	1.533		.219	1.404	.555
.090	1.510		.281	1.556	.673
.110	1.496		.344	1.632	.804
.130	1.486		.406	1.692	.882
.150	1.475		.469	1.726	.940
.170	1.450		.750	1.828	
.190	1.401		1.000	1.869	
			1.250	1.897	
			1.500	1.860	

TABLE 25. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 4.65$ - Continued

(b) $\alpha = -3.0^\circ$

Model station, $\frac{x}{r}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{r}$
	0	90	105	120	150	165	180	225	270	
.059							4.799			.059
.060							1.833			.060
.061							1.780			.061
.063							1.700			.063
.064							1.603			.064
.075	.053	.040	.037	.034	.033	.029	.029	.032	.036	.075
.085	.172	.136	.124	.118	.096	.093	.091	.097	.120	.085
.096	.221	.139	.130	.115	.110	.107	.111	.133	.138	.096
.119	.154	.135	.129	.122	.111	.108	.105	.108	.125	.119
.143							.006			.143
.154	.031	.017	.015	.012	.008		.005	.007	.014	.154
.171							.003			.171
.189							.002			.189
.206	.023	.011	.008	.006	.003	.002	.001	.004	.007	.206
.223							.001			.223
.241							.001			.241
.258	.016	.007	.004	.003	.001	.000	.001	.002	.025	.258
.275							.001			.275
.293							.001			.293
.310	.010	.002	.000	.000	-.000	-.000	.001	.000	.000	.310
.327							-.001			.327
.344							-.001			.344
.362	.006	-.001	-.002	-.002	-.001	-.002	-.000	-.001	-.002	.362
.379							.002			.379
.396	.007	-.004	-.003	-.002	.001	.001	.002	.001	-.004	.396
.408							.004			.408
.426	.026	.027	.059	.022	.014	.012	.012	.014	.024	.426
.451	.043	.029	.027	.026	.056	.019	.021	.019	.027	.451
.476	.041	.027	.024	.023	.022	.021	.021	.022	.024	.476
.501	.046	.022	.019	.020	.019	.019	.021	.023	.023	.501
.530	.012	-.004	-.005	-.006	-.007		-.004	-.006	-.007	.530
.548							-.007			.548
.566							.005			.566
.584	.007	-.003	-.006	-.006	-.008	-.007	-.005	-.006	-.007	.584
.602							-.006			.602
.620							-.005			.620
.638	.005	-.004			.003	-.006			.006	.638
.655							-.005			.655
.673							-.002			.673
.691	.004	-.006	-.006		-.005	-.003			.006	.691
.709							-.003			.709
.727							-.001			.727
.745	.004	.018			-.006	-.004			-.007	.745
.763							-.000			.763
.781							-.001			.781
.799							-.001			.799
.817	.005	-.007	-.006	-.006	-.001	-.002			.006	.817
.834							-.001			.834
.870							-.001			.870
.888							-.002			.888
.921	-.031						-.001			.921
.924	.005	-.007			-.006		-.002			.924
.947							-.002			.947
.964							-.000			.964
.981	.006	-.004	-.006			-.003			.004	.981
.984							-.001			.984
.998							-.006			.998
1.000	-.031						-.006			1.000

Nose rake			Base rake		
Orifice station, y , in.	Cp top	Cp side	Orifice station, y , in.	Cp top	Cp side
.010	.580		.031	.163	.091
.030	1.248		.094	.326	.207
.050	1.430		.156	.437	.289
.070	1.425		.219	.564	.363
.090	1.409		.281	.647	.412
.110	1.403		.344	.725	.476
.130	1.386		.406	.785	.559
.150	1.372		.469	.872	4.794
.170	1.345		.750	1.114	
.190	1.294		1.000	1.276	
			1.250	1.398	
			1.500	1.476	

TABLE 25. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC

MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 4.65$ - Continued(c) $\alpha = 0^\circ$

Model station, $\frac{x}{l}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{l}$
	0	90	105	120	150	165	180	225	270	
.059							1.840			.059
.060							1.812			.060
.061							1.750			.061
.063							1.678			.063
.064							1.556			.064
.075	.036	.040	.040	.041	.041	.041	.041	.040	.038	.075
.085	.124	.134	.135	.136	.132	.130	.128	.122	.119	.085
.096	.172	.149	.147	.146	.143	.142	.136	.131	.124	.096
.119	.123	.134	.135	.135	.135	.133	.132	.126	.119	.119
.143							.017			.143
.154	.017	.018	.017	.016	.017		.015	.013	.014	.154
.171							.011			.171
.189							.008			.189
.206	.011	.012	.010	.011	.011	.010	.008	.008	.007	.206
.223							.006			.223
.241							.006			.241
.258	.007	.007	.008	.007	.007	.006	.005	.004	.025	.258
.275							.005			.275
.293							.002			.293
.310	.003	.003	.004	.004	.002	.002	.002	.002		.310
.327							.001			.327
.344							.001			.344
.362	-.000	-.00	.000	.001	.000	-.000	.000	-.000	-.000	.362
.379							-.000	-.000	-.000	.379
.396	.002	-.001	-.001	-.000	.000	-.000	-.000	-.000	-.000	.396
.408							-.001			.408
.426	.026	.029	.062	.027	.027	.026	.025	.025	.024	.426
.451	.034	.032	.031	.032	.064	.031	.033	.032	.029	.451
.476	.030	.029	.029	.029	.029	.029	.029	.029	.028	.476
.501	.033	.027	.028	.028	.028	.027	.027	.027	.025	.501
.530	.003	-.003	-.004	-.003	-.004		-.003	-.004	-.005	.530
.548							.004			.548
.566							.003			.566
.584	-.000	-.002	-.003	-.003	-.004	-.006	-.002	-.005	-.006	.584
.602							.002			.602
.620							-.002			.620
.638	-.001	-.002			.006	-.004			-.005	.638
.655							.001			.655
.673							.001			.673
.691	-.000	-.003	-.004			-.004	-.003		-.003	.691
.709							.001			.709
.727							.001			.727
.745	.000	.020			-.003	-.004			-.003	.745
.763							-.002			.763
.781							.001			.781
.799							.002			.799
.817	-.000	-.004	-.003	-.003	-.004	-.002			-.001	.817
.834							.001			.834
.870							.002			.870
.888							.001			.888
.921	-.030									.921
.924	-.000	-.004								.924
.947										.947
.964										.964
.981	-.000	-.001	-.003							.981
.984										.984
.998										.998
1.000	-.032									1.000

Orifice station, $y, \text{in.}$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y, \text{in.}$	Cp top	Cp side
.010	.464		.031	.091	.103
.030	1.116		.094	.172	.165
.050	1.331		.156	.223	.236
.070	1.326		.219	.282	.289
.090	1.319		.281	.319	.338
.110	1.312		.344	.363	.409
.130	1.280		.406	.411	.444
.150	1.245		.469	.455	.511
.170	1.211		.750	.607	
.190	1.213		1.000	.665	
			1.250	.718	
			1.500	.794	

TABLE 25. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 4.65$ - Continued

(d) $\alpha = 3.0^\circ$

Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.059							1.823			.059
.060							1.814			.060
.061							1.772			.061
.063							1.698			.063
.064							1.611			.064
.075	.027	.037	.042	.045	.052	.055	.054	.046	.036	.075
.085	.088	.135	.144	.158	.167	.173	.175	.154	.118	.085
.096	.130	.159	.167	.183	.184	.184	.184	.165	.132	.096
.119	.095	.134	.140	.147	.158	.161	.161	.147	.122	.119
.143							.030			.143
.154	.009	.017	.021	.024	.028		.030	.020	.010	.154
.171							.026			.171
.189							.023			.189
.206	.004	.011	.013	.017	.021	.020	.021	.015	.006	.206
.223							.018			.223
.241							.015			.241
.258	.002	.006	.009	.011	.014	.014	.014	.008	.023	.258
.275							.012			.275
.293							.010			.293
.310	.002	.002	.005	.005	.008	.008	.008	.004	-.001	.310
.327							.006			.327
.344							.005			.344
.362	-.000	-.0	.000	.001	.004	.004	.003	.001	-.004	.362
.379							.003			.379
.396	.001	-.003	-.001	-.000	.002	.003	.003	-.001	-.005	.396
.408							.003			.408
.426	.024	.026	.063	.030	.037	.038	.038	.027	.017	.426
.451	.029	.029	.031	.034	.070	.041	.043	.035	.023	.451
.476	.023	.027	.029	.033	.039	.042	.041	.033	.024	.476
.501	.027	.029	.033	.041	.042		.043		.023	.501
.530	.001	-.004	-.003	-.000	.003		.003	-.000	-.006	.530
.548							.003			.548
.566							.003			.566
.584	-.003	-.002	-.003	-.000	.001	.002	.003	-.002	-.008	.584
.602							.003			.602
.620							.003			.620
.638	-.004	-.004			.006	-.000			-.007	.638
.655							.003			.655
.673							.003			.673
.691	-.002	-.004	-.004			.001			-.003	.691
.709							.001			.709
.727							.001			.727
.745	-.000	.019		-.002	.000		.001		-.007	.745
.763							.000			.763
.781							.000			.781
.799							.000			.799
.817	.000	-.008	-.005	-.003	.002	.000	.000	-.004	-.007	.817
.834							.000			.834
.870							.000			.870
.888							.000			.888
.921	-.031						.000			.921
.924	-.000	-.005			-.004		-.001			.924
.947							.000			.947
.964							.000			.964
.981	.001	-.004	-.006			.001	.001	-.005	-.005	.981
.984										.984
.998										.998
1.000										1.000
	-.031									

Orifice station, $y, \text{in.}$	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, $y, \text{in.}$	Cp top	Cp side
.010	.275		.031	.068	.100
.030	.946		.094	.114	.209
.050	1.227		.156	.149	.298
.070	1.210		.219	.190	.358
.090	1.196		.281	.206	.432
.110	1.196		.344	.236	.503
.130	1.185		.406	.269	.591
.150	1.174		.469	.298	.646
.170	1.141		.750	.404	
.190	1.097		1.000	.513	
			1.250	.582	
			1.500	.623	

TABLE 25. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 4.65$ - Continued

(e) $\alpha = 6.3^\circ$

Model station, $\frac{x}{c}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{c}$
	0	90	105	120	150	165	180	225	270	
.059							1.819			.059
.060							1.837			.060
.061							1.810			.061
.063							1.759			.063
.064							1.753			.064
.075	.019	.036	.042	.052	.069	.076	.077	.060	.036	.075
.085	.058	.137	.160	.187	.218	.229	.235	.194	.120	.085
.096	.092		.171	.193	.223	.228	.231	.199	.133	.096
.119	.069	.134	.149	.166	.190	.196	.199	.170	.125	.119
.143							.047			.143
.154	-.000	.016	.024	.032	.045		.052	.035	.012	.154
.171							.048			.171
.189							.046			.189
.206	-.002	.010	.015	.024	.036	.039	.043	.025	.006	.206
.223							.040			.223
.241							.035			.241
.258	-.001	.002	.008	.013	.025	.028	.031	.017	.022	.258
.275							.029			.275
.293							.026			.293
.310	.000	-.002	.002	.007	.017	.021	.023	.010	-.004	.310
.327							.021			.327
.344							.020			.344
.362	-.001	-.007	-.003	.001	.014	.015	.019	.005	-.009	.362
.379							.017			.379
.396	.001	-.009	-.004	.000	.012	.015	.017	.004	-.011	.396
.408							.017			.408
.426	.022	.012	.057	.032	.052	.059	.060	.035	.008	.426
.451	.028	.016	.023	.036	.085	.066	.067	.042	.011	.451
.476	.021	.017	.025	.038	.064	.072	.075	.046	.014	.476
.501	.024		.028	.041	.070	.078	.081		.013	.501
.530	.000	-.008	-.004	.004	.021		.026	.008	-.010	.530
.548							.023			.548
.566							.021			.566
.584	-.006	-.009	-.006	-.000	.014	.018	.020	-.003	-.012	.584
.602							.018			.602
.620							.018			.620
.638	-.007	-.012			.007	.012			-.013	.638
.655							.017			.655
.673							.017			.673
.691	-.004	-.013	-.009			.011	.015		-.000	.691
.709							.017			.709
.727							.017			.727
.745	-.002	.011		-.004	.011				-.015	.745
.763							.017			.763
.781							.017			.781
.799							.016			.799
.817	-.002	-.015	-.010	-.004	.011	.015	.016	-.001	-.015	.817
.834							.016			.834
.870							.016			.870
.888							.016			.888
.921	-.033									.921
.924	-.004	-.015		-.004						.924
.947										.947
.964										.964
.981	-.002	-.013	-.012			.010	.016		-.001	.981
.984										.984
.998										.998
1.000	-.033									1.000

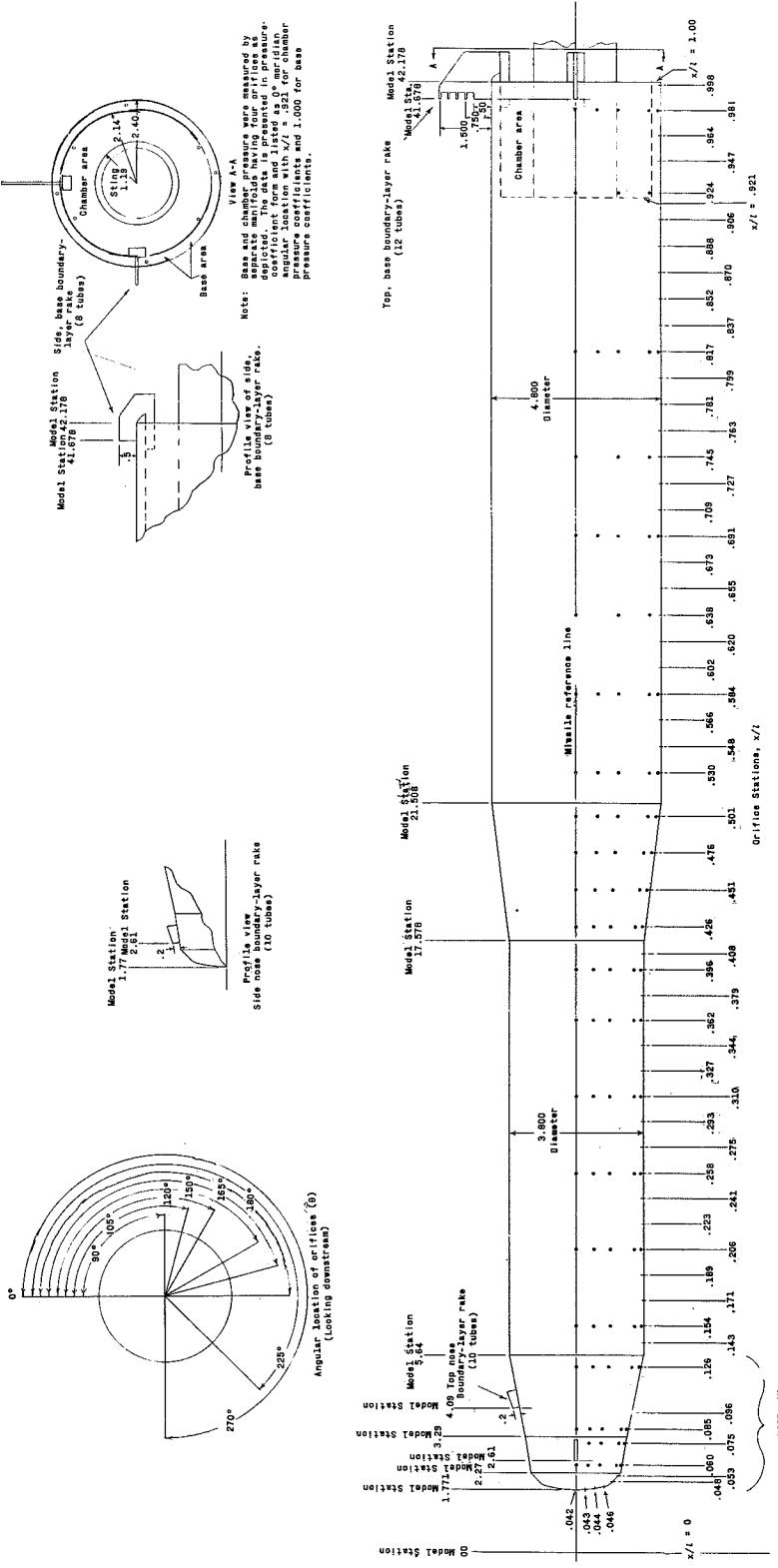
Orifice station, y, in.	Nose rake		Base rake		
	Cp top	Cp side	Orifice station, y, in.	Cp top	Cp side
.010	.144		.031	.091	.050
.030	.727		.094	.216	.269
.050	1.105		.156	.253	.441
.070	1.084		.219	.296	.559
.090	1.068		.281	.310	.683
.110	1.068		.344	.335	.805
.130	1.061		.406	.338	.877
.150	1.045		.469	.352	.948
.170	1.026		.520	.400	
.190	.969		1.000	.453	
			1.250	.511	
			1.500	.554	

TABLE 25. - PRESSURE COEFFICIENTS FOR AN INTERCONTINENTAL BALLISTIC
MISSILE FOR NOSE V WITH NATURAL TRANSITION AT $M = 4.65$ - Concluded

(f) $\alpha = 10.1^\circ$

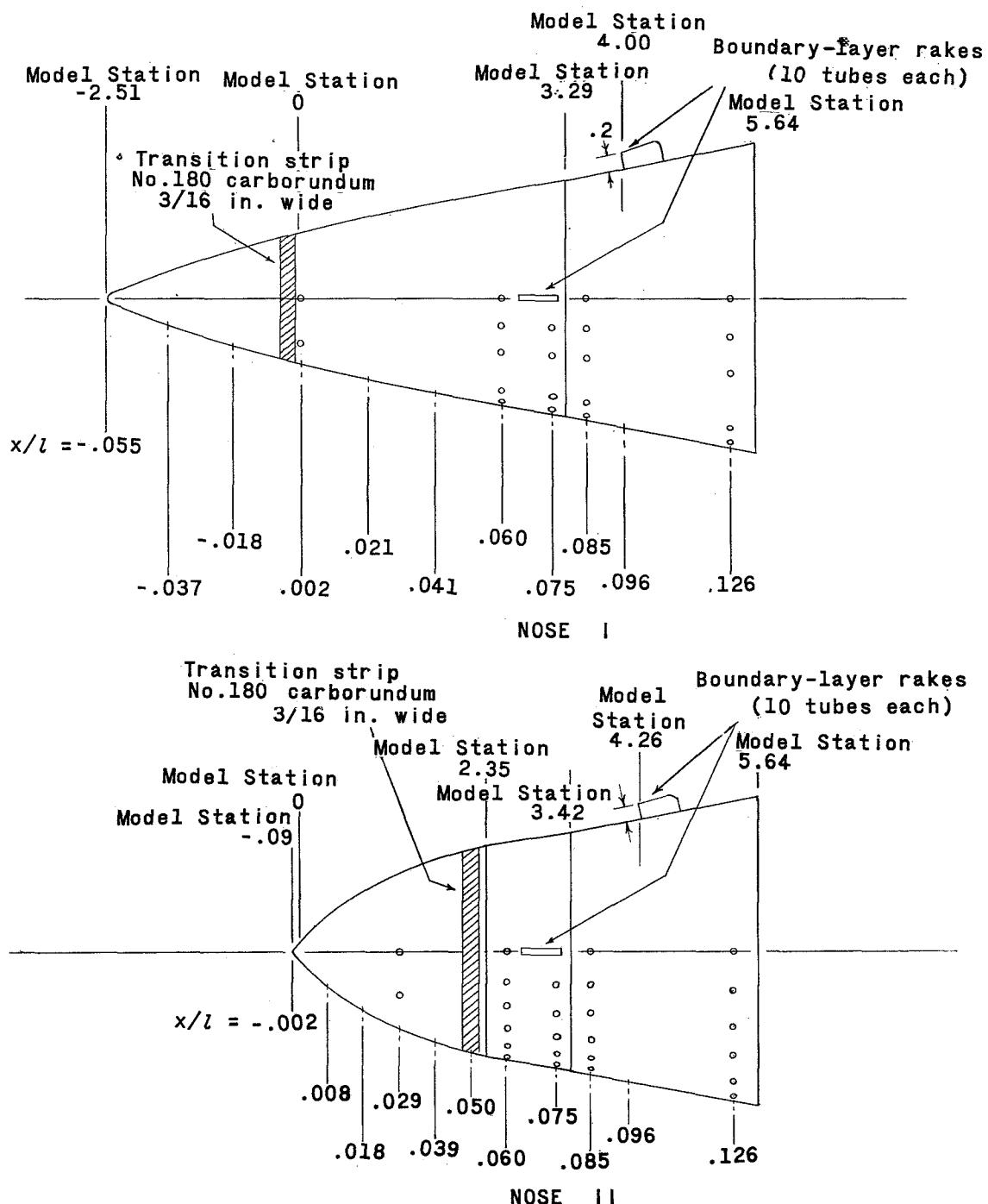
Model station, $\frac{x}{t}$	Cp for meridian angle, θ , deg -									Model station, $\frac{x}{t}$
	0	90	105	120	150	165	180	225	270	
.059							1.755			.059
.060							1.787			.060
.061							1.801			.061
.063							1.761			.063
.064							1.685			.064
.075	.011	.037	.050	.067	.095	.117	.118	.077	.035	.075
.085	.032	.134	.169	.219	.280	.300	.308	.239	.119	.085
.096	.055		.181	.215	.267	.280	.286	.233	.135	.096
.119	.043	.129	.153	.181	.228	.244	.248	.197	.120	.119
.143							.087			.143
.154	-.004	.016	.028	.046	.076		.089	.055	.012	.154
.171							.083			.171
.189							.074			.189
.206	-.003	.007	.017	.033	.058	.065	.067	.041	.004	.206
.223							.062			.223
.241							.058			.241
.258	-.001	-.0	.005	.020	.045	.054	.056	.029	.017	.258
.275							.055			.275
.293							.050			.293
.310	-.000	-.006	.001	.011	.036	.045	.048	.022	-.011	.310
.327							.047			.327
.344							.046			.344
.362	-.000	-.012	-.005	.005	.032	.041	.045	.017	-.015	.362
.379							.044			.379
.396	.002	-.014	-.007	.004	.032	.040	.044	.015	-.016	.396
.408							.044			.408
.426	.022	.002	.053	.038	.088	.105	.109	.055	-.006	.426
.451	.028	.005	.023	.051	.131	.130	.137	.072	-.000	.451
.476	.022	.007	.028	.059	.126	.144	.151	.085	.001	.476
.501	.022		.033	.063	.131	.151	.158		.001	.501
.530	-.000	-.012	-.002	.014	.049		.063	.026	-.016	.530
.548							.057			.548
.566							.054			.566
.584	-.007	-.013	-.007	.005	.036	.048	.051	.018	-.018	.584
.602							.051			.602
.620							.050			.620
.638	-.009	-.016			.011	.034			-.020	.638
.655							.051			.655
.673							.051			.673
.691	-.007	-.018	-.010			.035	.046		-.021	.691
.709							.051			.709
.727							.051			.727
.745	-.007	.009			.002	.035			-.021	.745
.763							.051			.763
.781							.051			.781
.799							.049			.799
.817	-.007	-.017	-.011	.000	.034	.045	.051	.013	-.021	.817
.834							.048			.834
.870							.049			.870
.888							.049			.888
.921	-.033						.048			.921
.924	-.008	-.019			-.000		.047			.924
.947							.045			.947
.964							.046			.964
.981	-.010	-.017	-.014			.032	.043		-.019	.981
.984							.046		-.021	.984
.998							.046			.998
1.000	-.033									1.000

Nose rake			Base rake		
Orifice station, y , in.	C_p top	C_p side	Orifice station, y , in.	C_p top	C_p side
.010	.080		.031	.131	.041
.030	.299		.094	.431	.334
.050	.874		.156	.398	.564
.070	.929		.219	.410	.703
.090	.903		.281	.414	.866
.110	.901		.344	.414	1.000
.130	.908		.406	.414	1.083
.150	.899		.469	.410	1.157
.170	.894		.750	.387	
.190	.878		1.000	.373	
			1.250	.405	
			1.500	.433	



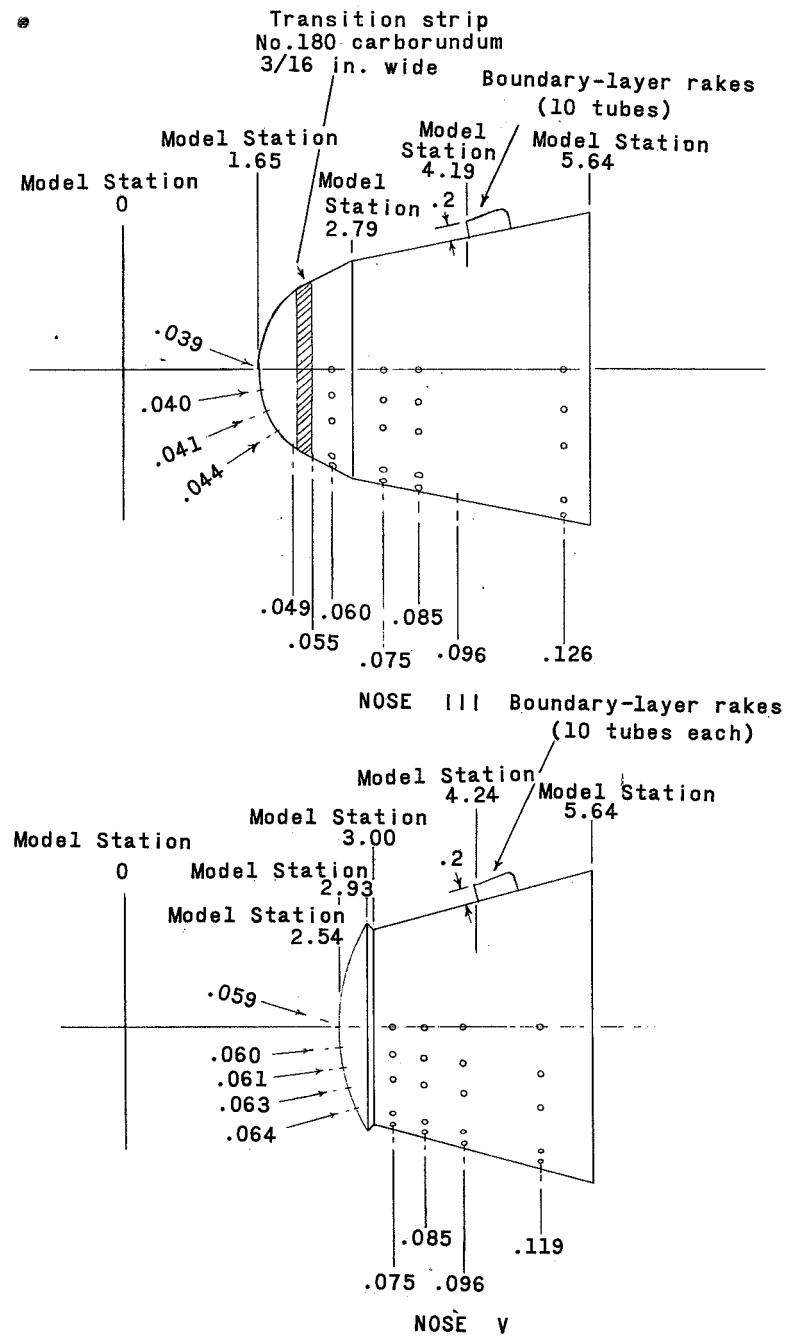
(a) Complete model with nose IV attached.

Figure 1.- Details of model of an intercontinental ballistic missile. (All dimensions are in inches unless otherwise noted.)



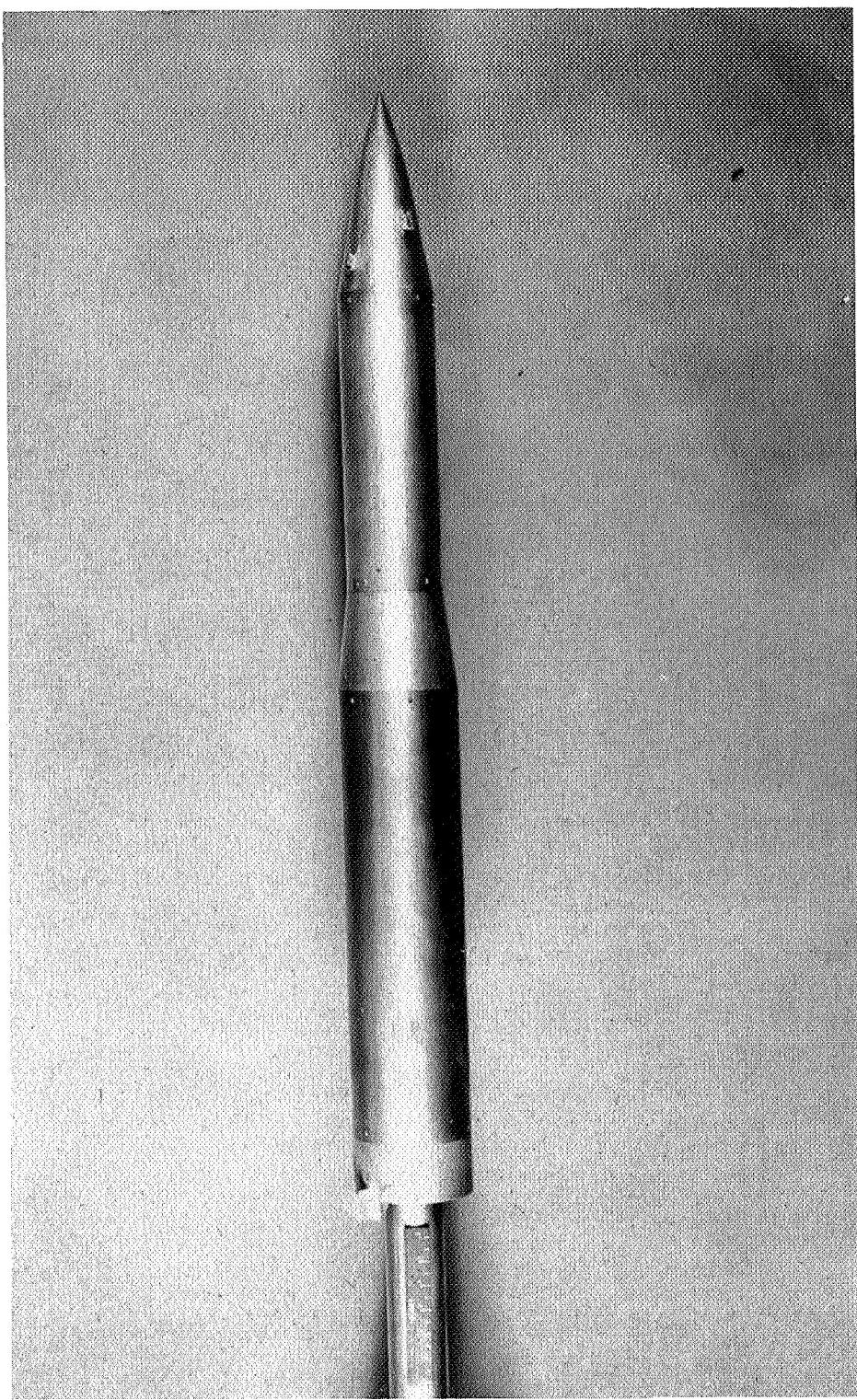
(b) Details of noses I and II.

Figure 1.- Continued.



(c) Details of noses III and V.

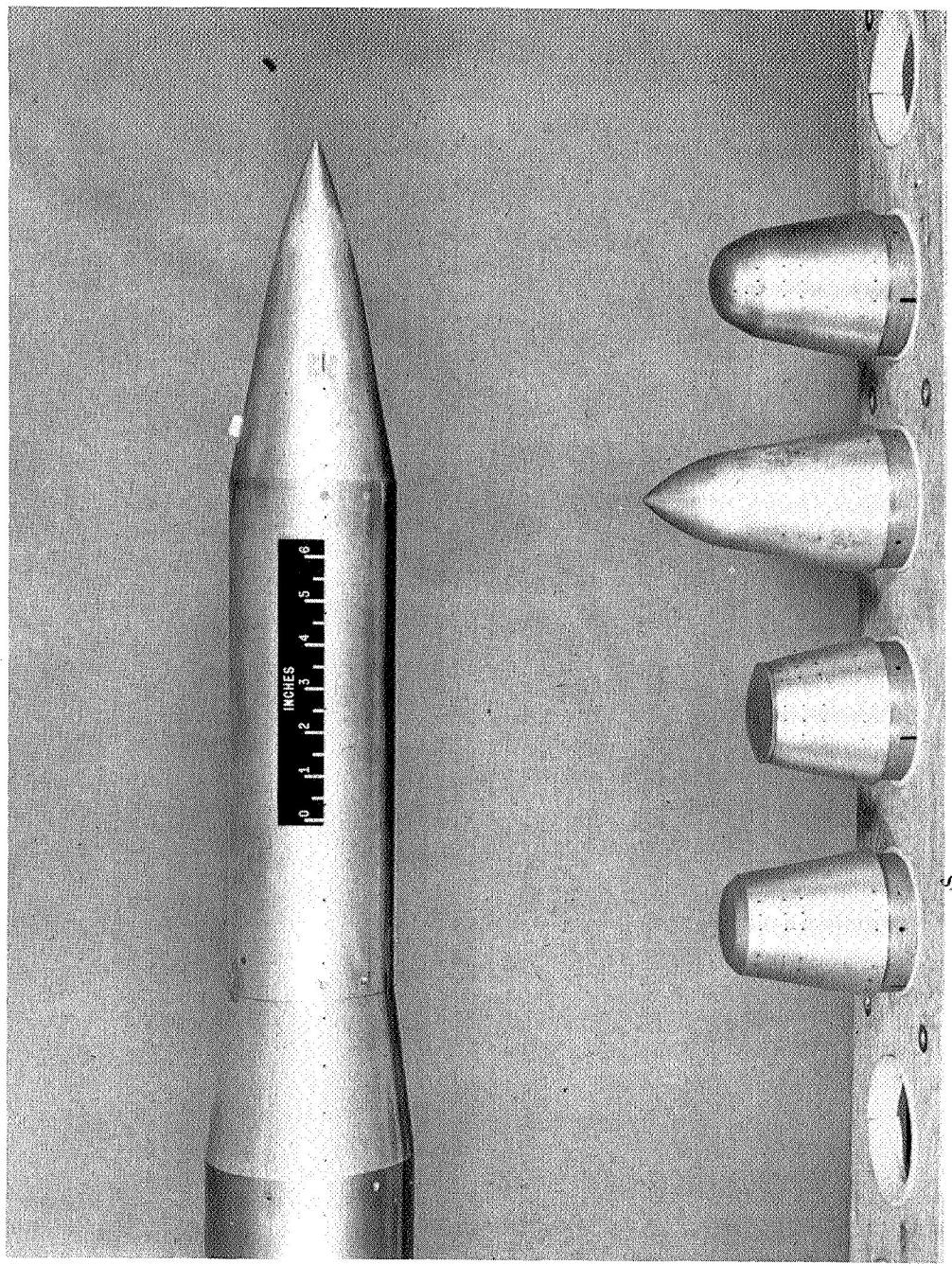
Figure 1.- Concluded.



(a) Complete model with nose I attached.

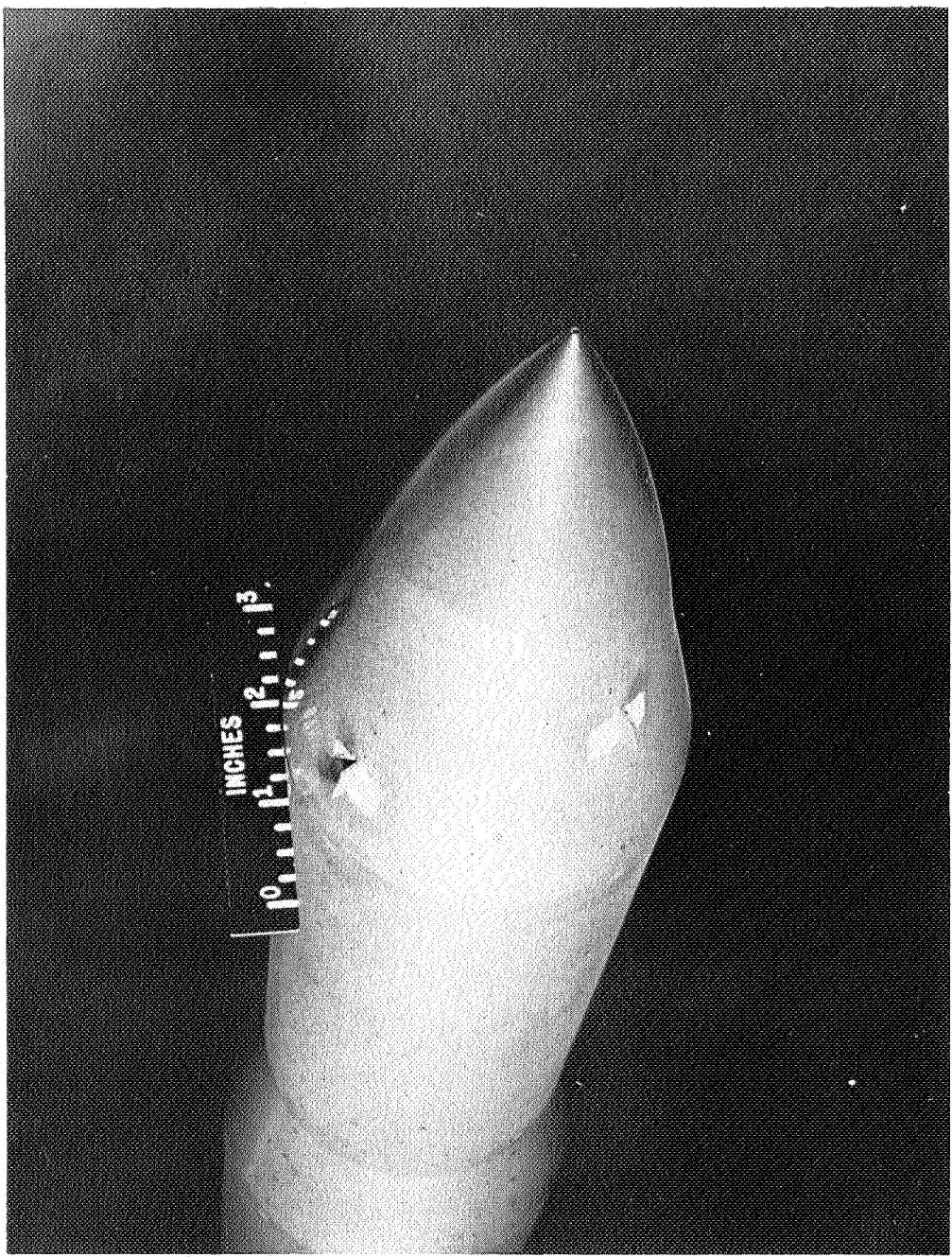
L-57-3067

Figure 2.- Photographs of test model of an intercontinental ballistic missile.



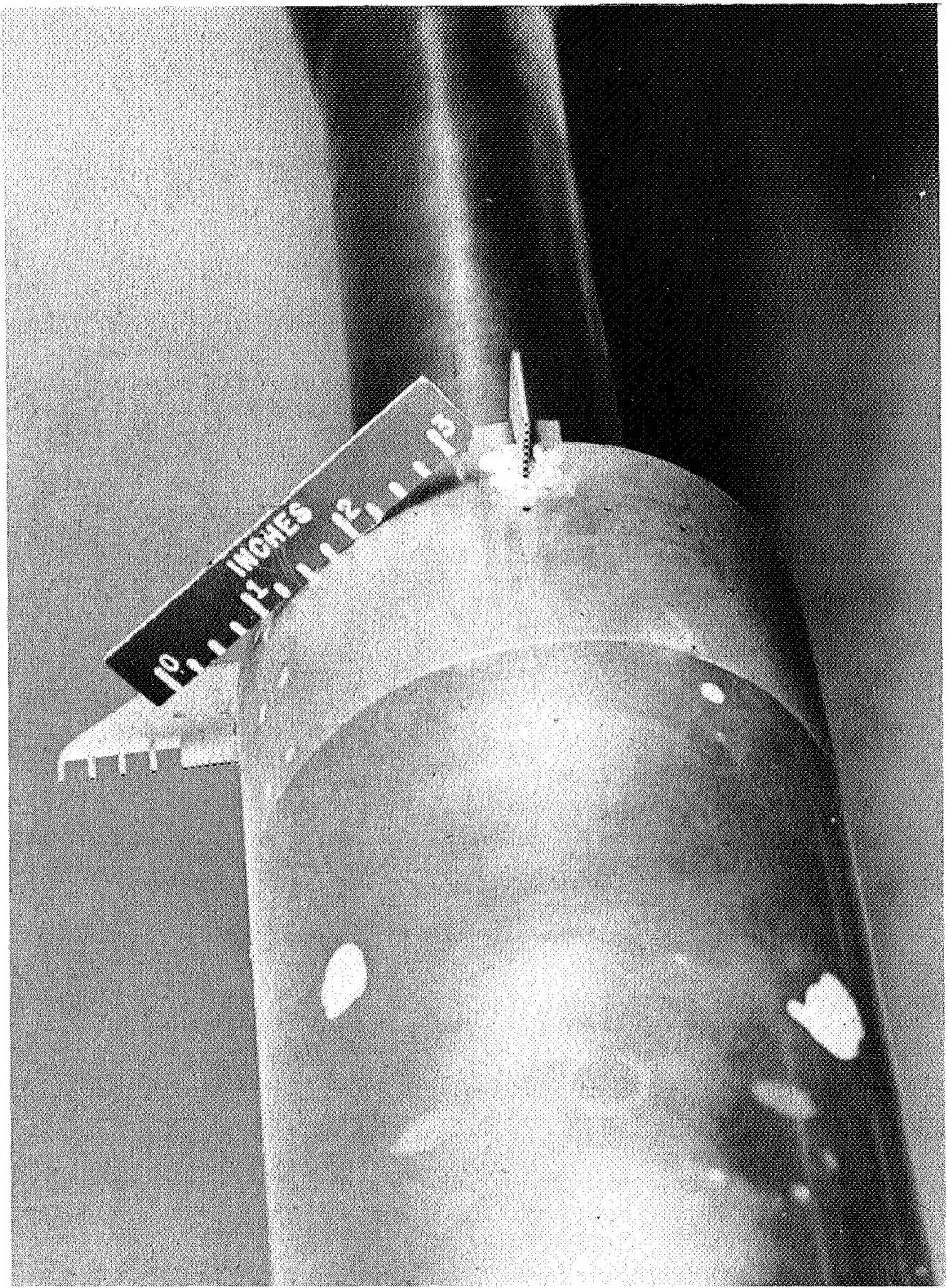
(b) Nose I attached to model and left to right noses IV, V, II, and III. L-57-3068

Figure 2.- Continued.



(c) Nose boundary-layer rakes installed on nose I. L-57-3069

Figure 2.- Continued.



L-57-3070

(d) Base boundary-layer rakes.

Figure 2.- Concluded.



M=1.57

(a) Nose I, natural transition.



M=1.57

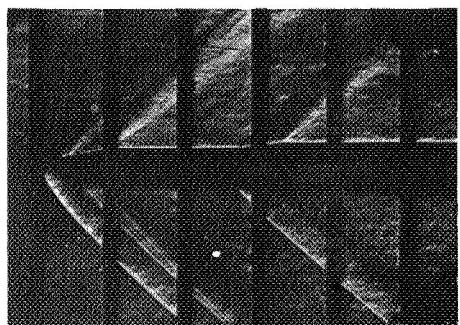


M=2.98

(b) Nose I, fixed transition.



M=1.57

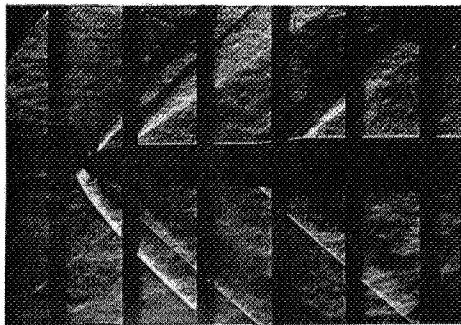


M=1.57

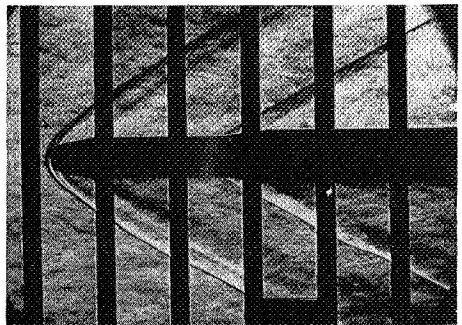
(c) Nose II, natural transition.

(d) Nose II, fixed transition. L-58-135a

Figure 3.- Typical schlieren photographs of a model of an intercontinental ballistic missile.



M=1.57



M=2.29



M=2.98



M=3.96

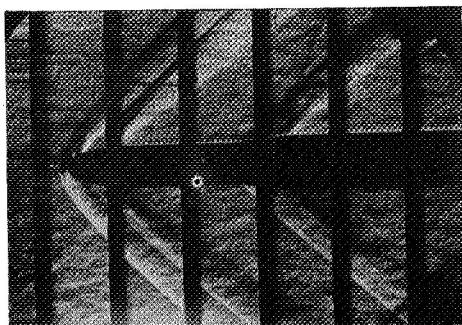


M=4.65

(e) Nose III, natural transition.

L-58-136a

Figure 3.- Continued.



$$M=1.57$$



M=2.29



M=2.98



M=3.96

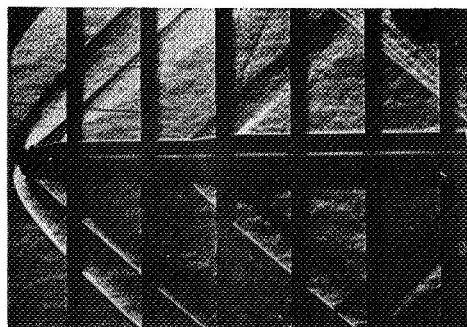


M=4,65

(f) Nose III, fixed transition.

L-58-137a

Figure 3.- Continued.



M=1.57



M=2.29



M=2.98



M=3.96

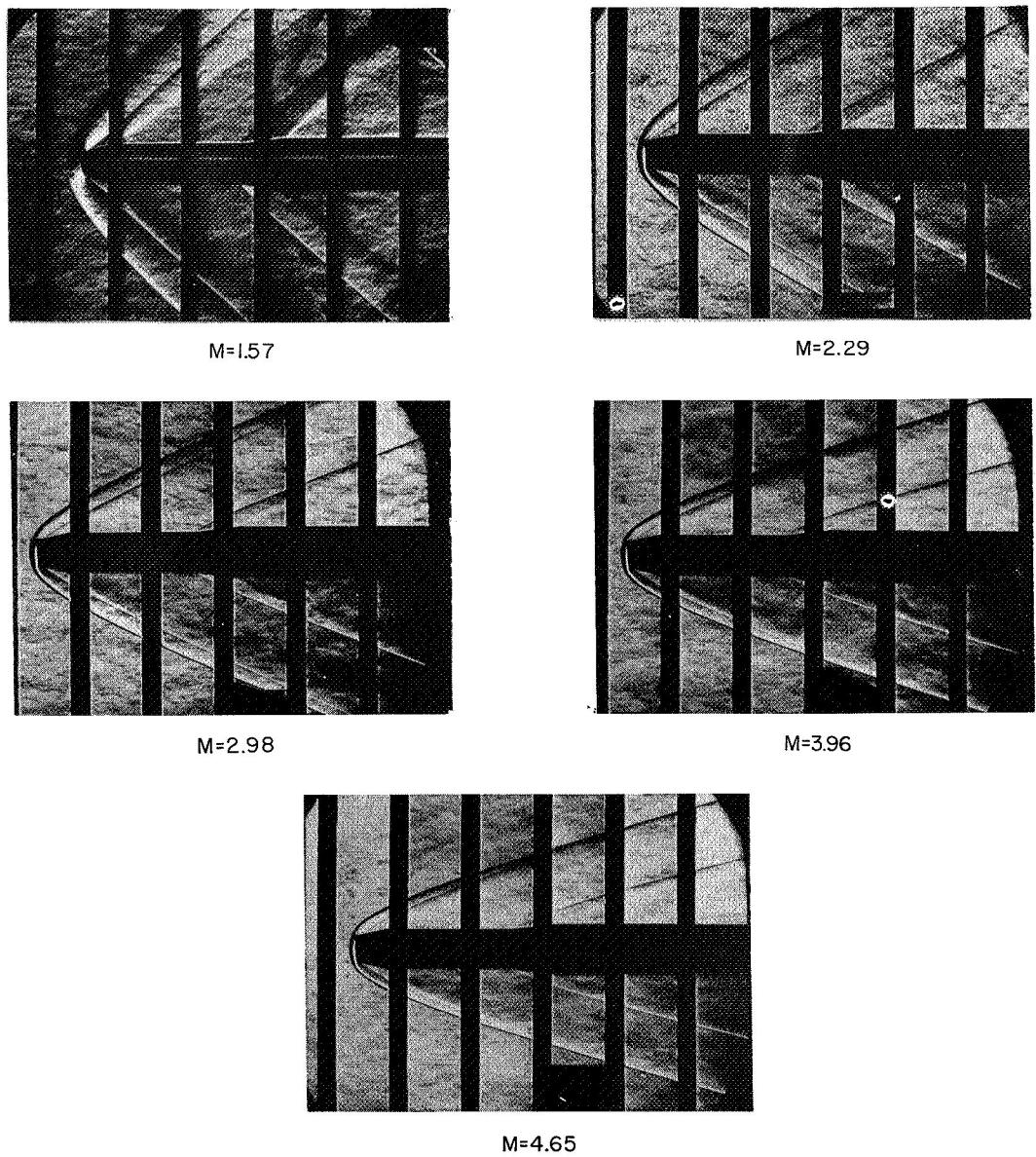


M=4.65

(g) Nose IV, natural transition.

L-58-138a

Figure 3.- Continued.



(h) Nose V, natural transition.

L-58-139a

Figure 3.- Concluded.

